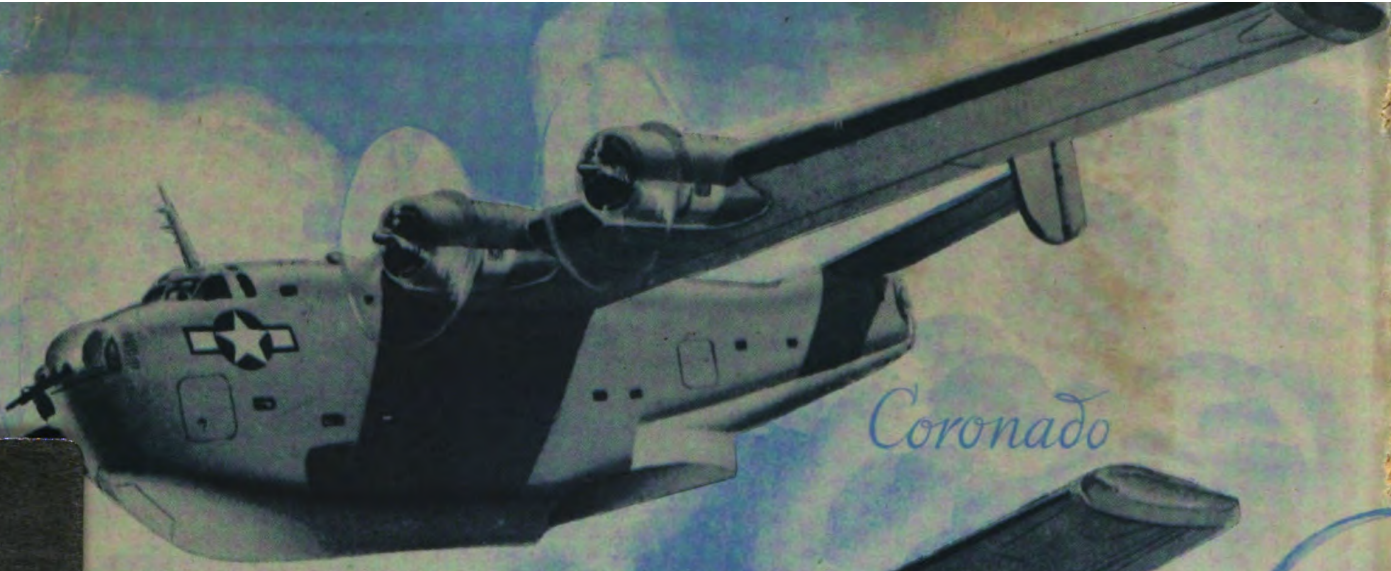
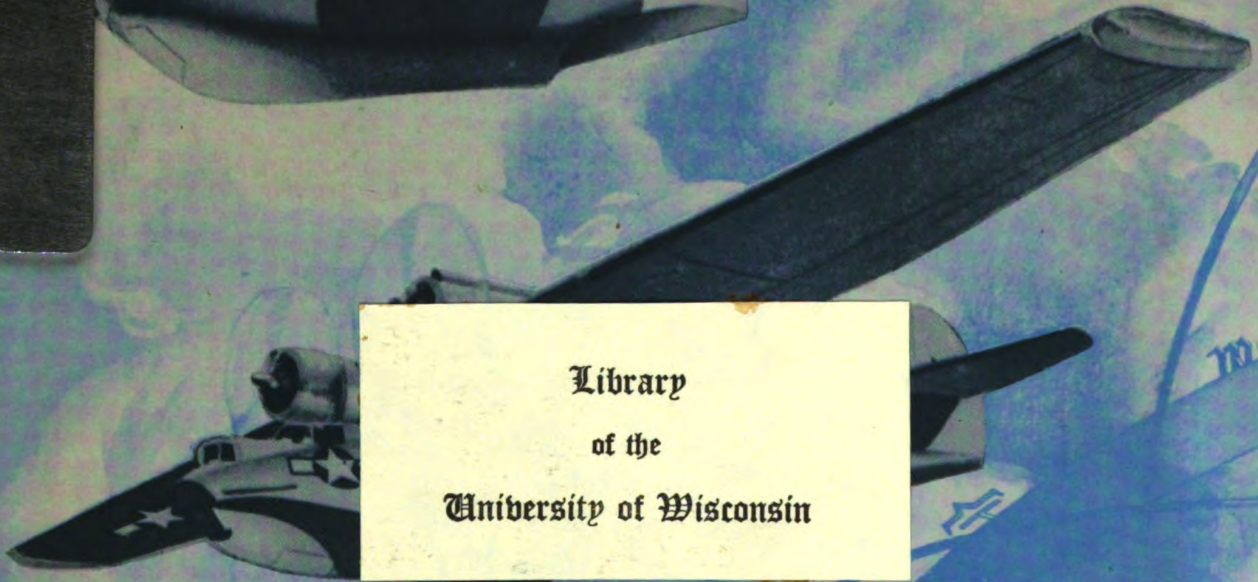


STS
.M432

AMERICAN AIR NAVIGATOR



Coronado



Catalina Amphibian

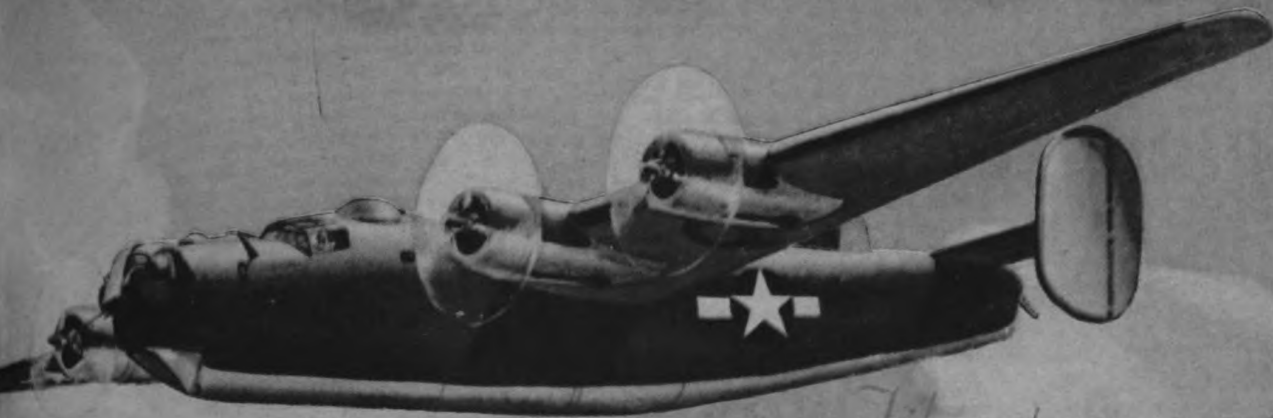


Vengeance



Liberator Express

Library
of the
University of Wisconsin



Liberator



Reliant



Sentinel "Flying Jeep"



Valiant



Crivello Flying B

AMERICAN AIR NAVIGATOR

ERRATA

- Page 60—after “for range reception . . . ,” add “in conjunction with homing using left-right indicator”
- Page 136—Problem Work No. 29—Date: “May 1, 1943.”
- Page 220—Problem Work No. 10—Answer No. 13 = $183^{\circ}/41$
D. R. REVIEW No. 1—Answer No. 6(a) TAS = 188 knots; Answer No. 6(b) = 191 knots. TAS.
- Page 221—Problem Work No. 14—Answer No. 4 = 52° ;
Answer No. 16 = 150° .
- Page 223—Problem Work No. 23—Answer No. 19, LHA = 122° E.



AMERICAN AIR NAVIGATOR

By
✓
CHARLES MATTINGLY
Chief Navigator
Consolidated Vultee Aircraft Corporation



Published by
CONSOLIDATED VULTEE AIRCRAFT CORPORATION
SAN DIEGO, CALIFORNIA

Copyright 1944
By Consolidated Vultee Aircraft Corporation
San Diego, California

Printed in
United States of America

573873

~~JUL 24 1942~~

573873

TABLE OF CONTENTS



CHAPTER 1	
EARTH AND CHARTS	1
CHAPTER 2	
INSTRUMENTS	13
CHAPTER 3	
DEAD RECKONING	29
CHAPTER 4	
RADIO NAVIGATION	53
CHAPTER 5	
CELESTIAL SPHERE	65
CHAPTER 6	
AIR ALMANAC	77
CHAPTER 7	
TIME	91
CHAPTER 8	
AIRCRAFT OCTANT	101
CHAPTER 9	
POSITION LINES	111
CHAPTER 10	
STARS AND THE WEATHER	139
CHAPTER 11	
SPECIAL PROCEDURES	155
CHAPTER 12	
YOU NAVIGATE TO HONOLULU	163
APPENDIX	189
GLOSSARY	211
PROBLEM ANSWERS	216
INDEX	223

Gift

The American Air Navigator is published by
Consolidated Vultee Aircraft Corporation
as a token of the organization's faith in the
brilliant future of global transportation.

PREFACE

"American Air Navigator" is intended primarily for the serious student of aerial navigation, and for those actually engaged in this interesting and vital profession. This book attempts to present air navigation as it really is today; it is strictly a training text for instructional purposes, and a reference source for professional navigators.

During twelve years of actual experience as a navigator, both surface and aerial, I have attempted to read carefully every book published on the subject of navigation and, in more recent years, books on aerial navigation. One characteristic seems common to most of them—an over-emphasis of the theoretical. Obviously, there is need for thorough coverage of theory, and many of these texts cover the subject as comprehensively as present limited knowledge of this science allows.

Acting on the conviction that there is need for a book which separates practical, applicable information from the vast field of theory, and which presents such information in compact form, I have attempted to collect into this volume all the data essential to successful aerial navigation as we know it today.

The fact that the book is laid out in outline form is no coincidence. Rather, it is the writer's belief that its outline form will make the volume an easy, intelligible teaching aid, and will permit rapid reference.

Appreciation is due here to the many navigators, both surface and aerial, who have assisted me in the compilation of this book. Special credit must be given Peter Selby, without whose help and encouragement this volume might never have reached publication. Likewise, David Hellyer gave valuable assistance in editing and correlating material in the book. For the high quality of the illustrations, thanks are due the Consolidated Vultee (San Diego Division) Service Illustration Section, under the direction of Charles Bundo, Jr.

C. D. M.

June 1, 1944
San Diego

INTRODUCTION

Half a dozen years ago, the professional aerial navigator was a newcomer to the aviation industry, and virtually unknown to the public. Non-stop flights across vast stretches of ocean still were regarded with awe, and few indeed had experienced the thrill of an aerial journey to Australia, India or London.

In 1940, Richard Archbold, renowned explorer and adventurer, accomplished one of the first extended, over-water flights "without incident" in a Consolidated Aircraft Corporation airplane, the PBY "GUBA." The flight of the Guba marked a milestone in the history of aerial navigation, and helped to blaze a trail which today is followed by hundreds of giant aircraft each month.

Pioneering with Archbold in this epic adventure were veteran pilot Russell Rogers, now director of Flight and Service for Consolidated Vultee Aircraft Corporation, and S. J. Barinka, superintendent of field operations for Convair at San Diego, who acted as the Guba's flight engineer. The experience these men gained on this pioneer flight, and on scores of later trips, helps make Consolidated Vultee's flight department today outstanding in the industry.

Forced by the second World War to regard the entire globe as a flying field, aviation today considers trans-oceanic flights little more than commonplace. At this moment, no spot on earth is more than 60 hours removed from your front door—thanks to aviation—and this "time-distance" rapidly is diminishing as the industry produces faster, longer-ranged aircraft.

So, just as yesterday he was a newcomer, today the aerial navigator is symbolic of the history-making strides taken by aviation since Pearl Harbor. His importance in the stirring drama of aviation stresses the fact that global air transportation is today a reality, where only yesterday it was a fond hope and dream.

In the "American Air Navigator," Charles Mattingly, Chief Navigator for Consolidated Vultee Aircraft Corporation, has recorded much of the valuable information and experience gained during his 12 years of navigating. Hundreds of Consolidated Vultee pilots, copilots, navigators, radiomen, and engineers have received instruction in aerial navigation under Mattingly's able tutelage, and the enviable safety record of Consolidated Vultee's over-water air transportation bears witness to the accuracy and thoroughness of this instruction.

Though "American Air Navigator" is published primarily as a textbook for training the corporation's flight personnel, it is fitting that Consolidated Vultee, as a leader and pioneer in the field of global transportation, should make this vital information available to all, in the interests of advancing world-wide air travel.

TOM M. GIRDLER,
Chairman of the Board,
Consolidated Vultee Aircraft Corporation,
San Diego, California.

AMERICAN AIR NAVIGATOR



☆ 1 ☆

EARTH AND CHARTS

AN INTIMATE, working knowledge of the earth's shape, size and geography is as essential to the well-informed aerial navigator as is a knowledge of aerodynamics to the successful pilot. It is not enough that the navigator know that the earth is round. He must understand what this spherical shape means in terms of his job; he must realize that the earth's very shape gives rise to many navigational problems which it will be his duty to solve.

The navigator must think of the entire earth as his workshop. In fact, the whole universe is his laboratory, for he must employ the stars and planets themselves in his daily work. No other calling in the aviation industry—and few in any industry—requires thinking and working in such universal concepts!

From the start, then, the navigator must prepare himself to learn a new language. He must break away from habits of thinking which have taught him to regard a neighboring city as "distant," for his work soon may make a 10,000-mile flight appear commonplace.

Realizing the necessity of thinking in global concepts, the navigator should study the following information earnestly and sincerely. If he adopts such an approach, he will find mere definitions will become living phrases to be added to his new vocabulary. A working familiarity with these terms, and a thorough understanding of their meaning, might some day mean the difference between life and death.

☆ 1 ☆

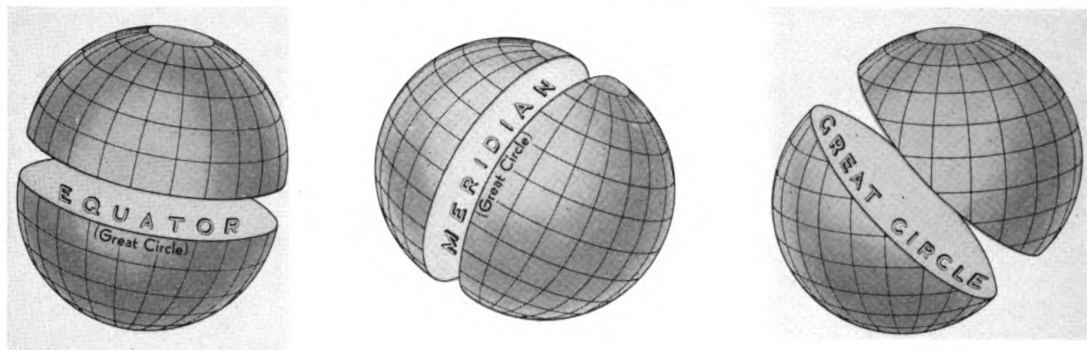


FIG. 1—GREAT CIRCLES

THE EARTH

Shape—The earth is an oblate spheroid; that is, a sphere slightly flattened at the poles. The polar diameter is about twenty-seven (27) miles less than the diameter at the equator. However, for purposes of navigation, the earth may be regarded as a sphere.

Size:

Polar diameter7,900 miles

Diameter at equator.....7,927 miles

TERMS

A Sphere is a body bounded by a surface, all points of which are equidistant from a point within, called the center.

A Great Circle is an imaginary circle on the earth's surface, the plane of which passes through the center of the earth. (Figure 1)

A Small Circle is an imaginary circle on the earth's surface, the plane of which does *not* pass through the center of the earth. (Figure 2)

Meridians are great circles passing through the earth's poles. (Figure 3)

The Prime Meridian is the meridian used as reference line for the measurement of longitude. Known also as the Greenwich meridian because it passes through the Naval Observatory at Greenwich, England, it was selected arbitrarily as a reference line and is so employed by navigators of most countries of the world. (Figure 3)

Longitude is angular distance East or West of the prime meridian, measured in degrees of arc from 0°-180°. (Figure 3)

The Equator is a great circle on the earth's surface lying midway between the poles. It is used as the prime reference line for the measurement of latitude. (Figure 4)

Parallels of Latitude are divisions of latitude parallel to the equator. They are small circles on the earth's surface. (Figure 4)

Latitude is angular distance North and South of the equator measured in degrees of arc from 0°-90°. (Figure 4)

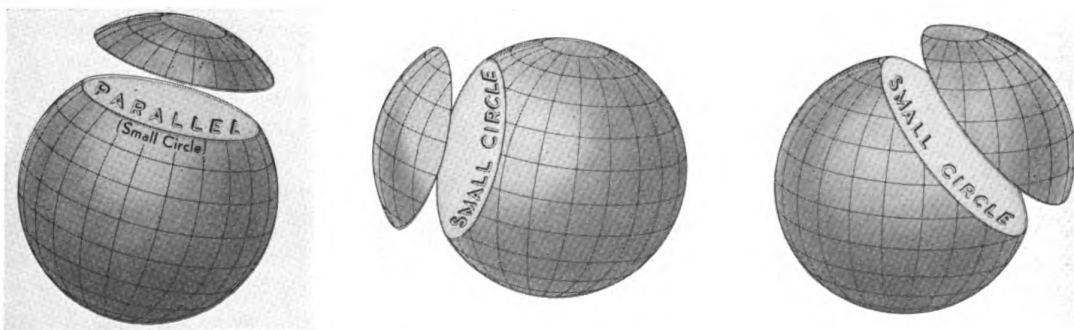


FIG. 2—SMALL CIRCLES

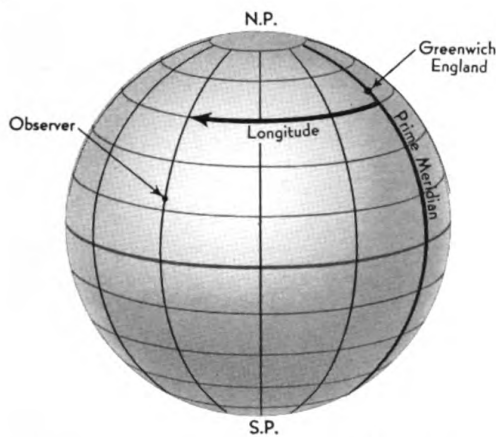


FIG. 3—MERIDIANS, PRIME MERIDIAN AND LONGITUDE

A **Statute Mile** is a unit of distance equal to five thousand, two hundred and eighty (5,280) feet, and is used only in over-land flying.

A **Nautical Mile** is a unit of distance equal to six thousand, eighty (6,080) feet, and is used in *all ocean flying*.

A **Knot** is a unit of speed and is equal to one nautical mile per hour.

A **Great Circle Track** is the path made good over the ground when flying a great circle course. It is the shortest distance between any two places on the earth's surface. (Figure 5)

Due to the convergence of the meridians toward the poles, the great circle track be-

tween two places will cross each meridian at a different angle, except at the equator or when the track coincides with a meridian.

A **Rhumb Line** is a line (or course) which intersects all meridians at the same angle. (Figure 5) Because it appears as a straight line on a Mercator chart (Figure 6) it is the course line most used in ocean aerial navigation. Actually, on the surface of the earth, it is a curved line which is sometimes known as a loxodromic curve, or equiangular spiral. On the equator, or along a meridian, the rhumb line coincides with the great circle track.

CHARTS

A **Chart**, or a map, is a graphic representation of a portion of the earth's surface laid out upon a plane surface.

When man first began to draw maps, the world he knew was very small indeed. His ideas about the world around him, like the maps he drew, were crude and incomplete.

At first his map-making experiments were, in all probability, attempts to satisfy a natural curiosity about his surroundings. In time, however, maps became vital necessities. Caravans to the East used them for guidance through strange lands. Ship captains felt more secure in navigating from port to port if the best available maps were aboard.

Then, as mariners gained more courage, and ventured farther and farther from sight of land, the need arose for a new type of map, a

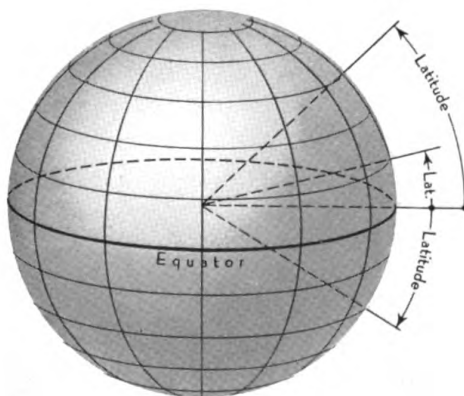


FIG. 4—EQUATOR, LATITUDE AND PARALLELS OF LATITUDE

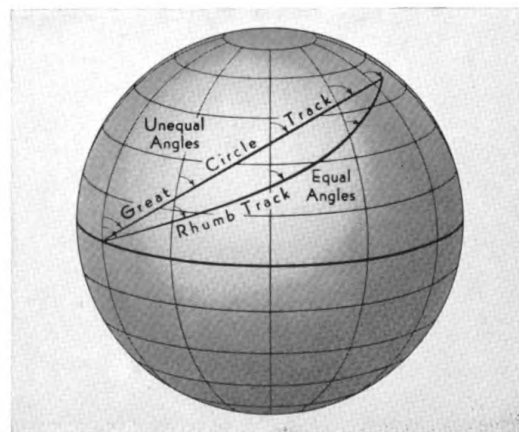


FIG. 5—GREAT CIRCLE AND RHUMB TRACK ON EARTH'S SURFACE

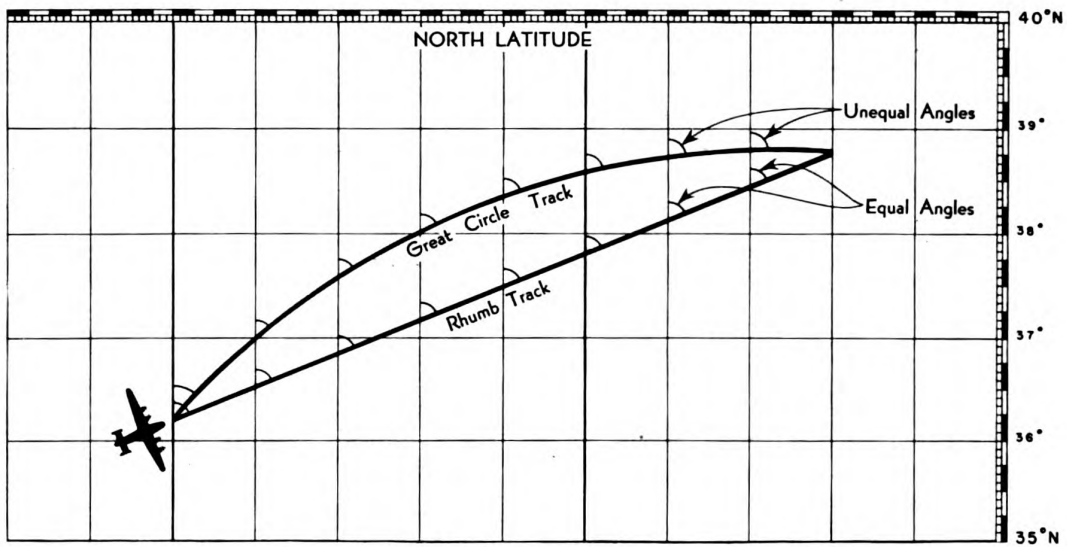


FIG. 6—GREAT CIRCLE AND RHUMB TRACK ON MERCATOR CHART

map which would represent large areas of water and adjacent or included land in some graphic manner. These maps, which dealt primarily with water areas, came to be known as charts.

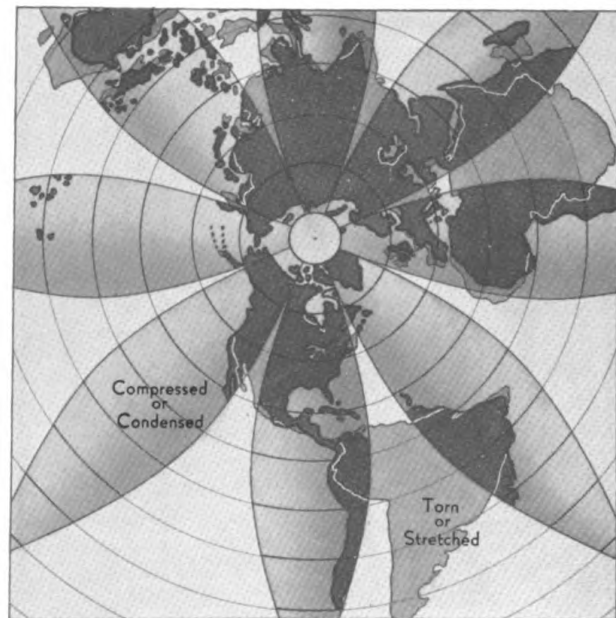
Thus arose the distinction between maps

and charts: a map is primarily concerned with the land, while a chart supplies information mostly about water and land bounded by water. This distinction makes the chart especially valuable to the navigator, for whom—in fact—the chart was designed.



Sphere

FIGS. 7-8
SURFACE FEATURES OF EARTH
DISTORTED ON PLANE SURFACE



Sphere Distorted into Plane

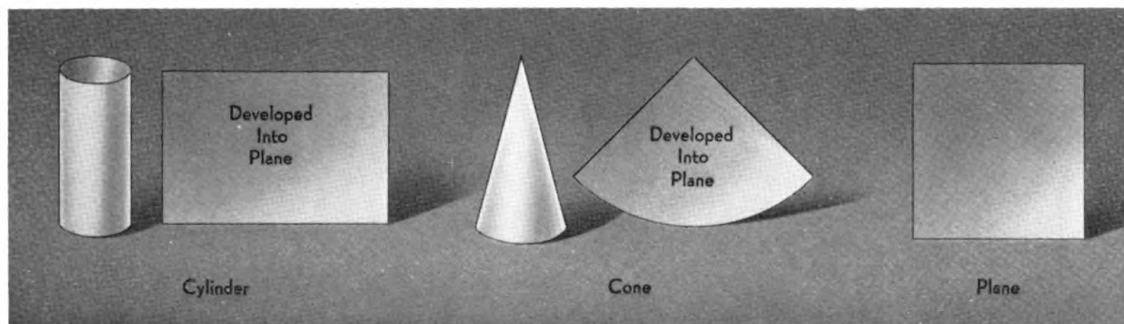


FIG. 9

FIG. 10

FIG. 11

DEVELOPMENT OF CHARTS

The Ideal Chart would be one which represented graphically the entire surface of the earth, exactly to scale, on a flat sheet of paper. However, the earth's surface bounds a sphere (Figure 7) which cannot be developed upon a plane surface for the same reason that a section of orange peel cannot be made to lie flat unless it is torn, stretched, or compressed (Figure 8). Therefore, a system of map making has been devised which permits the surface features of the earth to be projected upon surfaces capable of development into a plane.

Three Surfaces which may be developed into a plane surface are:

- Cylinder (Figure 9)
- Cone (Figure 10)
- Plane (Figure 11)

Chart Projections—Charts are transferred to developable surfaces by a method known as "projection." By projection is meant to project, from the center of the earth (theoretically) the surface features of the area to be charted upon a developable surface held tangent (touching) or nearly tangent to the earth's surface. Actually, the chart is made by mathematical computation. The principle of the projection may best be illustrated, however, by imagining the eye of the observer at the center of the earth.

A certain amount of distortion of surface features will remain even when charts are made by projection. Each of the three developable surfaces will produce a chart with varying degrees of distortion. For this reason, many different types of projection have been devised, each intended to meet a specific need by embodying certain desirable characteristics while

eliminating, so far as possible, characteristics which are less desirable.

The Four Principal Projections most commonly used by the aerial navigator are:

- Mercator
- Lambert
- Gnomonic
- Polyconic

MERCATOR CHART

Importance—Because its particular features make it best suited for ocean flying, the Mercator chart is the most important chart used in ocean navigation.

Description—The Mercator chart principle involves depicting surface features of the earth as they would appear if projected onto a cylinder held tangent to the earth at the equator (Figure 12). It is not a true geometric projection, however, since the projection point (eye) is considered to move along the earth's axis toward the poles in order that the distances between parallels of latitude will increase in the same proportion as the distances between meridians at the same latitude (Figure 13). Thus, the surface features of the earth when projected on the cylinder retain their shape, although they become proportionately larger as the latitude increases. The actual magnifying factor is equal to the secant of the latitude (a mathematical equation).

Desirable Features—The desirable features of the Mercator projection arise primarily from the fact that all meridians of longitude and parallels of latitude are represented as parallel straight lines, perpendicular to each other. As a result, positions are easily plotted on it, courses and bearings are straight lines easily measured, and charts of the same scale

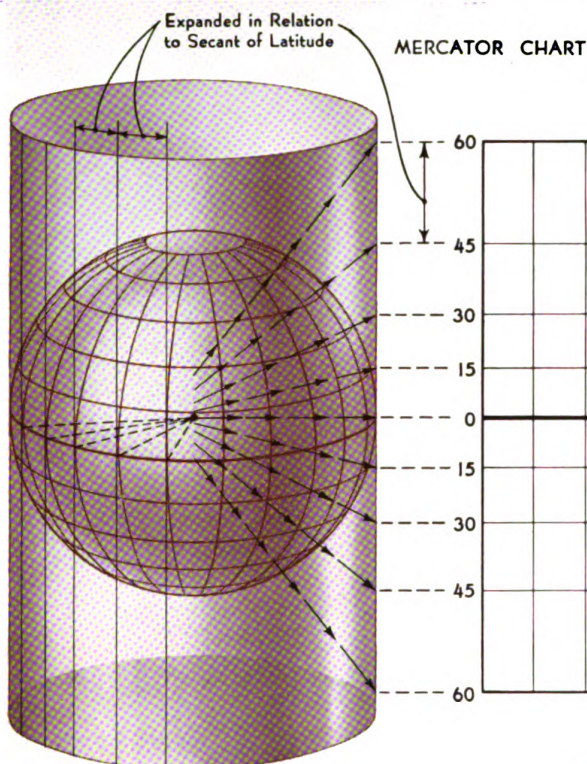


FIG. 12 Projection

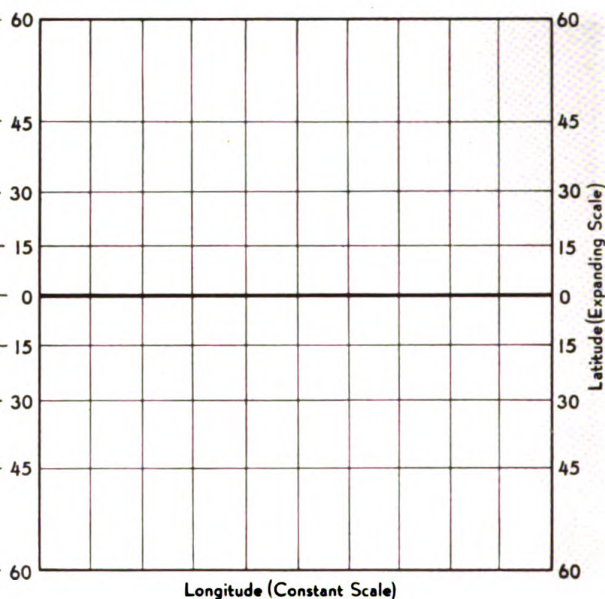


FIG. 13 Development

may be joined together accurately. Other advantages are a distance scale in nautical miles, simplicity of graphic construction, and the important fact that a rhumb line appears as a straight line (Figure 14).

Undesirable Features—Perhaps the least desirable feature of this projection is the expanding latitude, which results in a non-uniform distance scale as well as in great distortion in high latitudes. Undesirable also is the fact that a great circle track—actually the shortest distance between two points—appears as a curved line, hence radio bearings must be converted to Mercator bearings (rhumb lines) for plotting.

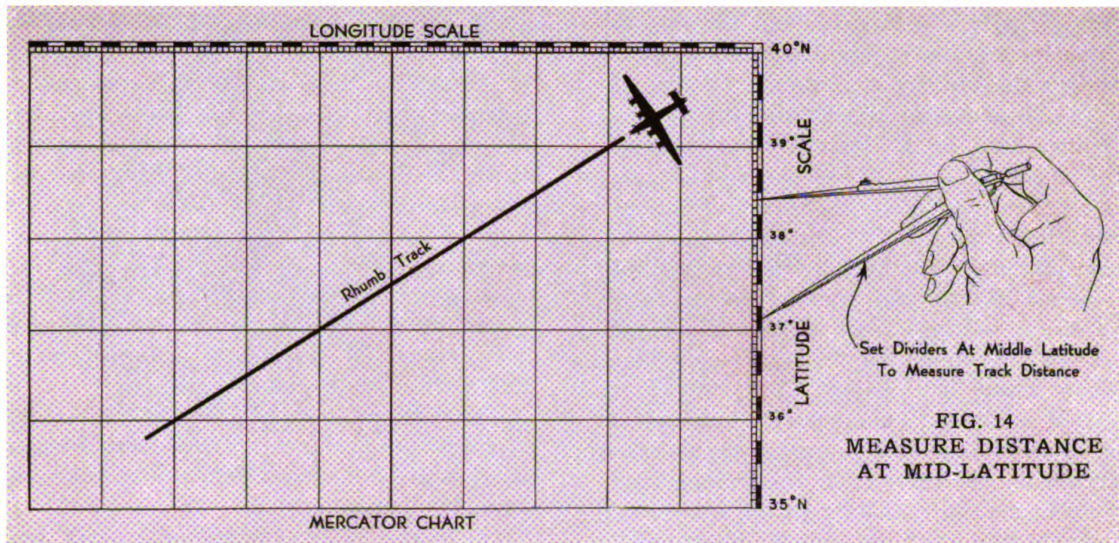
Measuring Distances—Since one minute of latitude equals one nautical mile, the latitude scale (Figure 14) is used to measure distance. However, due to the fact that latitude expands, it is necessary to *measure distance at the mid-latitude of the track being measured*. (Figure 14)

Graphic Construction—This method of

construction, by which the navigator may make his own chart, does not result in an exactly correct Mercator chart, although it is sufficiently accurate for obtaining a fix or plotting a position if the North and South extent is limited to 6° and latitude does not exceed 60° .

These limitations are, however, of a very general nature. Actually, for purposes of air navigation, sufficient accuracy can be obtained even when much greater extents of latitude are employed. The nearer the area to be covered by the chart is to the equator, the less distortion exists, hence the greater the area that can be included while retaining reasonable accuracy.

Graphic construction is a quick and easy method of making a chart, and charts so constructed may be used throughout the remainder of this book wherever needed. The reader is especially urged to utilize this method of construction to develop charts which he will need in connection with the problem work. Charts so constructed also can be used for actual navigation in the event the navigator should



find himself without the proper plotting chart.

Example: It is desired to plot a fix which is approximately 25° North latitude, and 170° West longitude. (Refer to Figure 15)

a. Draw a horizontal line near lower edge of paper.

b. Erect a perpendicular to this line near left edge of paper.

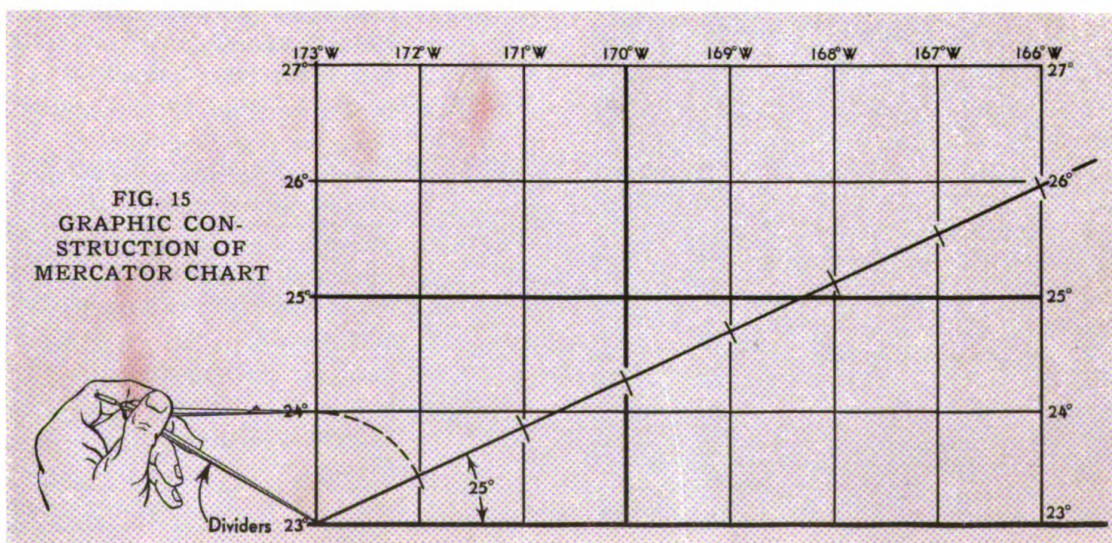
c. At the point of intersection of these two lines, draw a line making an angle of 25° with the horizontal line.

d. Using any convenient scale, set the dividers equal to 60 units (each a "minute")

and mark off 60-unit intervals on the sloping line, and also on the perpendicular.

e. Through these points draw lines parallel to the horizontal and perpendicular lines, and number them for latitude and longitude as per problem, making the mid-latitude 25° and the mid-longitude 170° .

f. Through the use of the 60-unit latitude scale, the latitude of any position on the chart may be quickly found, and distances easily measured. By setting the zero and 60-unit points of the reference scale to coincide with the meridians on either side of the point



whose longitude is desired, and aligning the scale edge to pass through the point, longitude can be read directly off the scale. (Figure 15)

Construction by Meridional Parts—Since the meridians on a Mercator chart are parallel, it can be shown that the longitude must have been expanded by the secant of the latitude. Hence, in order for the projection to remain conformal, the latitude must be expanded similarly. Therefore, in constructing an accurate Mercator chart, each minute of latitude has to be multiplied by the secant of the latitude at which it is located. But a minute of latitude is equal to a minute of longitude at the equator, and since the longitude does not vary on the Mercator chart, it is customary to establish the chart scale by letting a minute of longitude equal any convenient unit of length. The length of any minute of latitude then will be equal to the length of a minute of longitude multiplied by the secant of the latitude. Thus the length of a minute of latitude at 30° North is equal to a length of a minute of longitude times the secant of 30°. As an illustration, if 1' longitude = 1/16 inch, then 1' latitude at 30° N (or S) = 1/16 inch x secant 30°.

These expanded minutes of latitude, from the equator to the poles, are known as meridional parts, and the table of meridional parts (Table No. 5, Bowditch*) gives the sum of the

meridional parts from zero, at the equator, to the desired latitude.

Example: It is desired to construct a Mercator chart based on the following information: latitude 40° to 43° North, longitude 70° to 73° West, scale 1½ inch equals one degree of longitude at the equator, or 1/40 inch equals one minute of longitude at equator. (Refer to Figure 16)

a. Draw parallel North and South lines 1½ inch apart. Number these lines even degrees of longitude: 70°, 71°, 72°, 73° from East to West. These are the meridians.

b. Near the lower edge of the work sheet, erect a perpendicular to the meridians. Assume this line to be the principal parallel and number it 40° North latitude.

c. In the table of meridional parts, find meridional parts of principal parallel and all other parallels of latitude.

d. Compute separately the lengths of the meridians between each parallel of latitude and the principal parallel by multiplying the difference of meridional parts by the scale unit representing one minute of longitude:

2607.6 Meridional parts latitude 40°
(principal parallel)

2686.2 Meridional parts latitude 41°

78.6 Difference of meridional parts
x 1/40" Scale for 1' of longitude

1.97" Length of meridian between
latitude 40° and 41°

2607.6 Meridional parts latitude 40°
(principal parallel)

2766.0 Meridional parts latitude 42°

158.4 Difference of meridional parts
x 1/40" Scale for 1' of longitude

3.96" Length of meridian between
latitude 40° and 42°

2607.6 Meridional parts latitude 40°
(principal parallel)

2847.1 Meridional parts latitude 43°

239.5 Difference of meridional parts
x 1/40" Scale for 1' of longitude

5.987" Length of meridian between
latitude 40° and 43°

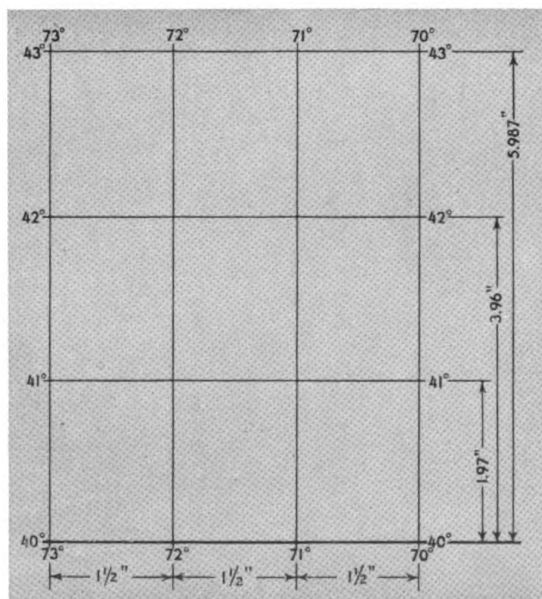


FIG. 16

*BOWDITCH, or Hydrographic Office Publication No. 9. Originally published by Nathaniel Bowditch in 1802, the volume is regarded as the "Bible" of surface navigation by most ocean navigators.

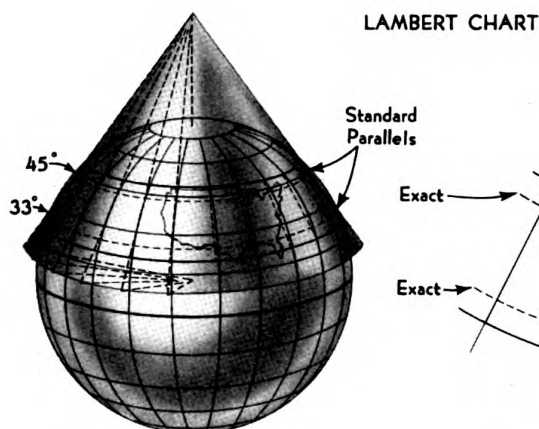


FIG. 17 Projection

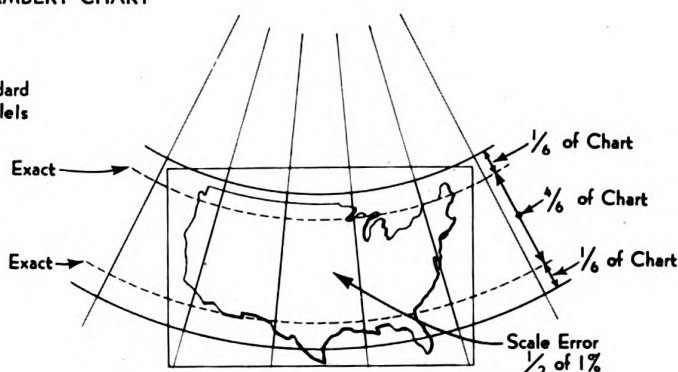


FIG. 18 Development

e. From principal latitude 40° measure off the computed meridian lengths (1.97"; 3.96"; 5.987"), and through these points draw lines parallel to the principal latitude 40° . Number these toward the pole 41° , 42° , 43° respectively.

Note: Minutes of longitude are found by dividing the degrees into 60 equal parts. Minutes of latitude must be computed from the principal parallel, the same as degrees, if extreme accuracy is desired.

LAMBERT CHART

Description—In the Lambert chart, the surface features of the earth are projected upon a right circular cone which is made to intersect the earth's surface at two parallels of latitude, called standard parallels of the chart (Figure 17). Along the standard parallels the scale is exact, but between them there is a slight dis-

tortion due to compressing or condensing of the features (Figure 18) and beyond them expansion or stretching of features occurs. Hence, in order to maintain as nearly as possible the same limits of error throughout the chart, the standard parallels are so chosen that one-sixth of the area to be projected is above, and one-sixth below, these parallels.

This chart was developed by Lambert in 1772, but it was not until the first World War that its possibilities were fully realized. At that time, the Allies used it for military maps because it afforded a maximum of accuracy in measuring directions and distances where the difference of latitude was not too great. Because of the accuracy with which it portrays topographical features, and because of the fact that, for all practical purposes, radio bearings may be drawn as straight lines, the Lambert

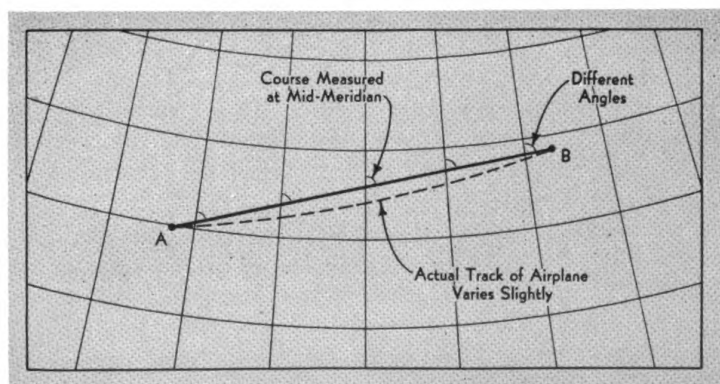


FIG. 19—MEASURING COURSE
ON LAMBERT CHART

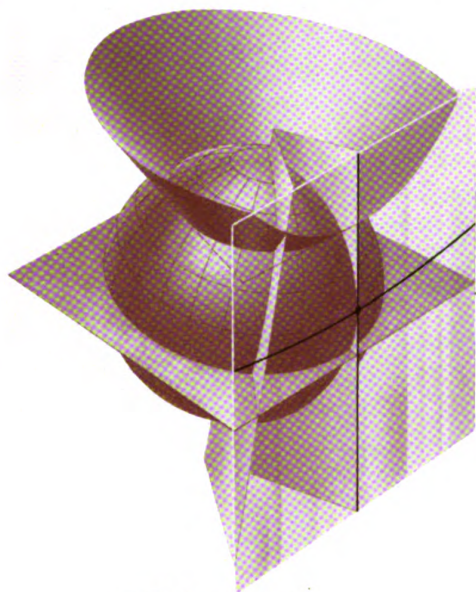


FIG. 20 Projection

GNOMONIC CHART

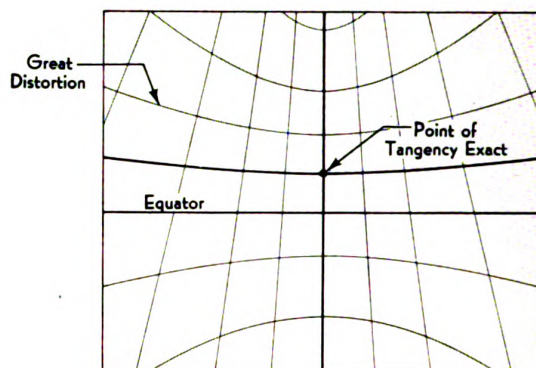


FIG. 21 Development

projection was selected by the Coast and Geodetic Survey for the development of aeronautical charts of the United States and Alaska.

Desirable Features—The scale is so nearly exact that distances may be measured directly, as though the scale were constant. A straight line very closely approximates a great circle, hence it may be considered the shortest route. This latter fact makes the chart extremely useful for radio navigation, as it is possible to plot radio bearings directly. Another advantage is that any number of charts may be joined together in any direction with perfect junctions.

Undesirable Features—Since on this projection the meridians converge, a straight course line will cross each meridian at a different angle (Figure 19). However, instead of flying a constantly changing course in order to follow a direct line, the practice generally is to measure the course at the mid-meridian. Though this method of measurement causes the actual track of the aircraft to differ slightly from the course plotted, it enables the pilot to hold a constant heading (Figure 19).

Plotting of position also is more difficult on this type of projection because the meridians and parallels of latitude are not shown as

parallel straight lines as on the Mercator projection.

GNOMONIC CHART

Description—The gnomonic chart is developed by projecting a portion of the earth's surface onto a plane held tangent to the earth at a point (Figure 20). At this point of tangency, the surface features will be exact, but distortion is great in all directions away from this point (Figure 21).

Desirable Features—This chart was developed primarily in order to show true great circle tracks as straight lines. Its principal use is in measuring the shortest distance between two points and in providing a graphic method of plotting the great circle track. The great circle track, drawn as a straight line on the gnomonic chart, is divided up into a series of chords, and the coordinates of these chords may then be re-plotted upon the Mercator chart for actual navigation.

Undesirable Features—Great distortion of the surface features in all areas except at the one point of tangency, and great difficulty in plotting positions because of a non-uniform scale are the principal drawbacks to this type of chart.

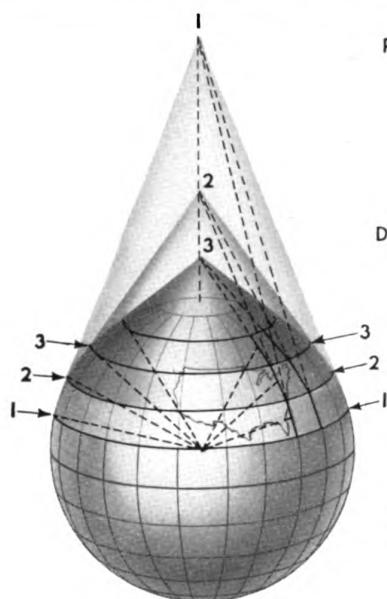


FIG. 22 Projection

POLYCONIC CHART

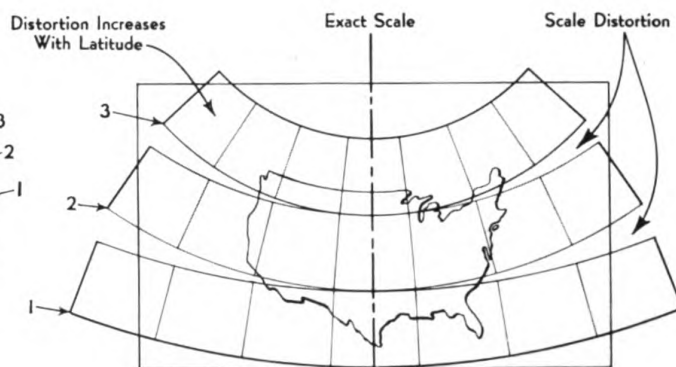


FIG. 23 Development

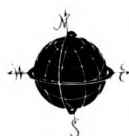
POLYCONIC CHART

Description—The polyconic chart derives its name from its method of development, which involves projecting from the center of the earth a portion of the surface features onto a series of cones, each tangent to the earth's surface at a different parallel of latitude, but having a common axis (Figure 22). This chart (Figure 23) is useful for small areas where accuracy of topographical features is most important, since it incorporates even less distortion error than the Lambert chart.

Desirable Features—The scale being almost exact near the central meridian, even for

large North and South areas, it is a highly desirable projection for pilotage charts, such as harbor and small coast line charts. It is used also by U. S. Army Engineers for fire control and tactical maps.

Undesirable Features—Use of the chart for practical navigation is limited to small areas for several reasons. The type of distortion inherent in this projection limits its use to charts of wide latitude and narrow longitude. On larger charts, positions are difficult to plot and courses are hard to measure because the meridians are curved and the parallels of latitude do not intersect them at right angles.



PROBLEM WORK

- No. 1 Make drawings showing principle of the Mercator, Lambert, and gnomonic projections. List desirable and undesirable features of each.
- No. 2 Construct graphically a Mercator chart for an area extending 2° North and South of latitude 25° N., and 2° East and West of longitude 127° W.



☆ 2 ☆

INSTRUMENTS

JUST as the entire globe is the aerial navigator's workshop, so his aircraft's navigational instruments are his most important tools. Those to be discussed in this chapter are the instruments actually employed by the long-range navigator for extended, over-water flights. Included are:

- Compass
- Altimeter
- Airspeed Indicator
- Thermometer
- Chronometer
- Drift Meter

Not included in this chapter are discussions

of the radio compass and the aircraft octant. Each of these is sufficiently important to merit detailed treatment in later chapters.

No attempt is made to describe the construction of instruments discussed in this chapter except where such descriptions are vital to an understanding of their actual use. Detailed mechanical descriptions are, for the most part, left to manufacturers' manuals, which are available to those interested in the construction and servicing of these units. Furthermore, all aircraft instruments are subject to such change that any text which attempted to describe them in detail might well be rendered obsolete overnight.

The stress in both text and illustrations, therefore, is laid on principle and use of these navigational "tools," rather than on their physical make-up.

COMPASS

Description — Both functionally and historically, the compass is first and foremost among all navigational instruments. Briefly, the compass is an instrument which points in a constant direction regardless of the aircraft's movements, and from which, by reference to a graduated card, the aircraft's direction of flight may be determined. The earliest-type compass probably was discovered when ancient navigators learned that a piece of lodestone or a magnetic needle, when floated on a cork in a bowl of water, would point in a constant direction, thus enabling them to navigate their sailing ships without reference to landmarks. Refinements on this early compass were made through the centuries, and the process of refinement continues today.

In recent years several other scientific principles have been utilized in attempts to develop a compass which would be even more reliable and constant than the magnetic compass. Among principles utilized are those of the gyro, and the movements of celestial bodies.

1. Gyro Compass—The gyro compass has proved highly successful aboard surface ships, but so far no model light and rugged enough for aircraft use has been developed. It has great future possibilities because it indicates the true North pole, and is, therefore, the ideal in compass construction. Its directive force is obtained mechanically by means of a rotating gyro which automatically aligns its axis with the true axis of the earth. The principle of this compass is derived from Foucault's Law, which states that a spinning body (the gyroscope) tends to swing around so as to place its axis parallel to the axis of any impressed force (the force in this case being the earth spinning on its axis).

2. Sun Compass — The sun compass is a special instrument designed for use in polar regions where the magnetic compass may be uncertain. It indicates direction by means of the sun, utilizing the principle of the sun dial. It has to be set continuously to local apparent

time, and the sun's shadow indicates direction of flight.

MAGNETIC COMPASS

Description — Though many attempts are being made to develop compasses operating on various scientific principles, the magnetic compass, greatly refined since the ancient navigator's floating lodestone, is still the most dependable and least liable to mechanical failure. For this reason, and also because it is the most fundamental of all navigational instruments, the magnetic compass should be thoroughly understood by the navigator.

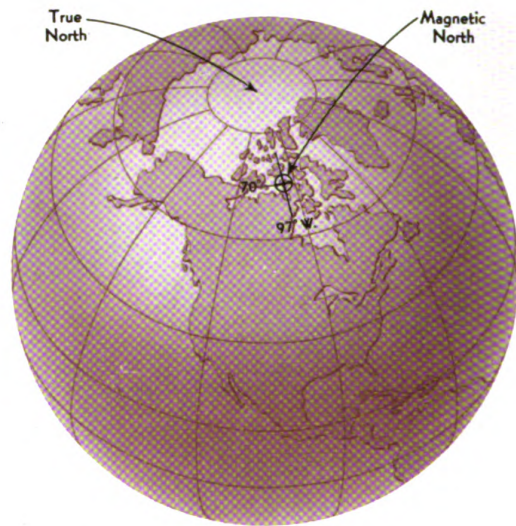


FIG 24—LOCATION OF MAGNETIC NORTH

The Directive Force of the magnetic compass is the magnetic field surrounding the earth. The magnetic compass, therefore, does not point to the true North pole but towards the magnetic North pole, which is approximately 1200 miles away from the true North pole, just north of Hudson Bay in about latitude 70° North, longitude 97° West (Figure 24). This difference between true North and magnetic North causes an error in the compass reading known as variation, which will be discussed thoroughly later on.

Principle — From elementary physics one learns that when two magnets are brought together, like poles repel and unlike poles attract

each other. Also, that the earth itself exhibits the properties of a great magnet, having North and South magnetic poles (Figure 25). Therefore, when a magnet is fully supported in space and is free to rotate in a horizontal plane, it tends to align itself with the earth's magnetic field, its ends pointing toward the North and South magnetic poles.

Dip—Notice in Figure 25 that the magnetic lines of force are parallel to the earth's surface at the magnetic equator, but that on either side of the equator they dip toward the poles, becoming vertical at the poles themselves. Because of this fact, the compass magnets also dip toward the poles appreciably in higher latitudes. For this reason the magnetic compass is not reliable in polar regions.

The Basic Construction of all magnetic compasses is similar, consisting of a card, graduated in degrees, to which is attached a magnet, or group of magnets, parallel to the North and South axis of the card. The whole card assembly, freely supported on a pivot, is contained in a non-magnetic bowl.

There Are Many Types of magnetic compass which have been developed for aviation and marine use, each designed for best performance under different conditions, but all constructed on the same basic principle. Two magnetic compasses, of different types, usually are installed in airplanes designed for long-range flying. They are:

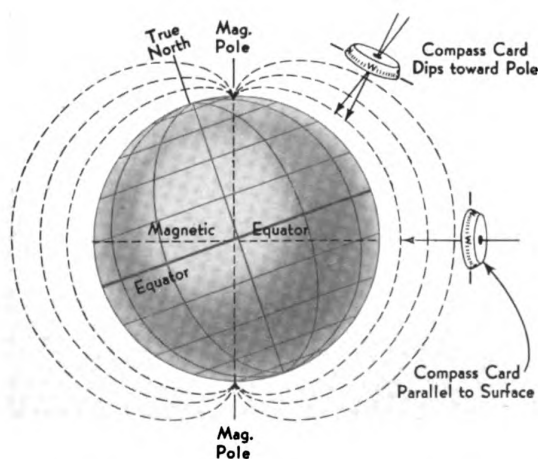


FIG. 25—EARTH'S MAGNETIC FIELD



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

FIG. 26—APERIODIC COMPASS

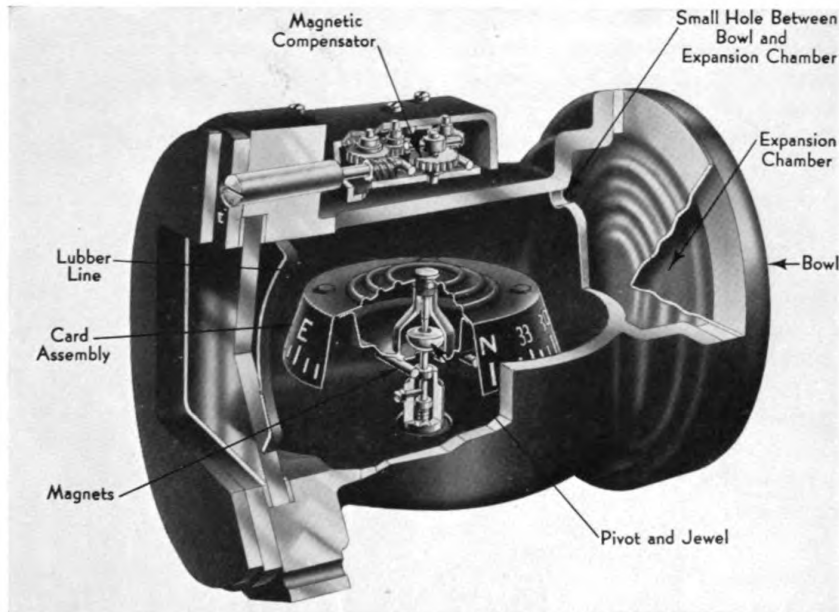
1. Aperiodic compass (the navigator's compass) (Figure 26).
2. Pilot's compass (Figure 27).

The aperiodic compass was designed especially for aerial navigation. It is easy to read and is so constructed that the compass card, when deflected from a position of equilibrium, will return to its heading slowly and positively and will not oscillate about the point of reading.

The Principal Parts of the magnetic compass (Figure 27) are the following:

1. **The Compass Card** usually is graduated in degrees. The four cardinal points (North, East, South and West) are lettered N, E, S, W. Some compasses, such as the aperiodic, show 1° graduations, but many of the pilots' type have only 5° graduations with every 30° numbered. The pilots' type compass is viewed from the side opposite its heading. The N marking, if viewed from the top, would actually be on the South end of the card. Because this compass is viewed "in reverse," the degree markings are backwards. Therefore, a change of heading causes the card to appear as though it were turning in the wrong direction.

2. **The Magnetic Elements** which supply the directive force to the card usually consist of a small bundle of hardened steel, magnetized needles. Magnetized needles are used because their magnetic influence is stronger than that of a single magnet. Two or more of these ele-



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

FIG. 27—PILOTS' TYPE MAGNETIC COMPASS

ments are secured to the underside of the compass card parallel to the North and South axis.

3. Pivot and Jewel—The pivot, made of hardened steel, balances the card and magnets on a jeweled support, usually a sapphire. This arrangement reduces friction and wear so that the card and magnets are as freely suspended as possible. A shock-absorbing spring keeps the pivot on the jewel and also insures maximum protection from vibration.

4. The Bowl is cylindrical. It is constructed of non-magnetic material to provide a housing for the card assembly and a container for the dampening fluid.

5. The Lubber's Line is usually a thin wire fixed to the center line of the bowl just clear of the freely-moving card assembly. Since the card always points in the direction of magnetic North, when the bowl is turned (as the airplane turns) the lubber's line, fixed to it, also turns and so indicates the amount of turn on the graduated card.

6. The Dampening Fluid is colorless, acid-free kerosene which completely fills the bowl. Its purpose is to dampen excessive oscillations of the card assembly, and to reduce shock, vibration, and friction of the pivot by supporting most of the weight of the card assembly. It

also prevents corrosion and keeps the jewel washed clean of foreign matter.

7. The Expansion Chamber is either a thin metal diaphragm, located in the liquid, or a hollow chamber in the top of the bowl, which keeps the bowl completely filled at all times by balancing the expansion of the liquid due to temperature changes.

8. Compensating Magnets usually consist of two sets of screw-adjustable magnets located either above or directly beneath the compass card. Both sets of magnets are arranged in the horizontal plane, but one set is placed so that it will lie on the fore and aft axis of the airplane, and the other so that it will lie on the athwartship axis. These magnets set up a compensating field which, when properly adjusted, greatly reduces the error known as *deviation*. Deviation is caused by local disturbing magnetic forces inherent in the aircraft itself, such as those created by engines, ferrous metals, and electrical circuits.

Location and Installation—The magnetic compass should be located where it may be easily read, but as far removed as possible from all sources of artificial magnetic disturbance. All direct current wires nearby should be closely twisted or they will set up a magnetic

field. The compass should be installed so that the planes of the lubber's line and the card pivot are vertical and parallel to the fore and aft axis of the aircraft. When so installed, the lubber's line will indicate the correct heading on the compass card, and the card assembly itself will be level in normal flight.

Compensation—To compensate a compass means to remove as completely as possible all error due to local disturbing magnetic forces existing in the aircraft. The procedure follows:

a. Before starting to compensate, place the airplane on some part of the field free from any artificial magnetic disturbance, such as power lines, pipes and steel structures.

b. Determine the correct magnetic heading of the airplane. There are two recognized methods of doing this:

By reference to a magnetic compass rose.
By means of a pelorus.

(1) **By Compass Rose.** This is the easiest and most common method. The magnetic compass rose is a large compass card laid out on the ground to indicate correct magnetic directions at every 15° or 30° throughout the 360° , starting at magnetic North. It is large enough so that the aircraft being compensated may be accurately headed on each of these 15° intervals.

(2) **By Pelorus.** The pelorus is a dummy compass card upon which sighting vanes may be arranged to take bearings on distant objects. These distant objects may either be objects on the earth, or celestial bodies (usually the sun). For compass compensation the magnetic bearing of the object to be used is first determined. The airplane may then be directed on any correct magnetic heading by reference to the pre-determined bearing. Knowing this method is of advantage because it may be used in the air to check compass error while flying a constant heading.

To continue compensation procedure:

c. Center compensating magnets, or, if they are the removable type, remove them.

d. Simulate all conditions of normal cruising operation and reduce to a minimum any deviation error caused by electro-magnetic influences (by shifting position of, and twisting together, direct current wires). With tail elevated and engines running, turn on and off the radio

receiver, radio transmitter, navigation lights, panel lights, Pitot-static tube heater, and generator. Check position of head sets, control positions and engine operation. Check individually and in practical combinations, on every 30° heading, the effect on the compass of these various influences. Any deviation remaining should not exceed 15° , and the ideal would be 0° . The compass is now ready for compensation.

e. Adjust compensating magnets to reduce remaining deviation as much as possible by the following procedure:

(a) Head the aircraft correct magnetic North. Adjust athwartship magnet (marked "N.S."), with a non-magnetic screwdriver, so the compass reads "N."

(b) Head the aircraft correct magnetic East. Adjust fore and aft magnet (marked "E.W.") so compass reads "E."

(c) Head the aircraft correct magnetic South. Adjust "N.S." magnet to *reduce by one-half* any error remaining on this heading.

(d) Head the aircraft correct magnetic West. Adjust "E.W." magnet to *reduce by one-half* any error remaining on this heading.

The deviation error in the compass now has been corrected for as much as possible, and the remaining error will have to be recorded and applied to all headings flown.

Calibration—Calibration is the determination and recording of existing deviation error, for use in flight. The amount of deviation error remaining in the compass on each compass heading is determined by swinging the aircraft through a complete circle and recording the error on each 15° heading. The error also should be recorded for actual flight conditions with radio and lights, on and off. With this information a curve of the deviation error is plotted which shows the amount of correction to be applied to any heading. A calibration graph is shown in Figure 28.

As is apparent from the above discussion, there are two basic errors in all magnetic compasses, namely, *variation* and *deviation*.

Variation—Variation is the angular difference, measured at the airplane, between *true North* and *magnetic North* (Figure 29).

The amount of variation error affecting a magnetic compass generally ranges from 0° to

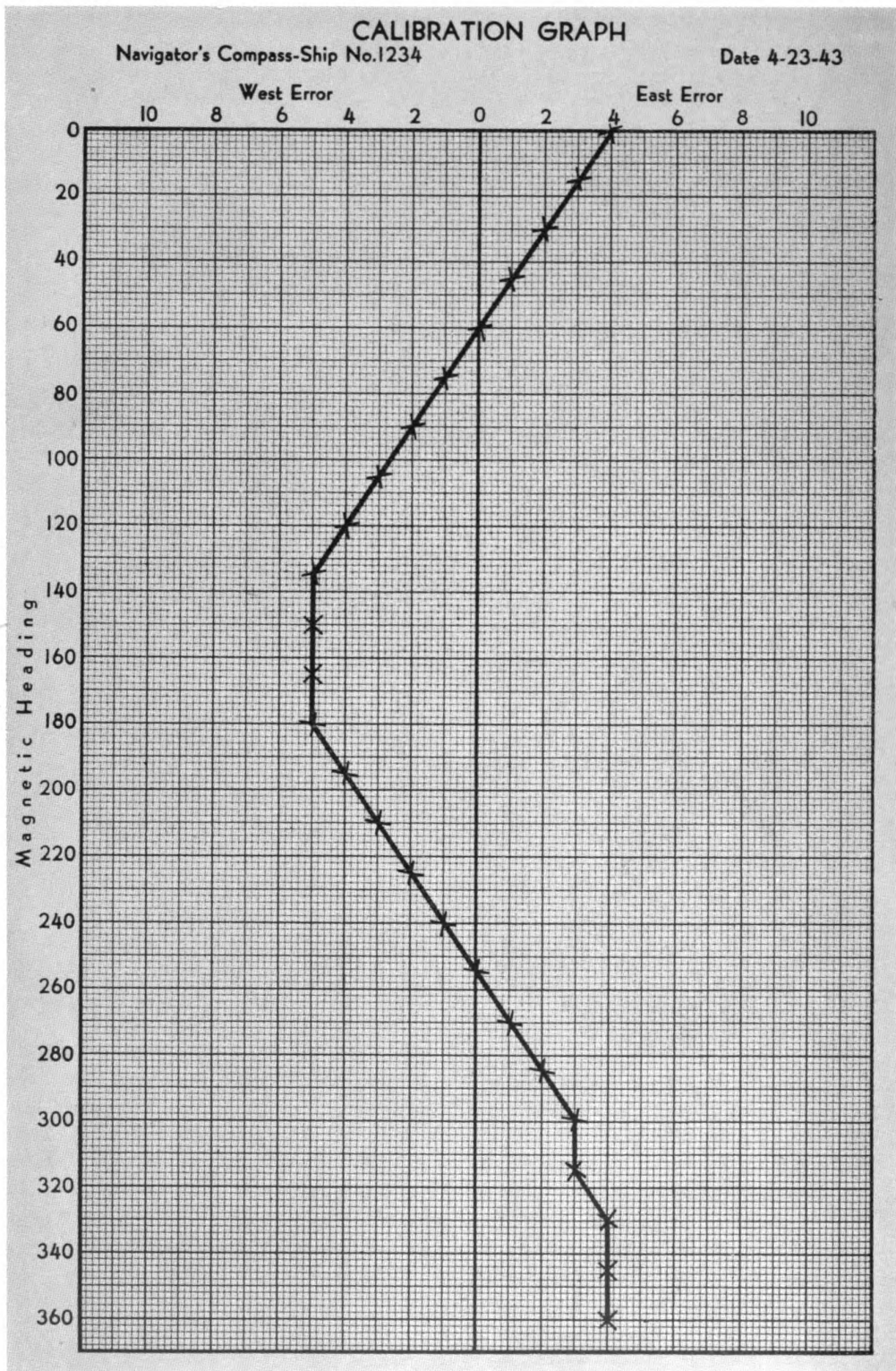


FIG. 28

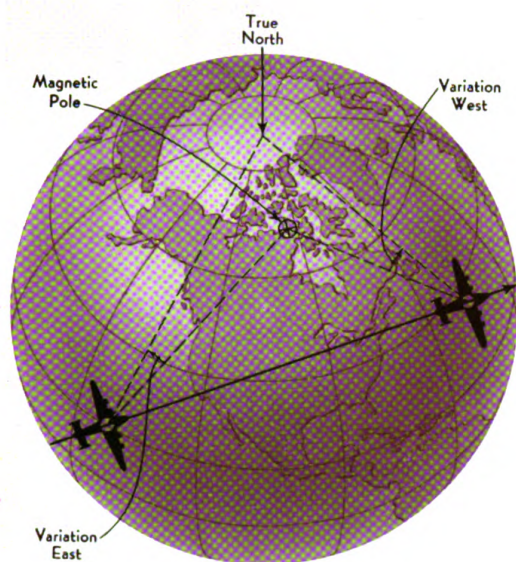


FIG. 29—VARIATION

about 40° , either East or West, depending upon an aircraft's geographical position. However, in polar regions variation up to 180° may be encountered. Besides changing with a plane's geographical position, variation also undergoes a *progressive annual change* which in the course of several years becomes appreciable, but which need not be considered for shorter periods.

The amount of variation for different geographical positions on the earth is indicated on all air navigation charts by lines connecting points of equal magnetic variation called *isogonic lines*. The line connecting points of zero variation, found wherever the plane's position is in line with the true and magnetic poles, is called the *agonic line*. The amount of annual change of variation also is noted on the face of the chart.

Deviation—Deviation is the angular difference, measured at the airplane, between *compass North* and *magnetic North* (Figure 30), compass North being the direction indicated by the compass needle.

Deviation error, as can be seen on the calibration chart (Figure 28), varies for each different heading of the aircraft. Therefore, it has to be applied on every change of heading. This error is recorded as accurately as possible.

However, many conditions in the air may cause it to vary. These may be local magnetic attractions in the area being flown over, or acceleration and swirl errors caused by speed, turns and vibrations, or mechanical failure of the compass itself. These additional errors may be reduced somewhat by lightly tapping the compass before reading, and only reading it when flying straight and level at a constant speed.

Note: Great stress has been placed by some writers of aerial navigation textbooks on the importance of knowing exact deviation. The author agrees that if the navigator were entirely dependent upon dead reckoning, such knowledge would be essential. He wishes to point out, however, that it seldom is necessary for the long-range aerial navigator to depend solely upon dead reckoning for more than a few hours. Usually it is possible to check the plane's position either by visual or celestial fixes. When this is possible, temporary changes in deviation from those tabulated cease to be a serious consideration, since any unknown compass error simply combines with the unknown wind to yield a total error which can be accurately determined between fixes. This would, of course, mean that the wind calculated would not be entirely accurate since it

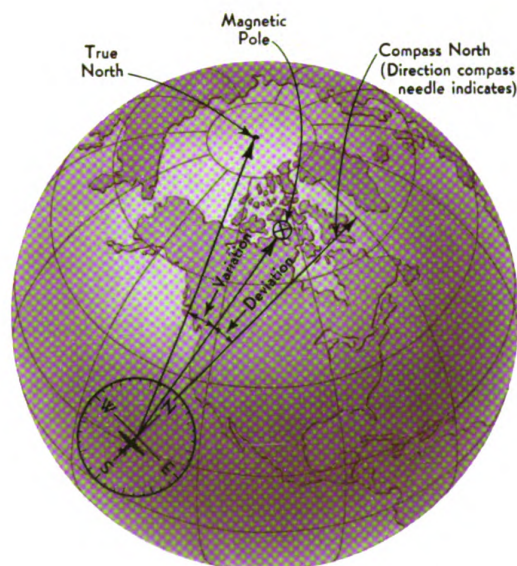


FIG. 30—DEVIATION

would include some deviation error not known, hence not allowed for. And while error in reporting the existing winds is certainly undesirable from a meteorological viewpoint, it in no way endangers the safety of the aircraft. Thus there is no real basis for the rather prevalent belief that an aircraft will become lost if exact deviation is not known. (Application of compass errors is covered in Chapter III).

ALTIMETER

Description—An aircraft altimeter (Figure 31) is an instrument which measures existing air pressure in the same way as does a barometer, the only difference being that the altimeter scale is graduated to read in feet of altitude rather than inches of mercury.

Indicated Reading Errors

1. **Constant Error** is instrument error of a permanent nature which can be calibrated, hence compensated for by proper application of the calibration correction to each instrument reading. Such error usually is due either to faulty construction or installation of the instrument. However, such error does not always exist.

2. **Variable Error** is caused by variations in atmospheric pressure and changes in temperature. This error likewise may be corrected if existing conditions are known. Since the instrument is actuated by atmospheric pressure, which is a variable factor, the manufacturer graduates the instrument card under artificial, simulated standard conditions.



Knob Adjusts Dial to Barometric Setting

FIG. 31—ALTIMETER

Standard conditions are:

At sea level—

Air pressure = 29.92" Hg.

Temperature = +15°C.

For change in altitude—

Air pressure decreases approximately one-half for each increase in altitude of 18,000 feet.

Temperature lapse rate is approximately 2°C per 1000 feet.

Due to changing weather phenomena, these standard conditions seldom are encountered. Therefore, it is apparent that an instrument graduated to standard conditions normally would read with a certain amount of error. To compensate the indicated reading for variable air pressure, a knob is provided on the altimeter which enables the dial to be rotated to agree with actual barometric pressure. Temperature variations are accounted for by calculation.

Altimeter Uses—The altimeter measures the altitude of the aircraft above any given point on the earth's surface. Therefore, it may be used for two purposes:

1. To measure the aircraft's height above an airport, regardless of the airport's altitude, or
2. To measure the height of the aircraft above sea level.

In over-ocean navigation, the navigator is interested primarily in the true altitude above sea level, obtained as follows:

a. Set altimeter to *standard sea level pressure* (29.92" Hg.). The altimeter then will read as *pressure altitude*.

b. Note outside *air temperature*.

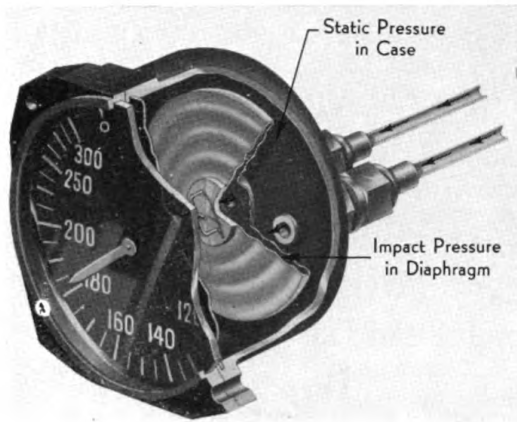
c. Again adjust altimeter to latest *known barometric pressure*. The altimeter then will read as *indicated altitude*. (This barometric pressure may be obtained by radio contact with the nearest ground station or from the isobars indicated on the latest weather map).

d. On the computer set *air temperature* opposite *pressure altitude*. Read *true altitude* opposite *indicated altitude*. (Computers vary slightly but give similar results).

AIRSPEED INDICATOR

Description—The airspeed indicator (Figure 32) is an instrument for measuring the aircraft's speed with reference to surrounding air.

The principle of operation is as follows: A tube, known as the Pitot (pee-tow) tube, is placed forward, outside of the aircraft, and clear of the slipstream. Impact air pressure upon this tube is relayed to a metallic diaphragm inside the instrument. The diaphragm also is affected by existing atmospheric density, and in order that the diaphragm will be responsive to true barometric pressure, a second



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

tube, called the static tube, is mounted alongside the impact tube. The static tube relays the actual, static atmospheric pressure outside the aircraft to the interior of the airspeed meter case surrounding the metallic diaphragm. The excess of impact pressure over static pressure expands the diaphragm, which in turn actuates a dial graduated in terms of air speed at sea level and under standard atmospheric conditions.

The impact tube and static tube mounted together are called the Pitot-static tube. In some newer installations, the Pitot and static tubes are mounted separately.

The airspeed indicator does not indicate the true air speed of the aircraft, mainly because of installation errors and the fact that it is actuated by variable atmospheric pressure. For this reason the following terms are employed to indicate what corrections, if any, have been applied.

Indicated air speed
Calibrated air speed
True air speed

Indicated Air Speed—Indicated air speed is the speed as read on the instrument.

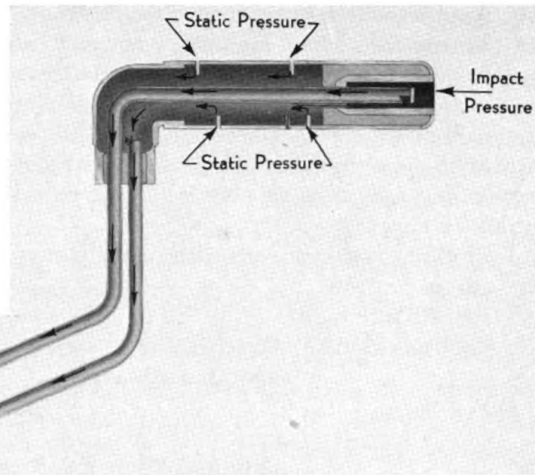


FIG. 32—AIRSPEED METER AND PITOT TUBE

It is based upon standard velocity head values at sea level and standard atmospheric pressure. It does not, however, take into account variations in pressure altitude, air temperature or Pitot-static tube installation errors. Therefore, it is necessary to apply certain corrections to obtain true air speed.

Some aircraft airspeed indicators read directly in knots, although most indicators read in statute miles. In ocean aerial navigation, all distances are measured in nautical miles. Therefore, if the aircraft is equipped with an instrument graduated to read in statute miles, convert the reading into nautical miles before applying further corrections.

Example: 175 m.p.h. indicated = 152 knots indicated.

Calibrated Air Speed—Calibrated air speed is simply *correct indicated air speed*, i.e., instrument reading corrected for *Pitot-static tube installation error* and minor instrument *calibration error*.

The instrument error usually is slight, but the installation errors in some aircraft are quite large, depending upon the location of the Pitot-static tube.

Several methods are employed for finding the corrections which must be applied to indicated air speed in order to determine calibrated air speed (correct indicated air speed).

One method often used is to fly over a measured course at various speeds, noting the elapsed time.

Another method is to trail a "static bomb," an instrument which indicates correct air speed, having previously been calibrated in a wind tunnel. Regardless of the method used, the calibrated air speed, when found, is recorded in each individual ship along with corresponding indicated air speeds for future reference in flight.

Flight performance manuals usually show the calibration thus:

For PBY:

Indicated air speed of 102 knots =
calibrated 109 knots.

For B-24:

Indicated air speed of 175 m.p.h =
calibrated 176 m.p.h.

Note: Calibration error correction for different speeds varies but slightly. Therefore, the same amount of calibration error correction is applied to all indicated air speeds.

True Air Speed—True air speed is the actual speed of the aircraft with reference to the surrounding air, and is obtained by *correcting the calibrated air speed (correct indicated air speed) for existing temperature and pressure altitude of the aircraft.*

The airspeed meter is graduated for sea level and standard atmospheric pressure, but since atmospheric density decreases as altitude increases, the excess of impact pressure over static pressure also decreases, causing the true air speed to be (nearly always) greater than the indicated air speed.

Air Speed Correction Problems:

Temperature = $+10^{\circ}\text{C}$.

Pressure altitude = 10,000 feet.

Example No. 1

Indicated air speed = 102 knots

Calibration error correction = $+7$ knots

Calibrated air speed = 109 knots

True air speed = 131 knots

Example No. 2

Indicated air speed = 175 m.p.h.

Calibration error correction = $+1$ m.p.h.

Calibrated air speed = 176 m.p.h.

Calibrated air speed = 153 knots

True air speed = 183 knots

Note: (Use the aircraft computer in converting m.p.h. into knots and in finding true air speed from calibrated air speed).

THERMOMETER

Description—The thermometer is an instrument used to measure the free air temperature outside the aircraft.

It operates on the principle that a substance expands or contracts with an increase or decrease of heat. The aircraft thermometer utilizes a thin, coiled, bi-metallic strip which on expanding or contracting actuates a dial graduated in degrees of temperature.

Two temperature scales commonly used are the Fahrenheit and Centigrade scales. They differ only in the size of graduation intervals.

The Fahrenheit scale is so graduated that 32° equals the freezing point of water and 212° equals the boiling point of water at sea level, under standard atmospheric conditions.

The Centigrade scale, more often used in aircraft, is graduated so that 0° equals the freezing point of water and 100° equals the boiling point.

CHRONOMETER

Description—A chronometer is a very accurate timepiece.

The aircraft chronometer is a watch of exceptionally fine construction and usually has a

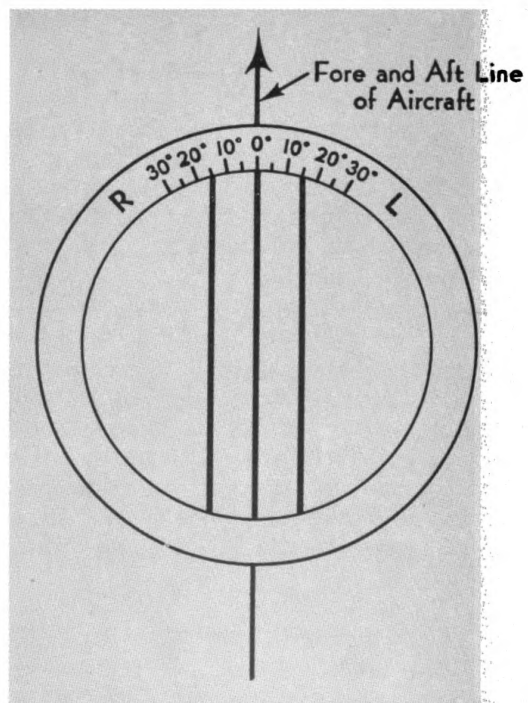
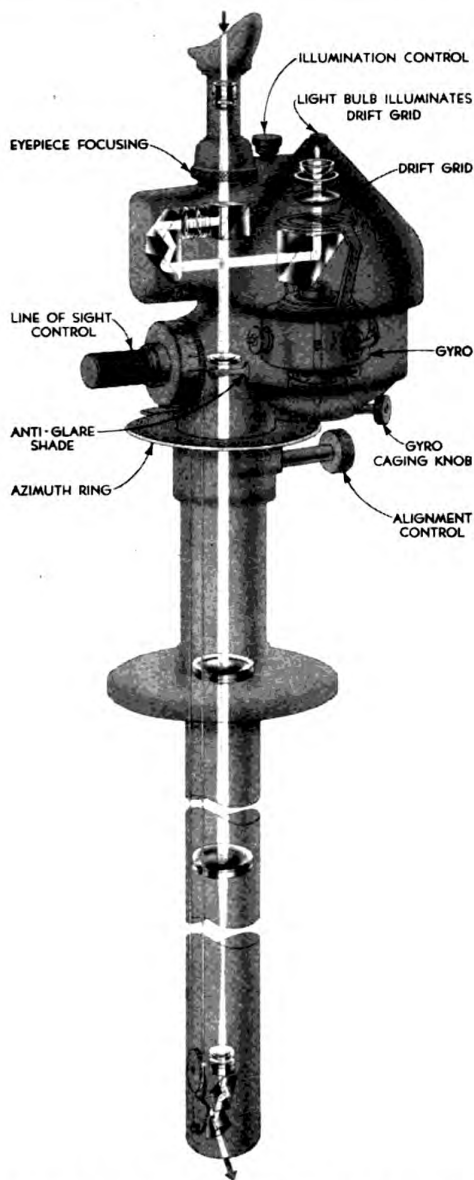


FIG. 33—SIMPLE DRIFT METER



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

FIG. 34—TYPE B-3 DRIFT METER

24-hour dial with a stop-second hand which enables it to be set for zero error. Since it is a precision instrument, it should be handled with the greatest of care. It should be wound at the same time each day, and kept in a case to protect it as much as possible from temperature changes and shock.

The chronometer is set to Greenwich civil time, and should be checked frequently either

by reference to a master watch or from radio time signals (ticks), in order to determine its "rate." The rate is the amount it gains or loses each day. As long as the rate is constant and not more than a few seconds each day, a record can be kept so that correct time will always be had by applying the amount of gain or loss since the last accurate check. If the rate is erratic, the watch should be sent to a jeweler for repair.

DRIFT METER

General—Drift represents the effect of crosswind on the aircraft's heading. If the wind's force and direction were accurately known it would be easy to calculate drift. Unfortunately, however, no reliable method yet has been devised to accurately pre-determine the wind force and direction for any proposed flight. Therefore the navigator must use all available methods to measure the aircraft's drift while in the air. The easiest method of doing this is by means of a drift meter, which is an instrument for determining the angle at which an aircraft is drifting to the right or left of its course.

The Simplest Type of drift meter (Figure 33) is a grid of parallel lines, installed in an opening in the floor of the aircraft. The center line, when parallel to the fore and aft axis of the aircraft, is marked 0° drift. In flight, the number of degrees to the right or left which the grid must be rotated in order to align the grid lines with the apparent motion of the ground, is the angle of drift.

A Refined Type of drift meter is the type B-3* (Figure 34). It utilizes the same principle as the type above described except that the ground is viewed through an optical system, and the grid is kept parallel to the ground by a self-erecting gyroscope. Besides an azimuth scale for reading drift or taking visual bearings through 360°, the B-3 meter incorporates a graduated line of sight control which can be used either for reading drift or measuring the vertical angles used in computing ground speed.

*A complete description of the B-3 drift meter may be found in Air Corps Technical Manual No. 205 or in the Bendix Instruments Manual.

Caution—Many types of drift meter have been devised, and their correct use makes accurate dead reckoning navigation possible. However, it should be remembered that a drift meter is a mechanical device, and therefore liable to error. The drift meter should be used whenever possible, but only as an added help to celestial navigation and other available navigation aids.

Use of Type B-3 Drift Meter to Measure Drift:

- Turn on the gyro and, when it has attained operating speed (3 to 5 minutes), release the gyro caging knob which is located under the gyro housing. The gyro should be released only in level flight and should be caged during maneuvers or when the gyro is not in use.
- Adjust the light rheostat so that the grid and the image on the ground are easily seen.
- Rotate the instrument until the apparent motion of a whitecap, object on the ground, or a signal device previously dropped from the aircraft follows the fore and aft grid wires. The drift angle then is read on the azimuth scale.

Note: By rotating the line of sight control, the ground object may be viewed directly beneath or to the rear of the aircraft.

Use of Type B-3 Drift Meter to Determine Ground Speed: Ground speed may be determined by the following formula after measuring the time (in seconds) it takes for an object on the ground to pass over a convenient angle, measured on the line of sight control.

Formula for ground speed—

$$\text{Ground speed} = \frac{\text{Altitude}}{\text{Time}} \times \text{Factor}$$

[For factor see factor table (Figure 35)]

One procedure is as follows:

- Determine the true altitude.
- Uncage gyro and allow five minutes for it to erect itself.
- With the line of sight control set at 0° (starting angle), sight a ground object on the center athwartship grid wire and follow its motion by revolving the line of sight control for ten seconds. (The line of sight control may be set other than at 0° at start and the time al-

lowed may be other than 10 seconds; however, the method cited simplifies solution.)

d. Read the angle on the line of sight control (finish angle) and obtain the factor by inspection of the factor table (Figure 35).

e. Compute formula.

Example No. 1

True altitude = 10,000 feet

Start on line of sight control 0°, finish 18°

Total time = 10 seconds.

$$\frac{10,000 \text{ ft.}}{10 \text{ sec.}} \times .1932 =$$

$$1,000 \times .1932 =$$

$$193.2 \text{ knots ground speed.}$$

Note: Factor (.1932) is for finish angle 18° and is found by interpolating between 15° and 20° as follows:

$$.159 = 15^\circ \text{ factor}$$

$$.216 = 20^\circ \text{ factor}$$

$$.057 = \text{difference}$$

Interpolating—

$$3/5 \text{ of } .057 = .0342$$

$$.159 = 15^\circ \text{ factor}$$

$$+.0342 = 3^\circ \text{ of difference}$$

$$.1932 = 18^\circ \text{ factor}$$

FACTOR TABLE FOR GROUND SPEED (In Knots)						
Finish Angle	Starting Angle					
	0°	10°	20°	30°	40°	50°
5°	.052					
10°	.104					
15°	.159	.054				
20°	.216	.111				
25°	.276	.172	.061			
30°	.342	.238	.126			
35°	.415	.310	.199	.073		
40°	.497	.392	.281	.155		
45°	.592	.488	.377	.250	.095	
50°	.706	.601	.490	.364	.209	
55°	.846	.741	.630	.504	.349	.140
60°	1.026	.922	.810	.684	.529	.320
65°	1.270	1.165	1.054	.928	.773	.564
70.9°	1.706	1.602	1.490	1.364	1.209	1.000

FIG. 35

Practical Alignment—For accurate drift meter readings it is necessary that the drift wires be parallel with the longitudinal axis of the aircraft, and that the azimuth scale read 0° forward and 180° aft when the drift wires are thus aligned.

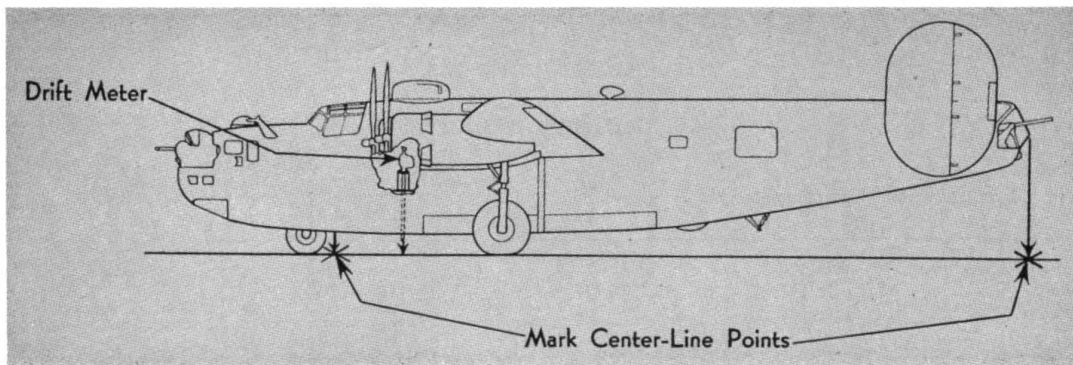


FIG. 36—ALIGNING DRIFT METER (SIDE VIEW)

This alignment may be accomplished quickly and accurately enough for practical navigation by means of a length of string and a plumb bob. The procedure is as follows: (see Figures 36 and 37)

a. Establish center line of aircraft by dropping a plumb bob from center of tail and center of nose (or center line behind nose wheel, if tricycle landing gear). (Figure 36)

b. Drop plumb bob from center of drift meter coverglass, and mark point vertically be-

neath on ground. Measure distance of this point from center line and then measure out a similar distance from the fore and aft center points. Through these three points hold a string parallel to center line (Figure 37).

c. Rotate drift meter, with gyro caged, sighting the string line fore and aft by turning line of sight control until parallel wires of drift meter are aligned with string line. Azimuth drift scale should then indicate 0° forward; if not, loosen screws and adjust azimuth ring.

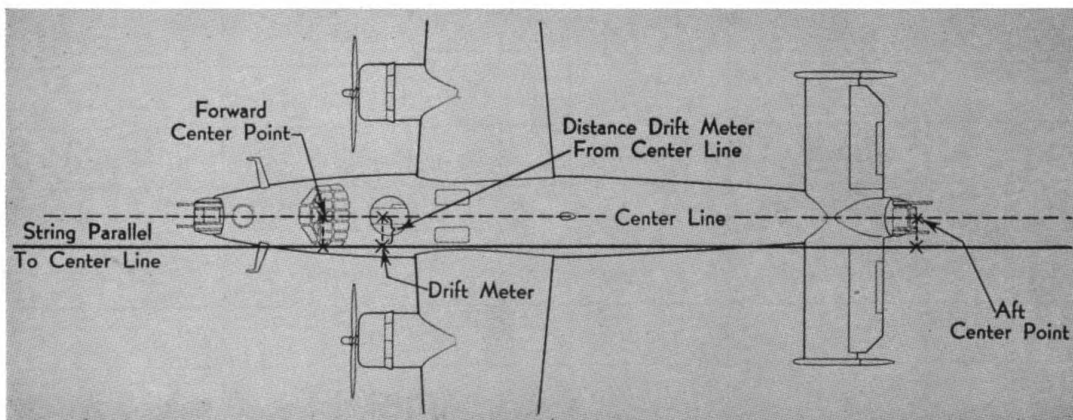
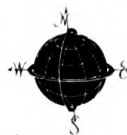


FIG. 37—ALIGNING DRIFT METER (TOP VIEW)



PROBLEM WORK

No. 3 Draw diagram of pilots' type magnetic compass and indicate the principal parts.

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 4

ALTIMETER CORRECTION

(Find true altitude. Use computer.)

No.	PRESSURE ALTITUDE	INDICATED ALTITUDE	TEMPERATURE	TRUE ALTITUDE
1	7500'	8000'	+ 10°C	
2	3300'	3000'	+ 15°C	
3	24,000'	23,000'	— 22°F	
4	10,000'	11,000'	0°C	
5	4500'	5000'	+ 15°C	
6	16,400'	17,000'	+ 14°C	
7	1700'	2000'	+ 10°C	
8	8000'	8400'	+ 18°C	
9	17,000'	16,500'	+ 32°F	
10	15,000'	14,000'	0°F	
11	5500'	6000'	+ 12°C	
12	1000'	1300'	+100°F	
13	2750'	3000'	— 7°C	
14	10,000'	9500'	+ 15°C	
15	8500'	8000'	+ 15°C	
16	6000'	5900'	+ 20°C	
17	7500'	7500'	0°C	
18	9500'	10,000'	— 5°C	
19	8000'	8500'	+ 42°F	
20	13,000'	12,400'	— 16°C	

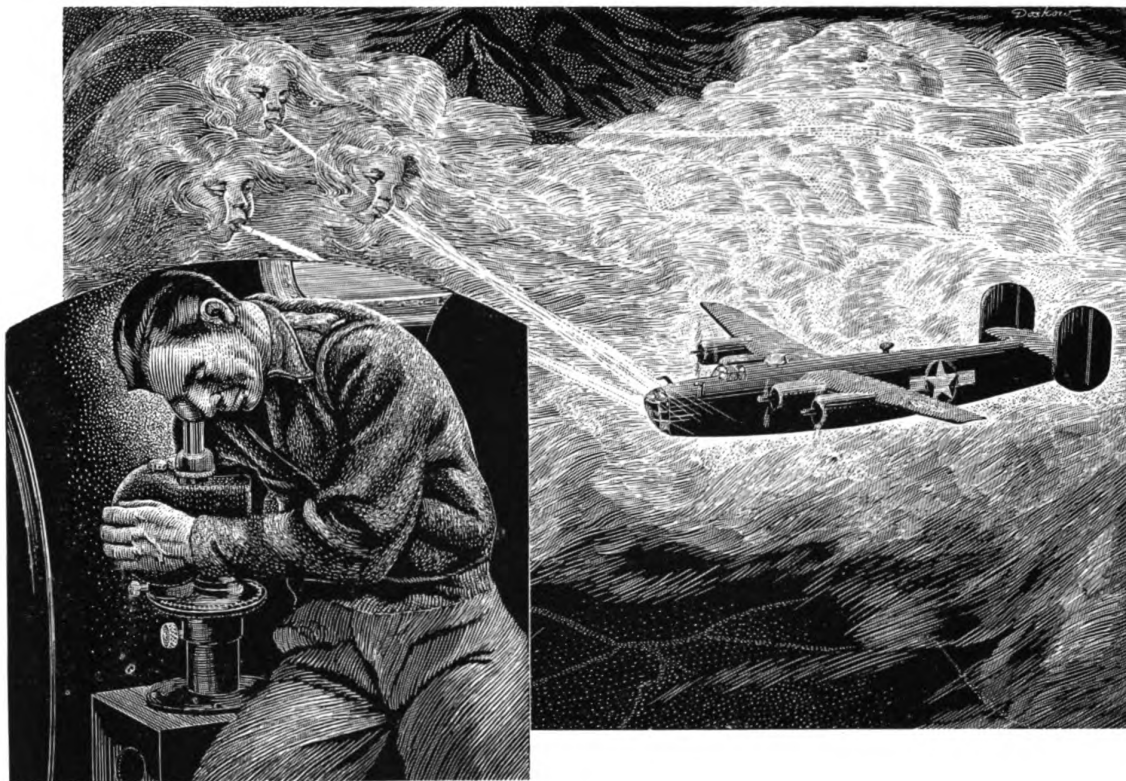
I N S T R U M E N T S

PROBLEM WORK NO. 5

AIR SPEED CORRECTION

(Find missing air speeds. Use computer.)

If	I. A. S. in m.p.h.	=	90	100	110	120	130	140	150	160	170	180	190	200	210
Then	C.A.S. in m.p.h.	=	90	102	113	124	135	146	157	168	178	185	196	204	212
No.	I.A.S. (m.p.h.)	C.A.S. (m.p.h.)	C.A.S. (knots)		PRESSURE ALTITUDE		TEMP.		T.A.S. (knots)						
1	142				11,000'		0°C								
2	160				8000'		— 3°C								
3					6000'		+42°F		185						
4					9000'		+21°F		191						
5	163				7000'		— 3°C								
6					3000'		+32°F		165						
7	145				6000'		+ 4°C								
8					9000'		+42°F		192						
9	158				2000'		+ 1°C								
10	139				4000'		— 6°C								
11	151				9000'		+18°C								
12					3000'		— 1°C		182						
13	165				6000'		+26°F								
14	180				14,000'		— 6°C								
15	186				11,000'		+14°C								
16	157				1000'		+76°F								
17					6000'		+19°C		186						
18					9000'		+ 6°C		195						
19	180				8000'		+14°F								
20	121				3000'		+23°F								



☆ 3 ☆

DEAD RECKONING

THE term "Dead Reckoning" is commonly supposed to have evolved from "Deduced Reckoning." First, "Ded. Reckoning" as an abbreviation, the form eventually became "Dead Reckoning" for convenience.

Dead Reckoning (DR) is the basic method of determining the location of an aircraft with reference to a known position by keeping an account or reckoning of the estimated distance flown and calculated track made good. It involves taking into account all available knowledge of time, speed, drift, heading, track, and weather conditions. If the direction and velocity of the wind actually were known, the dead

reckoning position likewise would be very accurate; however, weather data quite often is vague, and different navigators using the same information probably would obtain varying results.

One navigator, after making several flights across the South Atlantic with unusually correct DR calculations (in which luck certainly played a part), applied for employment with a different company. In his application he stated: "I feel I should be hired because of my unusual 'homing pigeon' instinct!" He wasn't hired, as his colorful claim was entirely without foundation.

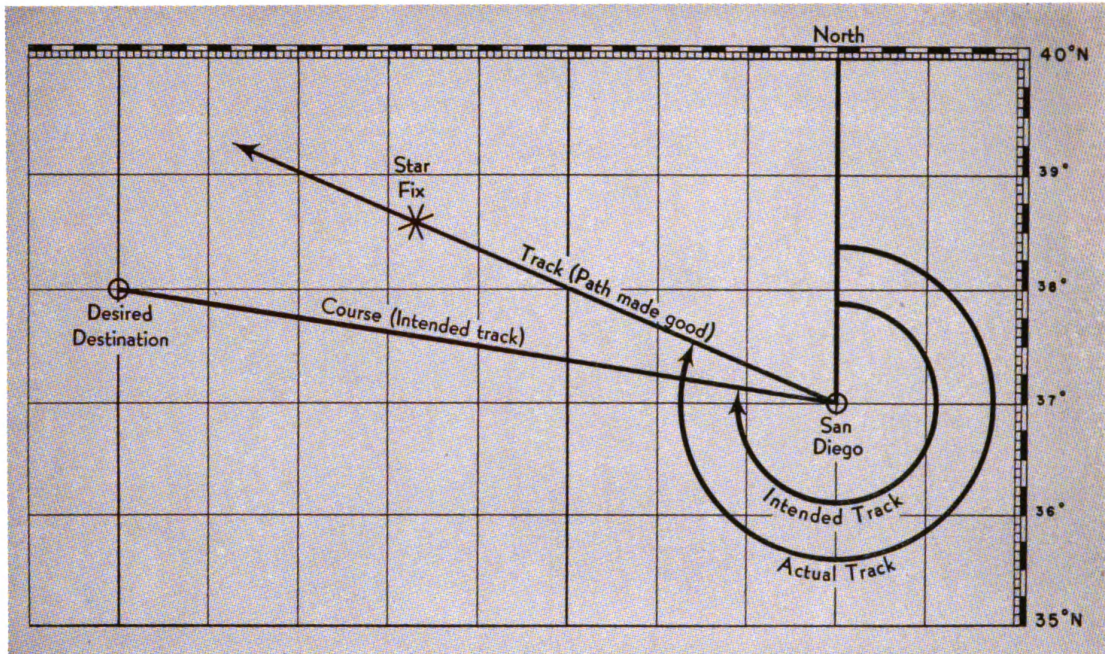


FIG. 38—TRACK AND COURSE

When a navigator continuously calculates fairly accurate DR positions, the "homing pigeon" instinct would more likely be the result of the following:

1. Complete understanding of all methods known of establishing and plotting positions.
2. A working knowledge of wind velocity triangles.

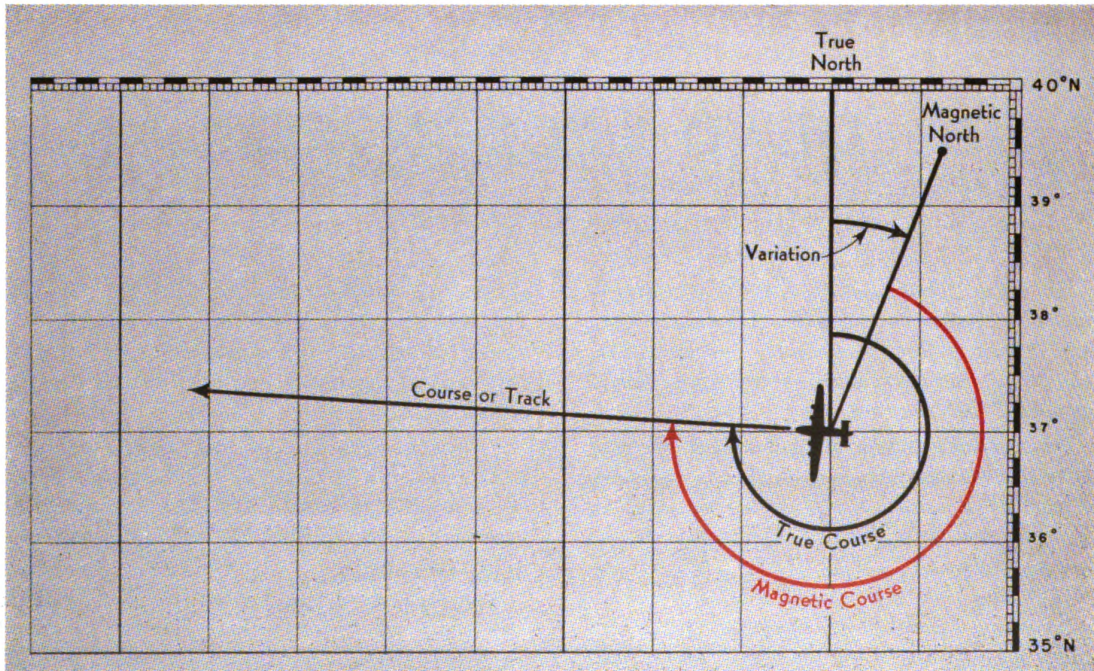


FIG. 39—TRUE AND MAGNETIC COURSE

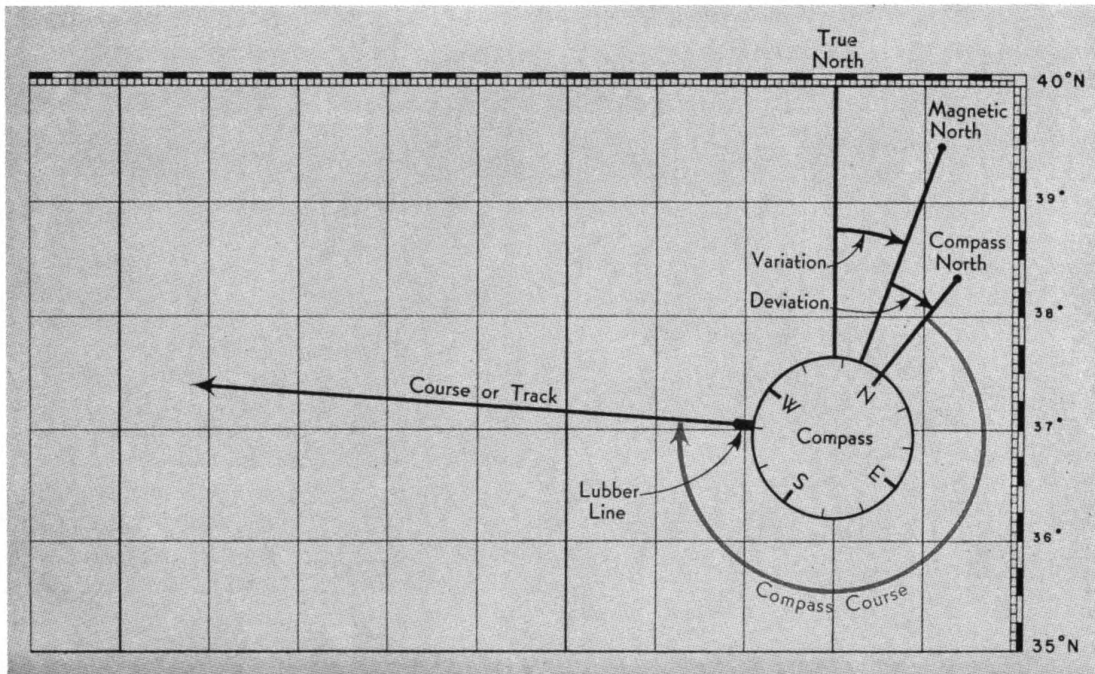


FIG. 40—COMPASS COURSE

3. A thorough proficiency in observing drift and the ability to correctly interpret values obtained.

4. Most important of all, the ability through reasoning and deduction from all available data regarding drift and speed, to determine his most probable position.

TRACK AND COURSE

Track is the *path* which the *aircraft actually makes good* over the ground. It is considered to be true track (i.e., measured from true North) unless otherwise indicated. (Figure 38)

Course is the *intended track*, i.e., the path which it is *intended* that the aircraft will follow in flying from one point to another over the ground. Unless otherwise indicated, it is considered to be true course. (Figure 38) Reference points from which courses are measured give rise to the following course terms:

1. **True Course** is the angular direction of the course measured from *true North*. It is obtained by applying both variation and deviation to compass course or variation to magnetic course. (Figure 39)

2. **Magnetic Course** is the angular direction of the course measured from magnetic North. It is obtained by applying variation to

true course or deviation to compass course. (Figure 39)

3. **Compass Course** is the direction indicated on the magnetic compass. It is the angular direction of the course with reference to compass North, and can be obtained from true course by applying both variation and deviation. (Figure 40)

Note: The term "compass course" is used here only for aiding the navigator in visualizing course corrections and for determining compass errors when the aircraft is stationary. In flight the aircraft does not fly a compass course because the direction indicated on the aircraft compass is actually compass heading. Compass heading (see page 33) also is affected by an additional course correction known as drift, which represents the effect of wind on the aircraft.

APPLYING COMPASS ERRORS

General—Orientation of the aircraft with respect to true North is most important to the navigator. However, it is imperative to bear in mind that only by applying the errors of variation and deviation to the magnetic compass reading can the aircraft's direction with respect to true North be obtained.

Thus it is apparent that true North is obtained *indirectly* from the magnetic compass reading. This simple fact sums up the principle of course conversions. The application of compass errors is not difficult, but must be mastered by the navigator if he is to be reliable.

Course Relationship—Course conversion is simplified once the navigator thoroughly understands the relationship of the various course terms to each other. This relationship may be outlined as follows:

True Course: *No error.*

Magnetic Course: *One error*—variation (from true course).

Compass Course: *Two errors*—variation and deviation (from true course).

Therefore

	(±)		(±)	
True	Var.	Magnetic	Dev.	Compass

Variation is error between true and magnetic.

Deviation is error between magnetic and compass.

Conversion "Jingle"—To simplify course conversions, the navigator should adopt some positive method which will:

1. Help him determine the correct application of errors.
2. Double check his course computations.

The following "jingle" may prove helpful:

"TRUE RIGHT EAST"

"TRUE LEFT WEST"

TRUE RIGHT EAST means—If error is *East* the more correct course is *right*. (Figure 41)

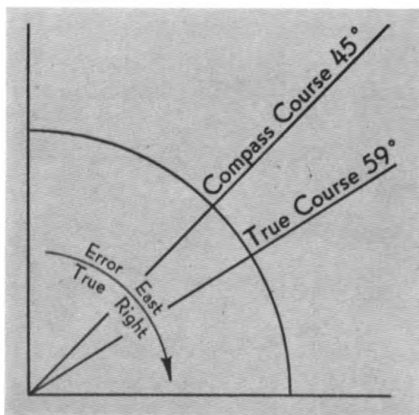


FIG. 41—"TRUE RIGHT EAST"

TRUE LEFT WEST means—If error is *West* the more correct course is *left*. (Figure 42)

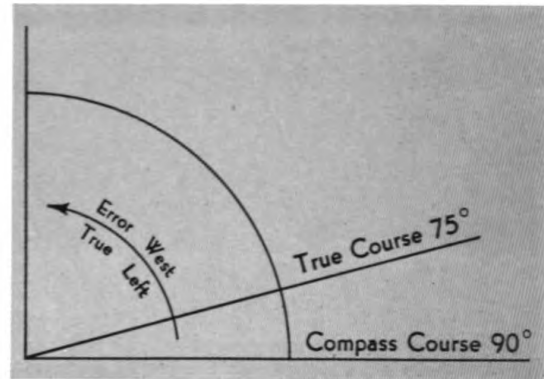


FIG. 42—"TRUE LEFT WEST"

In using the jingle, remember—True course contains no error and therefore is *more correct* than either magnetic or compass course.

Also remember—Magnetic course, containing one error, is *more correct* than compass course, which contains two errors.

Example No. 1

The compass course of an aircraft is 45° .

Required: True course.

Variation = 18° East

Deviation = 4° West

Total error = 14° East

The *true course*, therefore, is 14° right of the *compass course*, hence:

True course = 59°

Note: Variation was found on the chart; deviation on the calibration graph, for a course of 45° .

Example No. 2

The true course to be flown is 75° .

Required: Compass course.

Variation = 20° West

Deviation = 5° East

Total error = 15° West

The *true course*, therefore, is 15° left of the *compass course*, hence:

Compass course = 90° .

Note: Variation was found on the chart; Deviation, on the calibration graph for a course of 90° .

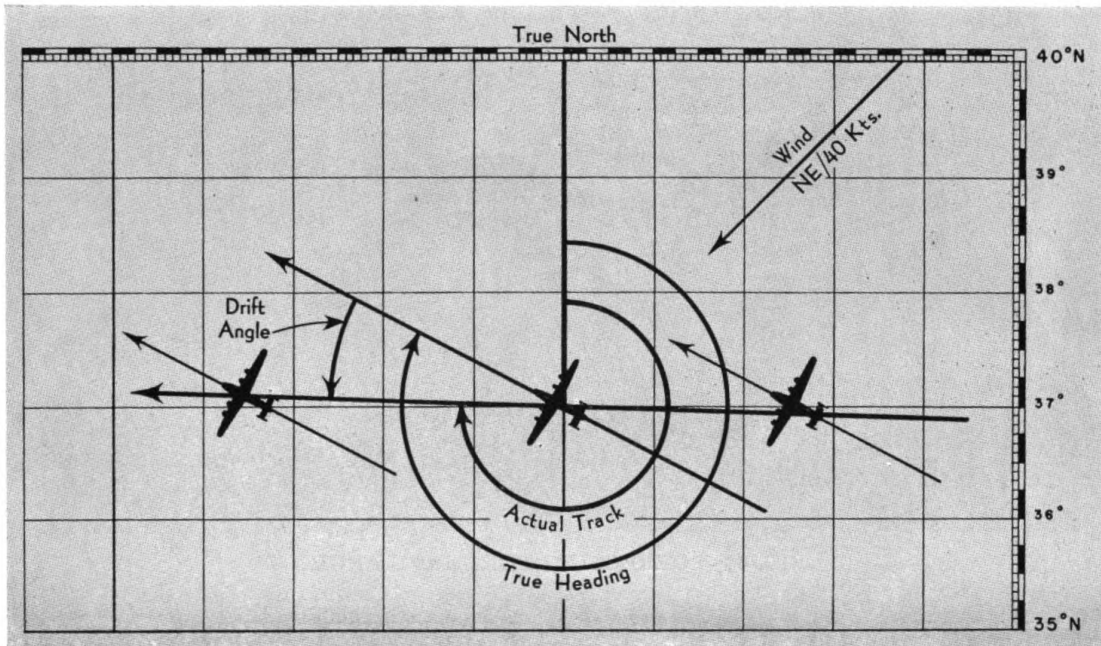


FIG. 43—DRIFT ANGLE AND HEADING

Sometimes compass errors are applied one at a time, as illustrated in examples No. 3 and 4.

Example No. 3

Compass course = 45°
 Deviation = -4° West (magnetic left)
 Magnetic course = 41°
 Variation = $+18^{\circ}$ East (true right)
 True course = 59°

Example No. 4

True course = 75°
 Variation = $+20^{\circ}$ West (true left)
 Magnetic course = 95°
 Deviation = -5° East (mag. right)
 Compass course = 90°

DRIFT ANGLE

Drift angle, or drift, represents the effect of crosswind on the aircraft and may be defined as the angular difference between heading and track.

HEADING

Heading is the direction in which the longitudinal axis of the aircraft is pointed and, except in the case of tail-wind, is always into the wind. (Figure 43) Unless otherwise indi-

cated, it is considered to be true heading. Reference points from which headings are measured give rise to the following heading terms:

1. **True Heading** is the heading of the aircraft measured from true North. It can be obtained by applying drift angle to true course.

2. **Magnetic Heading** is the heading of the aircraft measured from magnetic North. It can be obtained by applying drift angle to magnetic course.

3. **Compass Heading** is the heading of the aircraft measured from compass North. It can be obtained (theoretically) by applying drift angle to compass course. However, as the heading of an aircraft changes, the deviation changes. Therefore, since the drift angle may be large, a more accurate compass heading can be determined if the magnetic heading is computed first, and the compass heading is then computed by applying deviation to the magnetic heading.

Course-Drift-Heading Relationship—In Figure 44 it is apparent that when drift angle is applied to any course, the result is heading. Thus it may be assumed that the term "heading" indicates an additional error—drift angle—which for purposes of course conversion can be applied as an additional compass error.

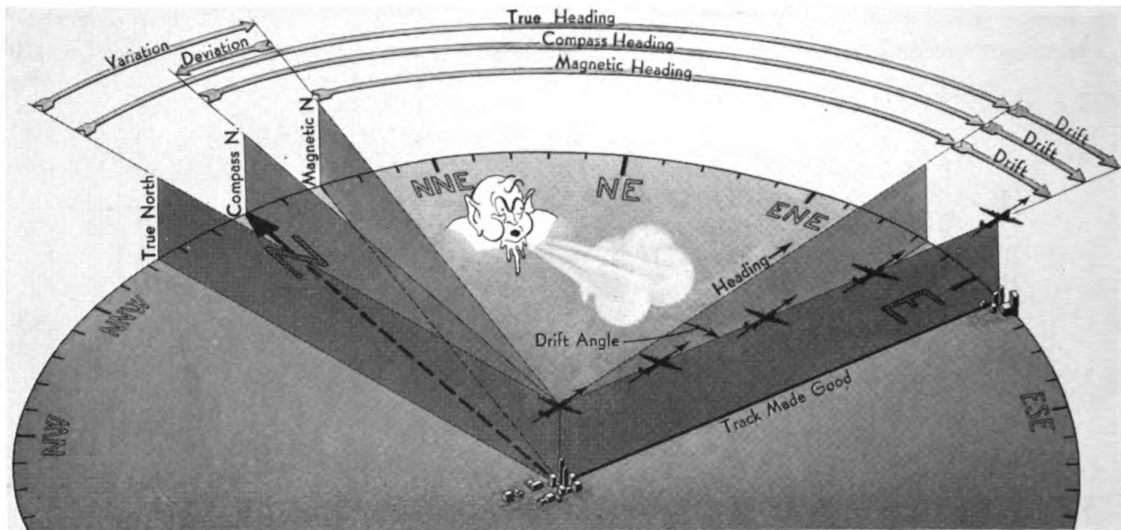


FIG. 44—CORRECTING FOR DRIFT ANGLE

Thus:

True heading: *One error*—drift (from true course).

Magnetic heading: *Two errors*—drift and variation (from true course).

Compass heading: *Three errors*—drift, variation and deviation (from true course).

Therefore, to find the compass heading necessary to fly in order to make good a desired true course, or to determine the true course being flown while steering a certain compass heading, the same jingle—**True Right East—True Left West**—holds good.

Example No. 1

To find compass heading necessary to fly in order to make good a desired true course:

True course	=	120°
Drift	=	5° Right (true course right)
True heading	=	115°
Variation	=	20° East (true right)
Magnetic heading	=	95°
Deviation	=	3° West (magnetic left)
Compass heading	=	98°

Example No. 2

To find the true course being flown while steering a certain compass heading:

Compass heading	=	320°
Deviation	=	6° East (mag. right)
Magnetic heading	=	326°
Variation	=	32° West (true left)
True heading	=	294°
Drift	=	14° Left (true course left)
True course	=	280°

SPEED

Ground Speed is the speed of the aircraft with reference to the ground.

Air Speed is the speed of the aircraft with reference to the surrounding air.

Note: Ground speed is measured along the true course or track because both ground speed and course or track are related to the ground.

Air speed is measured along the true heading, because both air speed and heading are related to the air.

Where a no-wind condition exists, ground speed and air speed are the same. Where there is wind present, they will differ by a vector amount equal to the velocity of the wind. ("Vector" is defined on page 35).

TIME—SPEED—DISTANCE

Time is the measure of duration which determines how long a plane has flown or can fly.

Time = Distance divided by Speed

Example:

480 nautical miles (Distance)	=	3.2 hours or
150 knots (Speed)	=	3 ^h 12 ^m (Time)

DEAD RECKONING

Speed is the rate of motion, or *ratio of time and distance*.

Speed = Distance divided by Time

Example:

$$\frac{480 \text{ naut. mi. (Distance)}}{3.2 \text{ hours (Time)}} = 150 \text{ knots (Speed)}$$

Distance is the measurement of space between two points which determines how far a plane has flown or can fly.

Distance = Time multiplied by Speed

Example:

$$\begin{array}{ccc} \text{(Time)} & \text{(Speed)} & \text{(Distance)} \\ 3.2 \text{ hours} \times 150 \text{ knots} & = & 480 \text{ nautical miles} \end{array}$$

Note: As can be seen from the formulas, time, speed, and distance definitely are related to each other, and each is the result of division or multiplication of the other two. It should be remembered, however, that only like units can be multiplied together or divided by one another to obtain correct results. That is, if distance is in nautical miles, speed will have to be in knots; and, if speed is in knots (nautical miles per hour) then the time must be in hours and *tenths of an hour*, not hours and minutes.

Minutes are converted into tenths of an hour by dividing them by sixty. Similarly, tenths of an hour may be converted into minutes by multiplying them by sixty.

Example:

$$\begin{aligned} \frac{1 \text{ minute}}{60} &= .017 \text{ hours} \\ .017 \text{ hours} \times 60 &= 1 \text{ minute} \\ \frac{2 \text{ minutes}}{60} &= .033 \text{ hours} \\ .033 \text{ hours} \times 60 &= 2 \text{ minutes} \\ \frac{3 \text{ minutes}}{60} &= .05 \text{ hours} \\ .05 \text{ hours} \times 60 &= 3 \text{ minutes} \end{aligned}$$

VECTOR DIAGRAMS

The term "vector," as used in dead reckoning, refers to a straight line whose *length*, drawn to any chosen scale, represents the *magnitude of a velocity*, and whose *direction* (usually shown by an arrowhead) indicates the *direction of the velocity*. Thus the velocity of an aircraft flying East at 120 knots can be represented graphically by a straight line 120 units in length and directed at an angle of 90° measured clockwise from true North.

Since wind, ground speed and air speed are

quantities which have direction and magnitude, they can be represented by vectors. When the three are combined graphically, the result is a *vector diagram*, or *triangle of velocities*. Given sufficient known conditions to permit construction of a vector triangle, the navigator can determine unknown elements by direct measurement. In air navigation, known conditions may be resolved into three basic triangles for determining unknown conditions.

Three Basic Types:

Type No. 1 (Figure 45)

Given: True course or track
True air speed
Wind

Find: True heading
Ground speed
Drift

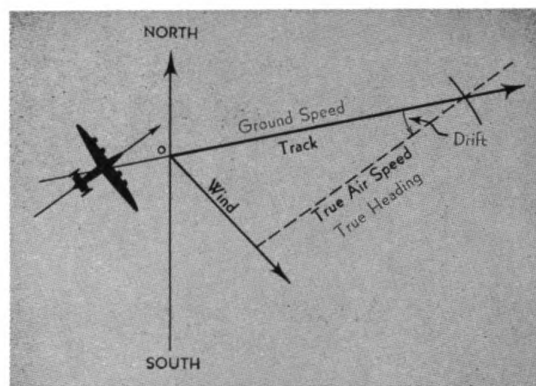


FIG. 45—VECTOR DIAGRAM: TYPE I

Type No. 2 (Figure 46)

Given: True heading
True air speed
Wind

Find: True course or track
Ground speed
Drift

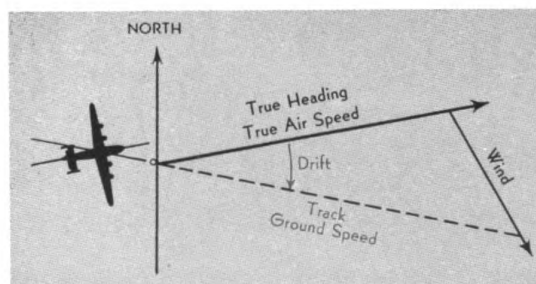


FIG. 46—VECTOR DIAGRAM: TYPE II

Type No. 3 (Figure 47)

Given: True course or track
Ground speed
True heading
Air speed
Find: Wind
Drift

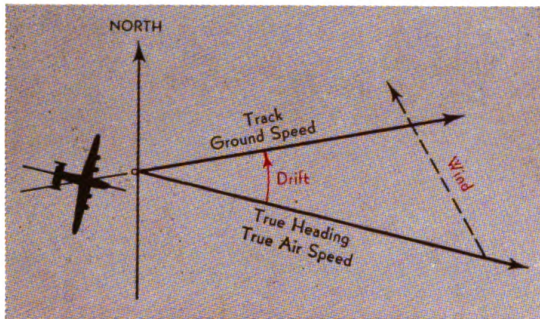


FIG. 47—VECTOR DIAGRAM: TYPE III

Sample Problems:

Type No. 1 (Figure 48)

Given: True course = 308°
True air speed = 150 knots
Wind = $20^\circ/35$ knots

Find: True heading
Ground speed
Drift

Type No. 2 (Figure 49)

Given: True heading = 250°
True air speed = 125 knots
Wind = $140^\circ/50$ knots

Find: True course or track
Ground speed
Drift

Type No. 3 (Figure 50)

Given: Track = 230°
Ground speed = 132 knots
True heading = 245°
True air speed = 175 knots

Find: Wind
Drift

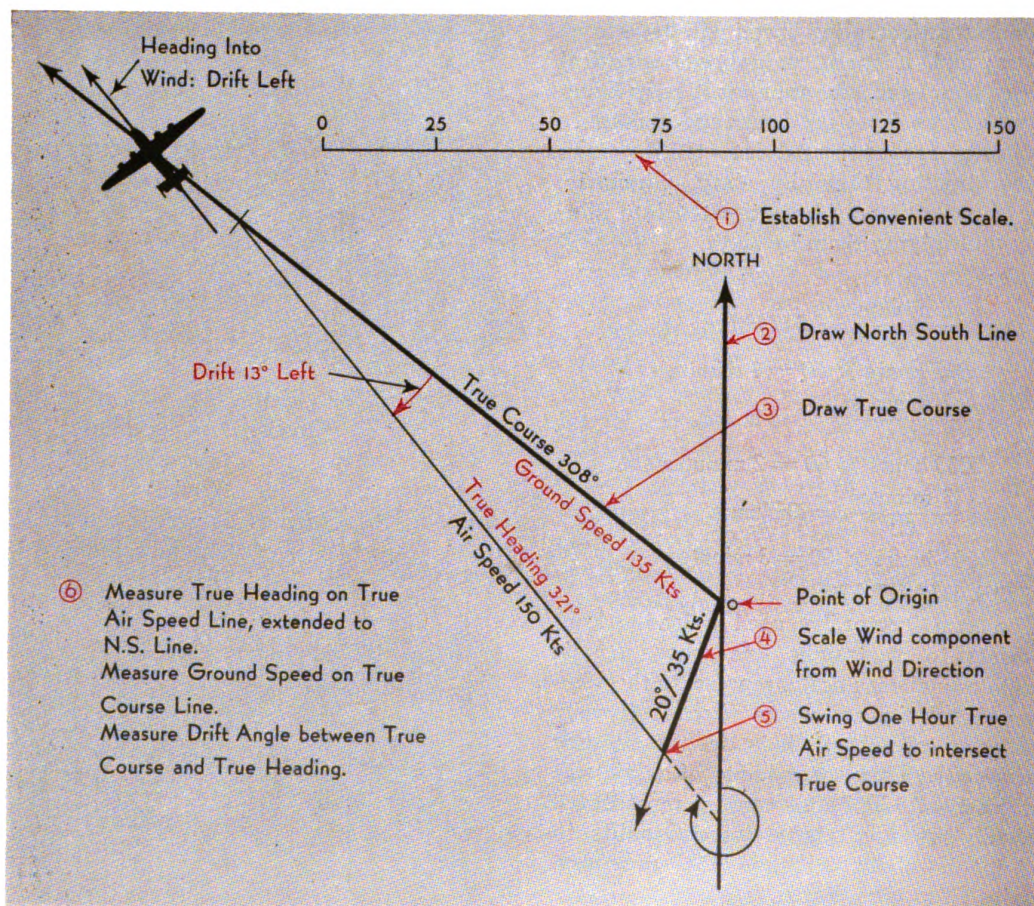


FIG. 48—APPLICATION OF TYPE I VECTOR DIAGRAM

DEAD RECKONING

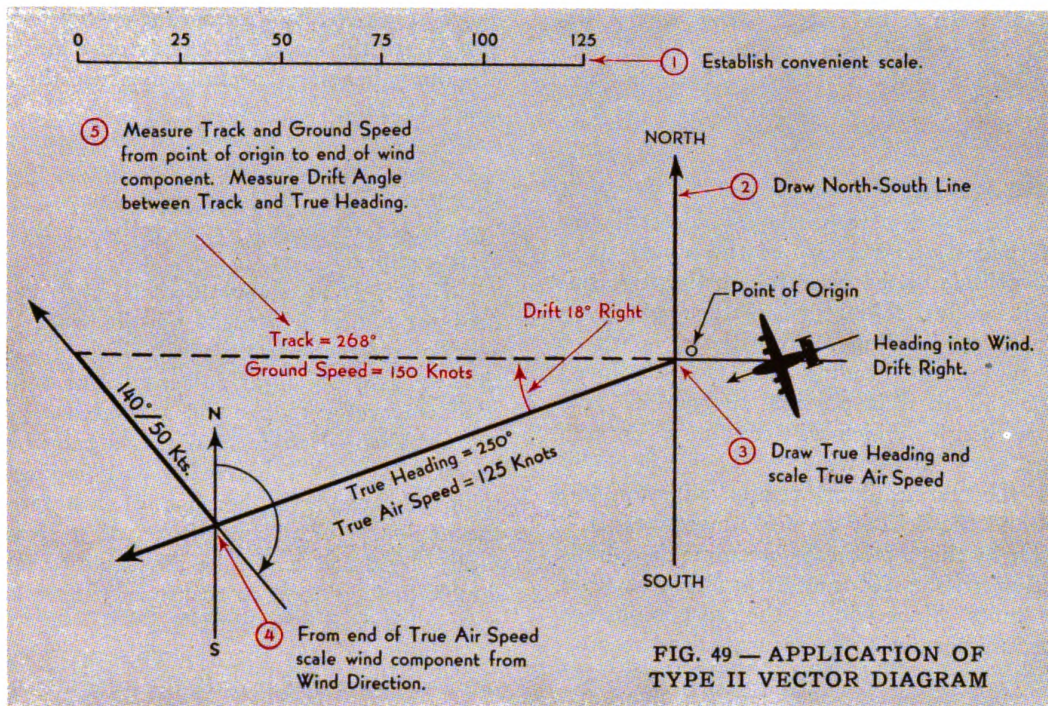


FIG. 49 — APPLICATION OF TYPE II VECTOR DIAGRAM

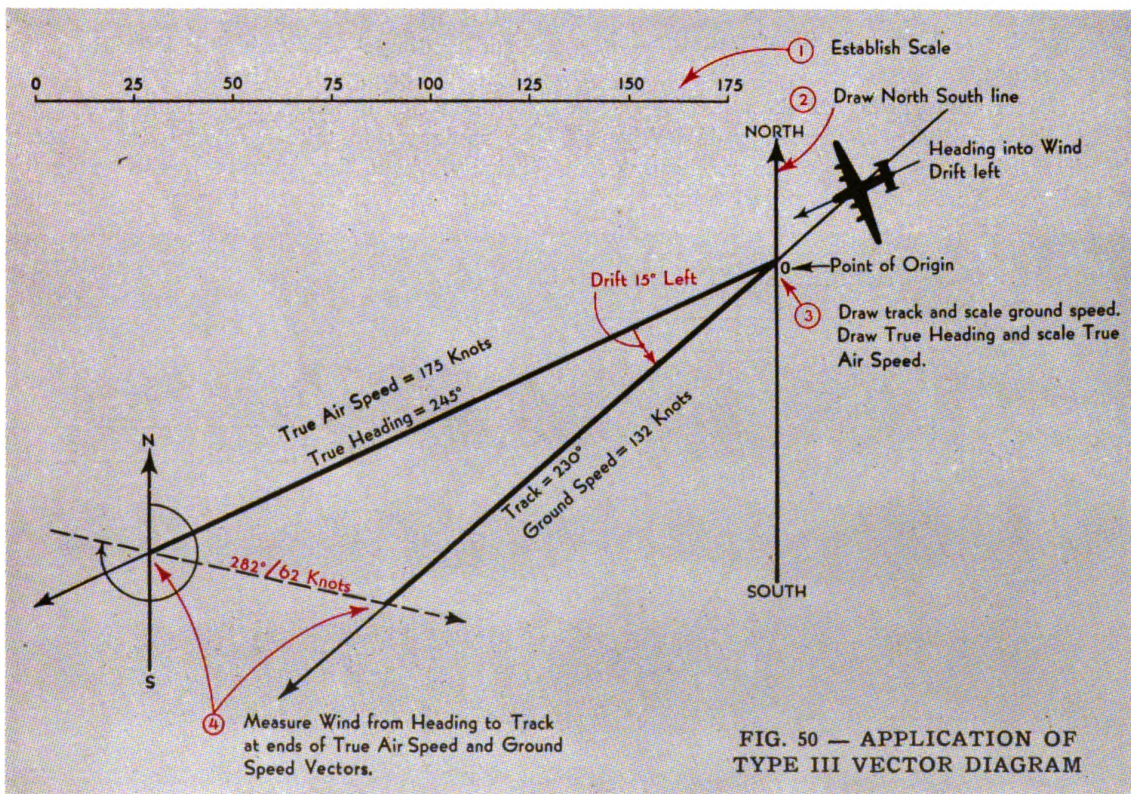


FIG. 50 — APPLICATION OF TYPE III VECTOR DIAGRAM

Note: In actual practice, wind vector triangles are worked on a computer, which permits solution of the problems with a minimum of effort. Knowing their construction, however, helps visualize wind effect.

DOUBLE DRIFT

Double drift is a method of determining the direction and velocity of the wind at the flight altitude of the aircraft. As the term implies, it involves taking drift meter readings on two separate headings. From the elements of drift angle, true heading, and true air speed thus obtained, it is possible to combine two vector triangles which will contain a common vector, equal to the wind.

If the two headings are approximately 90° apart and the drift is carefully measured, it is possible by this method to determine the wind's direction within 20° and its velocity within three knots. Since wind is the most important factor in determining an accurate dead reckoning position, a double drift is highly important to the ocean navigator, especially when the sky is overcast and navigation is entirely dependent upon dead reckoning.

Procedure—Figure 51 illustrates the following steps:

1. Determine *true air speed*.
2. From on-course heading turn 45° . When the aircraft is steady in this off-course heading, carefully measure the drift angle reading.

3. Turn 90° back toward the on-course heading. When the aircraft is steady on this second off-course heading, carefully measure the drift angle reading.

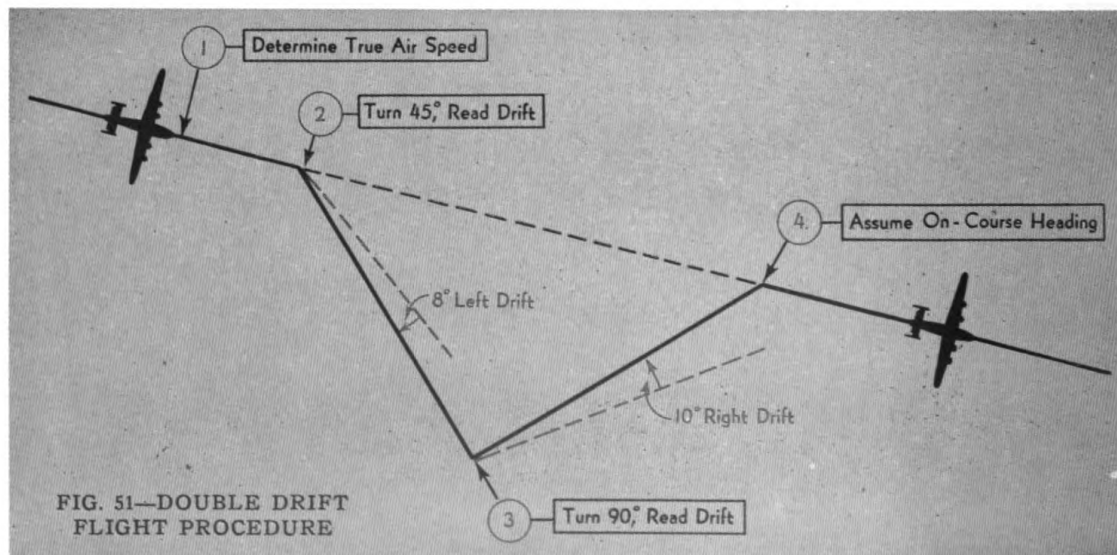
4. Turn 45° to assume on-course heading.

Then solve for wind by the graphic method shown in Figure 52 (sometimes known as the "wind star method"), or directly on the aircraft computer.

Note: If the double drift is taken during daylight hours, whitecaps on the waves will serve as reference points with which to line up the drift grid. At night, a flare must be dropped onto the water to provide a reference point.

Problem—Unable to obtain a celestial fix because the sky is overcast, and finding the radio undependable because of interference, the navigator of an aircraft, on an over-water flight, decides it would be advisable to take a double drift in order to assure maximum accuracy in his dead reckoning.

Upon gaining the captain's approval to proceed with a double drift, the navigator notes the aircraft's true air speed (170 knots) and the true on-course heading (105°). He then requests the captain to steady the aircraft on a true heading of 150° (which is 45° right of the on-course heading). After the aircraft has flown on this heading long enough to permit the navigator to take a careful drift reading (usually about five minutes), the navigator requests the captain to steer a true heading



DEAD RECKONING

of 60° (which is 45° to the left of the on-course heading). The navigator then takes his second drift reading, and after the aircraft has flown on this second off-course heading for the same number of minutes as were required for the first off-course leg, he advises the captain to return to the on-course heading of 105° .

Note: Flying an equal time on each off-course leg will bring the aircraft back to the original track.

Solution (Figure 52)

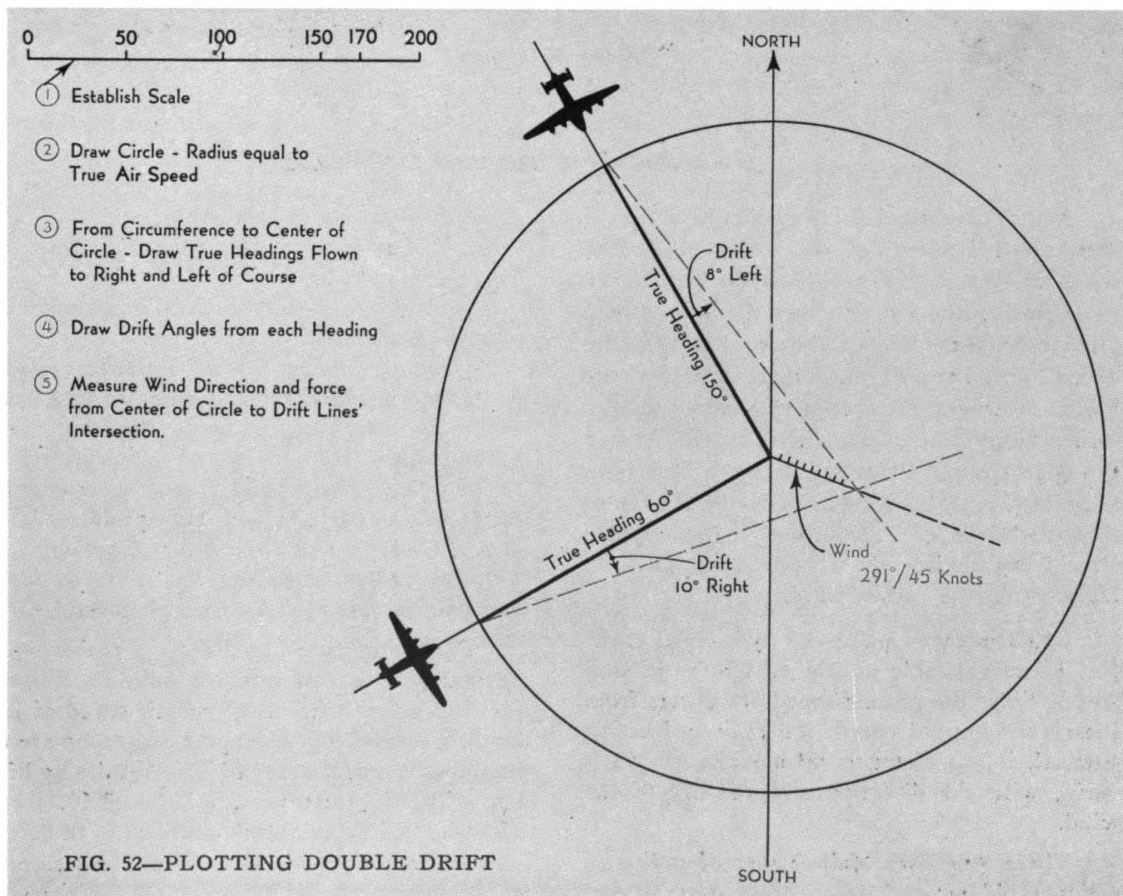
Given: On-course true heading 105°
 True air speed 170 knots
 First true heading 150°
 (45° right of on-course heading)
 Drift 8° left
 Second true heading 60°
 (45° left of on-course heading)
 Drift 10° right

Find: Wind—direction and velocity.

RADIUS OF ACTION

Radius of Action is the distance an aircraft can fly with a given amount of fuel, and under given wind conditions, and still return to its starting base or an alternate base. Radius of action is sometimes called, in ocean navigation, the "point of no return," or "splash point." Once the pilot has flown past this point he cannot safely return, and must reach his destination, or "splash"!

Safe Return means to allow sufficient fuel reserve for safe landing and for unforeseen conditions. In long range flying, from San Diego to Honolulu, for example, fuel for two hours flying deducted from the total flying time is considered a safe reserve above available fuel hours. On shorter flights, with proportionately less fuel, 25% usually is kept in reserve.



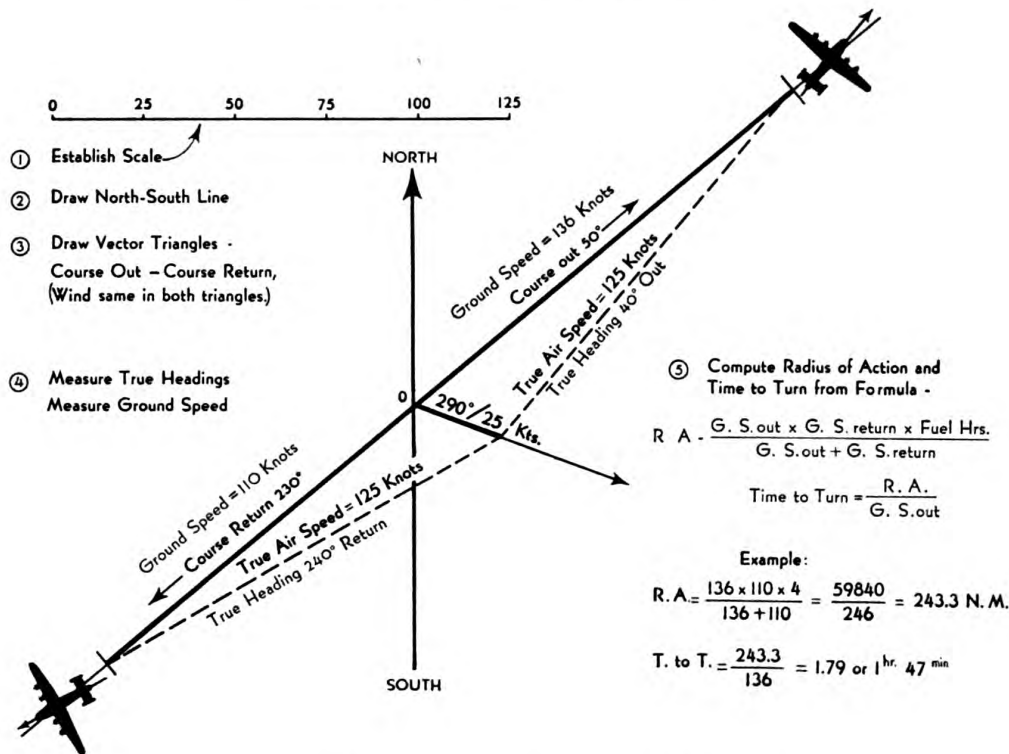


FIG. 53—RADIUS OF ACTION TO SAME BASE

Wind Conditions—The radius of action is determined before take-off, using estimated wind, and the wind is considered to remain the same during the trip out and the return trip. Obviously, the force of the wind affects the ground speed and heading on both legs of the flight. However, the change of heading for the return flight results in a different wind vector triangle. Actual winds encountered in a long range flight sometimes differ considerably from estimated winds. Therefore, the navigator should re-calculate the splash point while in flight, using the known wind.

The Distance a plane can fly is equal to the fuel hours available multiplied by the ground speed. Since the ground speed out differs from the return ground speed, the radius of action—which is the distance out a plane can fly—varies with the direction and velocity of the wind.

There Are Two Main Types of radius of action problem:

1. Returning to the same base.
2. Turning off-course to an alternate base.

In both problems the following information is determined:

Distance out (radius of action); time to turn; true heading out and true heading back or to an alternate; ground speed out and ground speed back or to an alternate. The true headings and ground speeds out and back result in two parts in each type problem, both of which may be solved by vector triangles.

The radius of action and time to turn, however, are solved by special formulas.

Returning to Same Base — Problem — A navigator of an aircraft containing sufficient fuel to fly five hours at a true air speed of 125 knots is notified by the captain that they are to search on a true course of 50° for a missing lifeboat. The forecast wind is from 290°, force 25 knots. The navigator's problem is to determine how far out his aircraft may search and still return safely with a one-hour fuel reserve.

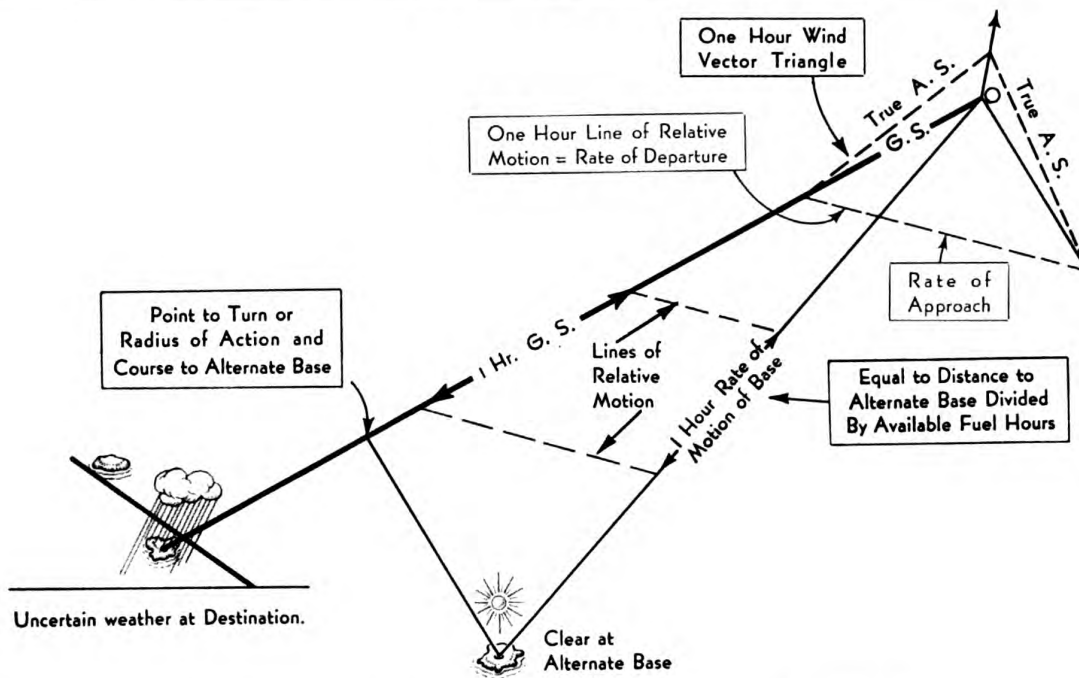


FIG 54—RADIUS OF ACTION TO ALTERNATE BASE: THEORY

Solution (Figure 53)

Given: Course out 50°
 Course return 230°
 (reciprocal)
 Fuel available 4 hours
 (5 hours total, 1 hour reserve)
 True air speed 125 knots
 Estimated wind $290^\circ/25$ knots

Find: Distance out (radius of action)
 Heading out
 Heading return
 Time to turn

Turning To An Alternate Base—This is primarily a safety problem and is employed to determine how far an aircraft can fly toward a desired destination and still return to an alternate base if necessary.

The problem is based on relative motion of the alternate base, which actually may be a moving base, such as an aircraft carrier, or a fixed base, which for purposes of the problem is considered to move. In either case, the aircraft and the alternate base are assumed to start at the same time, from the same position.

Each moves at its respective velocity for the same length of time—an amount equal to the number of available fuel hours—but on different tracks.

Figure 54 illustrates the tracks and relative motion of aircraft and alternate base.

Problem—A navigator of an aircraft containing sufficient fuel to fly seven hours at a true air speed of 150 knots leaves his base airport flying on a course of 250° toward a distant island. The forecast wind is from 200° , force 25 knots; however, the weather report shows a warm front approaching, making safe landing at the desired island uncertain. Weather conditions at an alternate island bearing 230° true and 55° nautical miles from his base airport are clear and expected to remain so. Therefore, since it is necessary to reach the distant island if at all possible, the captain decides to proceed, hoping to arrive before the front closes in. The navigator's problem is to determine how far they can proceed toward the desired island and still, if advised by radio it will be impossible to land, to turn off and reach the alternate island with a safe one and one-half hours of fuel in reserve.

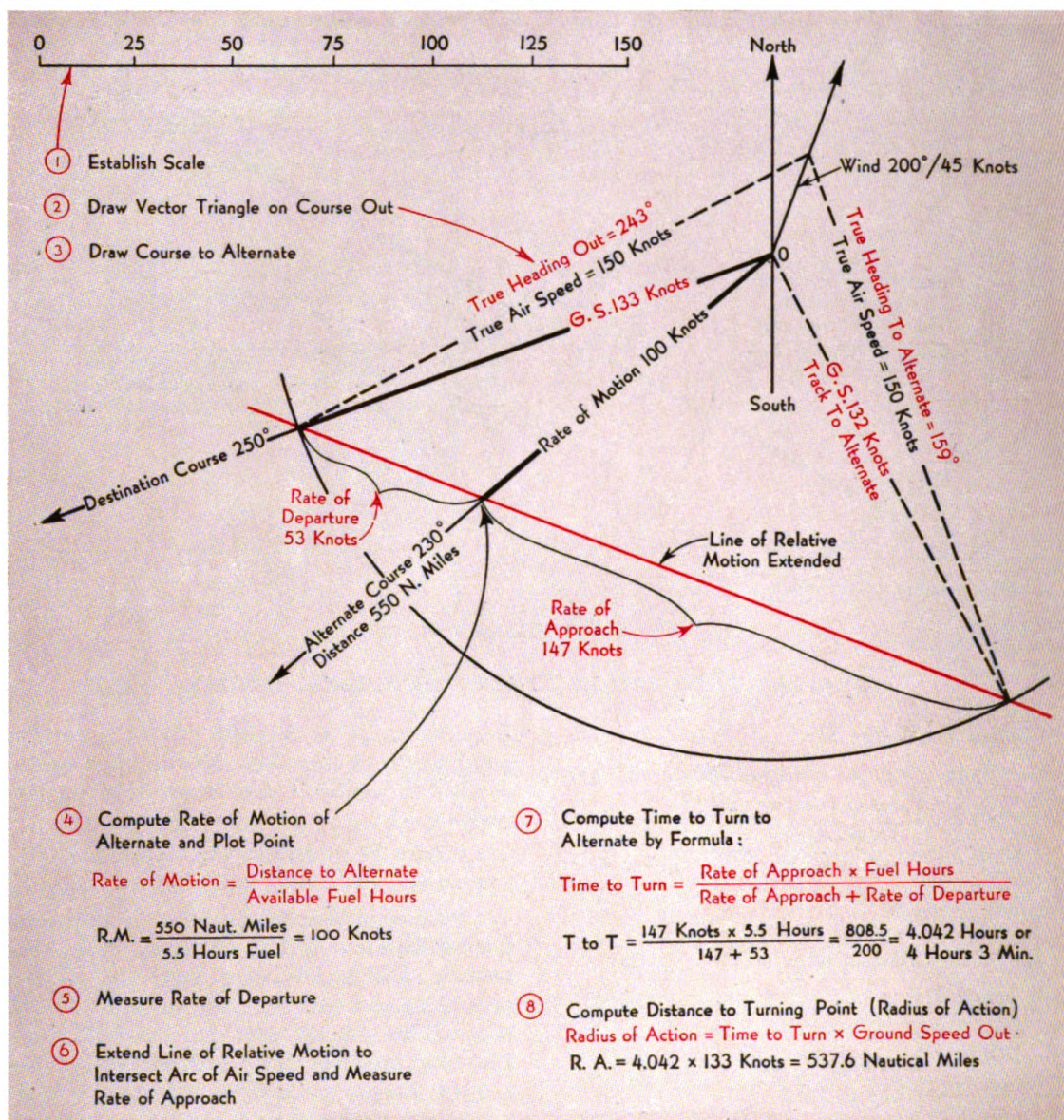


FIG. 55—RADIUS OF ACTION TO ALTERNATE BASE: EXAMPLE

Solution (Figure 55)

Given: Course to destination 250°
 Course to alternate 230°
 Distance to alternate 550
 nautical miles
 Fuel available 5½ hrs. (7 total)
 (1½ reserve)

True air speed 150 knots
 Estimated wind 200°/25 knots

Find: True heading out
 True heading to alternate
 Time to turn
 Distance to turning point

DEAD RECKONING

PROBLEM WORK NO. 6 APPLYING COMPASS ERRORS

(Find missing values.)

No.	TRUE COURSE	VARIATION	MAGNETIC COURSE	DEVIATION	COMPASS COURSE
1	92°		82°	2° E	
2		14° E	256°		259°
3	30°	6° E		1° E	
4		16° E	346°	0°	
5	165°		170°		175°
6	332°	11° W			339°
7	122°			3° W	125°
8		13° E		2° E	344°
9	1°		11°		10°
10	115°			4° E	96°
11		14° W		3° E	234°
12	225°		216°	3° W	
13		3° W	150°	4° W	
14	118°		129°		120°
15	322°	7° E			320°
16	187°	14° W		1° E	
17		16° E	296°		302°
18	334°			4° W	330°
19		14° E	349°	6° E	
20	96°		101°	1° E	

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 7

TRACK AND HEADING

(Find missing values.)

No.	Track True	Var.	Track Magnetic	Drift	Heading Magnetic	Dev.	Heading	
							Compass	True
1	117°	14° E		2° R		3° W		
2		16° W	247°		243°		241°	
3	47°		61°			3° W		51°
4	342°		332°		326°		326°	
5	96°	15° E			78°		71°	
6		16° W	292°		290°		299°	
7	111°	10° W				4° W		110°
8		14° E	228°		226°	3° E		
9	167°		151°	4° L			161°	
10		12° W		0°	221°		217°	
11		10° W		2° L	185°	0°		
12			182°		178°		183°	194°
13		17° W		7° R			120°	107°
14	356°		345°			2° E	344°	
15	6°		352°		353°	10° W		
16		16° E	284°	3° R			275°	
17	249°		261°	1° L		14° W		
18		14° W		4° R		5° W		218°
19				4° L	180°	10° W		204°
20		18° E	100°		96°		90°	

DEAD RECKONING

PROBLEM WORK NO. 8

TIME—SPEED—DISTANCE

(Find missing values. Use computer only for checking work.)

No.	TIME	SPEED (Knots)	DISTANCE (Nautical Miles)
1	1h27m	125	
2	1h34m	136	
3	2h02m	171	
4	0h44m	152	
5	4h13m	111	
6	2h16m	106	
7	3h57m	72	
8	0h52m	176	
9	2h02m		240
10	1h16m		200
11	6h26m		702
12	3h21m		476
13	2h14m		317
14	0h56m		173
15	4h03m		600
16		168	240
17		160	310
18		201	117
19		117	156
20		115	196

PROBLEM WORK NO. 9-A
VECTOR DIAGRAMS

(Draw vector triangles to find: Heading—ground speed—drift. All velocities in knots.)

No.	True Course	I.A.S.	Cali- bration Correc- tion	C.A.S.	T.A.S.	Pressure Altitude	Temp. C°	WIND		Var.	Dev.	HEADING		Ground Speed	Drift
								Dir.	Vel.			True	Compass		
1	85°				90			320°	30						
2	173°				110			84°	40						
3	140°				95			230°	25						
4	300°				120			53°	35						
5	208°				140			304°	20						
6	148°			177		5000'	+15°	223°	50						
7	170°			152		10,000'	+10°	90°	25						
8	240°			124		6000'	— 5°	308°	20						
9	37°			108		25,000'	— 3°	117°	25						
10	0°			170		4000'	+12°	270°	40						
11	350°	140	+8			14,000'	—30°	48°	34						
12	16°	150	+4			10,000'	+15°	131°	29						
13	100°	98	—3			20,000'	— 5°	355°	18						
14	60°	110	—4			16,000'	— 5°	142°	28						
15	330°	129	+6			8000'	+20°	79°	16						
16	204°	125	—5			4500'	+25°	85°	24	2°E	1°E				
17	31°	138	—2			3000'	+30°	286°	19	5°W	3°E				
18	76°	170	—1			5500'	—10°	138°	21	7°E	12°W				
19	190°	124	+2			7000'	+12°	251°	25	2°W	16°E				
20	270°	102	+3			6500'	+15°	49°	32	11°E	17°W				

DEAD RECKONING

PROBLEM WORK NO. 9-B

VECTOR DIAGRAMS

(Draw vector triangles to find wind and drift.)

No.	True Course	Ground Speed (knots)	HEADING		Dev.	Var.	I.A.S. (m.p.h.)	Calibration Correction	C.A.S. (knots)	T.A.S. (knots)	Pressure Altitude	Temp. C°	WIND		Drift
			Compass	True									Dir.	Vel. (kts)	
1	85°	130		75°						144					
2	145°	93		120°						110					
3	290°	110		280°						128					
4	194°	121		206°						118					
5	30°	101		44°						106					
6	48°	138	53°		2°W	4°E				145					
7	95°	119	87°		7°E	5°W				115					
8	210°	162	218°		3°E	2°W				128					
9	176°	178	176°		2°E	15°W				162					
10	300°	147	277°		4°E	3°E				144					
11	275°	109	290°		6°W	1°W			103		3000'	-10°			
12	60°	157	67°		2°W	6°E			153		2500'	-5°			
13	118°	143	94°		10°E	2°E			116		1900'	-7°			
14	4°	99	347°		12°E	2°W			76		20,000'	+3°			
15	17°	111	37°		2°W	0°			76		11,500'	+18°			
16	150°	141	155°		2°W	15°W	130	-6			7000'	+22°			
17	10°	126	12°		8°E	12°E	139	+7			6500'	-4°			
18	138°	101	145°		3°E	1°E	105	-4			2000'	+10°			
19	141°	117	141°		11°E	7°W	142	-6			3500'	-10°			
20	101°	129	101°		2°E	14°W	136	+2			1000'	-13°			

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 10

DOUBLE DRIFT

(Find wind by drawing "wind stars." All velocities in knots.)

No.	T.A.S.	FIRST HEADING	DRIFT	SECOND HEADING	DRIFT	WIND	
						DIRECTION	VELOCITY
1	90	258°	7°R	348°	0°		
2	120	220°	10°L	130°	1°R		
3	160	260°	6°R	170°	1°R		
4	160	0°	13°L	90°	4°L		
5	150	270°	4°R	180°	8°R		
6	130	315°	11°L	15°	6°L		
7	110	0°	0°	270°	10°R		
8	175	90°	8°L	180°	2°L		
9	165	45°	5°R	135°	10°R		
10	145	55°	10°R	145°	2°R		
11	155	75°	5°R	345°	5°R		
12	160	85°	5°R	175°	0°		
13	180	120°	13°L	30°	5°L		
14	135	140°	5°R	230°	8°R		
15	150	160°	8°L	70°	2°L		
16	170	180°	5°R	80°	6°R		
17	130	190°	7°R	280°	3°R		
18	185	200°	10°L	110°	6°L		
19	140	240°	7°R	150°	2°R		
20	120	220°	10°R	310°	4°R		

DEAD RECKONING

PROBLEM WORK NO. 11-A RADIUS OF ACTION TO A FIXED BASE

(Draw vector diagrams and find missing values.)

No.	True Course Out	Wind (knots)	C.A.S. (knots)	Temp. C°	Pressure Altitude	T.A.S. (knots)	Fuel Hours	Fuel Reserve	Fuel Available	GCT Take-off Time	True Heading		Radius of Action	Time to Turn
											Out	In		
1	82°	50°/30	182	+ 2°	6000'		6	10%		09:30				
2	310°	250°/30	160	— 3°	4000'		4	8%		10:20				
3	198°	280°/30	200	— 5°	5000'		3	30m		14:20				
4	250°	280°/30	140	— 4°	10,000'		10	5%		16:30				
5	270°	30°/24	125	+10°	3000'		4	10%		10:30				
6	200°	90°/36	111	—20°	8000'		4:30	30m		07:10				
7	50°	270°/10	110	+10°	9000'		9	1h		06:40				
8	40°	300°/30	90	—10°	5000'		5	—		15:15				
9	20°	270°/70	130	—20°	10,000'		8	25%		01:24				
10	135°	70°/44	175	0°	3000'		7	15%		00:45				

PROBLEM WORK NO. 11-B
RADIUS OF ACTION TO ALTERNATE BASE

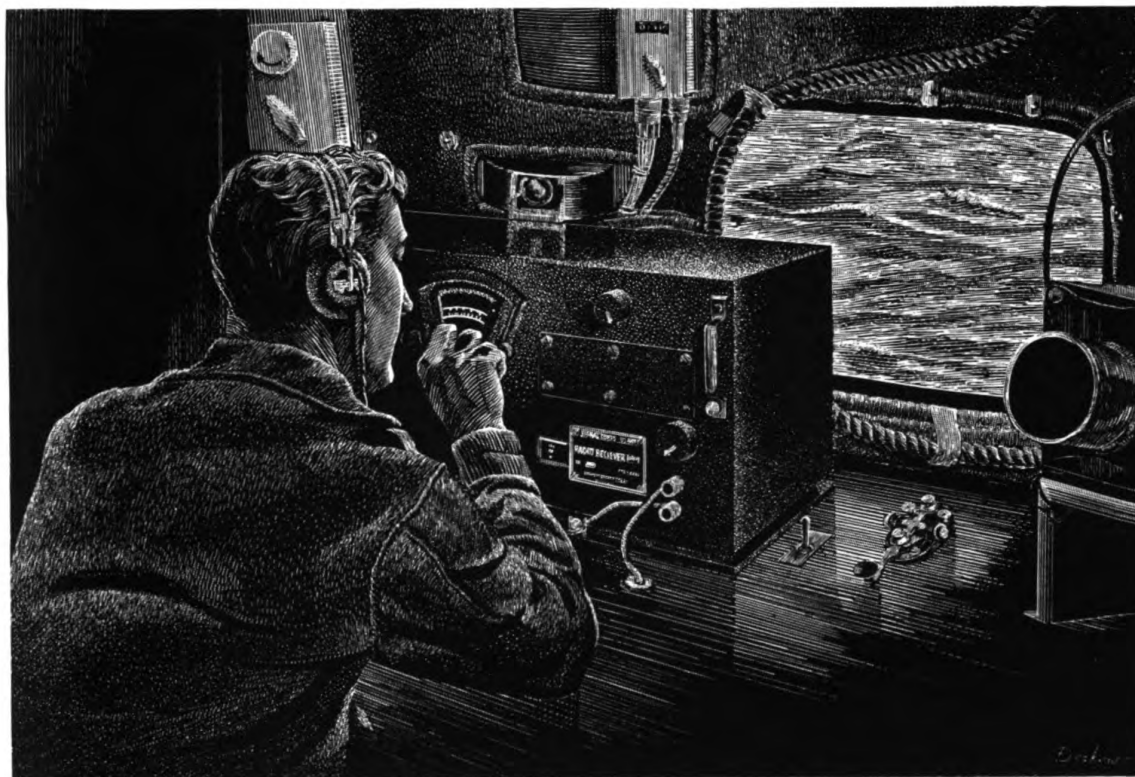
(Draw vector diagrams and find missing values.)

No.	TRUE COURSE TO		Distance to Alternate (Nautical Miles)	Wind	T.A.S. (knots)	Fuel Hours	Fuel Reserve	Fuel Available	TRUE HEADING TO		Distance Out	Time to Turn
	Destination	Alternate							Destination	Alternate		
1	50°	76°	280	270°/30	90	4	—					
2	240°	170°	180	180°/32	90	4	10%					
3	30°	340°	180	170°/20	80	5	10%					
4	270°	230°	245	45°/30	120	3.5	—					
5	130°	180°	200	210°/30	200	5	—					
6	350°	250°	180	270°/31	110	5	30 min.					
7	20°	190°	225	80°/60	100	6	20%					
8	110°	360°	600	220°/40	160	5	—					
9	190°	245°	300	30°/30	140	3	30 min.					
10	290°	200°	360	340°/15	110	4	10%					

DEAD RECKONING REVIEW TEST

1. Define :

(a) Latitude	(d) Variation
(b) Longitude	(e) Rhumb Line
(c) Great Circle	
2. Draw diagrams showing the projection and development of the Mercator, Lambert, and gnomonic charts, and list advantages and disadvantages of each.
3. Show how rhumb line track and great circle track appear on gnomonic chart and also on Mercator chart.
4. (a) Compass course is 302° , variation 10° W, deviation 4° E. What is magnetic course? What is true course?
 (b) True heading is 210° , variation 16° E, drift 4° left, deviation 3° W. What is true track? Compass heading? Magnetic track?
 (c) True course is 33° , variation 12° E, deviation 6° E, drift 4° right. What is true heading? Compass course? Magnetic course?
5. Pressure altitude is 8000 ft., indicated altitude 9000 ft., temperature $+8^\circ$ C. What is true altitude?
6. (a) Indicated air speed is 179 m.p.h., instrument correction $+2$ m.p.h., altitude 10,000 ft., temperature $+10^\circ$ C. What is calibrated air speed in knots? True air speed in knots?
 (b) Calibrated air speed is 157 knots, altitude 12,000 ft., temperature 0° C. What is true air speed in knots?
7. San Diego to position "A" is 510 nautical miles. Aircraft's true air speed is 180 knots, wind directly astern at 19 knots, time of departure is 03:10 GCT.
 - (a) At what time will plane reach position "A"?
 - (b) If aircraft arrives at position "A" at 07:30 GCT, what is the ground speed?
 - (c) If ground speed is 115 knots and aircraft arrives at position "A" at 09:10 GCT, what is the distance to position "A"?
8. (a) True air speed is 150 knots, wind $30^\circ/30$ knots, course 272° . What will be the estimated ground speed? True heading? Drift?
 (b) True heading is 60° , true air speed 100 knots, wind $350^\circ/25$ knots. What will be the estimated ground speed? Track? Drift?
 (c) Track is 195° , true heading 210° , true air speed 190 knots, ground speed 210 knots. What is wind? Drift?
9. While on a true heading of 230° , true air speed 120 knots, a navigator decides to take a double drift to determine the wind. His first true heading is 275° , drift 8° right. His second true heading is 185° , drift 10° left. What is direction and force of wind?
10. With 6 hours of fuel available over the required reserve, how far can an aircraft fly on true course of 220° , at a true air speed of 180 knots, wind $300^\circ/33$ knots and still return to its starting point? If the aircraft leaves San Diego at 09:15 GCT, what time should it turn around?



☆ 4 ☆

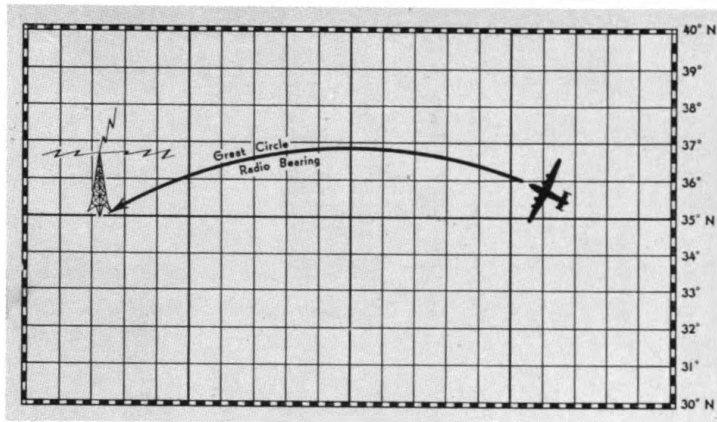
RADIO NAVIGATION

RADIO navigation is the method of determining an aircraft's position with relation to, or its direction from, known radio transmitting stations. To accomplish this orientation, advantage is taken of the directional characteristics of a loop antenna. However, because of interferences affecting transmission, and due to design limitations of present-day radio equipment, plus the fact that reception and use of radio waves are limited to the skill of the operator, radio navigation is not as accurate as celestial navigation or pilotage. Hence it should be used only as an *aid to navigation* in conjunction with all other available methods.

Radio Wave Characteristics—The radio bearing between a ground station and an air-

craft is a direct line over the surface of the earth; hence, it follows a great circle track (Figure 56). Drawn on a Mercator chart, the bearing would appear as a curved line bending toward the elevated pole (Figure 57).

The path of a radio wave, however, often is refracted, reflected, or otherwise caused to deviate from a great circle track by storms, by mountains, by passage over a coastline or by the phenomenon known as "night effect." *Night effect* is the change in reflection of the radio wave from the ionosphere, and is always encountered; however, the effect is more pronounced at sunrise and sunset (hence "night effect"), and may be recognized by a fluctuation in bearings. Night effect may be reduced



☆
 FIG. 57—RADIO BEARING
 APPEARS AS CURVED LINE
 ON MERCATOR CHART
 ☆

somewhat by varying the altitude of the aircraft, by taking an average of the fluctuations, or by selecting a station of lower frequency.

Value of Radio—Even with present-day limitations and the difficulties inherent in wave transmission, radio is an invaluable aid to navigation. With the development of better equipment and the continual increase in the number of ground stations throughout the world, it will continue to assume more importance.

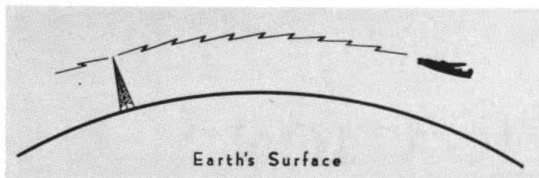


FIG. 56—RADIO BEARING FOLLOWS
 GREAT CIRCLE PATH

With radio, an aircraft can remain in constant communication with the ground as well as with other aircraft. By means of the radio direction finder (radio compass), bearings may be taken of a transmitting station, either on the ground or in another aircraft, with which the navigator can establish his position with relation to the transmitting station. By means of radio range stations, an aircraft's position can be determined so accurately that landings may be made even without visual contact with the ground.

RADIO COMPASS

Description—The radio compass is an instrument for receiving radio wave signals and measuring the *relative bearing* of the transmitting station from the aircraft's heading. The

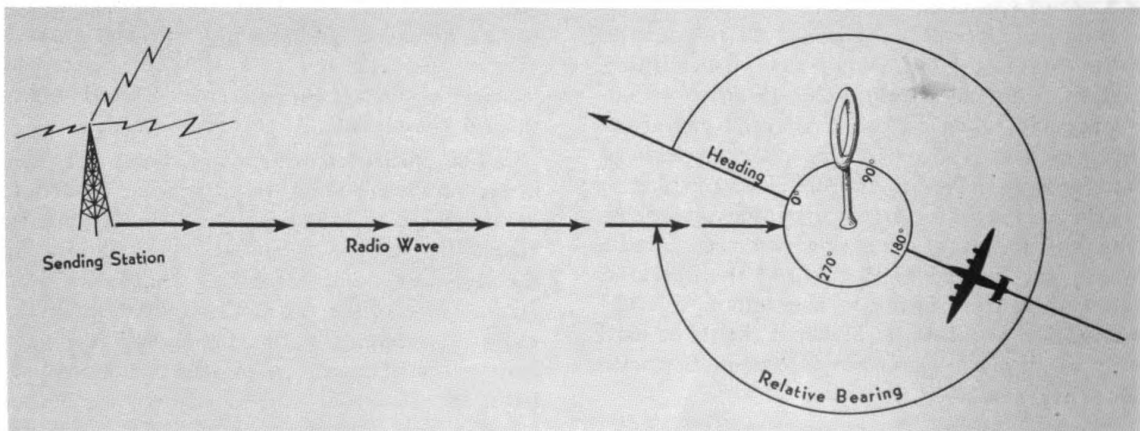


FIG. 58—ESTABLISHING RELATIVE BEARING WITH ROTATABLE LOOP

principle of operation involves the directional characteristics of a *loop antenna*. When the loop is at *right angles* to the direction of incoming radio waves, minimum signal volume is heard; maximum volume is heard when the loop is *parallel* to the incoming signal waves (Figure 58).

By use of the radio compass, direction of the transmitting station is established on an azimuth ring (Figure 59) by a pointer actuated by a rotatable loop antenna. The azimuth ring is graduated in a clockwise direction from 0° to 360°, the 0° and 180° points being aligned with the longitudinal axis of the aircraft. When the pointer is at 0°, the loop is at right angles to the longitudinal axis.

When the loop antenna is swung back and forth until the mean point of minimum volume is established, the azimuth dial then indicates the relative bearing of the transmitting station from the aircraft's heading. The point of minimum volume is used because, due to the antenna loop characteristics, that point is easier to detect than the point of maximum volume.

Loop antennas may be either manually or automatically operated.

Manually Operated Loop—The loop is rotated by a hand-operated crank. When the station signal received is at a minimum volume, the loop is at right angles to the radio wave received, indicating the *station's relative bearing*.



Courtesy Bendix Radio Division, Bendix Aviation Corporation

FIG. 59—AZIMUTH DIAL



Courtesy Bendix Radio Division, Bendix Aviation Corporation

FIG. 60—CONTROL PANEL

Automatically Operated Loop—When the station is correctly tuned in, an electric motor automatically rotates the loop, keeping it at right angles to the station from which the radio wave is emanating.

Note: If the crank mechanism or motor fails, it is possible in some installations to rotate the loop by hand. In such a case, the loop should be turned perpendicular to the aircraft's longitudinal axis. With the loop in this position, the aircraft can be pointed toward the transmitting station by altering the heading until the signal wave is at minimum volume. The bearing of the transmitting station is then indicated on the magnetic compass, relative to compass North.

BENDIX RADIO COMPASS

The following is the general procedure for operating the Bendix radio compass equipment as it is used by the ocean aerial navigator for homing to a required destination, for taking bearings, and for fixing the aircraft's position. Other installations are similar, and procedures are practically the same.

Master Switch—The master switch on the control panel (Figure 60) has four contacts: OFF, COMP (Compass), REC ANT (Receive Antenna), REC LOOP (Receive Loop), which

control all radio compass functions other than tuning and adjustment of signal levels, as follows:

1. OFF—No current is going through the set.
2. COMP—The equipment functions as a direction finding receiver connected to the loop and sense antenna, operating the *left-right indicator*.
3. REC ANT—The equipment is connected to the non-directional vertical antenna. As such, it is used for aural radio range reception and for obtaining communication reception.
4. REC LOOP—The equipment functions as a communication receiver connected to the directional loop antenna. This contact is used for aural-null bearings, aural-null homing, and for obtaining communication reception or listening to radio range reception during conditions of severe static caused by rain or snow.

Left-Right Indicator (Figure 61) — The left-right indicator is an instrument having a pointer in the form of a small conventionalized figure of an airplane which visually indicates "null" radio bearings. The pointer is actuated by the phase relationship of incoming radio waves. Minimum volume causes the pointer to remain at the center of the instrument, thus indicating that the loop antenna is at right angles to the incoming radio wave. When the transmitting station is ahead, rotation of the

azimuth dial or changes in aircraft's heading in one direction, will cause the left-right pointer to move in the opposite direction. If the pointer moves in the same direction, the station lies behind the aircraft.

Preliminary Operation

- a. Turn the *master switch* on.
- b. Carefully tune the receiver to the frequency of the desired radio station and adjust the *audio control volume*.
- c. Listen to the station's *call sign*, or *identification signal*, in order to make certain that the correct station has been tuned in.

HOMING

Procedure

1. FOR VISUAL HOMING (using left-right indicator).

- a. Adjust *master switch* to COMP.
- b. Set azimuth dial to read *zero*. The *needle* of the *left-right indicator* now points in the general direction of the transmitting station.
- c. Alter the heading of the aircraft to right or left so that the *indicator needle* is centered. The aircraft now will be pointing toward the desired station.
- d. If the needle is being continuously deflected right and left of the center, reduce the *compass control volume*, permitting the course to be followed more accurately.

2. FOR AURAL-NULL HOMING (using headphone)—This method of homing is used if the left-right indicator fails, or in case of severe rain static. It is called "flying a null course," and is identical with the method used in visual left-right homing, the only difference being that with this method an audible signal is received via the headphones, and the pilot alters the aircraft's heading to right or left to maintain the signal volume at a minimum.

If the signals are weak, turn the **CW** switch on.

Note: Homing is the easiest method of approaching a station on course. However, it should be used only when the aircraft is less than one hundred miles from the station. At greater distances, it would be more practical to plot radio compass bearings and establish compass headings to fly. This is true primarily



Courtesy Bendix Radio Division, Bendix Aviation Corporation

FIG. 61—LEFT - RIGHT INDICATOR

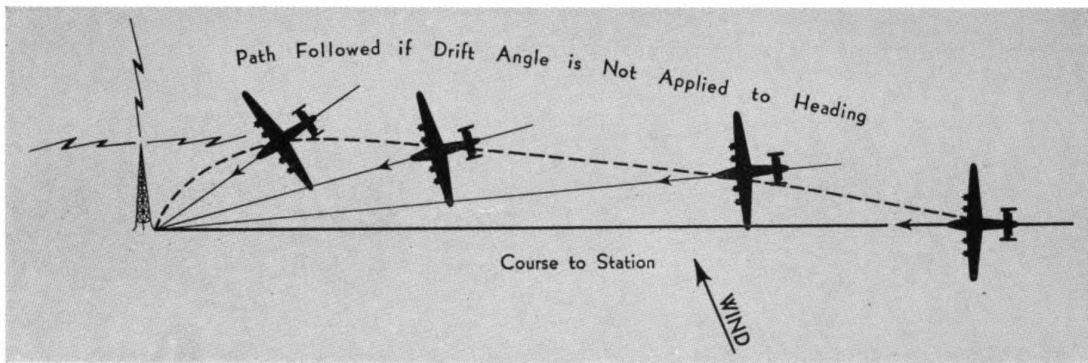


FIG. 62—EFFECT OF WIND IN HOMING

because of the effect of crosswind on the aircraft, for when homing with a strong beam wind blowing, the aircraft's heading has to be altered continuously into the wind in order to keep on the null course, or to keep the needle of the left-right indicator centered. Figure 62 illustrates the effect of wind in homing.

RELATIVE RADIO BEARINGS

Definition—A *relative radio bearing* is the angular direction of the transmitting station measured clockwise from the *true heading* of the aircraft.

Calibration Error—Practically all aircraft radio compass installations give definite errors in relative bearings. This error is called *calibration error* and must be applied to all *relative radio bearings* in order to find the *correct relative bearing*.

Calibration error is a result of the effect which the metallic fore and aft mass of the fuselage structure, athwartship mass of the wing structure, and other metallic masses in the aircraft have on the rotatable loop antenna. It is determined by a method similar to that used to find compass error, namely, by swinging the aircraft on a compass rose, or in the air, and recording the error found on each 15° heading throughout 360°. The 0° bearing, however, is obtained by heading the aircraft toward the transmitting station and taking the bearing over the nose of the ship. Hence the 0° point on the azimuth ring indicates the nose of the aircraft and not geographical North, as is the case with the magnetic compass.

From the data thus obtained, a calibration graph is drawn showing the amount of correction to be applied to any relative bearing.

Procedure

1. FOR VISUAL RELATIVE RADIO BEARING (using left-right indicator).

- a. Adjust *master switch* to COMP.
- b. Rotate *loop* until *needle* of left-right indicator is centered.
- c. Read *relative bearing* on the *azimuth dial* and apply *calibration error* to obtain *correct relative bearing*.

2. FOR AURAL - NULL RELATIVE BEARINGS (using headphones).

- a. Adjust *master switch* to position of REC LOOP.
- b. Rotate *loop* until the headphone volume decreases to a minimum.
- c. Read *relative bearing* on the *azimuth dial* and apply *calibration error* to obtain *correct relative bearing*.

True Bearings—In converting relative bearing to true geographical bearing, employ the following procedure:

- a. Note the *compass heading* of the aircraft at the instant the *relative bearing* is read.
- b. Convert *compass heading* to *true heading* by applying *variation* and *deviation*.
- c. Add *correct relative bearing* and *true heading*. Since true heading is the direction of the aircraft's heading from true North and the relative radio bearing is the direction of the transmitting station from the aircraft's heading, the sum equals the true geographical great circle bearing of the transmitting station from the aircraft.

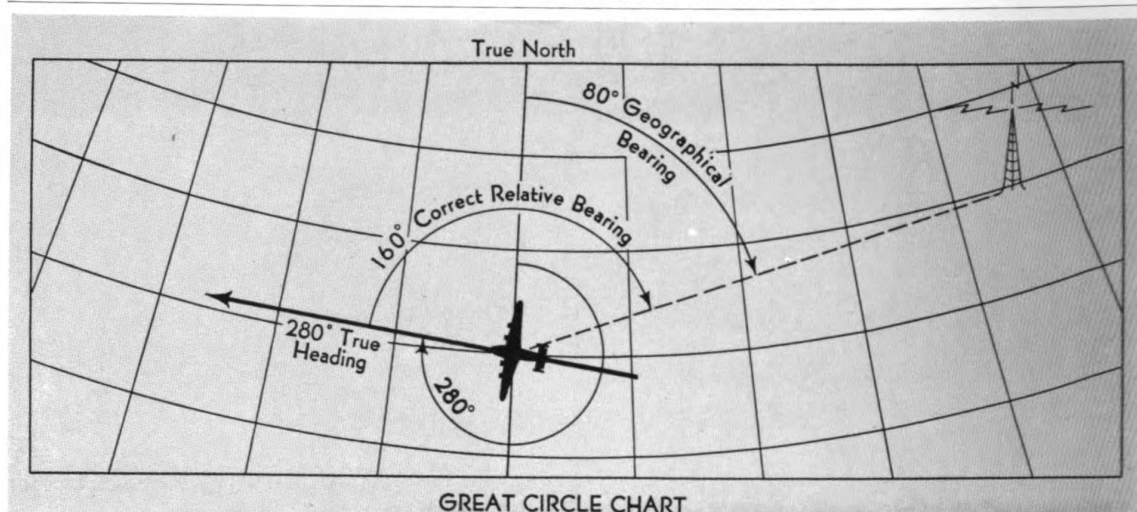


FIG. 63—DETERMINING GREAT CIRCLE BEARING

Note: If the sum of the relative bearing and true heading is over 360°, subtract 360°.

Example (Figure 63)

Compass heading	=	265°
Deviation	=	5° E
Magnetic heading	=	270°
Variation	=	10° E
True heading	=	280°
Relative radio bearing	=	155°
Calibration correction	=	+5°
Correct relative bearing	=	160°
True heading	=	280°
Sum	=	440°
Less 360°	=	-360°
Great circle bearing	=	80°
Mercator correction	=	+2°
Rhumb line bearing	=	82°

Mercator Correction—Since radio waves follow a great circle path which, on the Mercator chart appears as a curved line bending toward the elevated pole, a correction has to be applied to all radio bearings before they can be plotted as straight lines (rhumb lines) on the Mercator chart (Figure 65).

Correction Amount—The amount of correction to be applied in order to convert the great circle bearing into a rhumb line bearing varies with the *mid-latitude*, and the *difference of longitude* between the aircraft and the transmit-

ting station. This correction is gained by inspection of a radio bearing conversion table (Figure 64). The table should be part of the navigator's equipment and is in either tabular or graphic form.

Correction Sign—The *sign* of the correction is either *plus* or *minus* depending upon the position of the receiver with relation to the transmitting station. The sign of the correction for any position may be determined as shown in Figure 66. Thus, in Figure 64 or 66, when the receiver is West of the transmitting station (North latitude), the great circle bear-

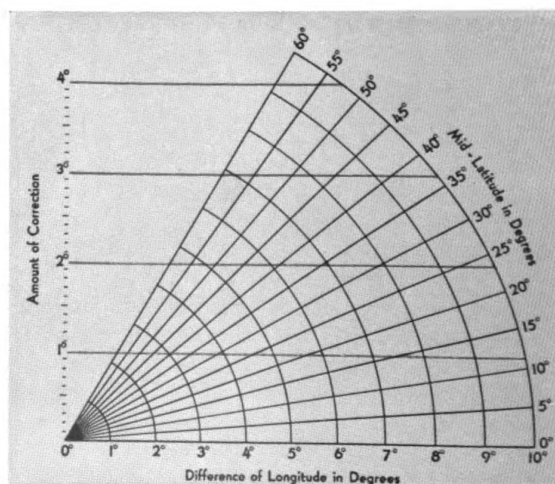


FIG. 64—MERCATOR CORRECTION DIAGRAM

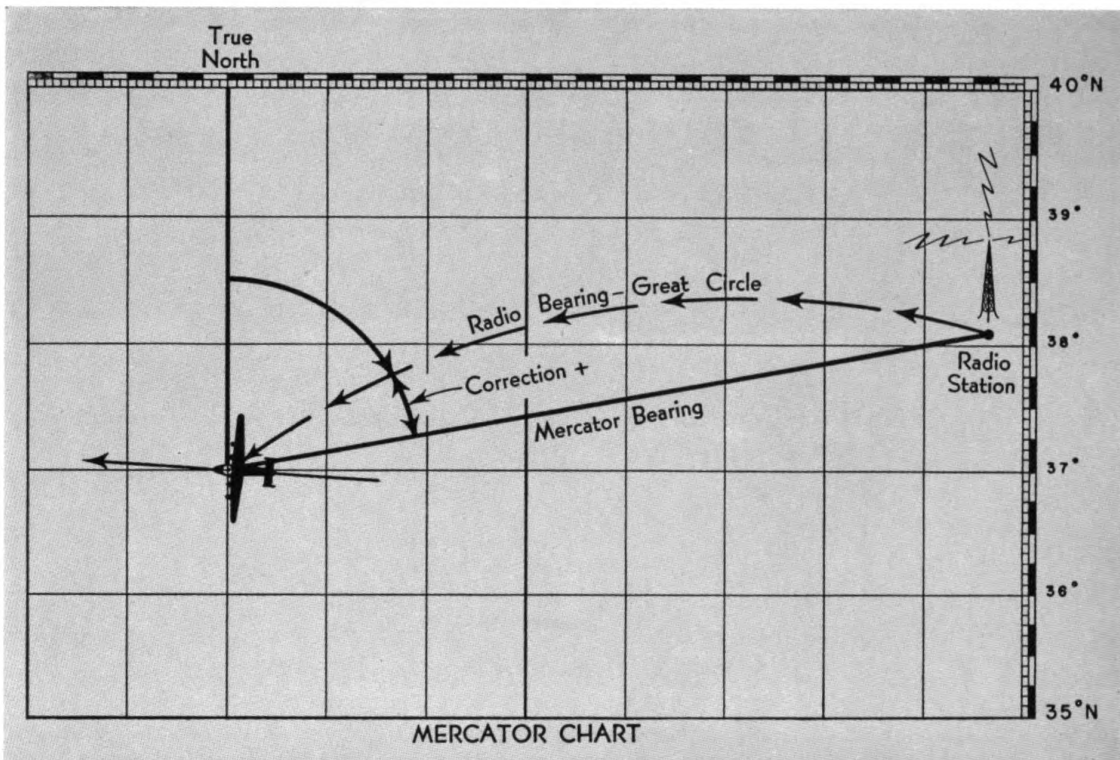


FIG. 65—CONVERTING GREAT CIRCLE BEARING TO MERCATOR BEARING

ing, which bends toward the poles, is less than the rhumb line bearing. Therefore, the correction is plus. In South latitude, the correction would be minus because the great circle there bends toward the South pole.

RADIO RANGES

Description—Radio range beacons are the safest and most widely-used form of directional guidance in the United States. Hundreds of radio range beacons mark every airway commercially flown in this country, and now, with

the establishment of island air bases, they aid the navigator throughout the world.

Radio range beacons transmit dot-dash (A) and dash-dot (N) signals in alternate quadrants. Neighboring quadrants overlap, forming a narrow beam identified by a continuous signal which is a result of the overlapping of the A and N signals. These narrow, wedge-shaped beams, or equisignal zones, indicate to the pilot that he is flying the safest on-course heading to approach or depart from the desired radio range station. The quadrant signals of

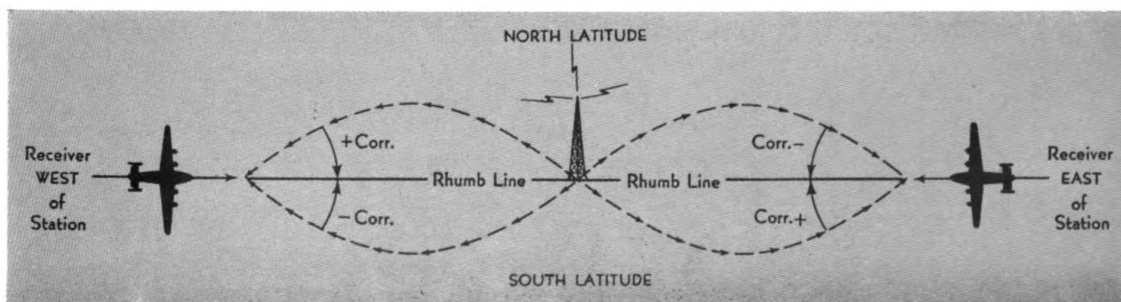


FIG. 66—DETERMINING SIGN OF MERCATOR CORRECTION

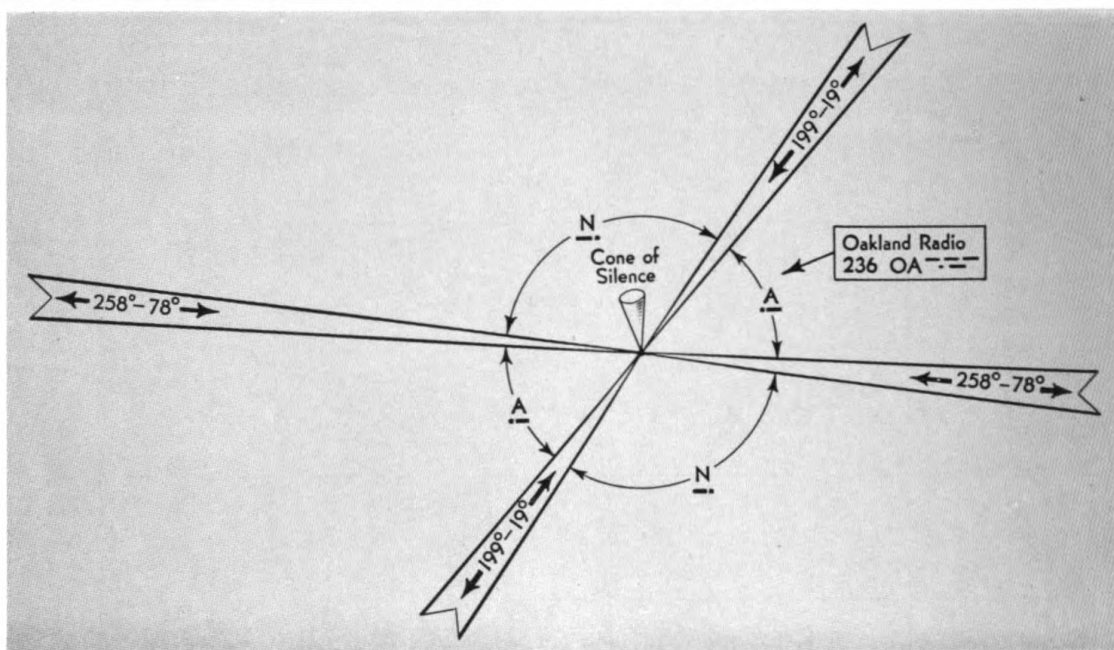


FIG. 67—TYPICAL RADIO RANGE

dot-dash and dash-dot (A and N) indicate to him the side of the equisignal zone from which he is approaching the station, or toward which he is drifting. When the radio compass is used for range reception, the master switch is on REC COMP. However, when the switch is on REC COMP, range reception is not entirely reliable. Therefore, the on-course range signal should be checked frequently by turning the switch to REC ANT, which is the best position for radio range reception.

Procedure—To follow a radio range course, it is necessary to have a chart (as illustrated in Figure 67) showing the radio range course and characteristics. The magnetic direction of the courses to and from the station are indicated, as are the locations of the A and N quadrants with relation to the various equisignal zones. The frequency and identification signals of the station also are noted. The procedure is as follows:

- a. Adjust the master switch to REC ANT.

Note: In case of rain-static, adjust master switch to REC LOOP and rotate loop parallel to radio wave for maximum signal strength.

- b. Turn aircraft so as to intercept the radio

range course. When *on-course*, the A and N signals will blend into a continuous dash, only interrupted by station identification.

- c. As the aircraft approaches the station the volume increases, but on arrival over the station, a decrease in volume occurs as the plane passes over what is known as the "cone of silence." If the master switch is on REC LOOP, the cone of silence will not be "heard."

USE OF RADIO BEARINGS

Properties and Limitations—True radio bearings are very useful aids to the aerial navigator who thoroughly understands their properties and limitations. These may be summarized as follows:

1. A true radio bearing is the direction of the transmitting station from the receiver relative to true North. If the receiver is an aircraft, the reciprocal of the bearing would be plotted through the known geographical position of the transmitting station.
2. The bearing is a position line; that is, it establishes the aircraft's position as somewhere on the straight line passing through the transmitting station. It does not, however, indicate the aircraft's distance from the station.

RADIO NAVIGATION

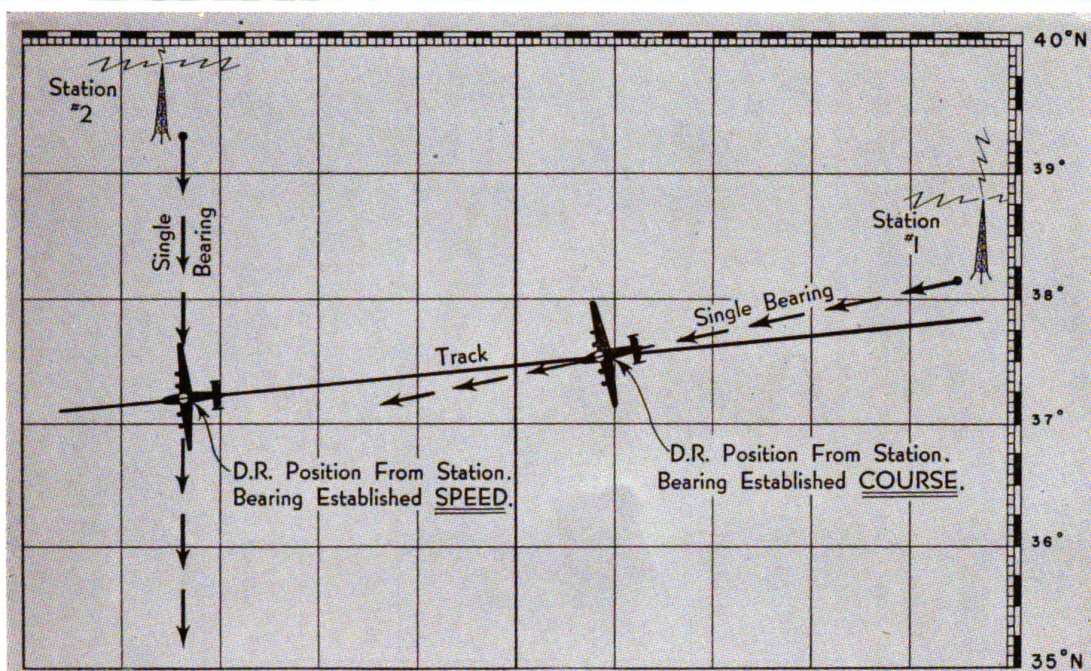


FIG. 68—POSITION DETERMINED BY SINGLE RADIO BEARING AND DEAD RECKONING

3. The bearing is a great circle, and if it is to be plotted on a Mercator chart, it must first be converted to a rhumb line bearing.

4. The bearing may not be absolutely ac-

curate; therefore, the navigator should analyze it carefully to determine the most likely position of the aircraft, allowing for any estimated errors.

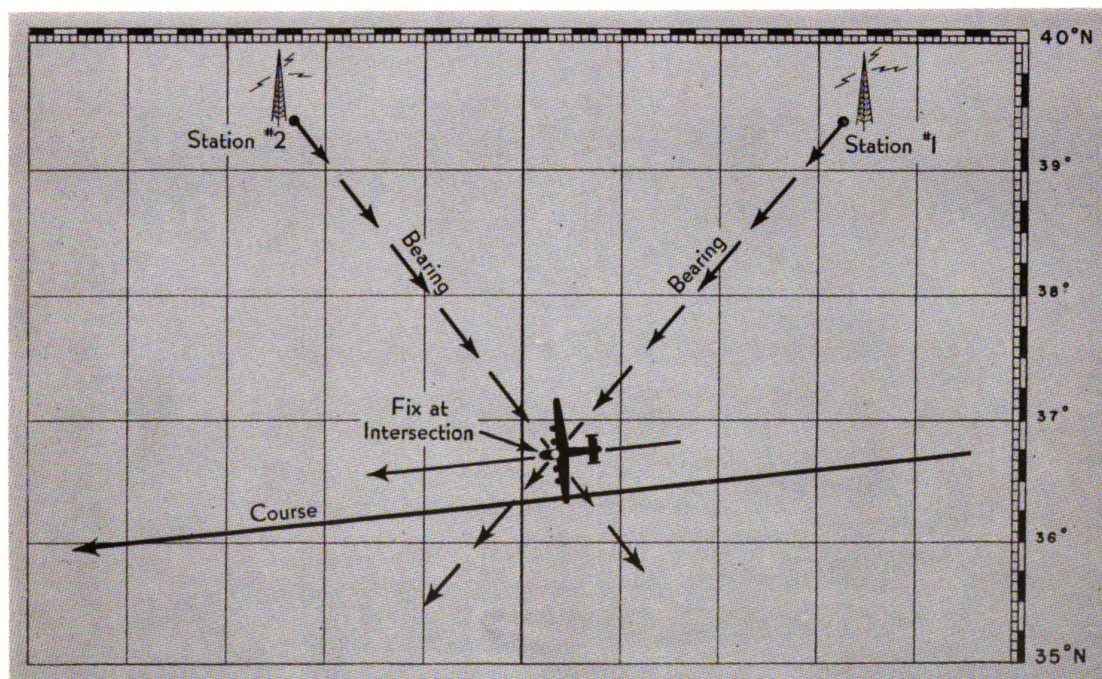


FIG. 69—POSITION DETERMINED BY INTERSECTION OF TWO RADIO BEARINGS

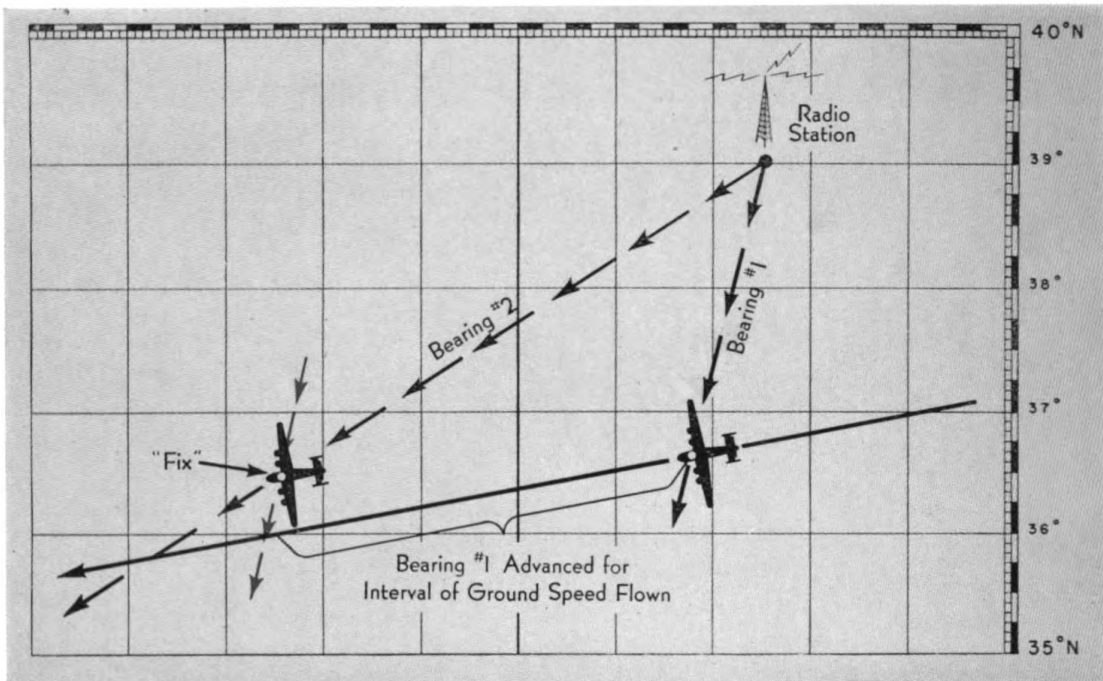


FIG. 70—POSITION DETERMINED BY TWO BEARINGS ON SAME STATION

Obtaining Fix—Methods most often used to determine the aircraft's position by use of radio bearing lines are as follows:

1. **Single Bearing and Dead Reckoning**

(Figure 68)—With only a single radio bearing the navigator may, by reasoning and deduction, estimate the aircraft's most probable dead reckoning position on the radio position line.

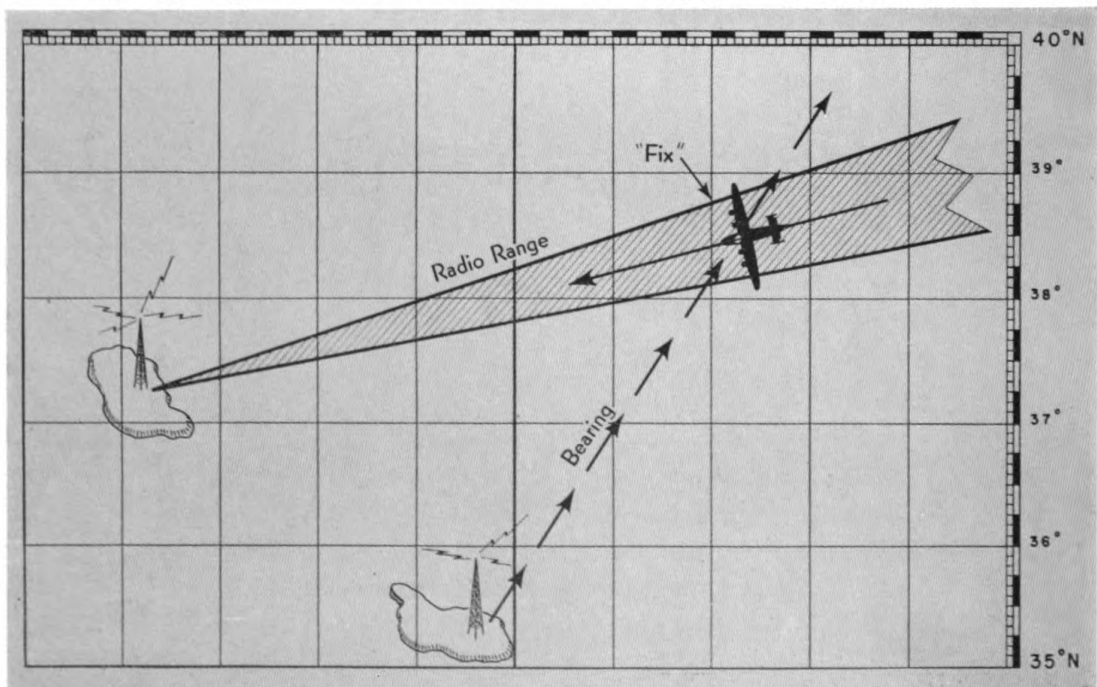


FIG. 71—POSITION DETERMINED BY RADIO BEARING AND ON-COURSE RANGE SIGNAL

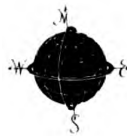
If the bearing is parallel to the track, it will serve to check the aircraft's track. If the bearing is at right angles to the track it will serve to check the aircraft's speed.

2. Two or More Bearings of Different Stations (Figure 69)—A single radio bearing establishes the aircraft's position as being somewhere on the bearing line. But when two or more bearings are taken of different stations at the same time, the only position possible to the aircraft is at the intersection of these bearing lines, hence its exact location is determined (assuming no errors).

3. Running Fix (two bearings of same station) (Figure 70)—With the track (or course) and ground speed of the aircraft known, a position fix can be determined by taking two bearings of the same station, allowing an in-

terval of time between bearings. During this time interval the bearing of the station will alter due to the change of position of the aircraft. And because the aircraft was located somewhere on each bearing line (at the instant the bearing was taken), the first bearing line can be advanced along the track a distance equal to the product of ground speed and time interval between bearings. The intersection of bearings establishes the aircraft's position.

4. Bearings Intersecting Range Courses (Figure 71) — Bearings intersecting range courses establish position of the aircraft in the same way as do two or more bearings taken of different stations. The radio range establishes the course of the aircraft, and a bearing taken on another station will intersect the range course, establishing the aircraft's position from the range station.



PROBLEM WORK

No. 12 Draw relative radio bearing diagram.

No. 13 Draw diagram showing how sign of Mercator correction is determined.

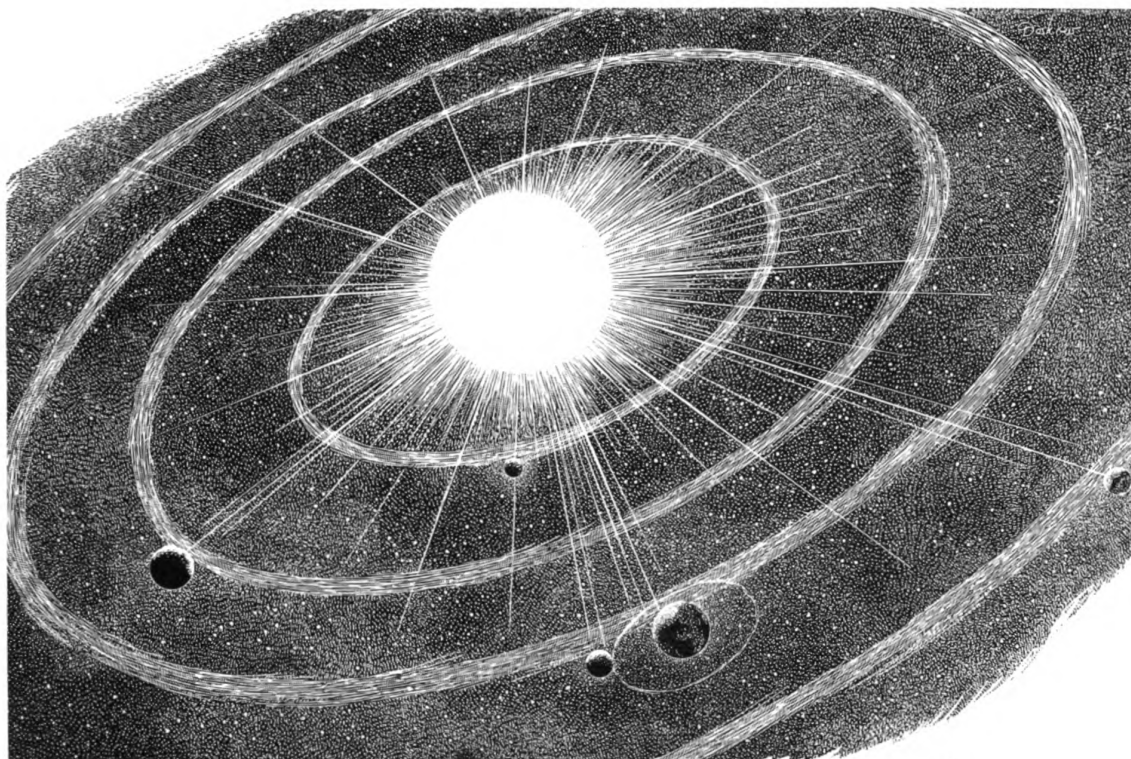
AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 14

RADIO BEARINGS

(Convert relative radio bearing to true Mercator bearing.)

No.	Relative Bearing	Calibration Correction	Corrected Relative Bearing	Compass Heading	Var.	Dev.	True Heading	True Great Circle Bearing	Mercator Correction	Mercator Bearing From Station
1	330°	-10°		270°	5° E	0°			-1°	
2	200°	+10°		310°	7° E	2° E			-1°	
3	25°	+10°		45°	15° W	1° W			-4°	
4	45°	+15°		185°	18° W	4° E			+1°	
5	100°	- 5°		140°	10° E	1° E			+3°	
6	360°	0°		77°	7° W	7° W			+4°	
7	180°	0°		295°	14° E	4° W			0°	
8	10°	+ 5°		315°	0°	3° E			+2°	
9	90°	0°		70°	5° E	2° W			0°	
10	270°	0°		360°	11° W	10° E			-2°	
11	350°	- 5°		100°	17° E	4° E			+4°	
12	20°	+ 7°		340°	13° W	6° W			-3°	
13	70°	+ 5°		50°	4° E	2° E			0°	
14	30°	+10°		210°	9° E	3° W			+2°	
15	320°	-15°		17°	11° W	2° W			-3°	
16	15°	+ 6°		289°	16° E	5° E			-1°	
17	105°	-11°		73°	10° E	6° E			-4°	
18	210°	+12°		359°	4° W	4° W			-3°	
19	50°	+ 7°		178°	11° E	2° W			-1°	
20	170°	- 6°		46°	1° E	1° W			+1°	



☆ 5 ☆

CELESTIAL SPHERE

WHEN surface features of the earth are visible, the position of an aircraft can be determined very accurately by pilotage (navigation with reference to prominent landmarks). Two other methods of determining the position of an aircraft—radio navigation and dead reckoning navigation—have been described in preceding chapters.

Each of these methods of navigation, however, is subject to certain limitations. In dead reckoning, for example, the direction and velocity of the wind seldom are accurately known, and as a consequence an element of error always is present in the determination of a DR position. This error is *cumulative*, that is, it

becomes progressively larger the farther the DR position is from a known position or fix. In long-range ocean flying it usually is impossible to check DR positions by reference to landmarks, such landmarks seldom being available. This fact makes navigation by DR unreliable, and navigation by pilotage impossible. Finally, mechanical factors and atmospheric conditions can cause radio bearings to be in error.

Thus it is apparent that any one, or all three of these methods of navigation might prove unreliable at any given time. Therefore, it is essential that the navigator preparing for long-range ocean flying learn still another method of determining his position—celestial

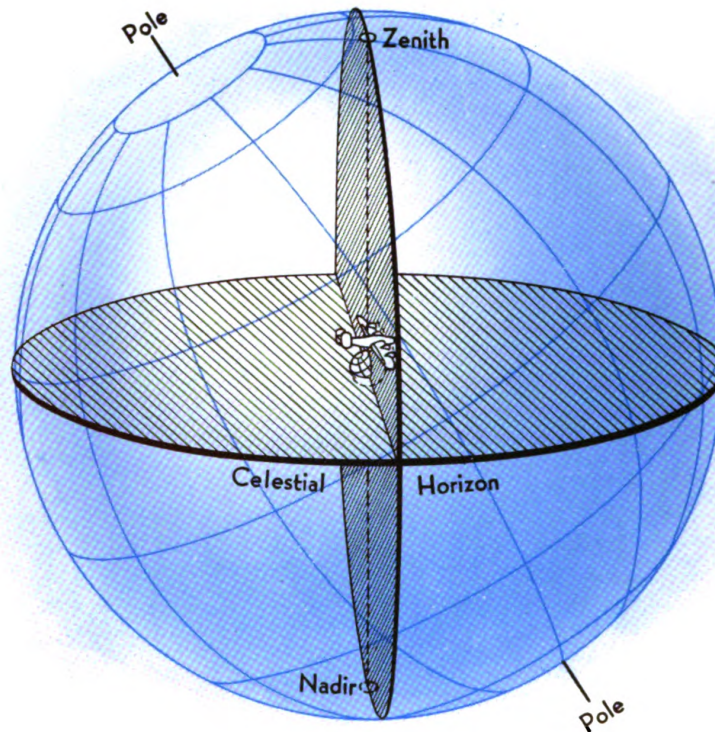


FIG. 72—ZENITH, NADIR, CELESTIAL HORIZON

navigation. Equipped with a knowledge of all four of these basic methods of navigation, he then can feel confident in flight over water or land, under any conditions.

The transition from navigation with reference to the earth to navigation with reference to celestial bodies is, in a sense, like stepping from solid earth into space. In "celestial," as the term is abbreviated, the navigator must acquire new mental habits. His vocabulary becomes enlarged with the addition of many new words and phrases. Even more important, his concept of the world and the universe undergoes a striking change. In learning celestial, the navigator must begin to reason in terms of infinity because, as pointed out earlier, the entire universe becomes his workshop. He learns to call the stars and planets by name, and to regard them as his assistants.

Though it appears complex at first glance, celestial navigation is, in reality, simplicity

itself. As in DR, a thorough working knowledge of the language of celestial is essential to a firm foundation in this method of navigation. The following terms, therefore, should be studied carefully and conscientiously, and it is strongly recommended they be memorized. It is not suggested that the navigator necessarily memorize the definitions as stated herewith, but that—more importantly—their *meaning* be committed to memory.

TERMS

Celestial Sphere—The celestial sphere is a vast globe of infinite radius, the center of which is considered to be located at the center of the earth, and upon which the celestial bodies are projected.

Actually, such a globe does not exist, but to an observer on the earth's surface looking up toward heavenly bodies whose distances from him are infinitely great, the universe ap-

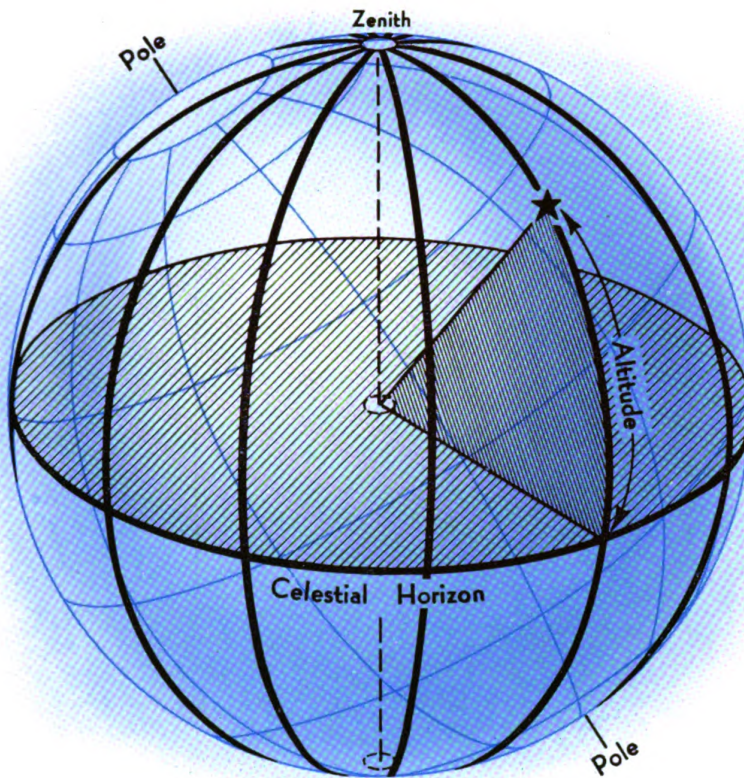


FIG. 73—VERTICAL CIRCLES, ALTITUDE

pears to be contained in a vast hollow sphere. The *distances* of these bodies from the earth do not interest the navigator, since their relative directions, not their distances, determine the observer's position.

Zenith—The zenith is that point on the celestial sphere directly above the observer (Figure 72). It is apparent, therefore, that as the observer changes position, his zenith moves with him. This is a most important concept, as upon it largely depends the navigator's ability to locate himself by reference to the celestial sphere.

Nadir—The nadir is that point on the celestial sphere directly beneath the observer (Figure 72). Since, like the zenith, this point is established by the observer's position, it too will move with him as he changes his position.

Celestial Horizon—The celestial horizon is a great circle on the celestial sphere whose plane passes through the center of the earth

at right angles to a line joining the zenith and nadir (Figure 72). It serves as a reference circle *from which* the *altitude* of a celestial body is measured, and *along which* the body's East-West position may be determined with reference to an observer.

Vertical Circle—Vertical circles are great circles on the celestial sphere passing through the zenith and nadir. These serve as reference circles *along which* the *altitude* of celestial bodies is measured, and hence are sometimes called circles of altitude (Figure 73).

The Prime Vertical is the vertical circle that passes through the East and West points of the celestial horizon (Figure 74).

Altitude—The altitude of a celestial body is the angular distance from the celestial horizon to the body, as measured upon the vertical circle passing through the body. (Figures 73, 74). It is one of two coordinates (known as the *horizon coordinates*) by means of which the po-

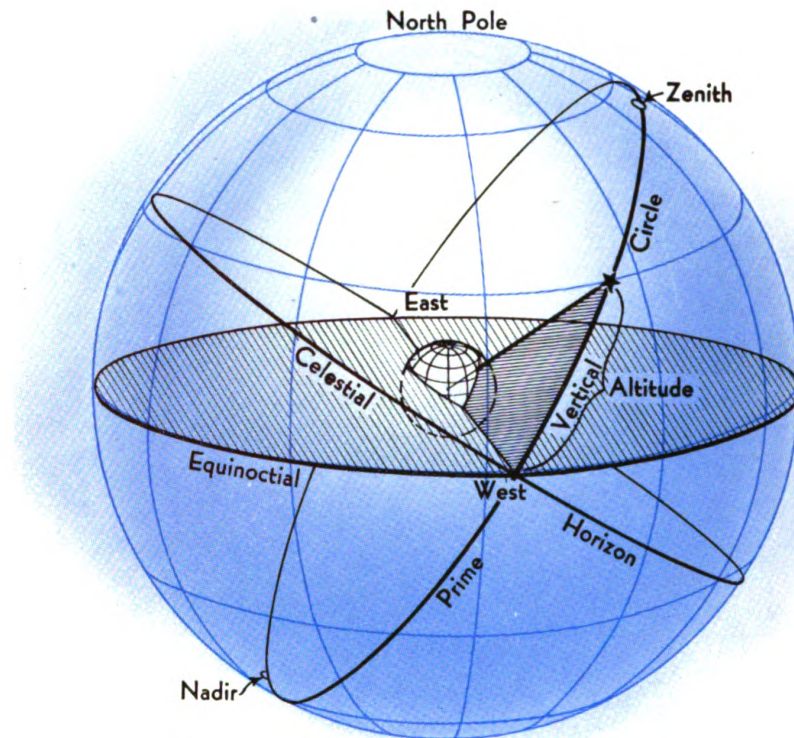


FIG. 74—EQUINOCTIAL, PRIME VERTICAL

sition of a celestial body may be located with reference to the observer's position.

Equinoctial, or Celestial Equator—The equinoctial, or celestial equator, is the great circle on the celestial sphere formed by extending the plane of the earth's equator to the sphere (Figure 74). It serves as a reference circle *from which* the North-South position of a celestial body is measured.

The equinoctial intersects the horizon at its East and West points.

Hour Circles—Hour circles are great circles on the celestial sphere passing through the North and South celestial poles. They serve as reference circles *along which* the *declination* of celestial bodies is measured, hence are sometimes called circles of declination (Figure 75).

Declination (dec.)—Declination of a celestial body is its angular distance from the equinoctial measured on the hour circle passing

through the body (Figure 75). It is named North or South according to its direction from the equinoctial. It is one of two coordinates (known as the *equinoctial coordinates*) by means of which the position of a celestial body may be located in space with reference to fixed points. Declination corresponds to latitude on the earth.

Celestial Coordinates—Coordinates of any kind are simply a means of locating an object with respect to two reference lines at right angles to each other. A familiar example of this is the locating of a house in a city. If a house is at the corner of 7th Street and 3rd Avenue, then its coordinates are 7 and 3. That is, it is 7 blocks North or South of the street used for North-South reference, and 3 blocks East or West of the avenue used for East-West reference. The house might also be located geographically by stating its latitude (North or South) and longitude (East or West). In

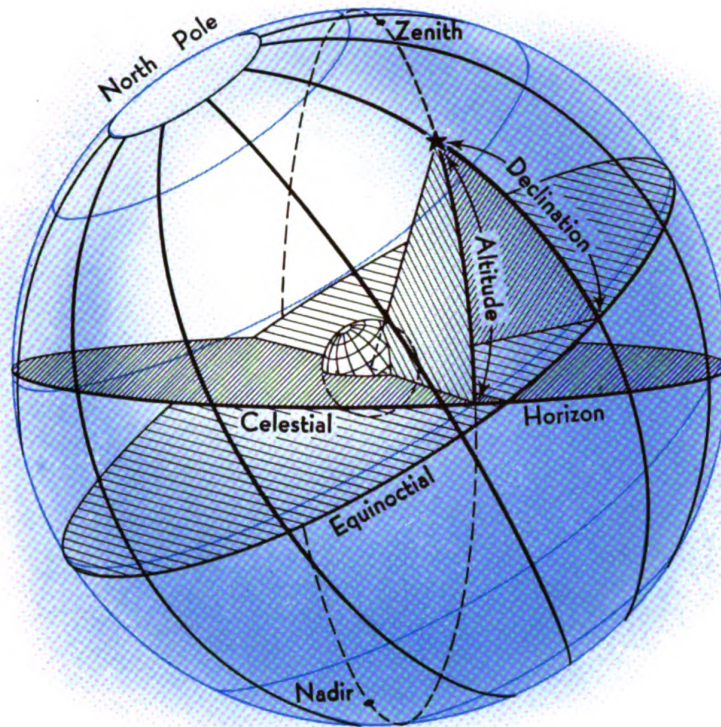


FIG. 75—HOUR CIRCLES, DECLINATION

the latter case, a different set of coordinates and reference lines would be employed than those used to establish position of the house in the city. There is a close parallel between the reference means used in the foregoing example, and those used to locate a celestial body in the sky.

For purposes of celestial navigation, two sets of coordinates, together with their reference circles, are used. The name of the primary reference circle used, in each case, gives rise to the name of the coordinate system. Thus, there is the horizon system of coordinates, and the equinoctial system.

The horizon system is established, basically, with relation to the observer's zenith. And since, as already explained (see definition of zenith), the zenith moves with the ob-

server, the horizon coordinates of a celestial body (altitude and azimuth) change as his position changes.

The equinoctial system, on the other hand, since it is derived from fixed points on the earth which are merely projected to the celestial sphere, constitutes a *fixed* system of coordinates. Hence the equinoctial coordinates (declination and hour angle), do *not* change with a change in the observer's position, but have definite values for any given instant of time. This makes possible the calculation of these values in advance, hence they can be tabulated for the navigator's use. (See Air Almanac, Chapter VI.) Using these tabulated values, the navigator can *compute* the horizon coordinates, which in turn make it possible for him to fix his position.

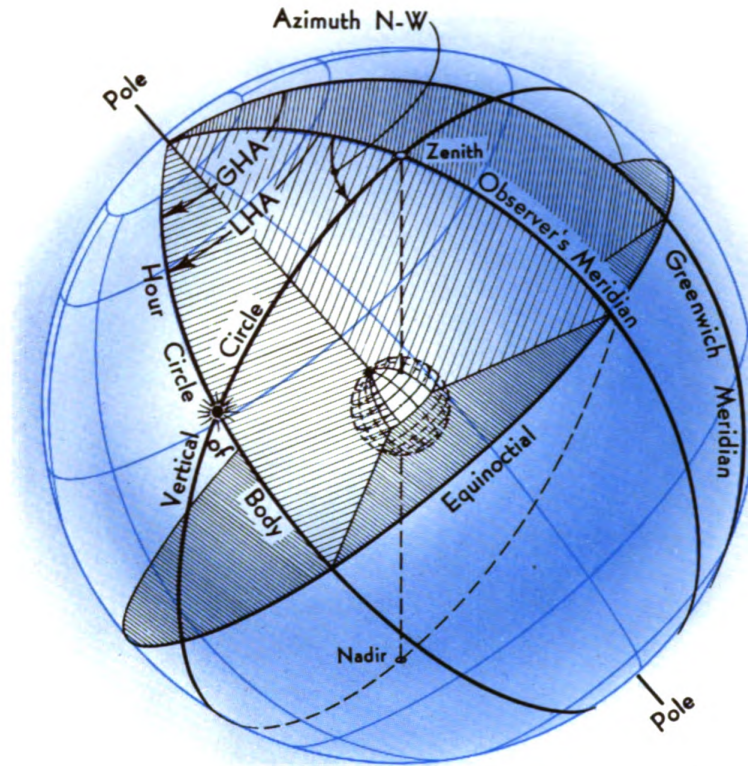


FIG. 76—AZIMUTH, LOCAL AND GREENWICH HOUR ANGLE

Azimuth (Az)—Azimuth is the bearing of a celestial body at the observer (Figure 76). It is the angle at the zenith between the celestial meridian of the observer and the vertical circle passing through the celestial body. It is measured from the elevated pole (North in North latitude, South in South latitude) and to the East or West through 180° (East when the celestial body is rising, and West when it is setting). Azimuth is the second of the two horizon coordinates by means of which a celestial body is located with reference to an observer.

Local Hour Angle (LHA)—The local hour angle of a celestial body is an arc of the equinoctial measured from the upper branch of the observer's meridian over West through 360° to the hour circle passing through the body (Figure 76). It also may be defined as the angle at the pole between the meridian of the observer

and the hour circle passing through the body. It is the second of the two equinoctial coordinates used to locate a body with relation to fixed points.

Greenwich Hour Angle (GHA)—The Greenwich hour angle of a celestial body is an arc of the equinoctial measured from the Greenwich meridian over West through 360° to the hour circle passing through the body (Figure 76). It also may be defined as the angle at the pole between the Greenwich meridian and the hour circle passing through the body. GHA corresponds to longitude on the earth.

Ecliptic—The ecliptic is the great circle path on the celestial sphere that the sun appears to follow due to the annual revolution of the earth (Figure 77). It is inclined to the equinoctial at an angle of approximately

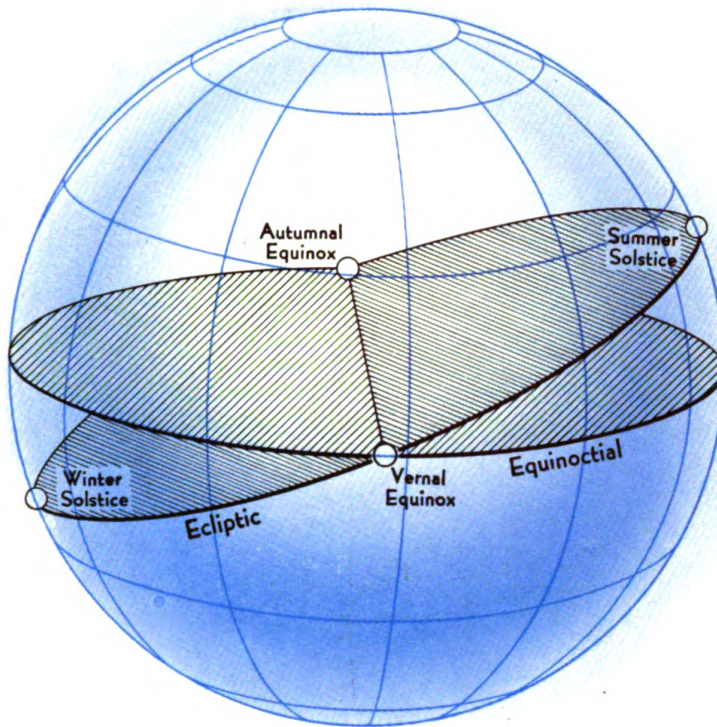


FIG. 77—ECLIPTIC, EQUINOXES, SOLSTICES

$23^{\circ}27'30''$. This inclination is called the obliquity of the ecliptic.

Equinoxes—The equinoxes are the points in the celestial sphere where the ecliptic intersects the equinoctial (Figure 77).

1. **Vernal Equinox** [called First Point of Aries (Υ)] is that point on the equinoctial which the sun passes in changing from South declination to North declination. It occurs about March 21st and marks the beginning of spring in the Northern hemisphere.

2. **Autumnal Equinox** (called First Point of Libra) is that point on the equinoctial which the sun passes in changing from North declination to South declination. It occurs about September 21st and marks the beginning of autumn in the Northern hemisphere.

Solstices—The solstices are the points on the ecliptic where the sun attains maximum declination, Northerly or Southerly (Figure

77). The point of maximum Northerly declination is termed Summer Solstice (about June 21st). The point of maximum Southerly declination is termed Winter Solstice (about December 21st). The occurrence of the summer solstice results in the longest period of daylight in one day throughout the year, and conversely the winter solstice brings the shortest period of daylight in one day.

Thus the position of the earth along the ecliptic (or *apparent* position of the sun) determines the seasons of the year. The reason for this is evident when it is realized that the seasons depend primarily on the amount of heat which the earth receives from the sun, and in turn, the amount of heat received depends upon the position of the earth relative to the sun. The earth's axis, it will be remembered, is inclined at an angle to the ecliptic, hence when the North pole is tilted *towards* the sun, the Northern hemisphere receives a maximum

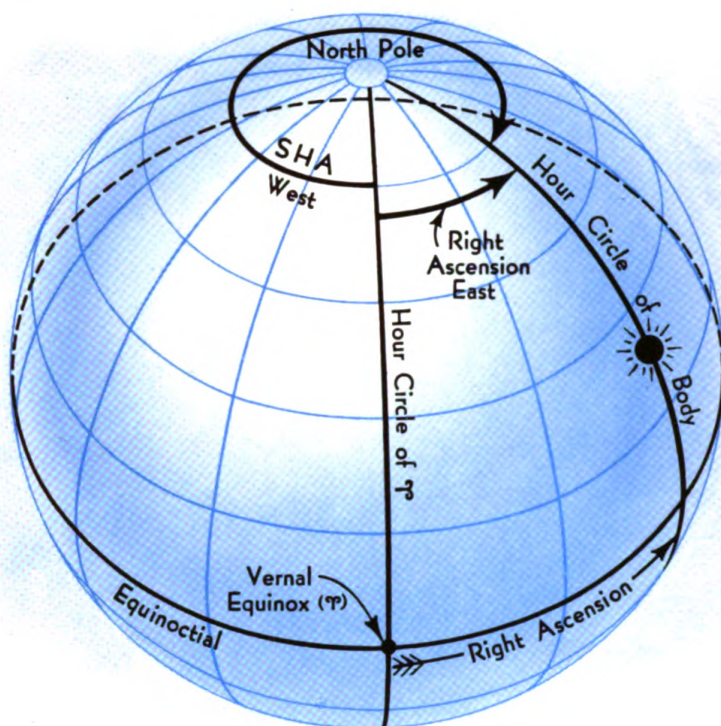


FIG. 78—RIGHT ASCENSION, SIDEREAL HOUR ANGLE

amount of heat per unit area and so experiences summer. When the North pole is inclined *away* from the sun, the Northern hemisphere receives a minimum amount of heat per unit area and so experiences winter.

Right Ascension (RA)—Right ascension is the angle at the pole between the hour circle of the First Point of Aries and the hour circle passing through the body, measured to the *East* through 24 hours or 360° (Figure 78). It also may be described as the arc of the equinoctial intercepted between these hour circles. Right ascension seldom is used in aerial navigation due to the convenience of modern air almanacs, which are tabulated in sidereal hour angle.

Sidereal Hour Angle (SHA) — Sidereal hour angle is related to right ascension in that SHA equals 360° minus RA (Figure 78). Stated as an equation:

$$\text{SHA} + \text{RA} = 360^\circ$$

It is the angle at the pole (measured to the *West* through 360°) between the hour circle of

the First Point of Aries and the hour circle passing through the body. Sidereal hour angle is used to find GHA of all star sights in aerial navigation.

Astronomical Triangle—The intersections of the observer's (celestial) meridian, the hour circle passing through a celestial body, and the vertical circle passing through a celestial body form a spherical triangle on the celestial sphere known as the *astronomical triangle* (Figure 79). The sides and angles of this triangle represent the relationship existing between the observer's position, and the horizon coordinates and equinoctial coordinates of a celestial body. If the approximate latitude of the observer is known, together with the hour angle and declination of the celestial body, then the altitude and azimuth of the body can be computed. The values thus obtained can then be used to determine the observer's position by methods to be discussed in later chapters.

The solution of the astronomical triangle, then, is the navigator's fundamental problem

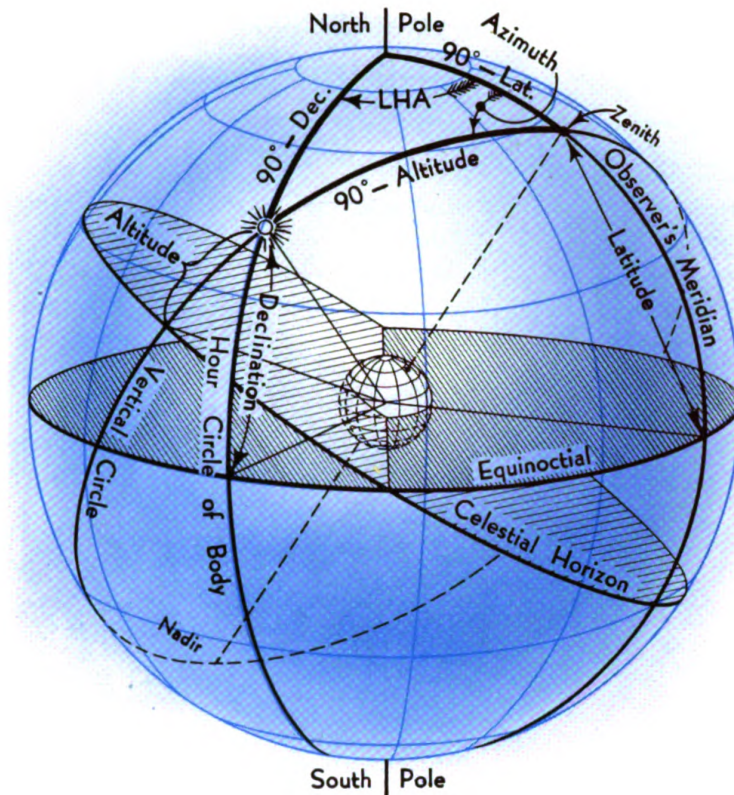


FIG. 79—ASTRONOMICAL TRIANGLE

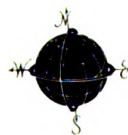
in celestial navigation. This solution is based on formulas of spherical trigonometry. It is not necessary for the aerial navigator to know these formulas, however, except for his own interest, as the various possible solutions of the astronomical triangle have been worked out for him and compiled into tables of pre-computed altitude and azimuth. From these, the desired information may be obtained by inspection.

The aerial navigator should learn how to draw the astronomical triangle, and should know its component parts, as this knowledge

will help him to become familiar with the terminology of celestial navigation. In this triangle, the local hour angle (which is the angle at the pole) is the most important angle, as it determines the observer's longitude.

In drawing this triangle, remember that it is formed by three great circles:

1. Hour circle passing through the body.
2. Celestial meridian passing through the zenith.
3. Vertical circle passing through the body.



PROBLEM WORK NO. 15

CELESTIAL SPHERE

(Show the following information.)

- | | | |
|------------------------|--------------------------|---------------------|
| 1. Observer's meridian | 6. Equinoctial | 11. GHA |
| 2. Greenwich meridian | 7. Ecliptic | 12. LHA |
| 3. Hour circle | 8. Vernal equinox | 13. Declination |
| 4. Celestial poles | 9. Longitude of observer | 14. SHA |
| 5. Zenith | 10. Latitude | 15. Solstices |
| | | 16. Right ascension |

PROBLEM WORK NO. 16

ASTRONOMICAL TRIANGLE

(Show the following information.)

- | | |
|--------------------------|--------------------|
| 1. Celestial poles | 7. Zenith distance |
| 2. Zenith and nadir | 8. LHA |
| 3. Equinoctial | 9. Azimuth |
| 4. Celestial horizon | 10. Altitude |
| 5. Astronomical triangle | 11. Declination |
| 6. Latitude | 12. Polar distance |



☆ 6 ☆

AIR ALMANAC

THE American Air Almanac is one of the aerial navigator's principal tools and, therefore, is deserving of careful study. The navigator should understand this indispensable volume thoroughly, because it provides him with the astronomical data required for aerial navigation. (See Appendix for sample Air Almanac pages.)

As compared with the Nautical Almanac, the information in the Air Almanac contains a certain amount of error. However, the average error is only about $0'.5$, which is so slight as to be considered negligible. This error has been allowed in order to permit presentation of the information in a more condensed and convenient form, thus making possible faster solution

of celestial navigation problems, the most important being the solution of the astronomical triangle.

As explained in the previous chapter, in order to solve the astronomical triangle, the navigator must know the equinoctial coordinates (GHA and declination) of a body (Figure 80). These have been precomputed and tabulated for him in the Air Almanac. Government astronomers are assigned to the important work of predetermining, for any given instant of GCT, the exact location of all heavenly bodies useful to navigators. The findings of these astronomers are published in advance in the Air Almanac, which is printed three times annually. All in-

☆ 77 ☆

formation needed for any given day's navigation is printed on front and back sides of a single page of the Almanac.

Note: To avoid the possibility of using day-old data by mistake, it is advisable to tear out each daily page as soon as it becomes obsolete.

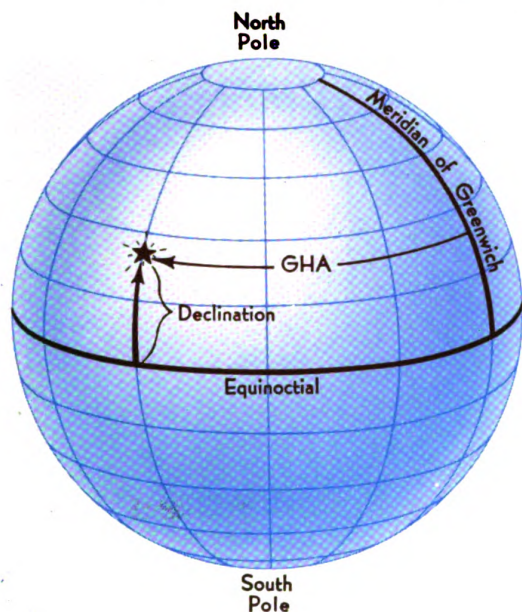


FIG. 80—EQUINOCTIAL COORDINATES

GHA AND DEC

Greenwich Hour Angle (GHA) and Declination (Dec.) of Sun, Moon, and Planets—On the daily page of the Air Almanac are listed the GHA of the sun, moon, and planets (most suitable for observation). Declination of the sun and planets is given in hourly intervals, while declination of the moon is tabulated in ten-minute intervals of GCT throughout 24 hours.

In taking this data from the Almanac, the correct declination will be the nearest tabulated value for any given instant of GCT. To obtain the correct GHA, however, it is necessary, first, to find the value listed for the preceding ten-minute interval of GCT, and then to add to it an interpolated amount of GHA for the additional minutes and seconds. These interpolation values may in turn be obtained directly from tables printed on the inside of the

front cover and also on the back of the star chart.

The moon has an interpolation table separate from the other celestial bodies due to its proximity to the earth and its relatively rapid movement.

GHA and Declination of a Star (Figure 81)

—In navigation the stars are considered to be fixed bodies in space. Therefore, their relative positions are always the same. This fact is used to advantage in the Air Almanac. Instead of giving the GHA of each individual star, the Almanac lists the daily positions of the *First Point of Aries* or *vernal equinox*, as measured by its Greenwich hour angle. On the inside of the back cover, the sidereal hour angles (SHA) of the 55 brightest stars are listed, and the SHA of an additional nine stars are listed on the opposite page. SHA, as explained earlier, is the position of a star with reference to the First Point of Aries, measured to the West from the First Point of Aries.

Hence:

$$\text{GHA (star)} = \text{GHA (Aries)} + \text{SHA (star)}$$

The change in declination of a star for the four-months period which the Air Almanac covers is practically negligible. Therefore, declination is listed opposite the SHA on the in-

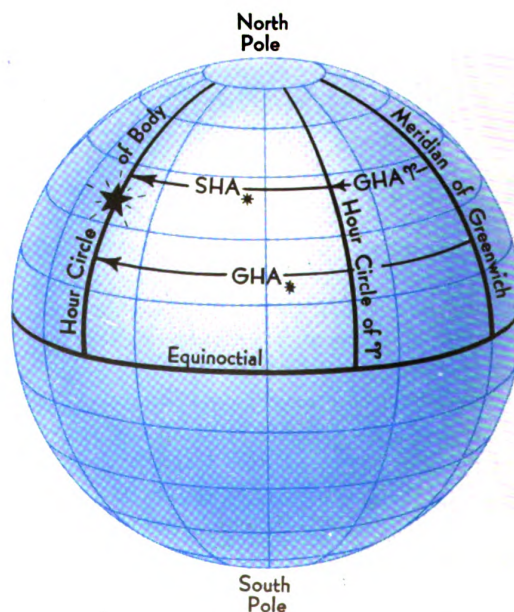


FIG. 81— $\text{GHA}^* = \text{GHA} \gamma + \text{SHA}^*$

side back cover, and it may be read directly for any given star.

Examples:

Jan. 1, 1943, GCT 10^h12^m22^s. Find GHA and declination.

Sun

GCT 10 ^h 10 ^m	GHA	331°41'
+ for 02 ^m 22 ^s	Corr.	36'
<hr/>		
GHA for 10 ^h 12 ^m 22 ^s		332°17'
Declination	S	23°03'

Saturn

GCT 10 ^h 10 ^m	GHA	187°27'
+ for 02 ^m 22 ^s	Corr.	36'
<hr/>		
GHA for 10 ^h 12 ^m 22 ^s		188°03'
Declination	N	19°36'

Moon

GCT 10 ^h 10 ^m	GHA	42°52'
+ for 02 ^m 22 ^s	Corr.	34'
<hr/>		
GHA for 10 ^h 12 ^m 22 ^s		43°26'
Declination	S	7°13'

Star

GCT 10 ^h 10 ^m	GHA	252°41'
+ for 03 ^m 22 ^s	Corr.	36'
<hr/>		
SHA* (Sirius)		259°21'
	GHA*	412°38'
		—360°
<hr/>		
GHA for 10 ^h 12 ^m 22 ^s		152°38'
Declination	S	16°38'

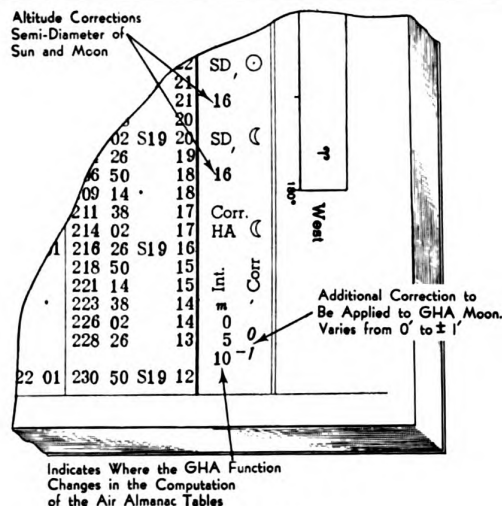


FIG. 82

Note: In lower right corner of the A. M. side of the daily page is the notation shown in Figure 82.

HOUR ANGLE DIAGRAMS

Description—The *hour angle diagram* is a graphic method of illustrating the relationship between the meridian of Greenwich, meridian of the observer, and the hour circle of a body. It is important because it aids the navigator in visualizing GHA, LHA, and longitude.

This relationship may be illustrated on a globe (Figure 83).

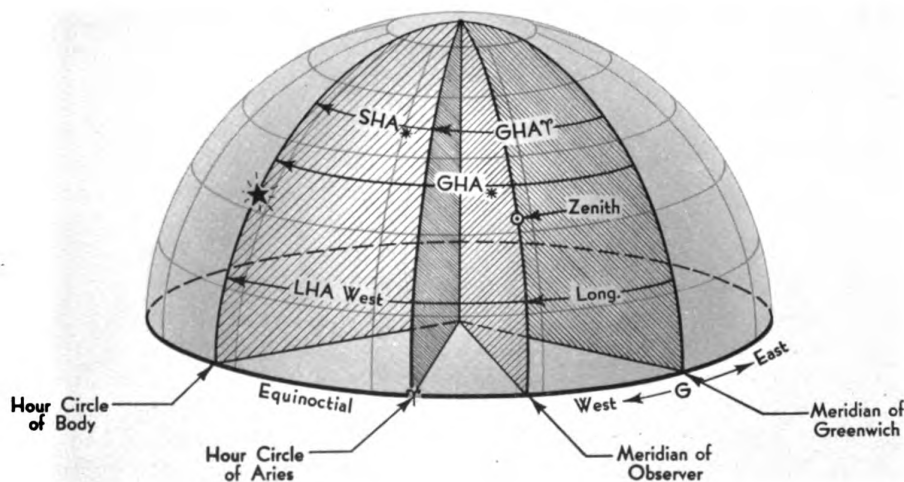


FIG. 83—RELATIONSHIP BETWEEN OBSERVER'S MERIDIAN, GREENWICH MERIDIAN AND HOUR CIRCLE OF BODY

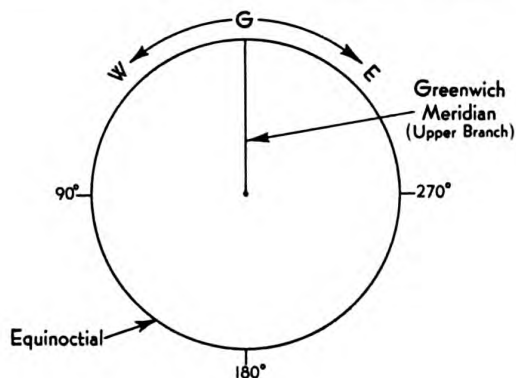


FIG. 84—BASIC HOUR ANGLE DIAGRAM

The easiest method of drawing hour angle diagrams, however, is on the plane of the equinoctial, in which case the globe is considered to be viewed from the South pole. The equinoctial thus appears as a circle, and the meridians or hour circles appear as straight lines radiating from the center. The meridian of Greenwich is drawn from the top of the circle to the center, and directions measured to the left are named West, and those measured to the right are named East (Figure 84). This diagram can be used to indicate any relationship of hour angle and longitude.

Examples:

As Used for Sun (Figure 85)

Given: Long. 160° East
GHA 125° West
Indicate: LHA of Sun

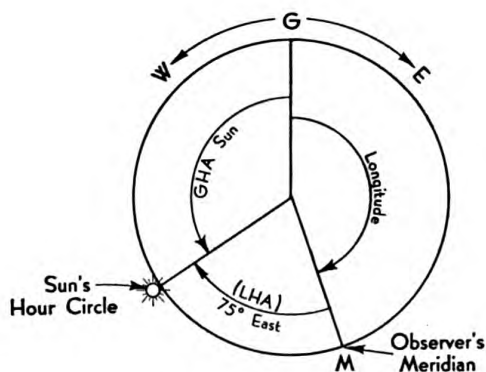


FIG. 85—HOUR ANGLE DIAGRAM FOR SUN

As Used for Star (Figure 86)

Given: Long. 170° West
GHA γ 120° West
SHA* 150° West
Indicate: LHA of Star

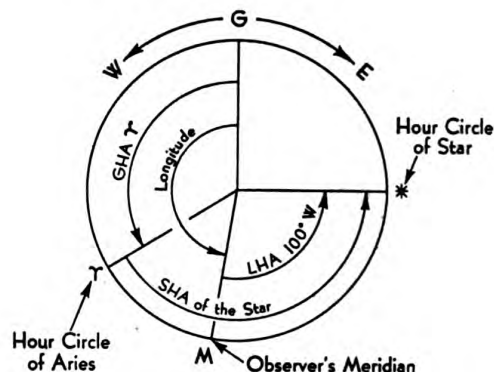


FIG. 86—HOUR ANGLE DIAGRAM FOR STAR

DEFINITIONS OF ALTITUDE

True Altitude (H_0)—True altitude is the arc of the vertical circle between the celestial horizon and the center of the celestial body. It is also the angle, measured at the center of the earth, between the celestial horizon and the center of the celestial body (see Figure 89).

Sextant Altitude (H_s)—Sextant altitude is the altitude of the celestial body as read on the sextant or octant.

Computed Altitude (H_c)—Computed altitude is the true altitude of a celestial body as calculated for an assumed position of the observer.

SEXTANT ALTITUDE CORRECTIONS

Instrument Correction (I.C.)—Instrument correction is the correction which must be applied to sextant readings because of mechanical errors in the instrument. (To be discussed more fully in Chapter IX).

Refraction (Ref.)—It is a proved principle of physics that light passing from one medium into another of different density is bent from a straight path.

When viewing a distant celestial body, the observer is in reality seeing a ray of light from that body. This ray of light follows a curved path through the earth's atmosphere due to the increasing density of the atmosphere near the

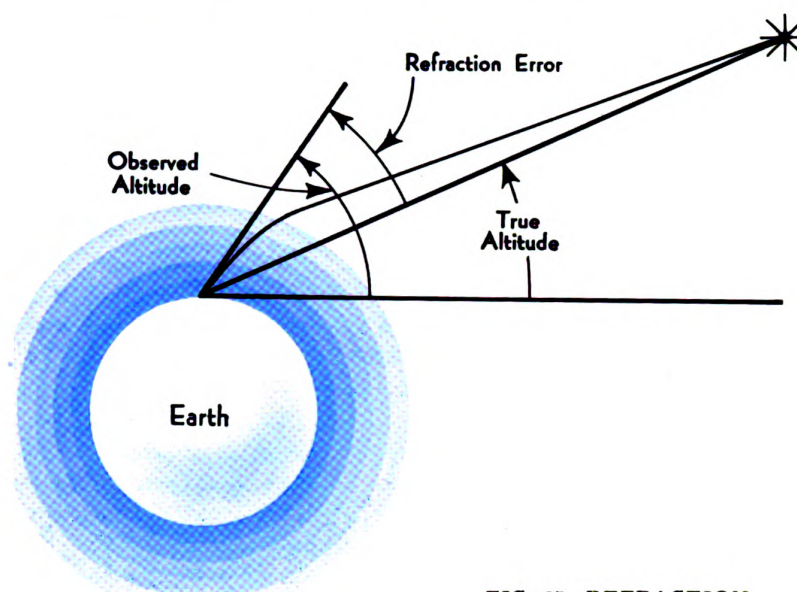


FIG. 87—REFRACTION

surface. In measuring the altitude of a celestial body the observer actually measures the altitude of a line tangent to this curve, and the difference between the last direction of the ray and its direction on entering the earth's atmosphere is the correction angle called *refraction*. Refraction causes the observed altitude (H_S) to be always greater than the true altitude (H_O), as shown in Figure 87.

Correction for refraction is, therefore, always *minus*. The higher the body, the less the refraction. When the body is directly overhead, refraction is zero. The amount of this correction is tabulated on the back outside cover of the Air Almanac.

Parallax (Figure 88)—Parallax may be defined as the angle at the celestial body subtended by the earth's radius.

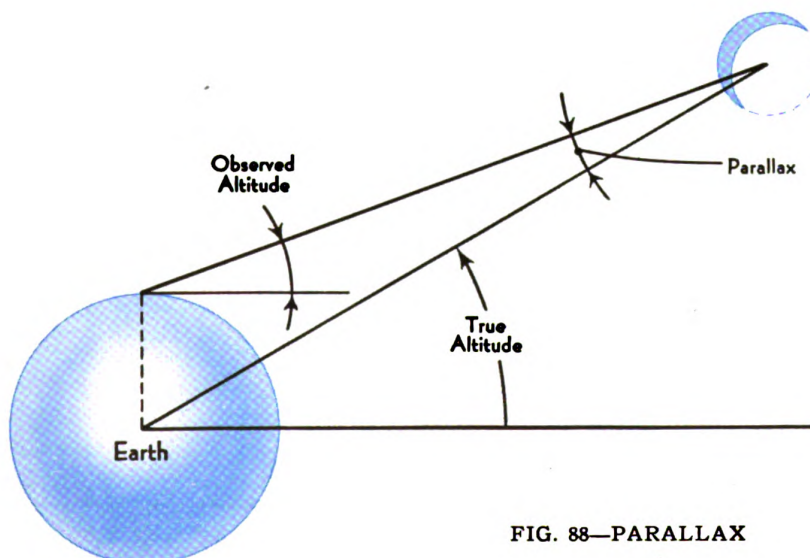


FIG. 88—PARALLAX

All navigation tables are computed on the assumption that altitude is measured from the center of the earth to the center of the celestial body. However, since the navigator measures the body's altitude from the earth's surface, theoretically a parallax correction must be applied in order to make the true altitude read as though the observer measured it from the center of the earth.

Actually, the moon is the only celestial body whose proximity to the earth makes a parallax correction necessary, all other bodies being so far distant as to make such a correction insignificant for air navigation purposes.

From examination of Figure 88 it is apparent that octant altitude always is less than true altitude. Therefore, correction for parallax always is plus. The amount of correction for various altitudes of the moon may be obtained by inspection from the A.M. side of Air Almanac daily pages. The greater the moon's

altitude, the smaller the parallax. When the moon is at zenith, parallax is zero.

Semidiameter and Dip (Figure 89)—Semidiameter and dip are altitude corrections seldom applied in aerial navigation, since they are encountered only when using the sea horizon as the horizontal reference plane. They only need to be considered, therefore, when using the marine sextant, or when using an aircraft octant to measure altitude above the sea horizon.

1. **Semidiameter** is the angle at the eye of the observer subtended by the radius of the celestial body. In using the sea horizon the navigator adjusts his octant to cause the upper or lower limb (outside edge) of the reflected image of the sun or moon to just touch (be tangent to) the visible sea horizon. Since the outside edge of the sun or moon is then being used to measure the altitude of the body, instead of the center of the body as required for true altitude, it is necessary to

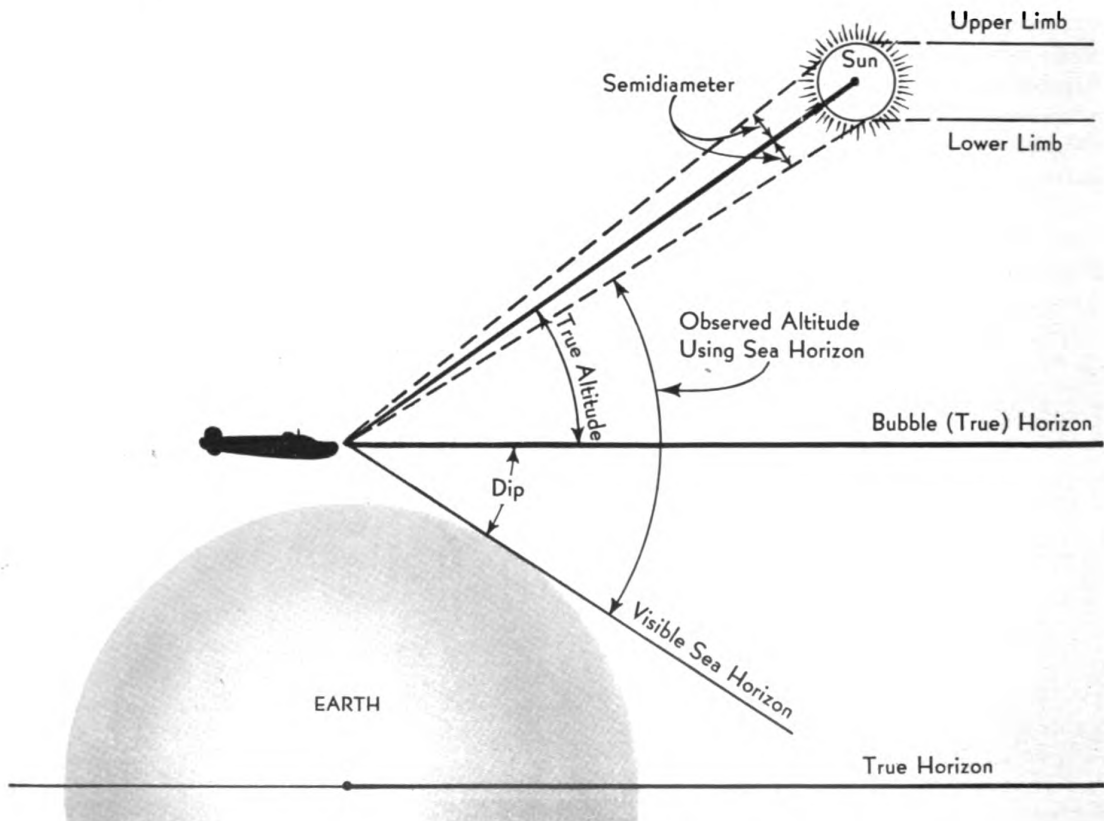


FIG. 89—DIP AND SEMIDIAMETER

apply a correction equal to half the body's diameter.

The Air Almanac gives the semidiameter correction of the sun or moon on the A.M. side of the daily page. This correction is either plus or minus, depending upon which limb of the body is used to measure the altitude. If the upper limb is used, the correction is minus. If the lower limb is used, the correction is plus.

2. **Dip** is the angle, at the eye of the observer, between the visible sea horizon and the true (celestial) horizon. In Figure 89, it is apparent that the altitude reading, using the visible sea horizon, is always greater than the true altitude; therefore, dip correction is always minus. The amount of correction is tabulated on the back cover of the Air Almanac.

Examples of Altitude Correction—On January 1, 1943, the following altitudes were measured with reference to the horizon while flying at an altitude of 8000 feet:

	SUN	PLANET VENUS	STAR ALTAIR	MOON
Observed Altitude	30°20'	60°15'	20°12'	28°35'
Refraction	— 01'	00'	— 02'	— 01'
Parallax				— 51'
CORRECTED ALTITUDE	30°19'	60°15'	20°10'	29°25'

On January 7, 1943, the following altitudes were measured with reference to the sea horizon while flying at an altitude of 200 feet:

	(Lower Limb) SUN	PLANET JUPITER	STAR RIGEL	(Upper Limb) MOON
Observed Altitude	15°45'	60°19'	30°55'	40°50'
Refraction	— 04'	— 01'	— 02'	— 01'
Parallax				+ 44'
Semidiameter	+ 16'			— 16'
Dip	— 14'	— 14'	— 14'	— 14'
CORRECTED ALTITUDE	15°43'	69°04'	30°39'	41°03'

POSITIONS OF PLANETS AND STARS ALONG THE ECLIPTIC

The ecliptic has been defined as the apparent path of the sun on the celestial sphere. Since all of the planets revolve around the sun, it is evident that the path of the planets across the celestial sphere also will follow, very nearly, the ecliptic. Actually, their paths never vary more than eight degrees on either side of the ecliptic.

In the Air Almanac, the diagram on the A.M. side of the daily page represents this path of the sun and planets in the celestial sphere. The sun is shown in the center, and through 180° on either side of the sun are shown the relative positions of the moon, the five planets Mercury, Venus, Mars, Jupiter, and Saturn, and the four bright stars Aldebaran, Antares, Spica, and Regulus (except when they are within five degrees of the sun).

This diagram is useful to the navigator in ascertaining positions of the planets at a glance. For instance, on January 1, 1943, Aldebaran and Saturn are shown in proximity. Recognizing Aldebaran in the sky, the navigator would know that the bright celestial body nearby was Saturn.

SUNRISE, SUNSET, MOONRISE AND MOONSET

On the P.M. side of the daily page are given tables for finding the times of sunrise, sunset, moonrise and moonset. These are tabulated in terms of local civil time and depend upon the latitude of the observer for correct interpretation.

With the nearest whole degree of the observer's latitude as an argument, the *LCT values* shown opposite the *two nearest tabulated latitudes* are noted and their difference obtained by

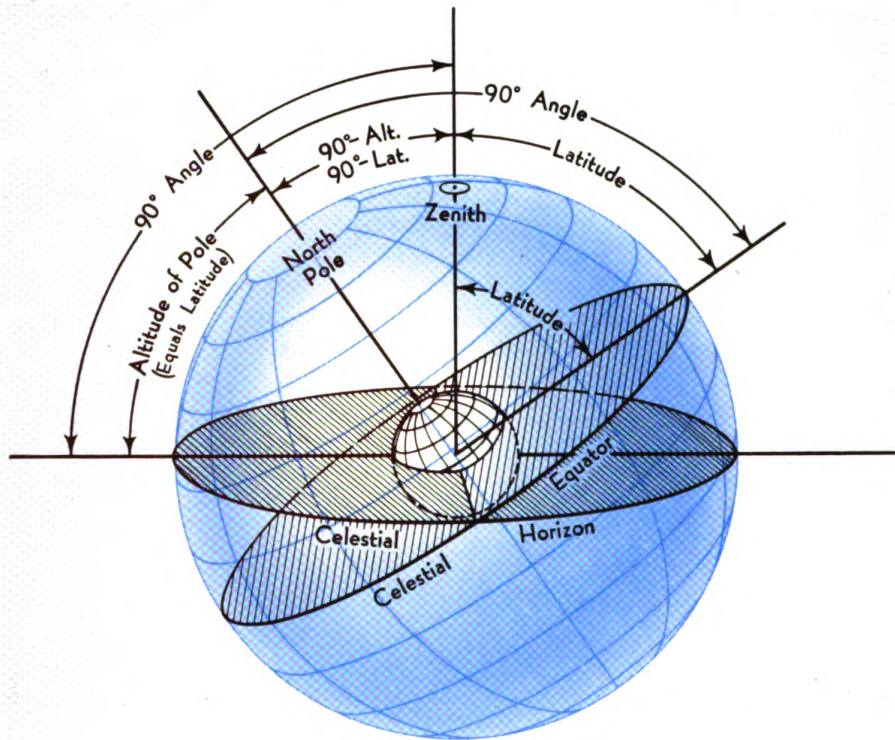


FIG. 90—ALTITUDE OF ELEVATED POLE EQUALS LATITUDE

subtracting the smaller from the greater. The *exact* LCT of the required rising or setting can then be obtained for the observer's latitude by interpolation.

If the GCT is desired (as is usually the case), it can be had by applying the observer's longitude (converted from degrees and minutes of arc to hours and minutes of time) to the LCT.

LATITUDE BY POLARIS

As shown in Figure 90, the altitude of the elevated pole above the celestial horizon is equal to the observer's latitude. This may be reasoned as follows:

1. Angle between zenith and equinoctial = Latitude.
Since angle between equinoctial and pole = 90° , and angle between zenith and horizon = 90° , then
2. Angle between zenith and pole = 90° minus alt. pole, or 90° minus latitude.
3. Hence, altitude pole = latitude.

From the diagram in Figure 90, it is apparent that the altitude of a star at the pole would be equal to the latitude. Polaris is known

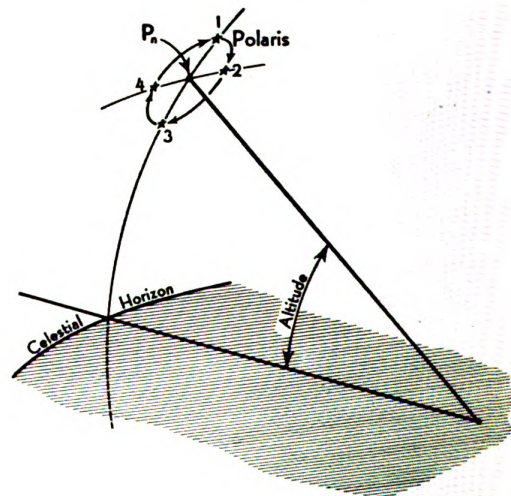


FIG. 91—POLARIS NOT EXACTLY AT P_n

AIR ALMANAC

as the North Star. However, it is not located exactly at the pole (P_n), but approximately one degree or sixty minutes away. As the earth revolves, the star Polaris appears to revolve around P_n (Figure 91).

When Polaris is in position No. 1 (Figure 90), the altitude of Polaris would be greater than the true altitude of P_n . When Polaris is in position No. 2 or No. 4, the altitude of Polaris would be equal to the altitude of P_n . With Polaris in any position other than No. 2 or No.

4—such as No. 1 or No. 3—a correction would have to be applied to the true altitude of Polaris to obtain the true altitude of P_n , hence the latitude.

The Polaris table in the Air Almanac (located on the back of the star chart) gives the amount of correction to apply to the true altitude of Polaris, in order to find latitude. The tables are computed to give the correction with relation to the local hour angle of the First Point of Aries ($LHA \cap$) from 0° to 360° , measured to the West.

Example: On January 1, 1943, at GCT $11^h20^m20^s$, in longitude $145^\circ18'$ West, the observed altitude of Polaris—using a bubble horizon—was $32^\circ15'$, and the height of aircraft 5000 feet.

GCT 11^h20^m	GHA \cap $270^\circ14'$	H_s	$32^\circ15'$
+ For 00^m20^s	Corr. $05'$	Ref.	$- 01'$
GCT $11^h20^m20^s$..	GHA \cap $270^\circ19'$	H_o	$32^\circ14'$
Longitude	$145^\circ18'$ West	LHA \cap Corr.	$+ 10'$
	LHA \cap $125^\circ01'$ West	LATITUDE	$32^\circ24'N$



PROBLEM WORK NO. 17

GHA AND DECLINATION

(Find GHA and declination. See Appendix for extracts from Air Almanac.)

No.	DATE	CELESTIAL BODY	GCT	GHA	DECLINATION
1	5- 1-43	SUN	00:17:10		
2	5- 5-43	SUN	22:08:07		
3	5-10-43	SUN	01:00:07		
4	5-15-43	SUN	00:51:17		
5	5-20-43	SUN	17:01:07		
6	5-30-43	JUPITER	18:01:07		
7	5-15-43	JUPITER	14:20:10		
8	5- 1-43	JUPITER	16:20:07		
9	5-10-43	JUPITER	13:24:10		
10	5-20-43	JUPITER	18:08:12		
11	5-30-43	MOON	22:48:30		
12	5- 1-43	MOON	23:00:07		
13	5-15-43	MOON	00:02:07		
14	5- 5-43	MARS	10:06:09		
15	5-15-43	MARS	12:05:00		
16	5- 1-43	FOMALHAUT	09:08:07		
17	5-10-43	BETELGEUX	13:39:07		
18	5-20-43	ARCTURUS	14:17:10		
19	5-15-43	DENEK	08:22:10		
20	5-30-43	ALTAIR	09:05:07		

PROBLEM WORK NO. 18

HOOR ANGLE DIAGRAMS

(Find GHA and LHA of the celestial body. Draw hour angle diagrams.)

No.	DATE	LONGITUDE	CELESTIAL BODY	GCT	GHA ARIES	SHA	GHA BODY	LHA
1	****	45°22'W	SUN	****			300°20'	
2	5-30-43	104°15'W	SUN	10:00:00				
3	5-15-43	40°00'E	SUN	16:23:00				
4	5-15-43	170°00'E	MOON	04:27:23				
5	5-10-43	117°11'W	MOON	08:20:00				
6	5- 5-43	105°30'E	MOON	19:24:00				
7	5- 1-43	5°30'E	JUPITER	18:42:30				
8	5-10-43	90°00'W	VENUS	22:24:00				
9	5-30-43	137°27'W	MARS	06:52:00				
10	****	15°00'W	DENEB	****	260°00'			
11	5- 1-43	178°50'E	SIRIUS	04:20:00				
12	5-15-43	70°30'W	ALTAIR	10:20:00				
13	5- 5-43	90°05'E	ALDEBARAN	23:00:00				
14	5-30-43	120°23'W	ANTARES	14:30:00				
15	5-10-43	20°15'E	FOMALHAUT	22:49:00				
16	5-15-43	2°15'W	PROCYON	16:05:00				
17	5- 1-43	29°39'E	NUNKI	02:08:00				
18	5-30-43	95°15'W	SPICA	09:41:00				

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 19
OCTANT ALTITUDE CORRECTIONS

[Correct sextant altitude (H_s) to true altitude (H_o). Bubble horizon used except where sea horizon is indicated.]

No.	DATE	CELESTIAL BODY	H_s	ALTITUDE	H_s CORRECTIONS				H_o
					Ref.	Par.	SD	DIP	
1	5- 5-43	SUN	49°12'	5000'					
2	5-10-43	SUN	57°49'	8000'					
3	5-15-43	SUN	36°18'	12,000'					
4	5-20-43	SUN	52°42'	6000'					
5	5-25-43	MOON	38°16'	8000'					
6	5-10-43	MOON	33°38'	12,000'					
7	5- 1-43	MOON	29°59'	10,000'					
8	5- 5-43	MOON	37°23'	9000'					
9	5-20-43	RIGEL	23°19'	11,500'					
10	5-25-43	CANOPUS	39°42'	2000'					
11	5- 1-43	SIRIUS	41°15'	4500'					
12	5-20-43	CAPELLA	27°29'	2000'					
13	5-10-43	JUPITER	42°39'	8500'					
14	5-15-43	SATURN	35°23'	14,500'					
15	5-25-43	VENUS	28°52'	8600'					
16	5-30-43	JUPITER	39°17'	11,000'					
17	5- 1-43	Lower-Limb SUN	30°42'	sea horizon 600'					
18	5-30-43	Upper-Limb MOON	45°12'	sea horizon 800'					
19	5-30-43	ALTAIR	42°21'	sea horizon 500'					
20	5-20-43	SATURN	41°38'	sea horizon 100'					

PROBLEM WORK NO. 20

LATITUDE BY POLARIS

(Find latitude.)

No.	DATE	GCT	H _s	I.C.	ALTITUDE	LONGITUDE	LATITUDE
1	5- 1-43	19:33:48	33°48'	—15'	14,000'	117°22'W	
2	5- 5-43	08:00:12	38°00'	— 8'	14,000'	133°42'W	
3	5-10-43	18:32:10	36°34'	—23'	14,000'	85°23'E	
4	5-15-43	03:40:00	38°43'	+ 3'	6000'	90°00'W	
5	5-20-43	17:30:14	30°06'	— 2'	10,000'	114°00'E	
6	5- 1-43	11:13:40	36°28'	— 8'	5000'	178°00'E	
7	5- 5-43	09:30:14	42°20'	+ 1'	12,000'	118°12'W	
8	5-10-43	23:30:00	25°00'	— 3'	5000'	152°40'E	
9	5-15-43	02:43:19	31°14'	— 7'	6000'	20°28'W	
10	5-20-43	17:14:42	24°19'	+ 1'	10,000'	114°26'E	
11	5- 1-43	02:10:30	33°58'	— 5'	8000'	117°11'W	
12	5- 5-43	06:25:10	25°42'	+ 5'	5000'	122°14'W	
13	5-10-43	03:26:14	44°07'	+ 8'	10,000'	121°14'E	
14	5-15-43	22:12:30	29°56'	+ 1'	6000'	94°10'W	
15	5-20-43	20:34:21	45°10'	— 1'	14,000'	176°41'E	
16	5- 1-43	14:31:00	36°54'	— 7'	9000'	69°00'E	
17	5- 5-43	19:11:23	28°04'	+11'	8000'	119°32'E	
18	5-10-43	05:16:13	43°11'	+ 3'	7000'	110°11'W	
19	5-15-43	11:02:00	33°02'	— 9'	11,000'	143°21'W	
20	5-20-43	19:51:29	39°55'	— 4'	9000'	178°21'E	

CELESTIAL REVIEW EXAMINATION NO. 1

1. Define :

- | | |
|-------------------------|--------------------------|
| (a) Altitude | (f) Declination |
| (b) Equinoctial | (g) Vertical Circle |
| (c) Local Hour Angle | (h) First Point of Aries |
| (d) Azimuth | (i) Ecliptic |
| (e) Sidereal Hour Angle | (j) Sphere |

2. Prove with the aid of diagram :

- (a) $LHA = GHA - \text{Longitude West}$
 (b) $GHA^* = GHA \mp + SHA^*$

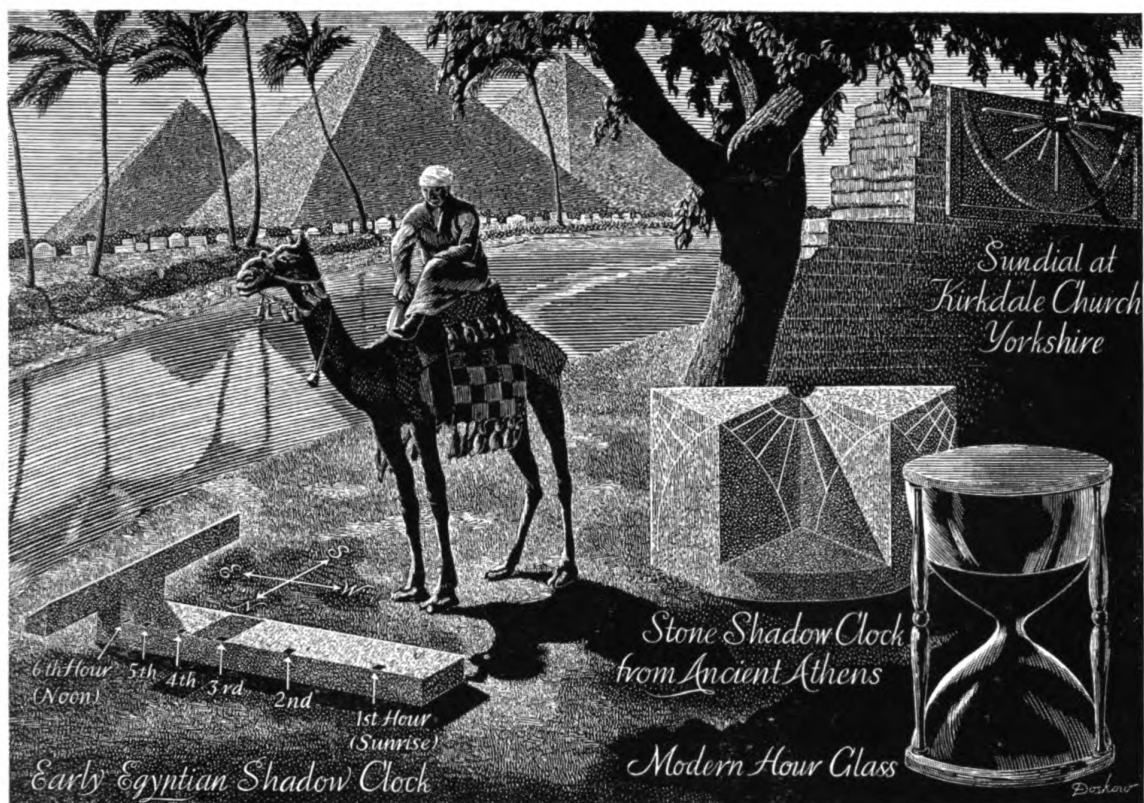
3. Find LHA to nearest whole degree; show solution and diagram.

- | | | |
|-----------------|------------------|------------------|
| (a) May 1, 1943 | (b) May 10, 1943 | (c) May 20, 1943 |
| Sun | Jupiter | Moon |
| GCT 08:21:14 | GCT 20:03:04 | GCT 17:50:00 |
| Long. 75° E | Long. 126° 10' W | Long. 117° 12' E |
| (d) May 5, 1943 | (e) May 15, 1943 | |
| Vega | Deneb | |
| GCT 04:12:10 | GCT 01:19:43 | |
| Long. 23° 30' W | Long. 100° E | |

4. Draw the Celestial Sphere showing the following:

- | | |
|-------------------------|-----------------------------|
| (a) Zenith | (e) Equinoctial Coordinates |
| (b) Nadir | (f) Astronomical Triangle |
| (c) Horizon Coordinates | (g) Ecliptic |
| (d) Celestial Poles | (h) Vernal Equinox |





☆ 7 ☆

TIME

THE navigator's concept of time differs radically from that held by the average citizen. To the man on the street, time affords a method of determining the beginning and end of a working day, or aids him in keeping an appointment. To be "on time" means, to the citizen, to arrive at his destination within a few minutes of the predetermined moment.

But the navigator has a deeper respect for time. "On time" to him means, literally, on time to the very second if possible, for a variance of seconds in his timekeeping will result in a miscalculation—perhaps a dangerous one—of his position.

Celestial navigation involves knowing, at any instant, the exact location in the celestial

sphere of all important celestial bodies with relation to the prime meridian of Greenwich. Lacking the exact time, the navigator would be unable to determine the exact location of these bodies. It follows, then, that he would be unable to determine the exact location of his aircraft, for his location is determined by reference to the position of these celestial bodies.

Definition—Time may be defined as a measure of duration, or the elapsed interval between two events.

Measurement of Time—A measure of time is afforded in the daily rotation of the earth on its axis, by observing the elapsed interval between two successive apparent transits of the sun or a star over a given terrestrial meridian

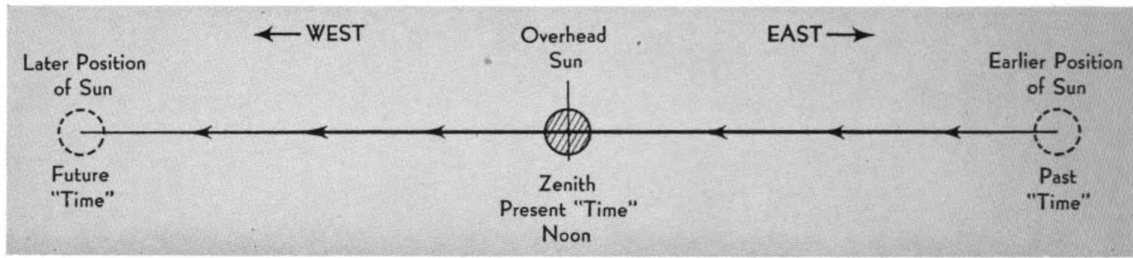


FIG. 92—APPARENT MOVEMENT OF SUN AFFORDS MEASURE OF TIME

(Figure 92). Although any meridian may be used, by international agreement the meridian through the Naval Observatory at Greenwich, England, was chosen as the prime meridian from which to reckon time.

Transit Instrument—Time is recorded at the observatory by a transit instrument. Basically, this instrument is a tube, the longitudinal axis of which is aligned with the meridian and the upper end of which contains a series of threads, parallel to the meridian. The lower end contains photographic equipment, which automatically photographs, on a moving plate, the exact instant that the celestial body chosen crosses the meridian (Figure 93).

Note: 1. The location of a celestial body West of Greenwich is determined by Greenwich hour angle.

2. The location of an observer either East or West of Greenwich is determined by longitude.

3. The location of a celestial body in relation to the observer is determined by local hour angle.

Relationship of Time, Hour Angle and Longitude (Figure 94)—GHA, LHA, and longitude are measured in degrees of arc along the equinoctial or equator, which are great circles equal to 360° of arc.

Time also may be converted into degrees of arc, because the unit of time (one day) is generated by the apparent movement of a body, in a circle of 360° of arc, around the earth. Therefore, time, hour angle, and longitude are related directly to each other.

Time may be converted into degrees of arc (hour angle or longitude), or arc (hour angle or longitude) may be converted into time by division or multiplication.

Conversion of Time Into Arc—The unit of time (one day) is divided into 24 hours of 60 minutes each, and each minute equals 60 seconds.

Since one day of 24 hours equals 360° of arc, the division of 360° by 24 hours proves that one hour equals 15° of arc.

Thus: One minute = $\frac{1}{4}^\circ$ (or $15'$) of arc.

Four minutes = 1° of arc.

One second = $\frac{1}{4}'$ (or $15''$) of arc.

Four seconds = $1'$ of arc.

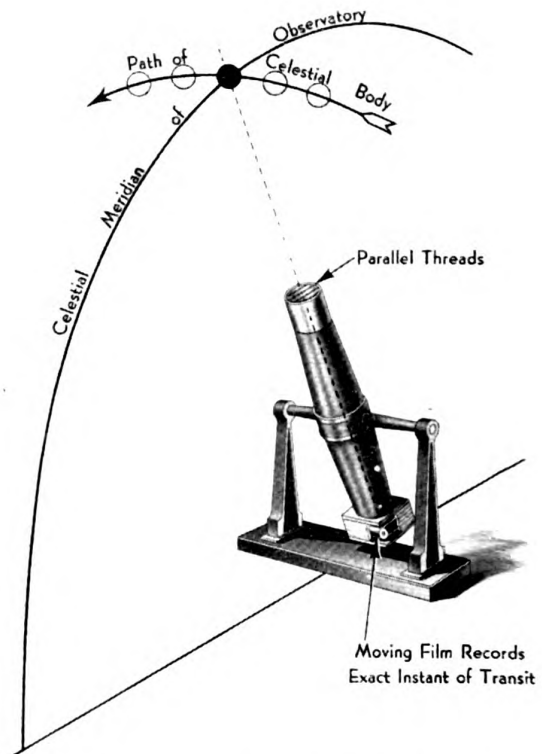


FIG. 93—OPERATION OF TRANSIT INSTRUMENT

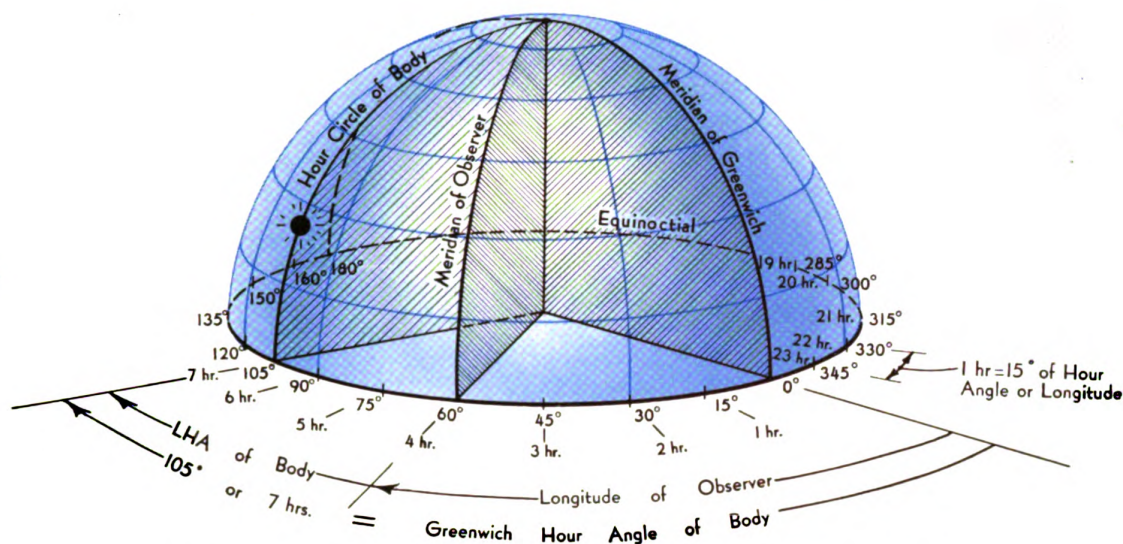


FIG. 94—RELATIONSHIP OF TIME, HOUR ANGLE, AND LONGITUDE

Three Kinds of Time—In addition to its daily rotation on its own axis, the earth also is describing a yearly revolution around the sun. The path it follows in this revolution is known as the earth's orbit. This *real* movement of the earth gives rise to an *apparent* movement of the sun. But since the earth does not move at a constant rate along its orbit, the apparent motion of the sun is not uniform, nor does it correspond exactly to the apparent movement of the stars. This variation in apparent movement gives rise to three kinds of time:

1. Apparent solar time, the unit of which is one apparent day.
2. Mean time or civil time, the unit of which is one civil day.
3. Sidereal time, the unit of which is one sidereal day.

APPARENT SOLAR TIME

Measurement—Apparent solar time is measured by the daily motion of the true sun.

The Unit—One apparent day is the interval of time between two successive *lower transits* of the sun over the same meridian. Obviously, the sun actually would have to be over the upper branch of the observer's meridian in order to measure the exact instant of transit. However, the unit, one day, containing twenty-four hours, is considered to have commenced

twelve hours earlier, or at the instant the sun crossed the lower branch of the meridian, at midnight. What actually is measured is *apparent noon* (Figure 95). Thus, time and hour angle differ by twelve hours, as hour angle is measured from the *upper* branch of the meridian.

Variation in Apparent Time—Clocks cannot be regulated to apparent solar time because the unit, one solar day, varies in length from day to day. This unequal rate is due to the obliquity of the ecliptic, and the lack of uniformity of motion of the earth in its orbit.

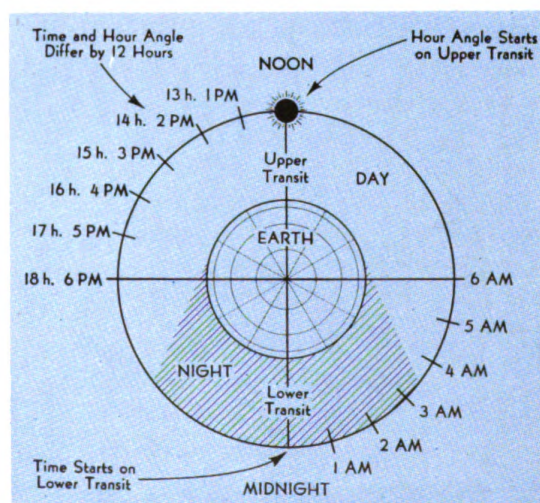


FIG. 95—UPPER AND LOWER TRANSITS OF SUN

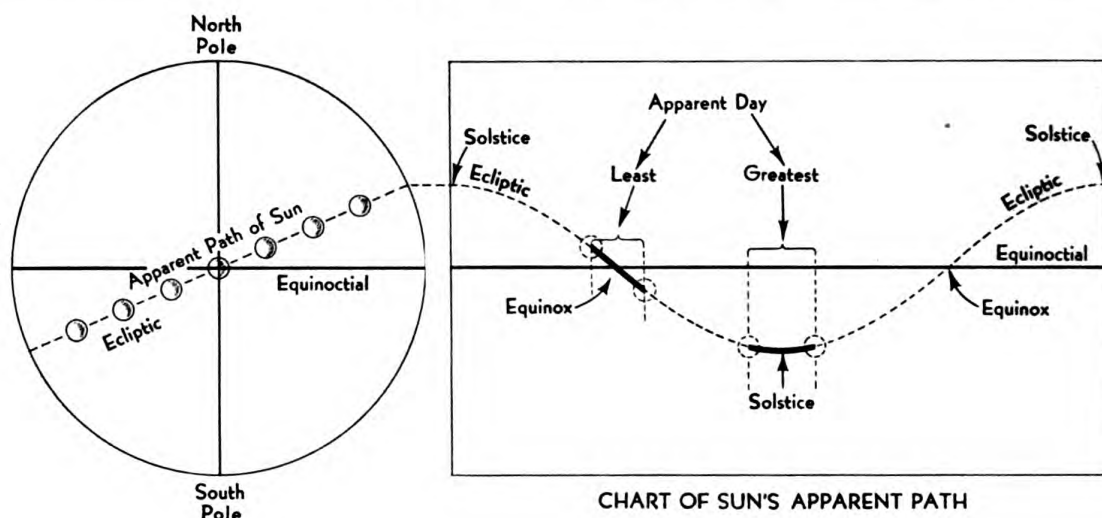


FIG. 96—VARIATION IN LENGTH OF APPARENT DAY DUE TO OBLIQUITY OF ECLIPTIC

Obliquity of the Ecliptic—Uniform time and hour angle are measured on the equinoctial. However, the apparent path of the sun is along the ecliptic, which is inclined at an angle of $23^{\circ}27'30''$ to the equinoctial. Hence, when the sun's motion is translated to the equinoctial, each apparent day varies in length (Figure 96).

Because of the angle of the apparent sun's path on crossing the equinoctial, the apparent day intercepts a smaller arc of equinoctial at the equinoxes than at the solstices, where the apparent path is parallel to the equinoctial.

Lack of Uniformity of Motion of the Earth in Its Orbit—In its yearly movement around the sun, the earth describes an ellipse.

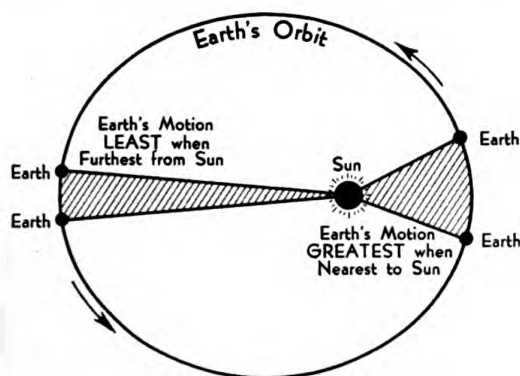


FIG. 97—EARTH'S RATE OF MOTION NOT UNIFORM

It has been demonstrated by Kepler's second law that in December, when the earth is closest to the sun (91,300,000 miles), the earth travels at a greater rate than when farthest away in July (94,000,000 miles). This uneven rate of travel of the earth causes the apparent motion of the sun to vary (Figure 97).

CIVIL TIME OR MEAN SOLAR TIME

Description—Civil life is mainly dependent on the hours of daylight. Therefore, the sun is the most logical reference point for measuring civil time. But since clocks cannot be regulated to the apparent sun's motion, a fictitious body called the *mean sun* is assumed to move along the equinoctial at a uniform daily rate, equal to the average rate of the *true sun* along the ecliptic.

Measurement—Civil time, or mean time, is measured by the daily motion of the mean sun.

Unit—One civil day is the interval of time between two successive lower transits of the mean sun over the same meridian. Clocks are regulated to civil time because it is uniform. Clocks used in civil life usually divide the day of 24 hours into two parts of 12 hours each. The 12-hour period measured from the instant of lower transit of the mean sun is called A. M. time, and the 12-hour period measured from the instant of upper transit is called P. M. time.

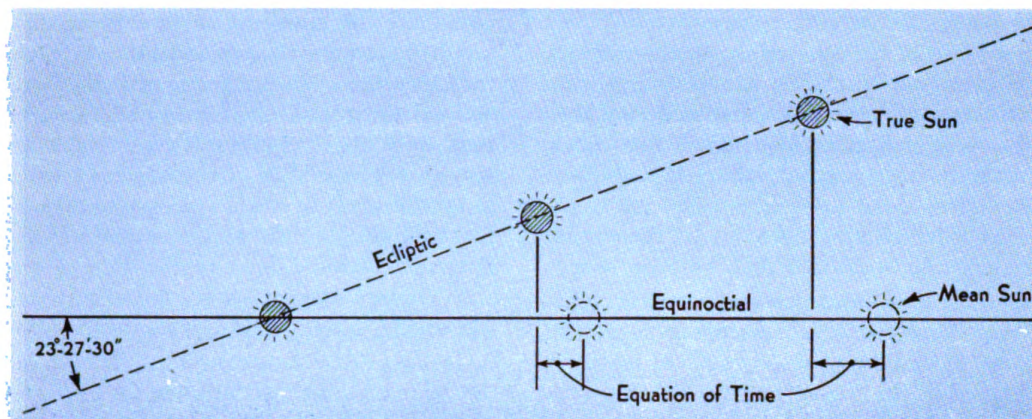


FIG. 98—RELATIONSHIP OF MEAN TIME TO SOLAR TIME

Chronometers used in navigation, however, usually are equipped with a 24-hour dial.

Equation of Time—The equation of time is the difference in hour angle between the hour circle of the true sun and the hour circle of the mean sun (Figure 98). The true sun sometimes is ahead and sometimes behind the mean sun

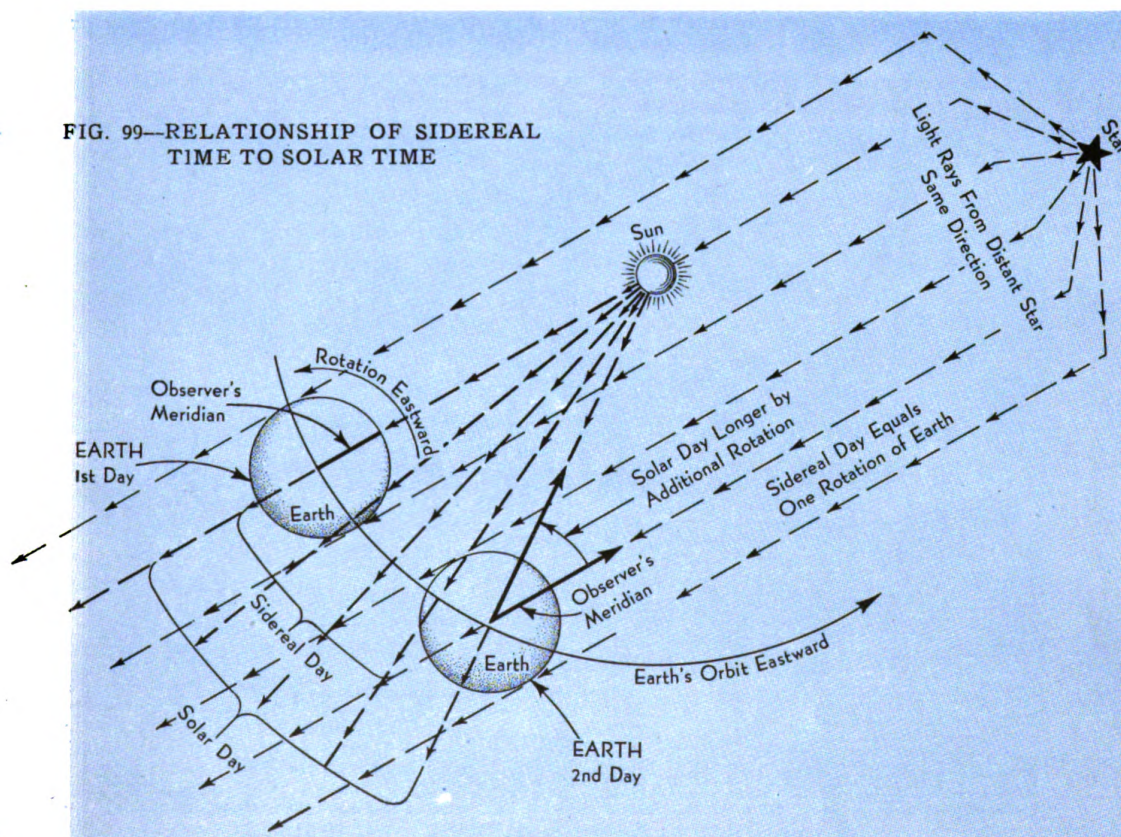
by an amount which varies from zero to about 16 minutes.

SIDEREAL TIME

Measurement—Sidereal time is derived from the apparent movement of the stars.

Unit—One sidereal day is the interval of

FIG. 99—RELATIONSHIP OF SIDEREAL TIME TO SOLAR TIME



time between two successive upper transits of the First Point of Aries over the same meridian. (The First Point of Aries, as previously defined, is one of two points of intersection of the ecliptic and equinoctial. Although it is a computed point in the celestial sphere invisible to an observer, it was selected as the point of origin for sidereal time because, in theory, it moves along the equinoctial at a uniform rate.)

Since the sidereal day starts at the instant the First Point of Aries crosses the upper branch of a meridian, sidereal time and the hour angle of the First Point of Aries are equivalent.

Difference Between Sidereal and Solar Time (Figure 99)—Units of sidereal time are of less duration than corresponding units of solar time because the sidereal day is measured from a star, whose distance is so infinitely great it may be considered a fixed point in space. By comparison, the solar day is measured from the sun, which appears to move due to the earth's yearly revolution around the sun.

Since stars are fixed, the sidereal day is generated solely by the earth's rotation, which is equal to approximately $23^{\text{h}}56^{\text{m}}$ of a mean

solar day. A mean solar day is nearly four minutes greater than a sidereal day because each day the earth moves in its orbit approximately 4° around the sun. Therefore, before two successive transits of the sun can be observed, the meridian of the observer will have to perform more than one complete rotation, corresponding to the apparent daily change in the sun's position.

The complete orbit covered by the sun is a mere pin point in comparison with the infinite distance of the nearest star. Therefore, the direction of a star is the same for any position of the earth in its orbit.

ZONE TIME

Description—The civil day for any position on the earth starts at the instant the mean sun crosses the lower branch of the meridian of that place. Therefore, *civil time* varies with the longitude of the place.

For convenience, and also in order to avoid endless confusion when traveling or in everyday business life, the entire region—as near as practical—extending $7\frac{1}{2}^{\circ}$ of longitude on either side of a designated standard meridian

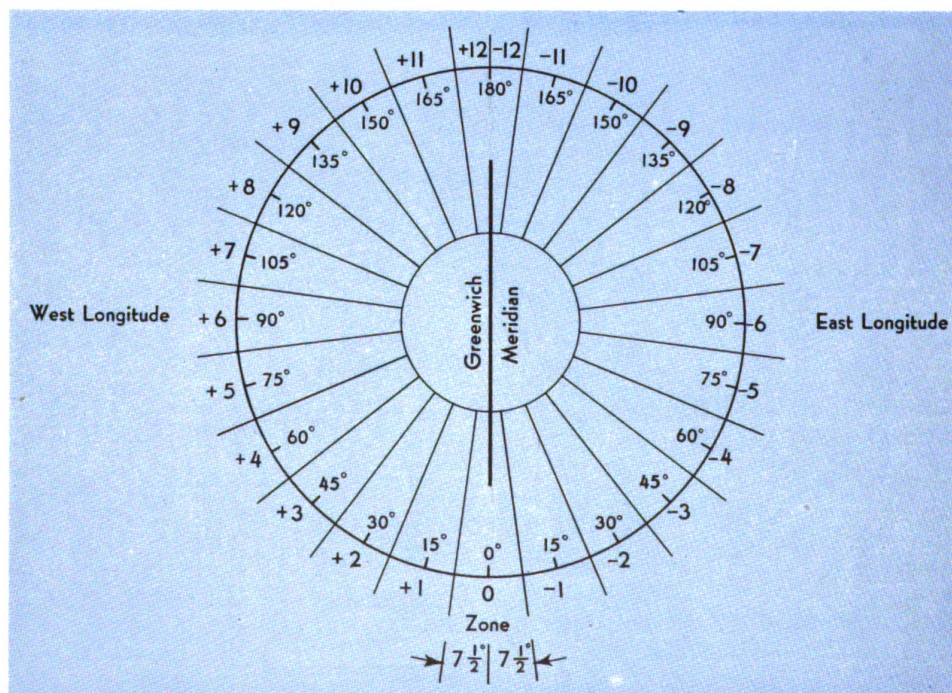


FIG. 100—DERIVATION OF ZONE DESCRIPTIONS

keeps its watches set to the time of the standard meridian. This time is called *zone time*.

The $7\frac{1}{2}^\circ$ of longitude on either side of the standard meridian form a zone 15° wide which equals the number of degrees of longitude the mean sun appears to travel in one hour. In order to develop zone time, the surface of the earth is considered to be divided into 24 of these one-hour zones, and the 24 standard meridians (15° of longitude apart) commence with the meridian of Greenwich as the prime meridian for reckoning time and longitude.

Zone Description (ZD) (Figure 100)—Zone description, abbreviated ZD, is the number of hours the zone is located East or West of the prime meridian of Greenwich. The twelve zones west of Greenwich are considered plus, and the twelve zones East of Greenwich, minus.

The zones in *West longitude* are called *plus zones* because, when an aircraft is in West longitude, the Greenwich civil time is *greater* than the local zone time, hence the ZD must be *added* to the *zone time* to find the GCT. This is due to the apparent westward movement of the

sun. When the sun is on the Greenwich meridian it has not yet crossed the meridians West of Greenwich, but it already has crossed the meridians East of Greenwich. As a result, when an aircraft is in *East longitude*, the Greenwich civil time is *less* than the local zone time and the zones are called *minus zones*, which means that the ZD must be *subtracted* from *zone time* to get the GCT (Figure 100).

International Date Line—The lower branch of the Greenwich meridian is known as the International Date Line, because an observer crossing this meridian will experience a date change.

Since zones extending to the West of Greenwich increase progressively to plus 12 hours, and zones East of Greenwich to minus 12 hours, obviously where they meet (at the lower branch of the Greenwich meridian) there exists an accumulated time difference of 24 hours or one day. An observer crossing from the plus 12 (West) zone into the minus 12 (East) zone would, therefore, gain one day. Conversely, crossing from East to West, he would lose one day.

Zone Time Conversion—If either the zone time or the Greenwich civil time is known, the other is easily computed by the following jingle:

Longitude West—Greenwich Time Best.
Longitude East—Greenwich Time Least.

Example No. 1

May 1, Longitude 75° West, Zone Time 10:20 A. M.
Find: GCT.

Longitude 75° West	
15° (one hour longitude)	= +5 hours = Zone Description
Zone Time =	10:20 A. M.
ZD =	+5:00
GCT	15:20 (Greenwich Time Best)

Example No. 2

May 1, Longitude 90° East, GCT 20^h30^m00^s
Find: Zone Time.

Longitude 90° East	
15°	= -6 hours = Zone Description
GCT =	20:30:00 (Greenwich Time Least; there-
ZD =	- 6:00:00 fore, ZD -6 hours is added
Zone Time	26:30:00 to GCT)
	-24:00:00
Zone Time =	2:30 A.M. May 2

LCT-GCT Relationship—As discussed previously, civil time is time based on the apparent passage of the mean sun over the lower branch of a meridian on the earth. If the Greenwich meridian is used, the result is GCT. If any meridian other than the Greenwich meridian is used, the result is known as local civil time. Since the relationship between any meridian on the earth and the Greenwich meridian is determined by longitude, it is logical to assume—as is actually the case—that LCT and GCT also are related by longitude. In other words, the local civil time for any meridian, at any given instant of time, differs from Greenwich civil time by the longitude of that meridian. This may be expressed as follows:

$$\text{GCT} = \text{LCT} + \text{West Longitude, or} \\ - \text{East Longitude.}$$

Conversely,

$$\text{LCT} = \text{GCT} - \text{West Longitude, or} \\ + \text{East Longitude.}$$

The above formulas may be used to determine either LCT or GCT if one of them, and the longitude, is known. Or, if both the LCT and GCT are known, then it is possible to solve for the longitude. Obviously, however, consistent terms must be used. If solving for time, then longitude must be converted from arc to time; if solving for longitude, then the difference between LCT and GCT must be converted to degrees of arc.

It is important to recognize that zone time is, in reality, no more than a special case of local civil time, the distinction being that in zone time, the entire area lying $7\frac{1}{2}^\circ$ on either side of a standard meridian uses the local civil time of that standard meridian as its time. Actually, every point within the area has its own local civil time. Thus the zone time of a place whose longitude is 16° is one hour different than the GCT, whereas its LCT differs from the GCT by one hour and four minutes, which represents the exact longitude of the place converted from arc to time.

Examples of LCT and GCT Problems:

Example No. 1

May 1, Longitude $83^\circ 30'$ West, LCT = 6:32 A. M.

Find: GCT.

Longitude $83^\circ 30'$ West, converted to time = $5^h 34^m$

$$\begin{array}{rcl} \frac{(75^\circ)}{(15)} & = & 5^h + (8^\circ \times 4^m = 32^m) + (30' \times 4^s = 120^s \text{ or } 2^m) = 5^h 34^m \\ \text{LCT} & = & 6^h 32^m \\ \text{Longitude} & = & 5^h 34^m \text{ West} \\ \hline \text{GCT} & = & 12^h 06^m \text{ (Longitude West—GCT Best)} \end{array}$$

Example No. 2

May 1, Longitude $83^\circ 30'$ East, GCT = $16^h 20^m 10^s$

Find: LCT.

Longitude $83^\circ 30'$ East, converted to time = $5^h 34^m$

$$\begin{array}{rcl} \text{GCT} & = & 16^h 20^m 10^s \\ \text{Long.} & = & 5^h 34^m 00^s \text{ East} \\ \hline \text{LCT} & = & 21^h 54^m 10^s \text{ or } 9^h 54^m 10^s \text{ P. M. (Longitude East} \\ & & \text{—GCT Least)} \end{array}$$



TIME

PROBLEM WORK NO. 21

LOCAL ZONE TIME

(Find arrival time and date in GCT and local zone time.)

No.	FROM	TO	DEPARTURE (GCT)	FLYING TIME	ARRIVAL TIME AND DATE	
					GCT	ZONE TIME
1	San Francisco	Honolulu	10:00 May 1	12h00m		
2	Canton	Honolulu	21:00 May 5	9h00m		
3	Christmas	Fiji	14:00 May 10	7h00m		
4	New Caledonia	Brisbane	11:00 May 15	6h30m		
5	Brisbane	New Caledonia	23:00 May 20	6h00m		
6	Fiji	Canton	10:30 May 25	8h15m		
7	Honolulu	San Francisco	23:00 May 30	13h00m		
8	Honolulu	Christmas	14:15 May 15	10h00m		
9	Christmas	Fiji	15:00 May 1	7h00m		
10	Fiji	New Caledonia	16:30 May 10	6h00m		
11	New Caledonia	Brisbane	09:00 May 5	6h30m		
12	Brisbane	New Caledonia	06:30 May 20	5h30m		
13	New Caledonia	Fiji	17:15 May 25	7h00m		
14	Fiji	Canton	16:00 May 15	7h00m		
15	Canton	Honolulu	07:00 May 30	11h00m		
16	Honolulu	San Francisco	14:30 May 1	13h00m		
17	San Diego	Honolulu	05:13 May 5	13h41m		
18	San Francisco	San Diego	22:00 May 10	3h00m		

ZONE DESCRIPTION (War Time) *

Brisbane	—10	Christmas Is.	+10½
San Francisco	+ 7	Fiji Islands	—12
Honolulu	+ 9½	New Caledonia	—11
Canton Island	+10½	San Diego	+ 7

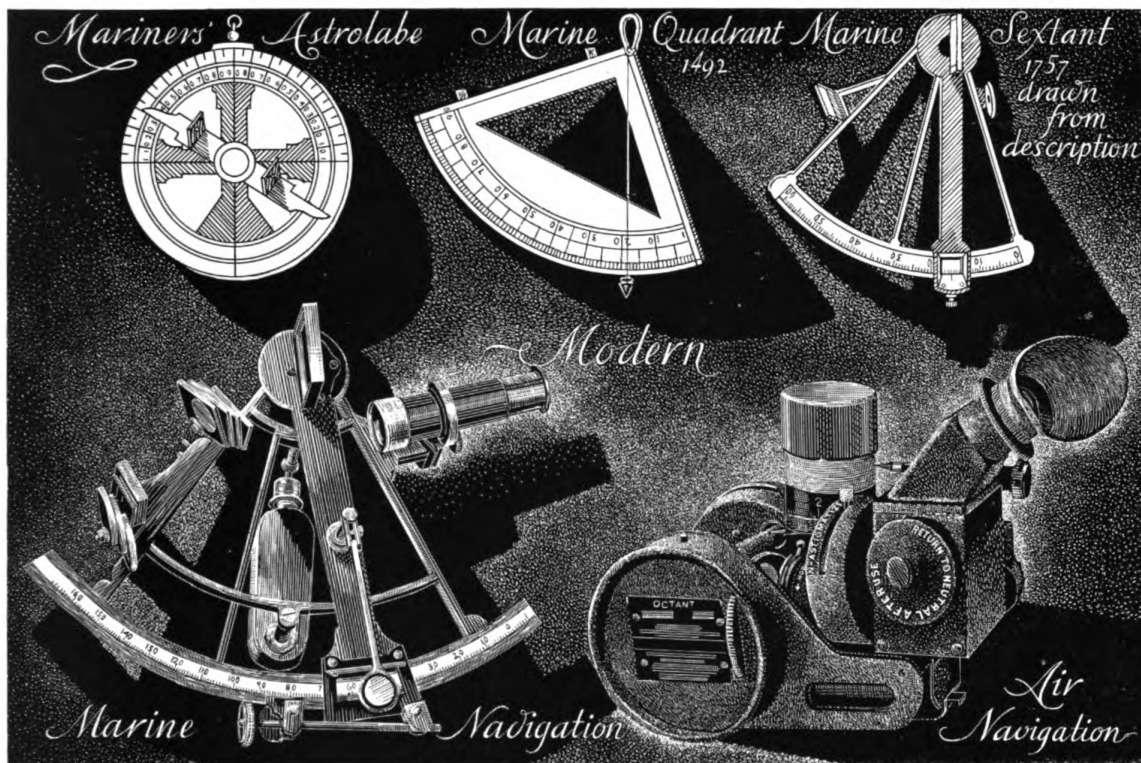
AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 22

LCT—GCT

(Find missing LCT or GCT, and date to nearest second.)

No.	LONGITUDE	GCT		LCT	
		TIME	DATE	TIME	DATE
1	90°W	22:30:20	May 1		
2	15°E	04:16:00	May 5		
3	110°15'W	05:55:00	May 10		
4	178°30'E	23:45:30	May 15		
5	117°11'W	16:43:04	May 20		
6	3°15'E	09:20:20	May 25		
7	145°30'W	08:25:30	May 30		
8	90°15'E	18:29:00	May 10		
9	170°16'E	09:05:06	May 1		
10	116°14'E	18:42:00	May 15		
11	173°40'W			06:40 P. M.	May 5
12	162°10'E			05:10 P. M.	May 1
13	93°20'E			12:42 P. M.	May 10
14	46°30'E			10:00 A. M.	May 15
15	73°40'W			07:56 P. M.	May 20
16	117°11'W			04:53 P. M.	May 25
17	173°20'E			12:00 Noon	May 30
18	45°00'W			02:16 P. M.	May 15
19	168°10'W			11:41 P. M.	May 5
20	101°11'E			01:32 A. M.	May 20



☆ 8 ☆

AIRCRAFT OCTANT

LIKE the modern aircraft compass, which is an up-to-date version of the medieval mariners' lodestone, today's aircraft octant is essentially a refinement of the early altitude-measuring devices used by such famed navigators as Columbus and Vasco de Gama. Columbus knew little of the science of celestial navigation as it is practiced today. History indicates that he relied almost exclusively upon dead reckoning, at which he was unusually proficient considering the crudeness of methods employed.

According to historians, the common quadrant was the only instrument of celestial navigation Columbus ever used. This primitive device was a quarter-circle of wood with sights

along one edge. A weight, swung from the apex of the instrument by a silk cord, caused the cord to give an altitude reading on the 90° scale when the sights were lined up on the heavenly body observed. It is easy to imagine the difficulties these early navigators encountered when trying to use such a device on their small, rolling, pitching ships. In fact, it is thought that Columbus never employed the quadrant for navigation, using it solely in attempts to determine the position of islands and other areas discovered in his journeys. Vasco de Gama, too, was equipped with some sort of altitude-measuring device, thought to be an astrolabe, and it is said that he always disem-

☆ 101 ☆

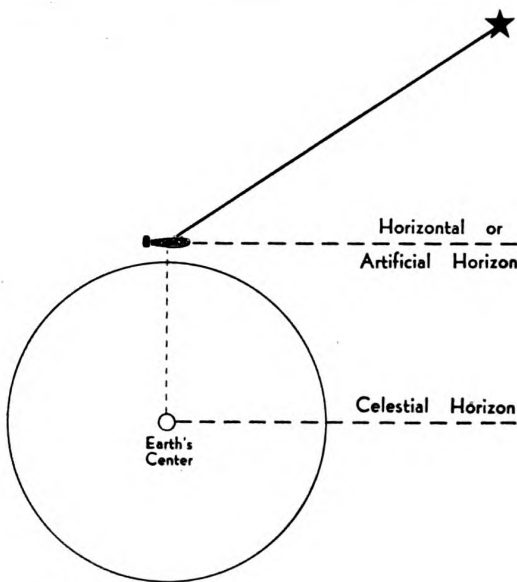


FIG. 101—OCTANT EMPLOYS ARTIFICIAL HORIZON

barked and hung the instrument on a tree in order to secure a steady rest when taking altitude sights.

The term "sextant" (sixth of a circle) first was applied to an optical instrument for measuring angular distances, invented by John Hadley in 1731. Hadley's device actually was an octant (eighth of a circle), but his instrument was enlarged by a Capt. Campbell in 1757 to one-sixth of a circle to meet the needs of navigation.

The marine sextant used for navigation of surface vessels is, generally speaking, not practical for aerial navigation since it relies upon a visible horizon from which to measure the altitude of a celestial body. Such a horizon seldom is available to the aerial navigator, and he must, therefore, use an instrument which contains its own horizon. For this reason the bubble octant was developed. Conversely, the

bubble octant is seldom useful on surface vessels because results obtained with it are not considered sufficiently accurate for surface navigation purposes.

OCTANT OR SEXTANT

The aircraft octant, or sextant, is an instrument for measuring the angle between any two visible objects. The principal use of the octant in aerial navigation is for measuring the altitude of a celestial body above an artificial horizon (a horizontal plane parallel to the celestial horizon) and in some cases, the altitude of a body above sea horizon (See Figure 101).

The terms "octant" and "sextant" are derived from the size of the angle such instruments are capable of measuring (Figure 102).

An octant measures angles up to 90° .

A sextant measures angles up to 120° .

A quintant measures angles up to 144° .

A quadrant measures angles up to 180° .

Most aircraft instruments are of the octant type, as the aerial navigator seldom has reason to measure an angle over 90° . A few, however, are of the sextant type.

ACCURACY OF AIRCRAFT OCTANTS

Obviously, it is impossible to measure exactly the altitude of a celestial body. The accuracy of measurement depends upon the refinement of the measuring instrument.

The theodolite, a high-precision instrument used in surveying, can measure angles as small as $1/1000$ th of a minute of arc.

A marine sextant, using the visible sea horizon, will measure altitude accurately to within tenths of a minute. Aircraft octants, however, which are made light and compact for easy handling, can be relied upon only to measure an altitude, from the sea horizon, to within one or two minutes, and when the artificial horizon is used, measurements may contain inaccuracies of from four to eight minutes or more.

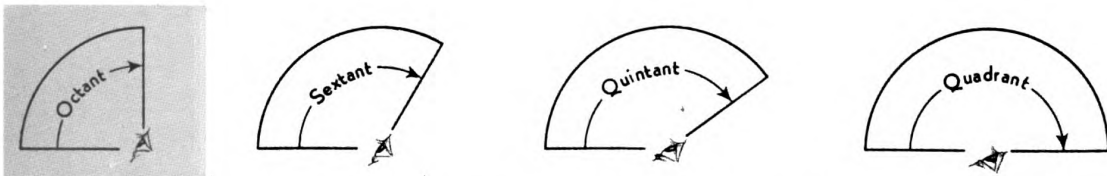
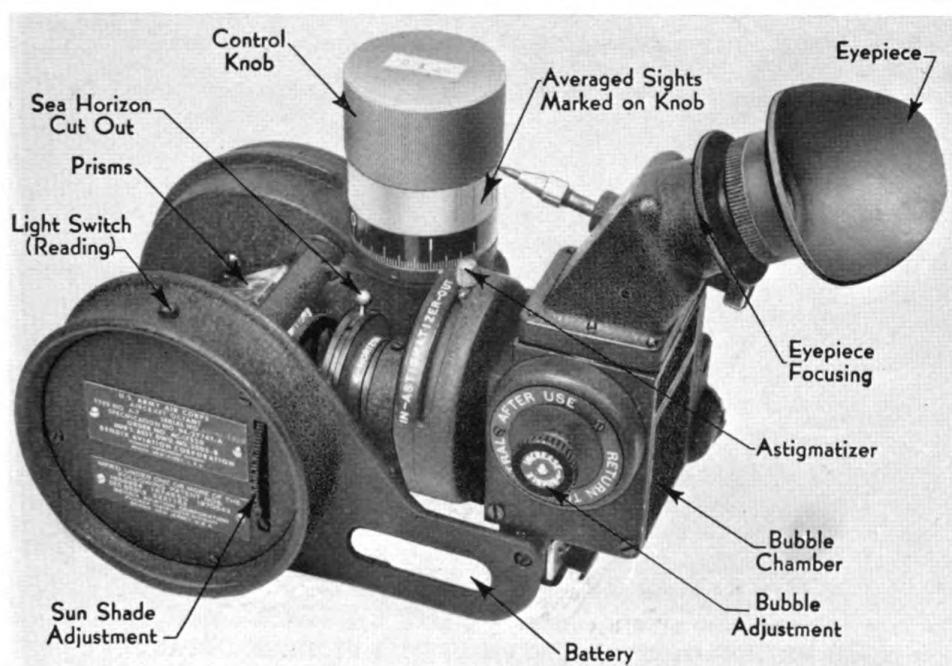


FIG. 102—ALTITUDE RANGE DETERMINES NAME OF INSTRUMENT



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

FIG. 103—BUBBLE OCTANT (Pioneer averaging type)

TYPES OF OCTANTS

Reasons for Different Types of Octants—

The ideal aircraft octant would be one which could be used to measure, with consistent accuracy, the altitude of a celestial body with a single observation.

This ideal is nearly accomplished when the altitude of a celestial body is measured with reference to the sea horizon. However, the sea horizon seldom is visible to the aerial navigator, and even when it is, it can be used only if the aircraft is at a very low altitude. Hence it is necessary, in aerial navigation, to employ an instrument which will measure the altitude of celestial bodies above an artificial horizon parallel to the true or celestial horizon.

The necessity of using an instrument which will measure accurately the altitude of a celestial body above an artificial horizon presents the main problem in present-day aerial navigation, and is the reason why so many different types of aircraft octant have been developed. Principal types which thus far have been developed are the gyro, pendulus, and bubble octants. Each of these utilizes the gravitational pull of the earth on a gyro, pen-

dulum or bubble in order to produce artificially a true horizon.

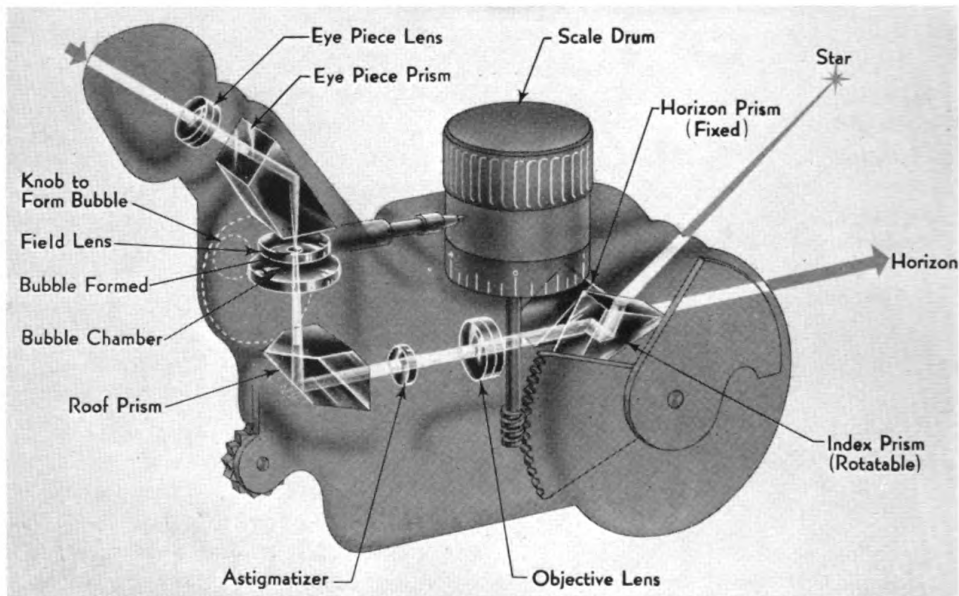
Gyroscopic Octant—The gyroscopic-type octant produces a true horizontal by means of a rotating gyro. This instrument appears to have possibilities of attaining the "single observation" ideal, but thus far no such instrument has been developed for aerial navigation which is adaptable from the standpoint of size, weight, and structural ruggedness.

Pendulus Octant—The pendulus-type octant produces a horizontal by means of a pendulum. No octant of this type has yet been developed suitable for aerial navigation.

Bubble Octant (Figure 103)—The bubble-type octant produces a horizontal by means of the gravitational pull on an air bubble, which is either permanently formed, or formed prior to use, in an enclosed chamber of liquid.

Nearly all aircraft octants are of the bubble horizon type, principally because of their simple and rugged construction, and ease of handling.

There are many bubble-type octants in use, and the aerial navigator should become familiar with as many of them as possible. Nearly all



Courtesy Eclipse Pioneer Division Bendix Aviation Corporation.

FIG. 104—OPTICAL SYSTEM OF THE PIONEER OCTANT

types now available are reliable when used correctly and handled with care. In choosing an octant, the navigator should ascertain if the instrument is constructed strongly enough to give consistently accurate readings under normal handling conditions. He also should consider the reliability of its bubble horizon, size and weight of the instrument, and most importantly, the ease with which it can be used to determine altitudes. The best type octants incorporate a simple and accurate averaging device, and allow a clear view of the sky so that the navigator can be certain he has the correct star in his field of vision. Another important consideration in the octant is the ease with which any error in the instrument can be determined and eliminated. An added help in the instrument is a means of observing the sea horizon, to permit check of index error while in flight (the procedure for which is explained later in this chapter).

OPTICAL PRINCIPLE OF OCTANTS

The principle underlying the operation of all altitude-measuring devices is essentially the same. A law of physics states that when a ray of light is reflected from a plane surface, the "angle of incidence" is equal to the "angle of reflection."

In the octant, the angle between a celestial body and the horizon (either the sea horizon or a self-contained artificial horizon) is measured by bringing into coincidence, at the eye of the observer (by means of an optical system consisting of prisms, lenses, and mirrors), the rays of light received directly from one, and by reflection from the other. The angle between the celestial body and horizon (altitude) then is read on a graduated scale by means of a counter (indicator), actuated by the movement of the reflecting surfaces.

Figure 104 illustrates the principle of the Pioneer Octant. The index prism reflects a ray of light to the observer's eye. The horizon prism is fixed and through it the horizon may be viewed directly. Normally, however, an artificial horizon, indicated by the bubble, is used. In this type of bubble octant the image of a celestial body is made to coincide with the bubble. In some instruments, however, the celestial body is viewed directly and the image of the bubble is made to coincide with the body.

DESCRIPTION OF BUBBLE HORIZON

Forming the Bubble—The bubble is formed beneath a concave lens, the plane of which is horizontal when the instrument is held vertically upright. Then, by means of prisms, the

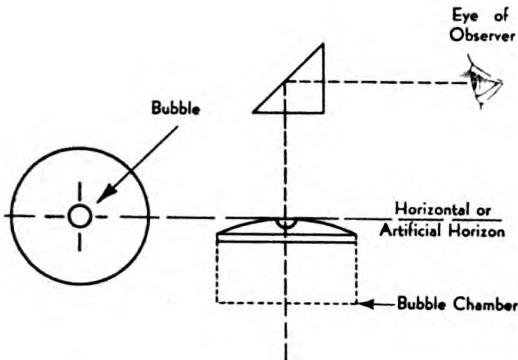


FIG. 105—BUBBLE'S INDICATION OF HORIZONTAL

eye views the bubble vertically through the bubble chamber (Figure 105).

The various type octants employ artificial bubble horizons which are either permanent, semi-permanent, or "make and break" type.

Permanent—The bubble is formed at the factory by adjusting the amount of liquid in the chamber. In this type of bubble horizon, temperature changes sometimes cause the bubble to disappear entirely. The bubble chamber, however, is usually a separate, interchangeable unit, and several chambers containing bubbles of various sizes are available in the carrying case.

Semi-Permanent—The bubble already is formed and may be enlarged or reduced as desired by a special adjusting mechanism.

"Make and Break" Type of Bubble—This is a very good type of bubble once the navigator has become accustomed to using it. The bubble must be formed before each observation, however, and as the size of the bubble tends to vary with temperature it is sometimes necessary to make adjustments during observation of a celestial body. The author used this type of bubble for over two years with excellent results, but a beginner should not attempt to form the bubble or change its size until he understands thoroughly the various characteristics of the bubble cell.

In this type of bubble arrangement there are two chambers completely filled with liquid: a bubble chamber with glass top and bottom which forms part of the optical system of the

telescope, and a diaphragm chamber with a small connecting passageway between the two. A control nut pulls the diaphragm out to form the bubble (see Figure 103).

Frequently there is formed in the diaphragm chamber a bubble too large to pass through the connecting passage into the bubble chamber, independent of the fact that there may or may not be another bubble visible in the bubble chamber. When this occurs, rotation of the control nut to form a bubble only increases the size of the bubble present in the diaphragm chamber, making it impossible for it to pass into the bubble chamber.

The presence of the bubble in the diaphragm chamber can be detected by the reaction of the diaphragm to rotation of the control nut. Thus when a bubble is visible in the bubble chamber, rotation of the control nut will alter the size of the visible bubble more slowly than when no bubble is present in the diaphragm chamber. This is due to the fact that change in pressure acts to alter the size of the bubble in the diaphragm chamber as well.

It is also noticeable that when there is no bubble visible in the bubble chamber, resistance felt when rotating the nut to form a bubble builds up gradually, as contrasted with a sudden building up of the resistance when another bubble is present.

Therefore, to properly form a bubble, the first step should always be (even though the bubble is visible) to hold the octant with the *control nut downward* at an angle of about 45° from the vertical and rotate the nut so as to put pressure on the liquid. If a bubble exists in the diaphragm chamber this will reduce its size, permitting it to pass into the bubble chamber. Sometimes when the bubble exists in the diaphragm chamber, it may be necessary to hold the instrument at this angle for a minute or two, shaking the octant from time to time until the bubble becomes small enough to pass into the bubble chamber.

Next, if no bubble has appeared, with the octant still in the inclined position *rotate the control nut just far enough to overcome the resistance* of the diaphragm, which should build up suddenly. When the bubble forms it will usually be accompanied by a sharp click, which can be felt on the control nut as the resisting force

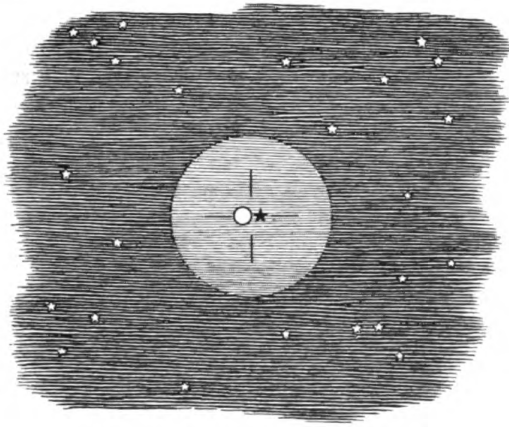


FIG. 106—OCTANT "FIELD"

is released. Then, immediately rotate the nut in the opposite direction in order to apply some pressure on the liquid. This will prevent the formation of too large a bubble and also reduce the size of any bubbles present in the diaphragm chamber so they can pass into the bubble chamber. Rotation of the control nut back and forth several times, each time releasing the suction on the liquid, will speed up the removal of all bubbles from the piston chamber. The control nut now may be set to produce a bubble of the proper size.

Collimation and Field—The bubble octant is so designed that when the image of the celestial body is caused to coincide, or match, with the center of the bubble, the angle as read on the octant will be the observed altitude of the celestial body. "Collimation" is the name given the condition of bubble and celestial body in coincidence.

The "field" of the octant is the small section of the celestial sphere which is visible in the octant's eyepiece (Figure 106). The larger the field of view, the easier it is for the navi-

gator to bring the image of the correct celestial body and the bubble into collimation.

Position of Bubble With Relation to Celestial Body—The optics of the bubble octant are such that collimation need not necessarily take place in the middle of the field, although the center is most desirable.

In Figure 107 the bubble and sun are shown in collimation in various parts of the field.

In No. 1 is shown the best position in the field.

In No. 2, error will be practically zero when collimation occurs anywhere on a vertical line passing through the center of the field.

In No. 3, large errors (about 5') will occur if collimation is effected anywhere in the field except on the vertical passing through the center. This appears contrary to reason, but actually, when collimation occurs along the vertical passing through the center of the field, the octant is aligned with the *vertical circle* passing through the celestial body. Collimation anywhere else indicates that the octant is being held in such a way as to increase the angle of altitude (Figure 108).

OCTANT ERRORS

Description—Aircraft octants are subject, basically, to two types of error: *Instrument error* and *bubble acceleration error*. Instrument error may be classed as a fixed calibration error, as it is caused by improper alignment of the measuring mechanism. It is fixed because it is an error which may be eliminated or, if not eliminated, accurately determined and recorded as *Instrument Correction* (I. C.) to be applied to each observation. Bubble acceleration error, however, is an observation error which is not fixed but varies with each observation taken.

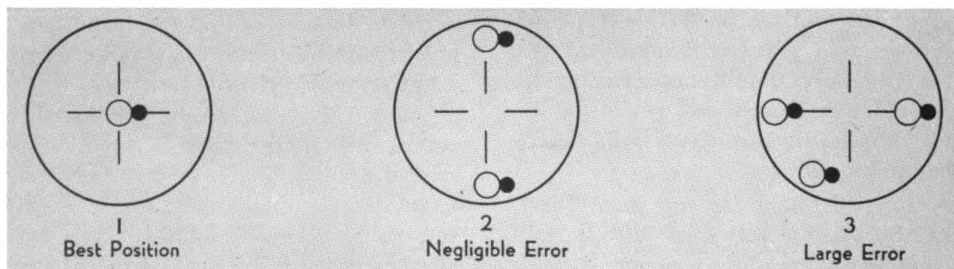


FIG. 107—COLLIMATION

The bubble is freely suspended in liquid in order that the gravitational attraction of the earth will cause it to indicate the horizontal. But, unfortunately, the bubble also responds to the acceleration of the aircraft which frequently causes it to indicate a false horizon.

Instrument Error—Instrument error may be due to *index* or *bubble error*, or to a combination of both.

1. **Index Error**—Index error results when movements of the index prism or mirror are improperly recorded on the graduated scale. When the index prism, which actuates the counter, is in proper alignment (i.e., parallel to the horizon prism), the counter should read zero. If it does not, there is index error present.

To check for index error, a celestial body or distant object on the earth, such as the sea horizon, is viewed directly through the horizon prism. The reflected image of this object, as seen in the index prism, is brought into coincidence with the object itself. This will cause the two prisms to be placed in true alignment, and the counter should read zero. If it does not, the counter or index prism may be adjusted. However, in most octants this requires **extremely** accurate handling. Most navigators prefer to record the amount of error, and to apply this correction to each observation.

For example, if the counter read 5' off the scale (less than zero), all altitudes as read on the octant would underread 5'. Therefore a 5' correction should be *added* to all observations. On the other hand, if the counter read 5' on the scale (greater than zero), a 5' correction should be *subtracted* from all observations.

2. **Bubble Error**—Instrument bubble error results when the bubble chamber is out of line with the optics of the instrument, which causes the bubble to indicate a false horizontal when the acceleration of gravity alone is influencing it.

Bubble error may be determined by mounting the octant on a steady surface, selecting some distant point on the same level as the sextant (either a point determined by a surveyor's level or the sea horizon) and causing the reflected image of such a point to coincide with the bubble's center. Counter should then indicate zero. If it does not, the error may be elim-

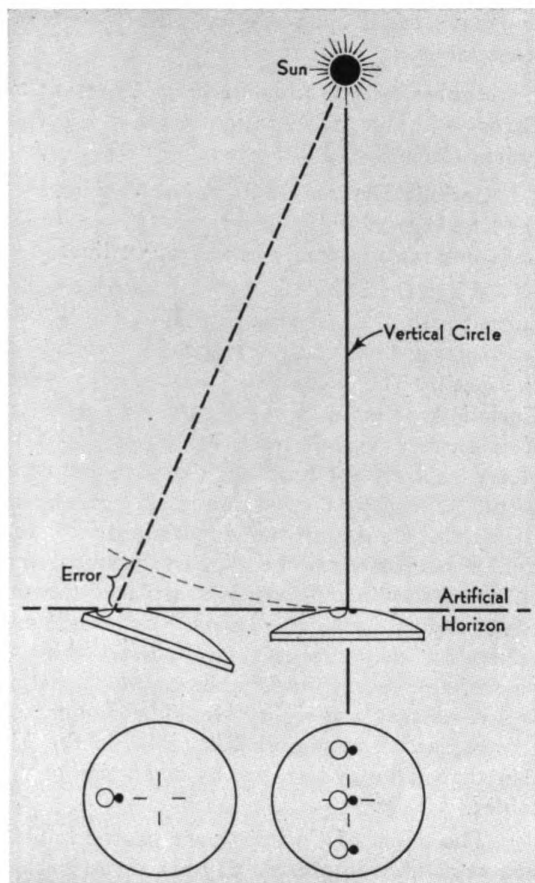


FIG. 108—DEVIATION FROM VERTICAL CAUSES ERROR

inated by adjusting the zero reference line of the scale, by realigning the bubble chamber, or by adjusting the mirrors.

Note: If the sea horizon is used, the dip correction must be taken into consideration.

Since the primary purpose of the octant is to measure the altitudes of celestial bodies from the bubble horizon, most navigators prefer to know the *total* instrument error. Total instrument error can best be determined by plotting a pre-computed curve showing the exact altitude of a celestial body for a period of several hours. The octant is then placed on a rigid support and the altitude of the body measured about every 15 minutes. The amount by which the octant reading differs from the true altitude as shown on the curve is, therefore, the amount of instrument error which, if not eliminated by adjusting the zero reference line of the scale,

will have to be applied to all future observations taken.

Bubble Acceleration Error or Observation Error—Bubble acceleration error is of two types: Coriolis and Transient.

Coriolis Error—A level bubble indicates the true vertical only when the bubble is at rest or is moving uniformly in a straight line.

When the bubble is moving uniformly in a curved path, the zenith indicated by the bubble is displaced toward the center of curvature of the path. The physicist Coriolis, for whom Coriolis type error is named, proved that even if a surface vessel (and hence an aircraft) steered a constant heading, the path followed would be slightly curved due to the rotation of the earth. Hence, altitudes measured with the bubble octant would be slightly in error, and the resultant lines of position displaced toward the center of curvature. In the Northern hemisphere the curvature is always toward the left side of the course, and in the Southern hemisphere toward the right side. The amount of this correction is compiled in tables in the Air Almanac and may be noted by inspection of the tables.

The effect of Coriolis error on the bubble octant as used in present day aerial navigation is, however, almost negligible. And since practical air navigation is not yet an exact science, the navigator need not be overly concerned with finding the exact Coriolis correction to apply to each line of position. With the development of faster long-range aircraft, and especially when flying in higher latitudes, it probably will prove necessary to apply Coriolis correction.

Transient Error—Transient error is error caused by accelerations or sluggish movement of the bubble due to slipping, skidding, varying airspeed and turning of the aircraft. This error is always present when observations are taken in the air, and is the principal reason why determining the true altitude of a celestial body is the most difficult and least accurate step in celestial navigation.

Transient errors in a single observation are frequently as great as 1° or 2° (60 to 120 miles). In order to reduce these errors to a minimum several precautions may be taken:

First, observation should be taken as nearly as possible from the center of gravity of the aircraft.

Second, the bubble should be adjusted to correct size, which is about twice the size of the sun as seen through the eyepiece. If the bubble is too large it will bounce around rapidly from one part of the field to another, making it very difficult to bring the body and bubble into coincidence in the center of the field. If the bubble is too small it will tend to lag, thus indicating a false horizon.

Third, averaging a number of readings taken over a period of time will give the mean of all possible horizontals the bubble indicated during the course of observation. Hence accuracy of the mean altitude increases in direct proportion to the time and number of sights taken; however, much depends upon the condition of the air and the skill of the navigator. With skill, 10 sights taken over a period of two minutes will give very good results.

HANDLING THE OCTANT

The aircraft octant is of rugged construction for aircraft use, but it should be remembered that it is a delicate instrument and should be handled with extreme care at all times. Intelligent use of most aircraft octants requires extensive practice (at least 50 observations) before a navigator begins to acquire a confident "feel" of the instrument. With practice, and by exercising all care humanly possible in taking an observation, he will consistently obtain good results.

The navigator should know all the parts of his octant, how they work and why, and before beginning an observation be sure they are all functioning properly. Prisms, mirrors, and eyepiece should be kept clean, and the eyepiece focused correctly. Also, the bubble should be adjusted to correct size. If a night sight is to be taken, the sun shades should be clear of the field of view, the batteries adjusted, and the light in perfect working order. The procedure in taking a sight is substantially as follows:

1. Scan the sky and select the stars to be "shot," choosing them so that the resulting position lines will intersect to provide a clear idea of the ship's position.



FIG. 109—AIRCRAFT OCTANT IN USE

Note: Selection of a star for purposes of a fix is important. Will it provide a speed line, course line, or check on other lines? Stars which are dim and hard to see, or stars which are obscured by occasional clouds should be shot first, as this will lessen the time between sights. If the sky is clear it is usually best to shoot the course line star first, then check line star, and the speed line star last.

2. Set the approximate altitude of the star on the octant and then make sure you have the right star in the field of view. Check by moving the field of view slightly to see relation of the star to other stars nearby, or check it by lining it up with the ribs in cockpit which hold the glass panes.

3. Select a position in the plane as near the center of gravity as possible from which the celestial bodies will be easily visible.

4. When ready to shoot, notify the captain, so that he may concentrate on holding the plane on as constant and steady a heading as possible.

5. Start shooting. It is best always to take three different star observations whenever possible. Establish a system and always shoot the same way thereafter in order to gain proficiency. Speed is desirable, but not essential. Never compromise accuracy for speed.

The author's system when using a non-averaging instrument, is as follows:

1. With the approximate altitude set, check the time on the chronometer, which is fastened to the octant.

2. Note the even minute, and when the second hand comes around to it, start shooting.

3. Record the observed altitude once every 15 seconds for two minutes and 15 seconds, each time causing the body to move out of coincidence a distance of several bubble diameters (up or down) and then realigning them. Thus ten sights are obtained, the mean altitude being found by adding up the minutes of altitude and then moving the decimal point over one digit. The mean time is obtained by adding one minute, seven seconds to the starting time.

Example:

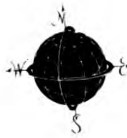
Arcturus	Altitude
Time Start = 23:46:00	47° 30'
Add = 01:07	25'
Mean Time = 23:47:07	25'
	15'
	30'
	25'
	10'
	25'
	25'
	35'
	24.5'
Index Error = +5'	
	H _s = 47° 29.5'
	Ref. —1'
	H _o = 47° 28.5'

When using an averaging octant the procedure is somewhat simpler. Note the time and start shooting on the even minute. While keeping the body and bubble in continuous coincidence, move the recording lever at regular intervals of two or three seconds for a period of two minutes. At the end of this time the average altitude is found by adjusting the index control so that the middle point of the marks left by the recorder is opposite the recorder point (usually a pencil lead) or other reference mark. The altitude is then read directly from the scale. The average time will be the starting time plus one minute.

In addition to the aforementioned methods of averaging sights there are devices which provide sight averages automatically. Several makes of octants feature various types of automatic averagers. One such averager is the so-called "cone" type, on which a constantly mov-

ing stylus records the altitude of the celestial body on a small sleeve of graph paper which slips over a cone-shaped holder, mounted on the altitude control knob. While the observer keeps the body and bubble in collimation over a two-minute period, the stylus records a curve on the graph paper. Examination of this curve provides the body's average altitude.

Another automatic averager records on a speedometer type dial the instantaneous average altitude of a celestial body for a period of two minutes. At the end of this time a shutter closes, indicating termination of the two-minute period. The final reading on the averager represents the average altitude of the sight for the period. This is usually a very accurate average as it actually represents an average of 120 sights (one per second for two minutes).





☆ 9 ☆

POSITION LINES

AS WAS stated in the preceding chapter, the development of the sextant marked an important milestone in the science of navigation. Though the instrument made it possible to measure the altitude of a celestial body more accurately, mariners still were able to determine only their approximate position, and then only after performing much tedious computation.

Like many a mariner who had sailed before him, Capt. Thomas H. Sumner, an American shipmaster, was acutely aware of the fact that the altitude of heavenly bodies could help men find their positions on the surface of the earth. In December, 1837, Capt. Sumner finally made

a discovery which launched the science of navigation into its most important era. This discovery is known as the "line of position," though in honor of its discoverer, it is often referred to as the Sumner line. It has been termed by one authority "the most important principle in modern navigation."

Though Sumner's contribution was a vital one, it remained for a French naval officer, Marcq Saint-Hilaire, to combine the solution of the astronomical triangle with his DR position and so obtain a line of position faster and more directly. His contribution is generally considered the cornerstone of celestial navigation as it is practiced today.

☆ 111 ☆

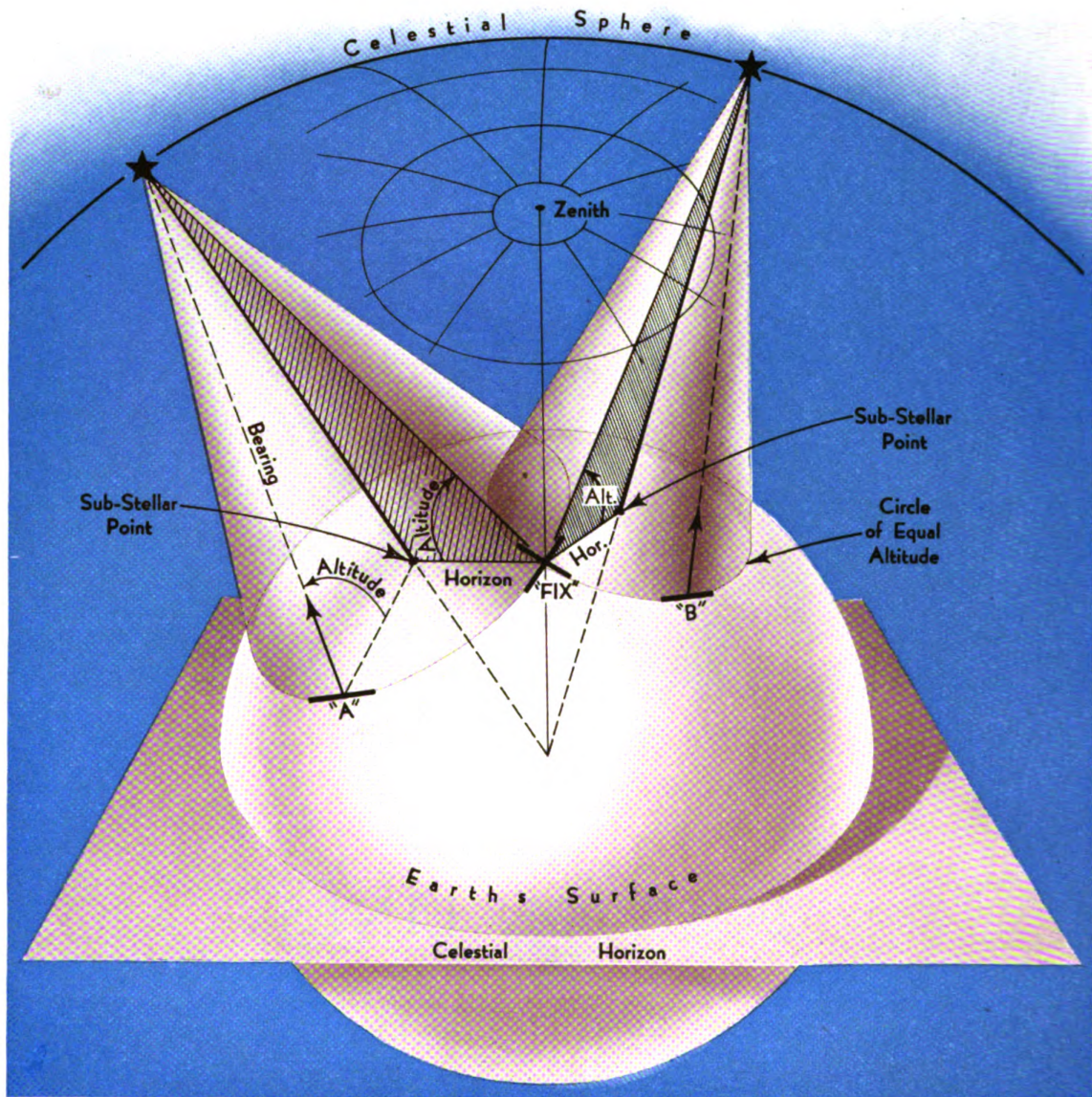


FIG. 110—CIRCLES OF EQUAL ALTITUDE

THEORY

Circle of Equal Altitude—A circle of equal altitude (Figure 110) is a small circle on the earth's surface, from any point of which (at the same instant of time) the altitude of a celestial body would be the same.

A Line of Position is a small segment of a circle of equal altitude. Being a portion of a circle, it is actually a curved line. However, since the circle is usually very large, for purposes of navigation this segment is considered

a straight line perpendicular to the azimuth, (bearing of the celestial body), as illustrated in Figure 110 at positions "A," "B," and at "Fix."

Position Line Fix—As will be noted in Figure 110, the true altitude of a celestial body determines the radius of the circle of equal altitude, or the observer's position with relation to the *geographical position*. It also will be noted that the observer could be located anywhere on this circle, such as at position "A" or

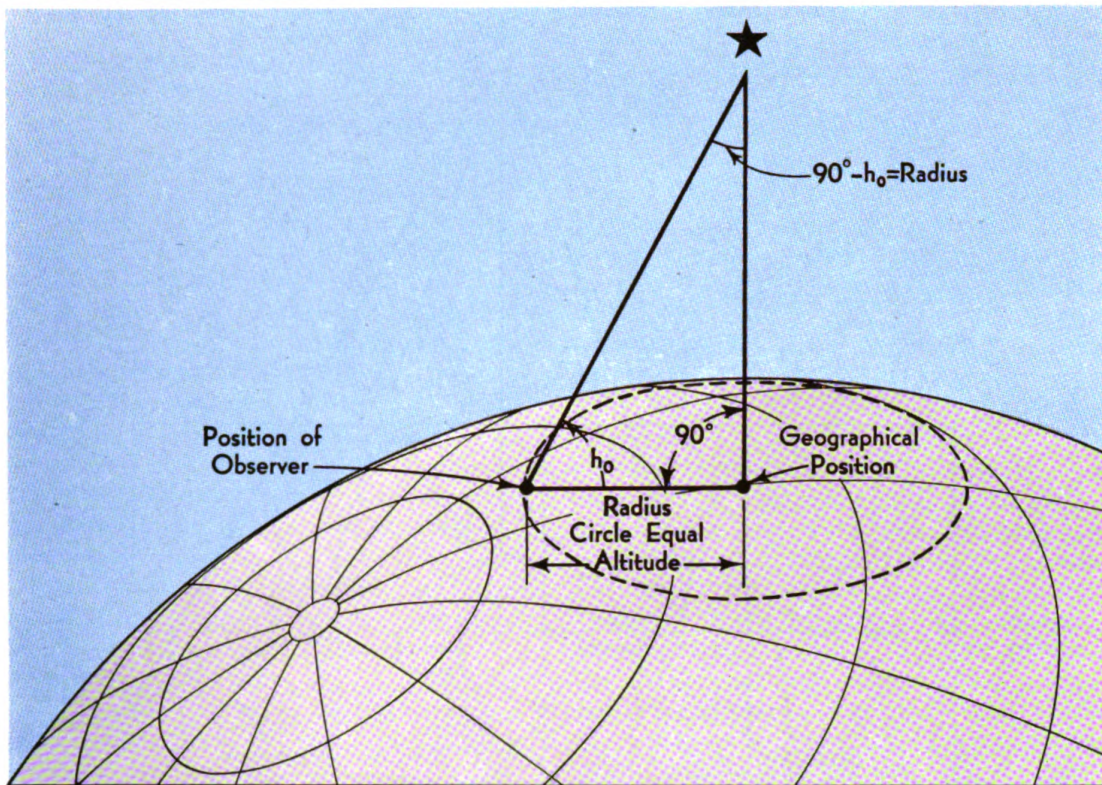


FIG. 111—RADIUS CIRCLE OF EQUAL ALTITUDE = $90^\circ - h_0$

at the fix. If the altitudes of two or more celestial bodies are measured, since the observer's position is somewhere on each circle of equal altitude, he can only be located at one of the intersections of these circles. And though these circles will cross at two points, only one of these intersections may logically be the navigator's position, the other usually being so remote that it can be disregarded. The point of intersection at which the observer is located, when plotted on the chart, is called a "fix," or known position of the aircraft for the time of sight.

The Geographical Position (G.P.) is that point on the earth's surface directly beneath a celestial body and is, therefore, the center of the circle of equal altitude (Figure 110). It is sometimes called the sub-solar, sub-lunar or sub-stellar position to indicate sun, moon or star as the celestial body used.

Since declination is celestial latitude and

Greenwich hour angle is celestial longitude (both contained in the Air Almanac for any instant of time), the geographical position of a celestial body may be located exactly on the earth's surface, using declination as latitude and GHA as longitude.

Observer's Relation to Geographical Position—Since the geographical position is directly beneath the celestial body, a right triangle is formed between the observer's position, the geographical position, and the celestial body itself. And since the three angles of a triangle total 180° , the radius of a circle of equal altitude is equal to 90° minus the altitude of the celestial body (Figure 111).

On the earth's surface, one degree (60 minutes) of arc of a great circle is equal to 60 nautical miles. Therefore, the angle at a celestial body (converted into minutes) equals the radius of the circle of equal altitude in nautical miles, or the *distance of the observer from the*

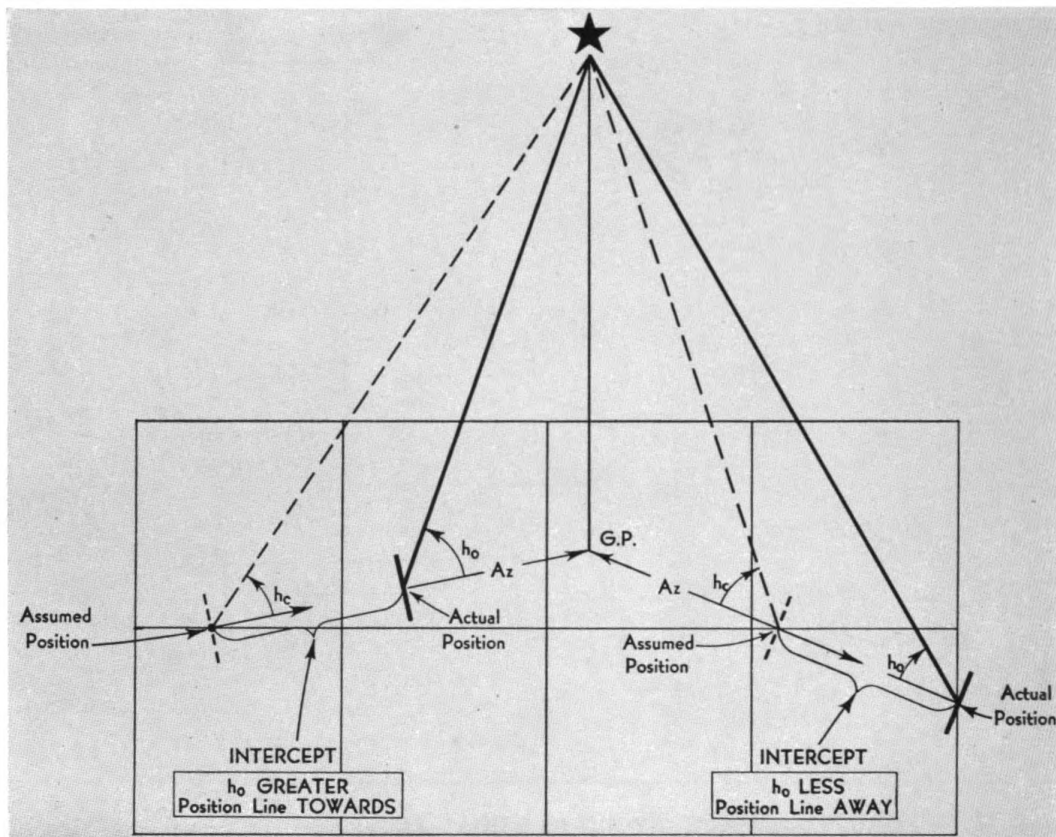


FIG. 112—ALTITUDE INTERCEPT

geographical position. The observer's position is determined then as somewhere on the small segment of the circle of equal altitude perpendicular to the azimuth line, such as at "A," "Fix," and "B" (Figure 110).

For a celestial body near the zenith, the circle of equal altitude would be small and could be plotted directly on the chart. However, for each minute the altitude *decreases* the observer's distance (radius of circle of equal altitude) *increases* one nautical mile. As a result, for the majority of sights the circle of equal altitude is too great for convenient plotting. Moreover, the azimuth cannot be measured closely enough for accurate navigation.

For these reasons the astronomical triangle is solved, and the observer's position relative to the celestial body's geographical position is computed by reference to tables of pre-computed altitude and azimuth. These tables give the navigator the true altitude and azimuth of any celestial body, at any instant,

for any given position on the earth's surface. Tables of this type have been printed in many different forms, each striving to minimize the amount of work involved. However, results obtained by any one of several accepted methods will be similar.

General Procedure of computing a line of position is as follows:

a. The navigator measures the altitude of a celestial body with his octant, and at the same instant notes the GCT. The observed altitude (H_s), when corrected for sextant and altitude errors, becomes the *true altitude* (H_o) of the celestial body from the aircraft's actual position at the exact time of observation.

b. The GCT, by reference to the Air Almanac, gives the *GHA* and *declination* of the body, thus locating its geographical position for that instant of time.

c. Next, by reference to a table of pre-computed altitude and azimuth, the navigator determines the true altitude and azimuth, at the time of

POSITION LINES

sight, for his *assumed* or *dead reckoning* position, which is estimated as near to the actual position of the aircraft as possible.

d. Comparing the two altitudes (Figure 112), if both the observed true altitude (H_O) and the computed altitude (H_C) are equal, the aircraft is somewhere on the line of position passing through the assumed or dead reckoning position. For each minute that true altitude (H_O) is *greater* than computed altitude (H_C), the line of position of the aircraft is one nautical mile *toward* the geographical position, measured from the assumed position. Conversely, for each minute that true altitude (H_O) is *less* than computed altitude (H_C), the line of position of the aircraft is one nautical mile *away* from the geographical position, measured from the assumed position.

This distance of the aircraft's actual line of position *toward* or *away* from the geographical position is called the *altitude intercept* ("a").

H.O. 214—"H.O. 214" is the designation for a group of books published by the U. S. Hydrographic Office (H.O.). These contain tabulated solutions of the astronomical triangle, so arranged as to yield to the navigator his computed altitude and azimuth by inspection. The books are rather large in size, and because each book covers only ten degrees of latitude, the navigator is required to take several volumes on an average flight. However, these tables are widely used because they require little computation, and thus reduce the chance of mathe-

matical error. This is very important to the long range aerial navigator, because, after flying several hours at high altitudes, he is apt to make mistakes due to fatigue.

Note: A newer method, known as H.O. 218, and developed by the British, also is very popular. The tables are similar to H.O. 214; however, they are a little easier to use because of the fact that the declinations of 22 stars have been included, eliminating the necessity of interpolation of declination for these stars. Otherwise, H.O. 218 is essentially the same as H.O. 214 for solutions of any celestial body whose declination is from 0° to 28° .

Use of Tables—The H.O. 214 tables may be used to find the true altitude and azimuth of a body for an aircraft's estimated *dead reckoning position*; however, in order to reduce the amount of interpolation required, they are generally used to find the altitude and azimuth for an *assumed position*. The assumed position is found by taking the nearest whole degree of dead reckoning latitude and the value of dead reckoning longitude which will yield a whole degree of local hour angle.

Arguments to Enter the Tables—The arguments necessary to enter the tables are:

1. Nearest whole degree of latitude.
2. Nearest whole or half degree of declination.
3. Nearest whole degree of local hour angle.

The values obtained are *altitude*, *azimuth*, and Δd ("delta d"). The information is arranged in the following order: Altitude— Δd — Δt —azimuth (see Figure 113).

Δt is disregarded, as it represents the change in altitude for a change of one minute of arc of hour angle, whereas in using assumed position the hour angle employed is a whole degree. Δt is used in obtaining altitude when working from a DR rather than an assumed position.

Δd , however, represents the change in altitude for a change of one minute of arc of declination, and since the exact declination of a celestial body usually will differ from the tabulated declination, the Δd value is multiplied by the number of minutes of declination difference to obtain the correction to altitude. (This correction may be had by inspection from

3° 00'		3° 30'		H.A.	Lat. 32°
Alt.	Az.	Alt.	Az.		
100.0	180.0	54.30	180.0	00	
99.9	178.3	54.29	178.3	1	
99.8	176.5	54.27	176.6	2	
99.7	174.8	54.23	174.9	3	
99.6	173.1	54.17	173.2	4	
99.5	171.3	54.11	171.4	5	
99.4	169.6	54.02	169.7	6	
99.3	168.0	53.52	168.2	7	
99.2	166.3	53.41	166.5	8	
99.1	164.6	53.29	164.8	9	
99.0	163.0	53.15	163.2	10	
98.9	161.6	52.99	161.6	11	
98.8	160.3	52.82	159.9	12	
98.7	158.4	52.64	158.4	13	

FIG. 113

the multiplication table arranged on the inside back cover of the volume or, with a little practice, it may be easily computed).

The application of this correction is determined by examination of the tabulated altitudes. If the values of altitude are *increasing* as the tabulated value used approaches the exact declination, the correction is *added* (indicated by a plus sign); if the values of altitude are *decreasing*, the correction should be *subtracted* (indicated by a minus sign). The altitude thus obtained is the *computed* altitude (H_c) for the *assumed position*.

Procedure to obtain line of position by H.O. 214 is as follows:

- Find the true altitude (H_o) by correcting the observed altitude (H_s) for octant errors (I.C.) and altitude errors.
- Find Greenwich hour angle and declination for the instant of Greenwich civil time of sight from the Air Almanac.
- Assume value of longitude nearest to dead reckoning position that will yield a whole degree of local hour angle. Note direction of LHA, East or West, as this determines whether the body is rising or setting, and is used to name the azimuth.

LHA has been described as the angular distance of the celestial body from the observer.

It is easily determined by simple arithmetic and the hour angle diagram, or by use of the following formula:

$$LHA = GHA - \text{West Longitude}$$

(Add 360° to GHA if necessary)

$$LHA = GHA + \text{East Longitude}$$

Note: The above formulas will result in values of LHA to 360° . However, in using H.O. 214 (or any other method of solving the astronomical triangle), LHA is measured only to 180° and named East or West with reference to the observer. Thus, if the value obtained by use of the formulas is less than 180° , it is named West; if, on the other hand, the value obtained is greater than 180° , it must be subtracted from 360° and then named East.

Note also that in West longitude LHA is always found by subtraction. Hence, the minutes of assumed longitude must equal the minutes of GHA in order for the LHA to come out in whole degrees.

In East longitude, the LHA is always found by addition, hence the minutes of assumed longitude when added to the minutes of GHA must total 60 minutes or one degree for the LHA to come out in whole degrees.

The following examples show how LHA is obtained.

Example No. 1 (Figure 114)

$$\begin{array}{rcl} \text{DR Position} & = & \text{Latitude } 30^\circ 12' \text{ North} \\ & & \text{Longitude } 117^\circ 23' \text{ West} \\ \text{GHA} & = & 93^\circ 19' \\ \text{Assumed Long.} & = & 117^\circ 19' \text{ West} \\ \text{LHA} & = & 24^\circ \text{ East} \\ \text{—or—} & & \\ \text{GHA} & = & 93^\circ 19' \\ & + & 360^\circ \\ & = & 453^\circ 19' \\ \text{Assumed Long.} & = & 117^\circ 19' \text{ West} \\ \text{LHA} & = & 336^\circ 00' \text{ West} \\ & \text{from} & 360^\circ \\ \text{LHA} & = & 24^\circ \text{ East} \end{array}$$

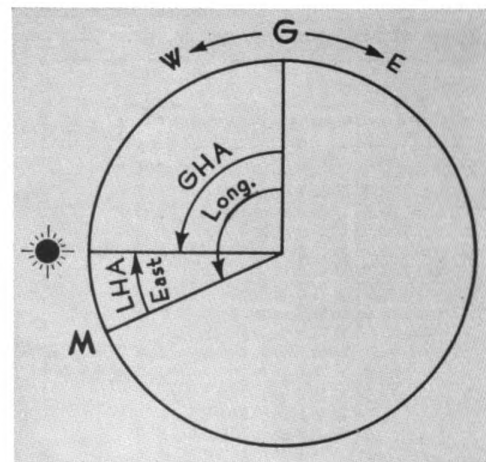


FIG. 114—LHA = GHA — WEST LONGITUDE

POSITION LINES

Example No. 2 (Figure 115)

DR Position = Latitude $33^{\circ}20'$ North
 Longitude $160^{\circ}15'$ East
 GHA = $269^{\circ}28'$
 Assumed Long. = $160^{\circ}32'$ East
 $\quad\quad\quad 430^{\circ}00'$
 $\quad\quad\quad -360^{\circ}$
 LHA = 70° West

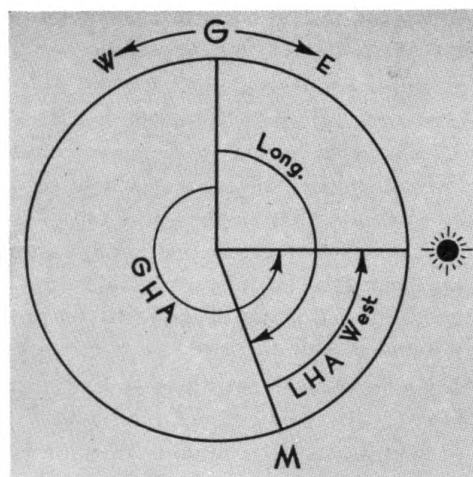


FIG. 115—LHA = GHA + EAST LONGITUDE

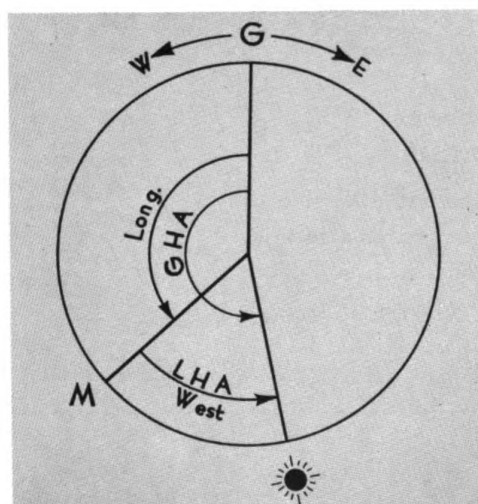


FIG. 116—LHA = GHA - WEST LONGITUDE

Example No. 3 (Figure 116)

DR Position = Latitude $34^{\circ}36'$ North
 Longitude $132^{\circ}00'$ West
 GHA = $190^{\circ}51'$
 Assumed Long. = $131^{\circ}51'$ West
 $\quad\quad\quad 59^{\circ}$ West

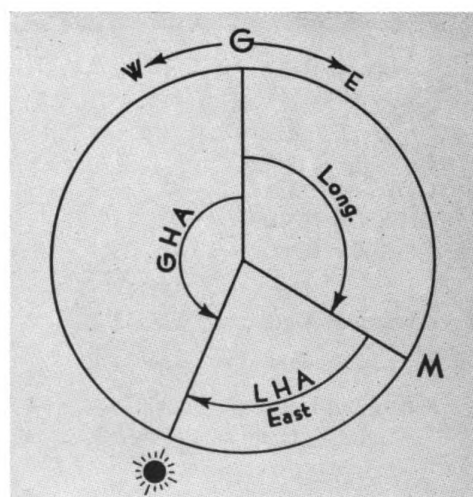


FIG. 117—LHA = GHA + EAST LONGITUDE

Example No. 4 (Figure 117)

DR Position = Latitude $31^{\circ}45'$ North
 Longitude $120^{\circ}50'$ East
 GHA = $158^{\circ}47'$
 Assumed Long. = $121^{\circ}13'$ East
 $\quad\quad\quad 280^{\circ}00'$
 from 360°
 LHA = 80° East

d. To continue procedure, assume latitude, nearest whole degree to DR position.

e. Enter proper volume of H.O. 214 and find assumed latitude. Note whether declination is *same name* as latitude or *contrary* to latitude. Find column of nearest whole or half degree of declination as shown at top of page (according to whether declination is same or contrary name to latitude), obtain values for altitude, Δd , and azimuth opposite the proper LHA found at side of page.

f. Name azimuth according to latitude and LHA.

g. Multiply Δd by number of minutes of declination difference and apply correction to

tabulated altitude in order to obtain computed altitude (H_C) for assumed position.

h. Compare H_C with H_O and mark altitude intercept *toward* or *away* depending on whether H_O is greater or less than H_C .

Caution—The most important factor in computing lines of position is accuracy. Therefore:

1. Write all work legibly.
2. Establish some such form as used in the following problems, and use it consistently.
3. Underline *assumed longitude*, *assumed latitude*, *azimuth*, and *intercept*, as these values are used to plot the line of position.
4. Name latitude and declination to insure use of the correct table.

SAMPLE PROBLEMS

The following problems illustrate the use of H.O. 214. In each example the navigator measured the altitude of the celestial body with a bubble octant while flying at an altitude of 5000 ft.

Example No. 1

May 1, 1943, GCT $18^h10^m12^s$, H_s sun $63^\circ32'$, I.C. $+1'$, DR latitude $30^\circ12'$ North, DR longitude $117^\circ23'$ West.

SUN

GCT $18^h10^m12^s$

GHA for $18^h10^m = 93^\circ14'$

+ for $00^m12^s = 03'$

GHA = $93^\circ17'$

Assumed Long. = $117^\circ17'W$

LHA = $24^\circ E$

Assumed Lat. = $30^\circ N$

Dec. = $14^\circ58'N$

Enter H.O. 214
(Same name)

LHA	$24^\circ N$
Lat.	$30^\circ N$
Dec.	$15^\circ N$

$H_s = 63^\circ32'$

I. C. = $+1'$

Ref. = 0

$H_O = 63^\circ33'$

$H_C = 63^\circ18.6'$

a = $14.4'$ toward

$\Delta d = .61$ (x $2' = 1.2'$)

H = $63^\circ19.8'$

Δd corr. = $-1.2'$

$H_C = 63^\circ18.6'$

Az = **N $118.9^\circ E$**

POSITION LINES

Example No. 2

May 1, 1943, GCT 05^h55^m00^s, H_S sun 24°30', I.C. -3', DR latitude 33°20' North, DR longitude 160°15' East.

SUN

GCT 05^h55^m00^s

GHA for 05^h50^m = 268°13'

+ for 05^m00^s = 1°15'

GHA = 269°28'

Assumed Long. = 160°32'E

430°

-360°

LHA = 70° W

Assumed Lat. = 33° N

Dec. = 14°49'N

Enter H.O. 214
(Same name)

LHA 70°W

Lat. 33°N

Dec. 15°N

H_S = 24°30'

I. C. = -3'

Ref. = -2'

H_O = 24°25'

H_C = 24°37.2'

a = 12.2' away

Δd = .50 (x 11' = 5.5')

H = 24°42.7'

Δd corr. = -5.5'

H_C = 24°37.2'

Az = N 87.7°W

Example No. 3

May 1, 1943, GCT 0^h40^m35^s, H_S sun 33°40', I.C. +5', DR latitude 34°36' North, DR longitude 132°00' West.

SUN

GCT 0^h40^m35^s

GHA for 0^h40^m = 190°42'

+ for 00^m35^s = 09'

GHA = 190°51'

Assumed Long. = 131°51'W

LHA = 59° W

Assumed Lat. = 35° N

Dec. = 14°45'N

Enter H.O. 214
(Same name)

LHA 59°W

Lat. 35°N

Dec. 15°N

H_S = 33°40'

I.C. = +5'

Ref. = -1'

H_O = 33°44'

H_C = 33°38.5'

a = 5.5' toward

Δd = .54 (x 15' = 8.1')

H = 33°46.6'

Δd corr. = -8.1'

H_C = 33°38.5'

Az = N 95.1°W

Example No. 4

May 1, 1943, GCT 22^h32^m10^s, H_S sun 16°12', I.C. -6', DR latitude 31°45' North, DR longitude 120°50' East.

SUN

GCT 22^h32^m10^s

GHA for 22^h32^m = 158°14'

+ for 02^m10^s = 33'

GHA = 158°47'

Assumed Long. = 121°13'E

280°00'

from 360°

LHA = 80° E

Assumed Lat. = 32° N

Dec. = 15°02'N

Enter H.O. 214
(Same name)

LHA 80°E

Lat. 32°N

Dec. 15°N

H_S = 16°12'

I. C. = -6'

Ref. = -3'

H_O = 16°03'

H_C = 16°14.4'

a = 11.4' away

Δd = .50 (x 2' = 1.0')

H = 16°13.4'

Δd corr. = +1.0'

H_C = 16°14.4'

Az = N 82.2°E

Example No. 5

May 1, 1943, GCT 21^h13^m10^s, H_s moon 35°02', I.C. = 0, DR latitude 30°15' South, DR longitude 120°30' West.

MOON

GCT 21^h13^m10^s

GHA for 21^h10^m = 166°20'
+ for 3^m10^s = 46'

GHA = 167°06'

Assumed Long. = 120°06'W

LHA = 47° W

Assumed Lat. = 30° S

Dec. = 0°18'N

Enter H.O. 214
(Contrary name)

LHA 47°W
Lat. 30°S
Dec. 0°N

H_s = 35°02'

I. C. = 0

Par. = +48'

Ref. = -1'

H_o = 35°49'

H_c = 36°00.9'

a = 11.9' away

Δd = .62 (x 18' = 11.2')

H = 36°12.1'

Δd corr. = -11.2'

H_c = 36°00.9'

Az = S 115.0°W

Example No. 6

May 1, 1943, GCT 11^h12^m20^s, H_s star Rigel 14°01', I. C. -2', DR latitude 32°20', South, DR longitude 130°20' East.

STAR RIGEL

GCT 11^h12^m20^s

GHA for 11^h10^m = 26°01'

+ for 2^m20^s = 35'

SHA Rigel = 282°03'

GHA = 308°39'

Assumed Long. = 130°21'E

439°00'

-360°00'

LHA = 79° W

Assumed Lat. = 32° S

Dec. = 8°16'S

Enter H.O. 214
(Same name)

LHA 79°W
Lat. 32°S
Dec. 8°S

H_s = 14°01'

I. C. = -2'

Ref. = -3'

H_o = 13°56'

H_c = 13°40.2'

a = 15.8' toward

Δd = .52 (x 16' = 8.3')

H = 13°31.9'

Δd corr. = +8.3'

H_c = 13°40.2'

Az = S 88.9°W

Example No. 7

May 1, 1943, GCT 17^h53^m10^s, H_s Mars 33°50', I. C. +4', DR latitude 34°50' South, DR longitude 160°30' East.

MARS

GCT 17^h53^m10^s

GHA for 17^h50^m = 143°25'

+ for 3^m10^s = 48'

GHA = 144°13'

Assumed Long. = 160°47'E

305°00'

from 360°

LHA = 55° E

Assumed Lat. = 35° S

Dec. = 8°57'S

Enter H.O. 214
(Same name)

LHA 55°E
Lat. 35°S
Dec. 9°S

H_s = 33°50'

I. C. = +4'

Ref. = -1'

H_o = 33°53'

H_c = 33°35.8'

a = 17.2' toward

Δd = .59 (x 3' = 1.8')

H = 33°37.6'

Δd corr. = -1.8'

H_c = 33°35.8'

Az = S 103.7°E

POSITION LINES

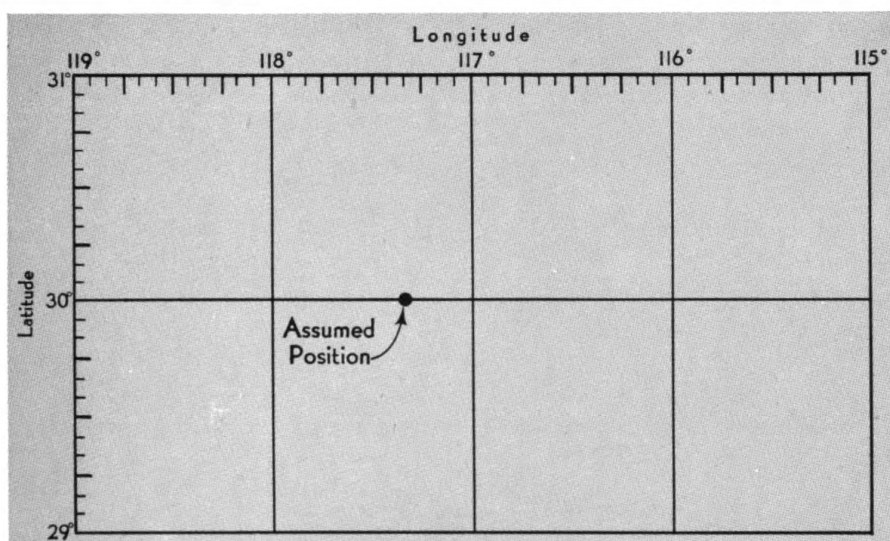


FIG. 118—PLOTting ASSUMED POSITION

PLOTting LINES OF POSITION

Step One — Plotting Assumed Position —

Since the solution of the astronomical triangle by altitude-azimuth tables locates the small segment of the circle of equal altitude (line of position) with relation to an assumed position, the first step in plotting the line of position (L.O.P.) is to locate on the chart the assumed

position used in the solution. Thus, if the assumed latitude = 30° North and the assumed longitude = $117^\circ 20'$ West, this position would be plotted as shown in Figure 118.

Step Two—Plotting Az—The second step is to plot the azimuth, or bearing of the body, through the assumed position as shown in Figure 119.

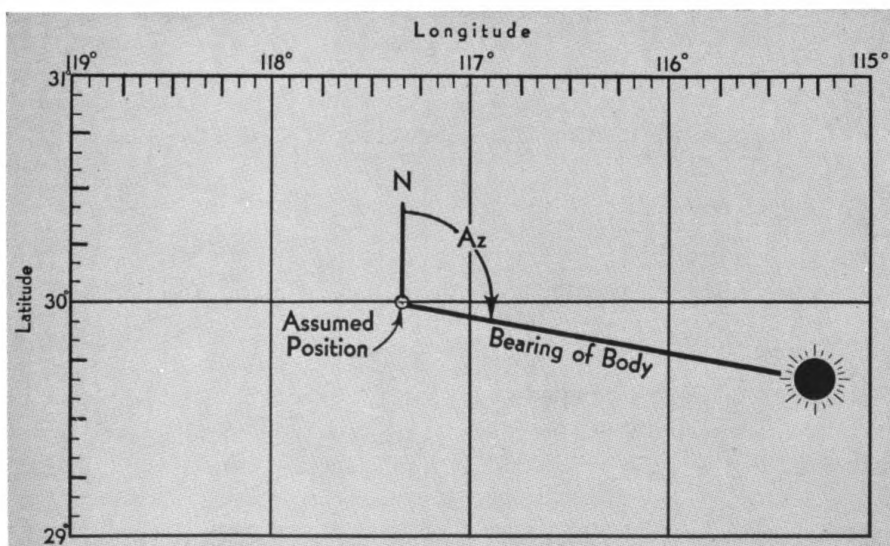


FIG. 119—PLOTting AZIMUTH THROUGH ASSUMED POSITION

Note: Bearings usually are measured from North (0°) through 360° . Such bearings are called true bearings (Z_n). However, pre-computed bearings of celestial bodies (as found in H.O. 214), are generally tabulated in terms

With this in mind, the navigator should pay particular attention to the azimuth direction, because plotting azimuth incorrectly is the most frequent error made by navigators in plotting a line of position. One of the most

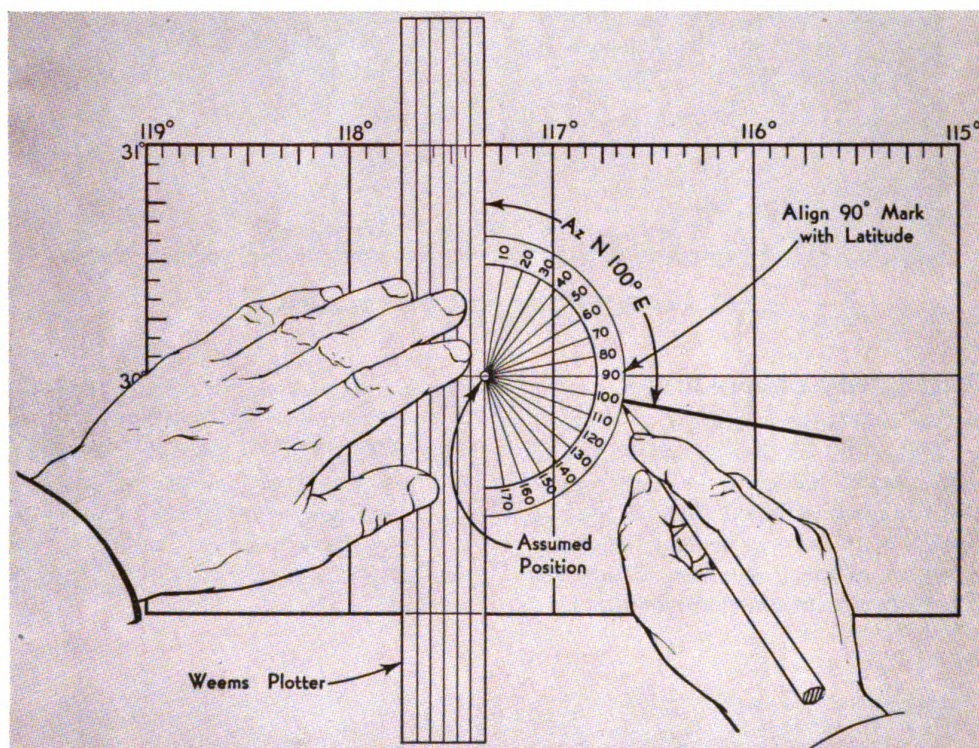


FIG. 120—PLOTING AZIMUTH WITH WEEMS PLOTTER

of azimuth, thus greatly simplifying the tables and eliminating much unnecessary compilation work. When azimuth is so employed, the bearing of a body is measured from the *elevated* pole towards the East or West through 180° . If the observer is in North latitude, the elevated pole is the North pole; if he is in South latitude, the elevated pole is the South pole. Azimuth is measured to the East if the body is rising, or to the West if it is setting.

practical methods which the author has yet found is to use a Weems plotter, which has an azimuth scale graduated to read from 0° to 180° .

As illustrated in Figure 120, the center of the Weems plotter is placed over the assumed position with the 0° mark pointed toward the elevated pole. Azimuth is then easily measured East or West according to the LHA of the body.

POSITION LINES

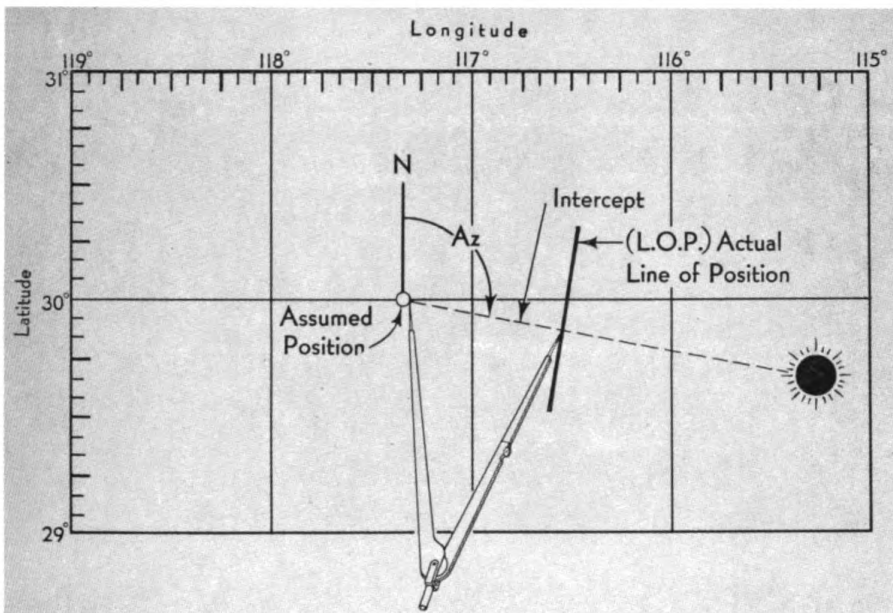


FIG. 121—MEASURING ALTITUDE INTERCEPT AND PLOTTING L.O.P.

Step Three—Plotting Actual Line of Position (Figure 121)—The third step is to plot the actual line of position. With the dividers set

at a distance equal to the altitude intercept, locate a point on the azimuth line a distance from the assumed position equal to altitude

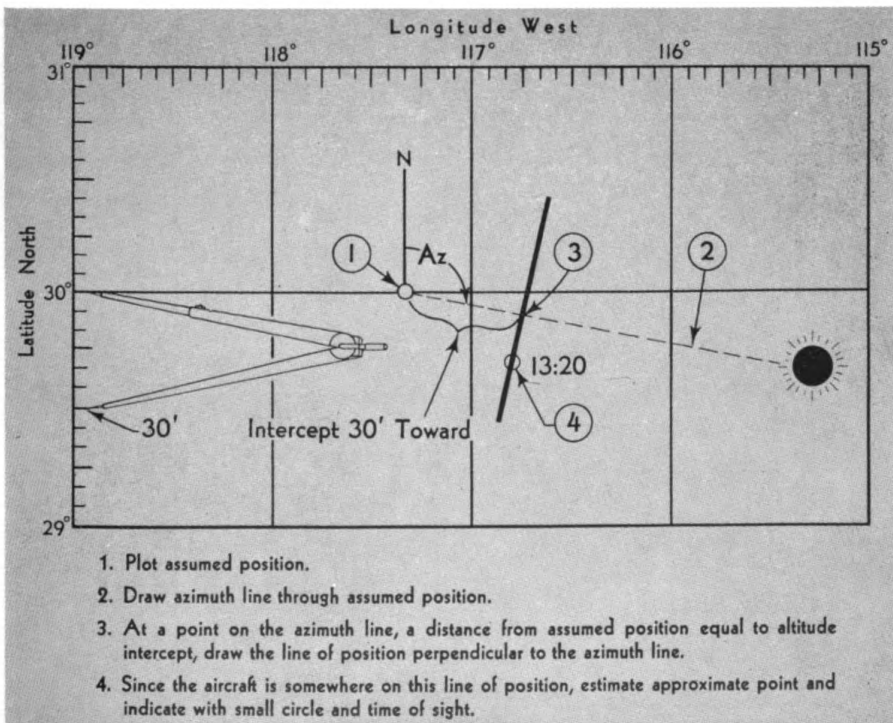


FIG. 122

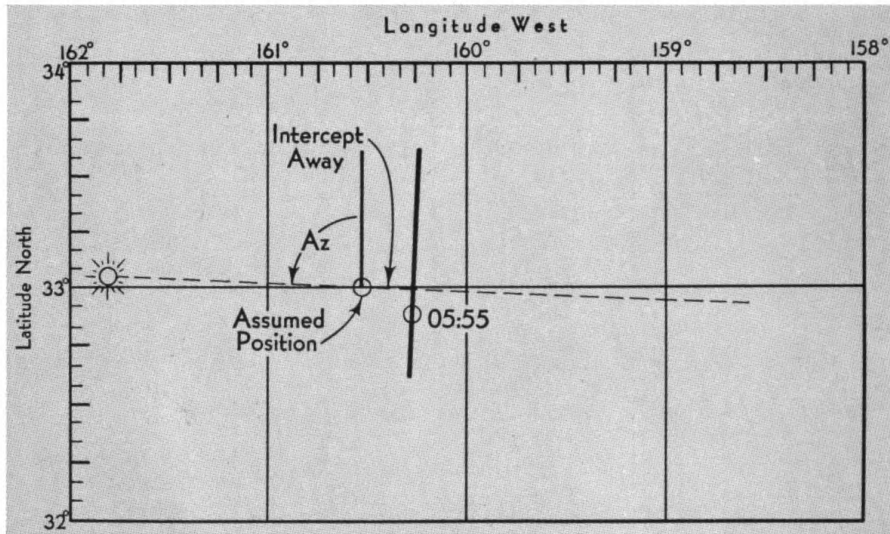


FIG. 123

intercept, and through this point erect the actual line of position perpendicular to the azimuth line.

Sample Problems—The following examples show how lines of position are plotted, using H. O. 214.

Example No. 1 Sun (Figure 122)

Time of sight.....GCT 13^h20^m
 Assumed position....Latitude 30° North
 Longitude 117°20' West
 AzimuthN 100°E
 Altitude Intercept....30 miles toward

Example No. 2 Sun (Figure 123)

Time of sight.....GCT 05^h55^m
 Assumed position....Latitude 33° North
 Longitude 160°32' West
 AzimuthN 87.7°W
 Altitude Intercept....13.2 miles away

Example No. 3 Moon (Figure 124)

Time of sight.....GCT 21^h13^m10^s
 Assumed position....Latitude 30° South
 Longitude 120°30' West
 AzimuthS 115°W
 Altitude Intercept....11.9 miles away

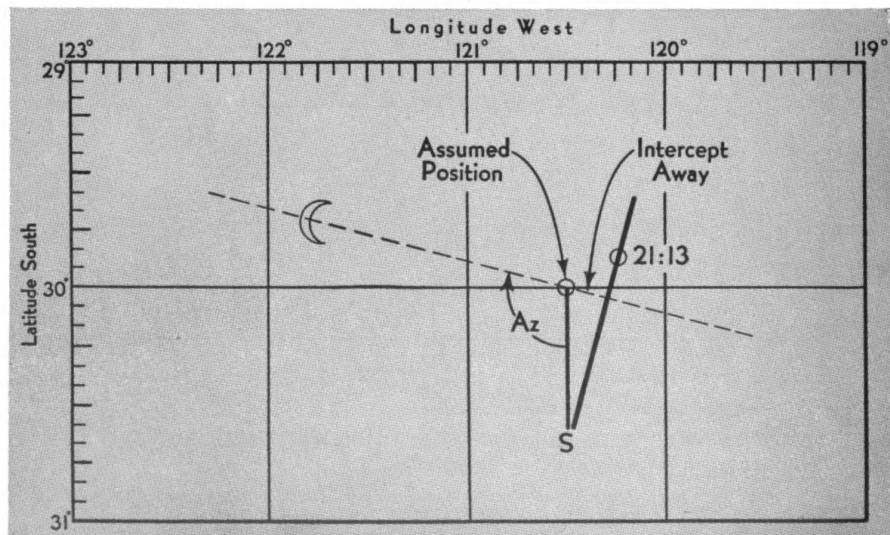


FIG. 124

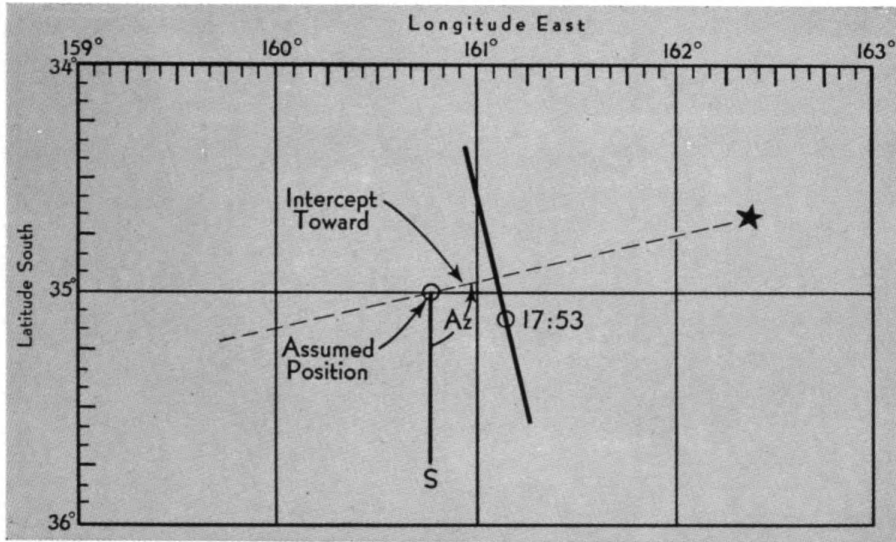


FIG. 125

Example No. 4 Planet (Figure 125)

Time of sight.....GCT 17^h53^m10^s

Assumed position....Latitude 35° South

Longitude 160°47' East

AzimuthS 103.7°E

Altitude Intercept...17.1 miles toward

SINGLE LINE OF POSITION

The single line of position, although it definitely locates the aircraft's position in one direction only, can be very useful to the navigator who understands its value and correctly interprets the information which it gives him.

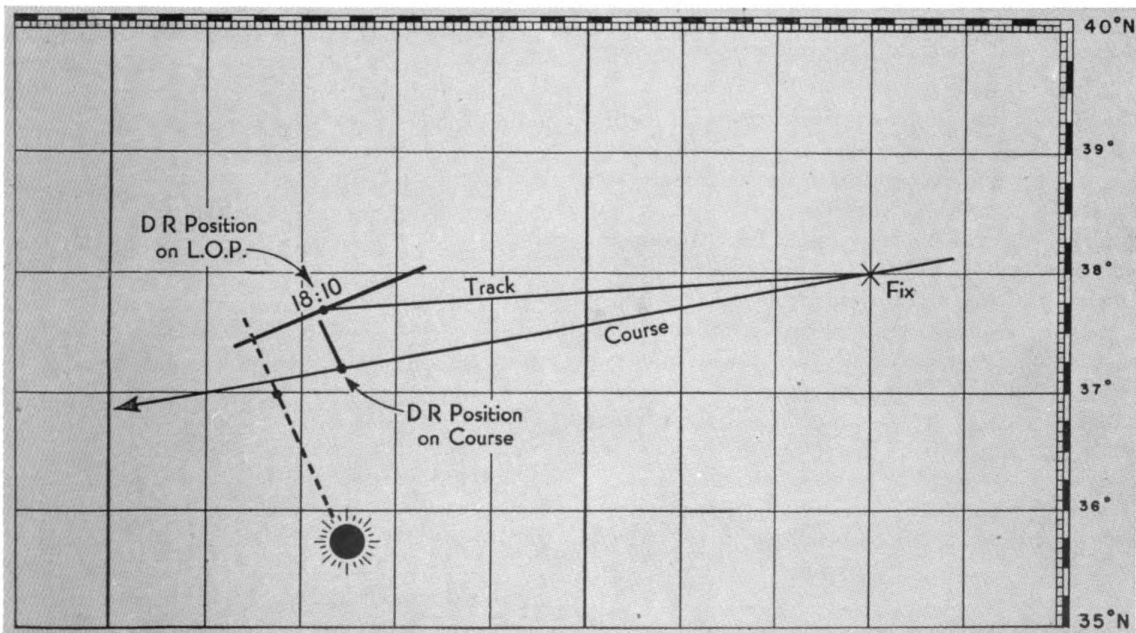


FIG. 126—SINGLE L.O.P. PARALLEL TO COURSE PROVES TRACK

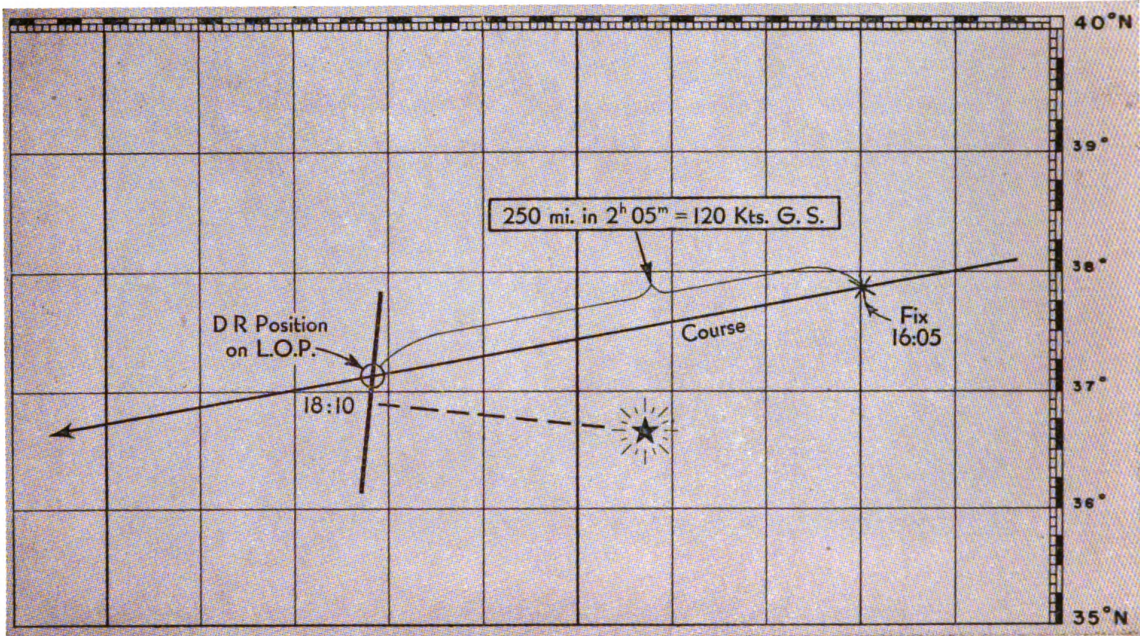


FIG. 127—SINGLE L.O.P. PERPENDICULAR TO COURSE PROVES GROUND SPEED

Determining Track—If the single line of position is parallel to the aircraft's course, it will indicate the amount of drift and the track being made good. (Figure 126)

Determining Ground Speed—If the single line of position is perpendicular to the aircraft's course, it will indicate the ground speed being made good. (Figure 127)

Advancing Lines of Position—The navigator, having plotted a line of position for the instant of observation, quite often finds it advisable to advance or retard the line to a different time. This is possible because, since the aircraft is known to be somewhere on the line of position at the instant of observation, it may be assumed that all points on the line of position are moving as the aircraft moves and at a common rate along the same course, much as would a row of aircraft whose wings were joined together. Therefore, if the line of position is advanced or retarded *along the course* to a dead reckoning position, all the points of the original line of position move in the same direction, and through the same distance. (Figure 128)

Hence to advance (or retard) a line of position to a different time, move the point where

it intersects the course line along the course a distance equal to the minutes of ground speed flown, and through this advanced (or retarded) point draw a new line of position parallel to the original line. (Figure 129)

The accuracy of the advanced line of position depends on how accurately the ground speed and track are known. For lines of position advanced or retarded from three to twenty minutes, errors involved are negligible, and the new line may be considered as accurate as the original line of position. However, if the line is advanced more than one hour it may contain appreciable error due to the fact that the dead reckoning ground speed is based upon an estimated wind. Hence, such lines should be used with appreciation of all possible errors of the dead reckoning course and ground speed.

POSITION FIXES

Fix by Two Sun Lines—Often the aerial navigator must depend solely on single position lines (such as sun lines) in order to fix his position. In this procedure, the navigator shoots the sun and obtains single position lines at elapsed intervals which should not be more than one hour apart. During this elapsed interval the sun's azimuth is changing continuously,

POSITION LINES

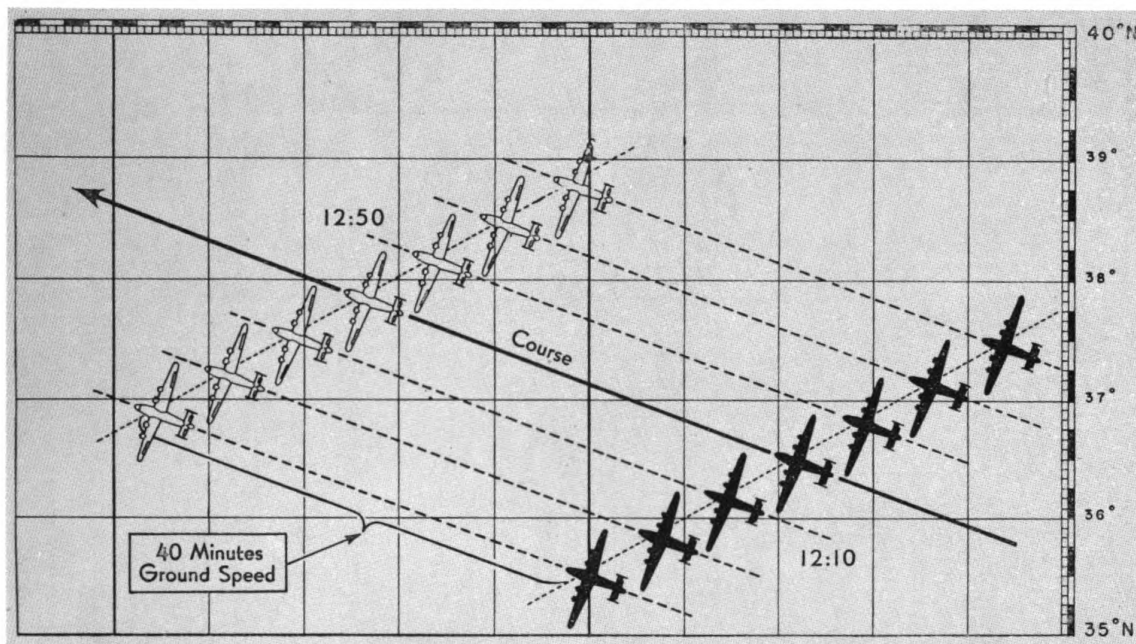


FIG. 128—ALL POINTS ON L.O.P. MOVE PARALLEL TO COURSE

which likewise causes the direction of the position line to change. Therefore, by advancing the previous position line to the latest position line, a cross of lines will be obtained which will thus fix the aircraft's position. (Figure 130)

For such a fix to be of any value, the navigator must carefully analyze all possible errors in the track and ground speed used to advance the line of position, because the azimuth change usually is very slight. This type fix is most reliable when the azimuth change is rapid, such

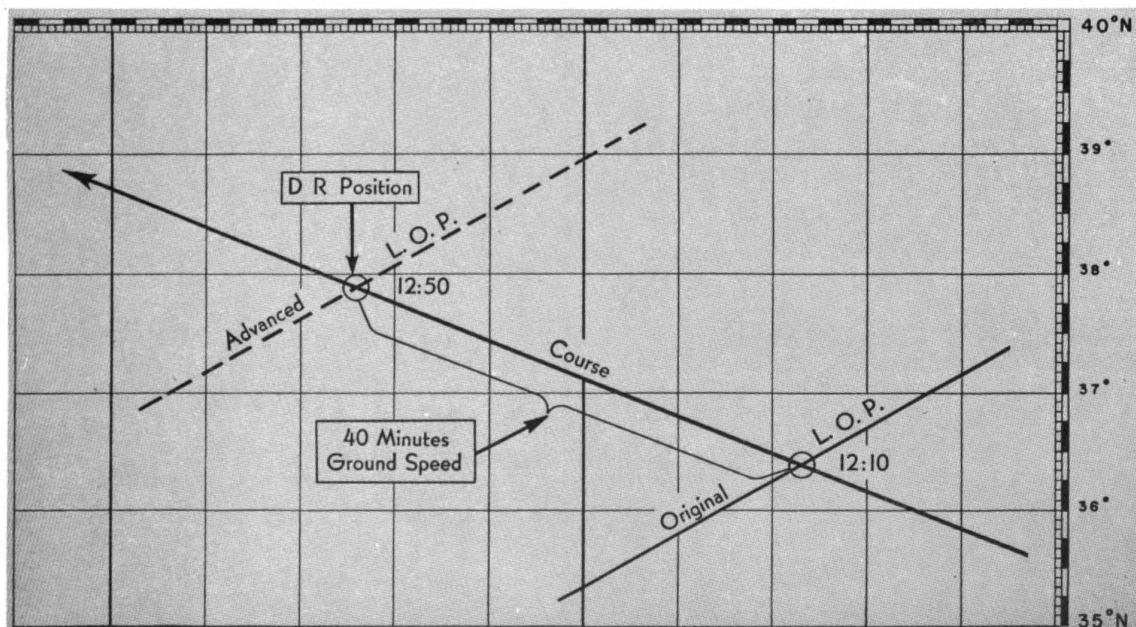


FIG. 129—ADVANCED L.O.P. DRAWN PARALLEL TO ORIGINAL L.O.P.

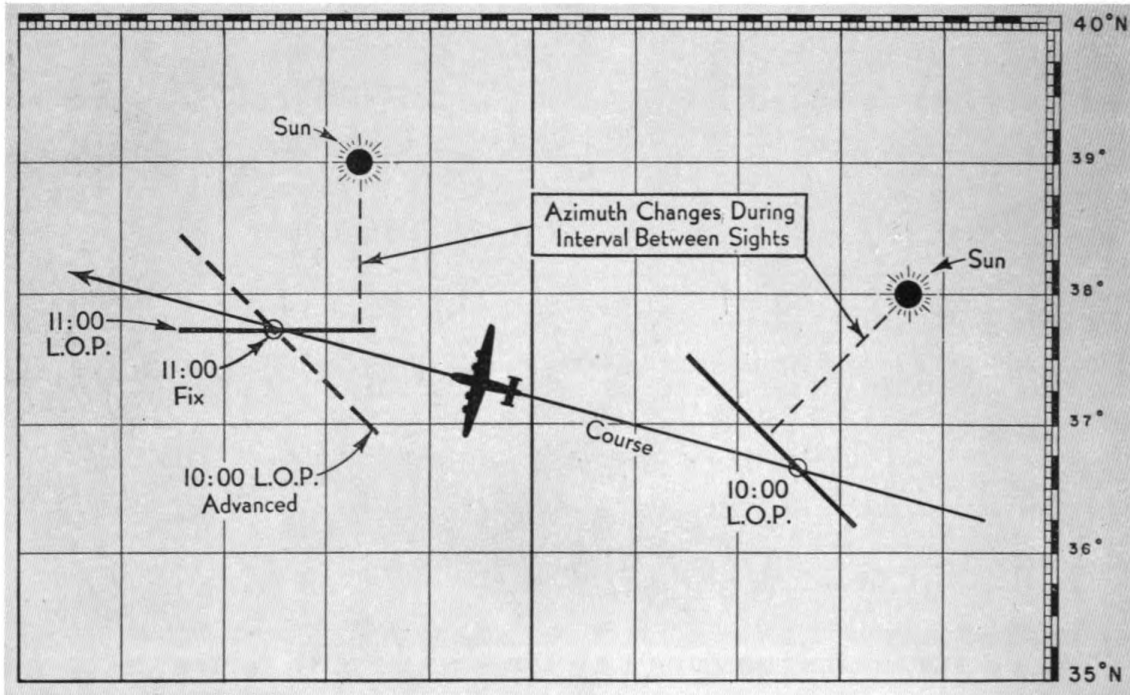


FIG. 130—RUNNING FIX BY TWO SUN LINES

as when the body is near the observer's celestial meridian. When this occurs, the body may be observed at shorter intervals and the resultant angle of "cut" of position lines will be greater.

Fix by Any Two Position Lines — Any bearing, such as a visual bearing of a mountain peak or a radio bearing, also provides a position line since it definitely locates the aircraft's po-

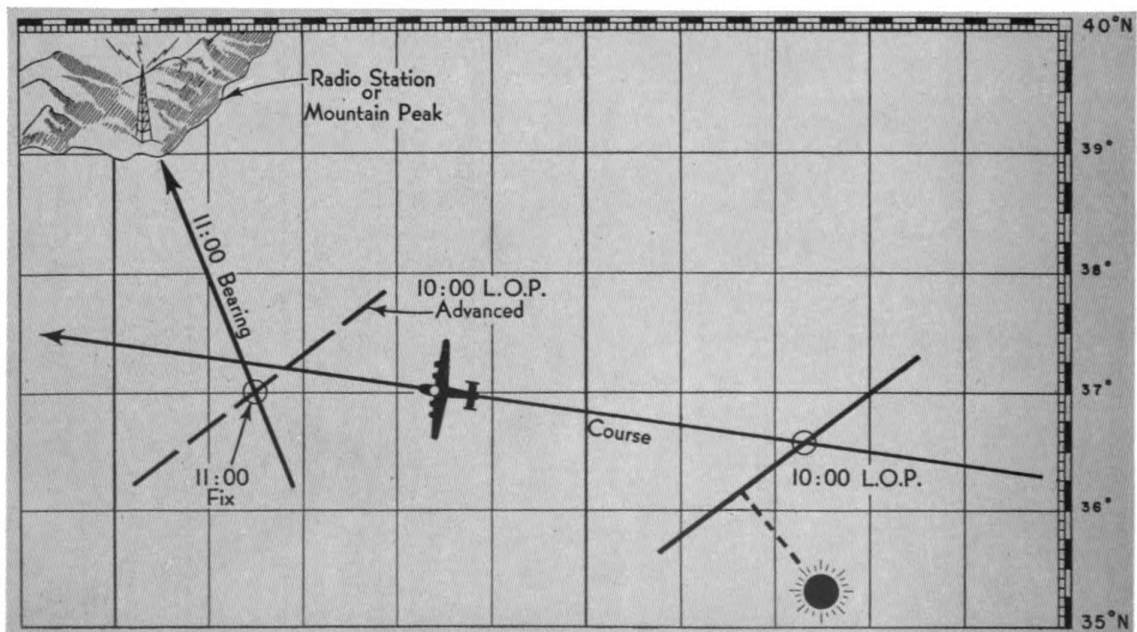


FIG. 131—RUNNING FIX BY ANY TWO POSITION LINES

POSITION LINES

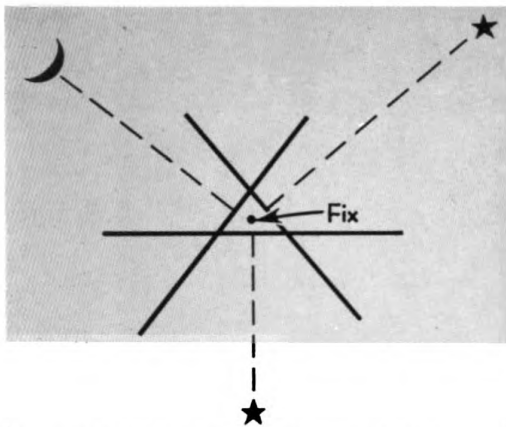


FIG. 132—POSITION TRIANGLE

sition in one direction (i.e. somewhere on the bearing line). Therefore, bearings may be advanced to celestial position lines, or celestial position lines may be advanced to bearings in order to obtain a fix. (Figure 131)

Three Star Fixes—The three star fix is by far the most desirable and most accurate method of celestial navigation.

To obtain a fix, three celestial bodies whose

azimuths differ (by at least 60° if possible) are observed, and the resultant position lines, when plotted on the chart, should intersect at a point or small triangle which definitely locates the aircraft's position. (Figure 132).

Since it is physically impossible to take three sights simultaneously, the observations are taken with as little time interval between as possible. The lines of position are then plotted, and advanced or retarded to a common time. (Figure 133)

In actual practice, as illustrated in Figure 134, the last line of position (11:00) is usually plotted on the chart first. Then, since only the common time fix (11:00) is desired, the earlier lines of position (10:40 and 10:50) are only lightly indicated at their origin, to facilitate their advance to the common time (11:00).

Track and Ground Speed Between Fixes—

A fix, whether it is a celestial fix or a visual fix, determines the location of the aircraft at the time of the fix. Therefore, when fixes are taken at intervals, the track, ground speed, drift, and wind between the fixes may be easily and accurately determined.

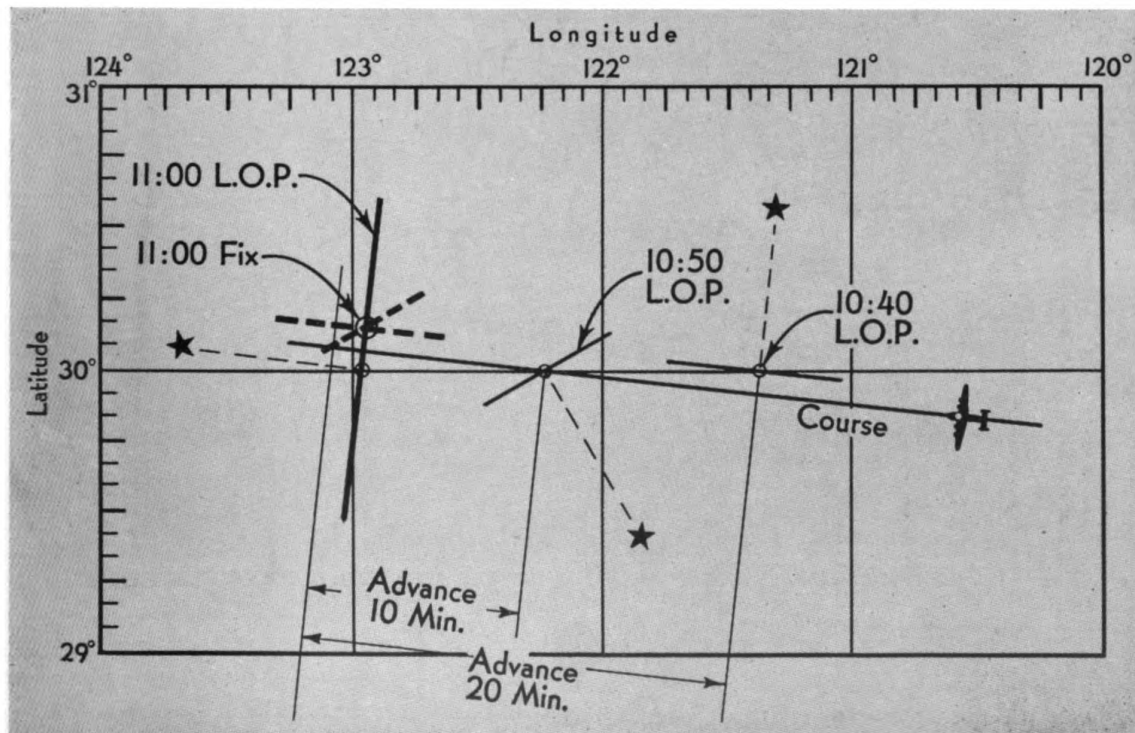


FIG. 133—THREE STAR RUNNING FIX

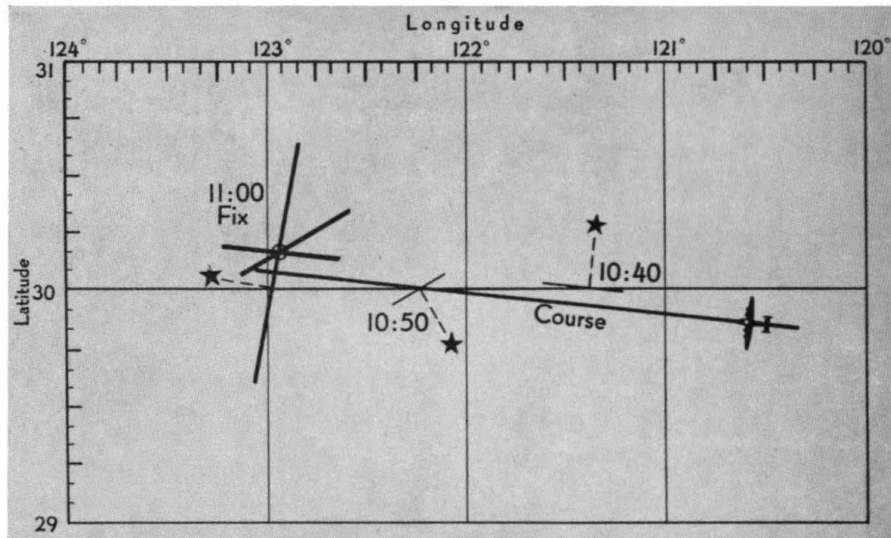


FIG. 134—PRACTICAL PLOTTING OF THREE STAR FIX

The track is determined by a straight line connecting two successive fixes. The ground speed is found by dividing the distance between fixes by the time interval flown.

The wind is then the resultant vector component between the track and ground speed

and the true heading and true air speed. (Figure 135)

In actual practice, the track and distance flown between fixes are measured on the chart.

The ground speed, drift, and wind are usually solved on the computer.

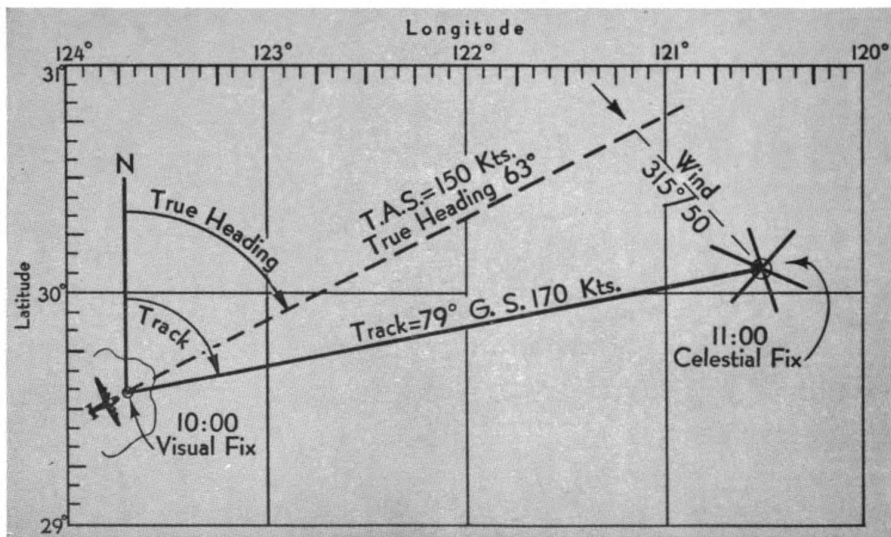


FIG. 135—DETERMINING TRACK, GROUND SPEED AND WIND BETWEEN FIXES



PROBLEM WORK

No. 26—Plot Problem Work No. 25.

POSITION LINES

PROBLEM WORK NO. 23

ASSUMED POSITION—LHA—DECLINATION

(Find assumed position, LHA and declination.)

No.	DATE	DR POSITION		GCT	CELESTIAL BODY	ASSUMED POSITION		LHA	Dec.
		Lat.	Long.			Lat.	Long.		
1	5- 1-43	34°10'N	110°32'E	22:22:00	ARCTURUS				
2	5- 5-43	23°15'S	172°14'W	22:14:00	SUN				
3	5-10-43	14°21'N	117°42'W	20:19:00	JUPITER				
4	5-15-43	22°18'N	178°31'E	02:19:32	SUN				
5	5-20-43	11°01'S	124°19'E	19:37:07	RIGIL KENT.				
6	5-30-43	14°36'N	117°51'E	12:38:00	PROCYON				
7	5- 1-43	32°51'N	168°41'W	00:31:14	SUN				
8	5- 5-43	18°13'S	104°32'E	10:52:00	VENUS				
9	5-10-43	15°14'S	168°14'E	12:15:00	NUNKI				
10	5-15-43	32°08'S	114°39'W	23:19:00	MOON				
11	5-20-43	18°54'N	127°27'W	03:16:00	SPICA				
12	5-30-43	7°15'S	92°00'W	00:10:00	ALDEBARAN				
13	5- 1-43	19°43'N	116°37'W	05:36:00	BETELGEUX				
14	5- 5-43	4°04'N	152°19'W	21:50:00	RIGEL				
15	5-10-43	16°31'S	86°49'E	05:16:00	CANOPUS				
16	5-15-43	31°19'S	178°19'W	02:16:20	SUN				
17	5-20-43	20°14'N	117°22'W	02:50:00	BELLATRIX				
18	5-30-43	17°51'S	172°15'E	23:18:45	SUN				
19	5- 1-43	14°19'S	179°12'W	03:50:00	RIGIL KENT				
20	5-10-43	23°31'N	162°14'W	06:53:19	PROCYON				

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 24

ALTITUDE AND AZIMUTH (H. O. 214)

[Find altitude (H_c) and azimuth. Use volume IV, H. O. 214.]

No.	LATITUDE	DECLINATION	LHA	H_c	AZIMUTH
1	33°N	13°48'N	65°W		
2	36°N	23°06'N	48°E		
3	39°N	11°20'S	61°E		
4	38°N	22°40'N	35°W		
5	34°N	16°24'N	41°W		
6	38°S	57°32'S	29°W		
7	31°N	8°43'N	51°E		
8	35°N	26°18'S	40°W		
9	31°N	19°29'N	30°E		
10	38°N	7°24'N	38°W		
11	31°N	45°57'N	26°E		
12	32°N	45°05'N	42°W		
13	31°S	29°56'S	19°E		
14	36°S	28°10'N	29°W		
15	35°N	5°22'N	43°E		
16	32°S	12°15'N	39°E		
17	33°S	8°16'S	40°W		
18	31°N	16°38'S	27°E		
19	39°N	10°52'S	46°W		
20	32°N	38°44'N	28°E		

POSITION LINES

PROBLEM WORK NO. 25

ASSUMED POSITION—INTERCEPT—AZIMUTH

(Find assumed position, altitude intercept and azimuth by H. O. 214.)

No.	CELESTIAL BODY	DATE	GCT	H ₀	DR POSITION		ASSUMED POSITION	Int.	Az.
					Lat.	Long.			
1	NUNKI	5-1-43	12:18:30	19°40'	35°02'S	156°03'E			
2	CANOPUS	5-5-43	05:44:13	71°49'	35°10'S	153°29'E			
3	MOON	5-10-43	03:01:35	18°31'	35°25'S	154°15'E			
4	SUN	5-15-43	02:58:00	29°58'	29°50'N	160°00'W			
5	JUPITER	5-20-43	23:58:36	58°02'	30°01'N	158°15'W			
6	ANTARES	5-20-43	10:19:10	31°00'	30°25'N	161°12'W			
7	VENUS	5-10-43	06:11:18	65°42'	39°20'N	152°28'E			
8	CAPELLA	5-1-43	04:34:35	83°04'	38°45'N	151°25'E			
9	SUN	5-15-43	02:15:58	69°12'	39°12'N	151°12'E			
10	SUN	5-5-43	15:38:38	17°17'	32°40'N	135°45'W			
11	MOON	5-20-43	11:58:31	30°09'	33°05'N	136°00'W			
12	MARS	5-15-43	20:32:29	27°30'	33°15'N	136°12'W			
13	JUPITER	5-1-43	05:05:05	20°43'	36°25'S	145°20'W			
14	MOON	5-10-43	02:24:00	34°58'	36°05'S	146°12'W			
15	SUN	5-5-43	19:35:39	30°00'	35°45'S	146°50'W			
16	MOON	5-20-43	11:06:00	45°30'	31°40'S	155°20'E			
17	PROCYON	5-15-43	08:19:39	36°38'	31°58'S	157°01'E			
18	SUN	5-1-43	01:08:00	42°39'	32°12'S	156°18'E			
19	MOON	5-10-43	03:32:56	25°50'	37°45'S	133°20'W			
20	JUPITER	5-25-43	00:56:38	29°59'	38°01'S	135°02'W			

PROBLEM WORK NO. 27
ADVANCING SINGLE LINES OF POSITION

(Plot single line of position and advance required number of minutes or to GCT indicated.)

No.	DATE	GCT	CELESTIAL BODY	Ho	DR POSITION		Az	Intercept	Ground Speed	Track	ADVANCE L.O.P	
					Lat.	Long.					Time	N.M.
1	5- 1-43	06:00:00	SUN	49°35'	30°44'N	49°09'E			180	270°	5 Min.	
2	5- 5-43	13:10:20	SUN	32°17'	35°14'S	11°49'E			99	135°	20 Min.	
3	5-10-43	02:32:06	SUN	23°15'	37°52'S	178°40'W			177	32°	45 Min.	
4	5-15-43	18:45:26	SUN	67°30'	33°09'N	82°03'W			110	60°	45 Min.	
5	5-20-43	15:14:05	SUN	34°50'	35°17'S	56°32'W			198	253°	15:30:00	
6	5-30-43	09:21:17	SUN	30°05'	34°57'S	17°18'E			164	350°	09:30:00	
7	5- 1-43	08:27:13	SUN	45°01'	39°05'N	8°27'E			120	179°	08:30:00	
8	5- 1-43	03:30:00	MOON	26°03'	33°15'S	29°52'E			107	99°	10 Min.	
9	5- 5-43	21:14:30	MOON	63°01'	32°49'N	98°10'W			154	177°	21:30:00	
10	5-10-43	02:57:05	MOON	23°32'	37°14'S	117°05'W			114	240°	03:00:00	
11	5-20-43	23:45:01	MOON	28°18'	35°51'N	10°30'W			202	38°	24:00:00	
12	5-30-43	03:05:50	MARS	33°01'	34°51'S	117°10'E			107	260°	03:10:00	
13	5- 1-43	02:00:47	VENUS	42°16'	35°21'N	117°57'W			89	42°	02:10:00	
14	5- 5-43	10:10:00	JUPITER	41°03'	33°12'N	150°40'E			180	355°	10:30:00	
15	5-10-43	09:31:10	MARS	27°01'	37°05'N	70°15'W			160	180°	09:40:00	
16	5-20-43	15:36:40	POLARIS	36°05'	35°50'N	152°21'W			119	180°	15:40:00	
17	5-30-43	18:23:07	ARCTURUS	54°01'	36°05'N	14°07'E			148	90°	18:30:00	
18	5- 1-43	08:42:50	VEGA	54°09'	38°44'N	117°05'W			89	270°	08:50:00	
19	5- 5-43	21:00:58	ALTAIR	26°30'	32°57'S	169°40'E			107	106°	21:06:00	
20	5-10-43	15:16:00	DUBHE	49°30'	35°07'N	117°50'E			126	136°	15:30:00	

POSITION LINES

PROBLEM WORK NO. 28

THREE STAR FIXES

GROUND SPEED 180 knots—TRACK 242°

(Find line of position for each star. Advance lines of position to obtain "FIX" for time of third star in each group.)

No.	DATE	GCT	CELESTIAL BODY	H ₀	DR POSITION	FIX
1	5- 1-43	01:04:20	VEGA	46°39'	LAT. 32°10'N LONG. 8°41'W	
		01:10:00	DENEK	26°28'		
		01:15:12	ALPHECCA	83°04'		
2	5- 5-43	04:13:41	RIGEL	62°31'	LAT. 32°14'S LONG. 137°14'E	
		04:19:30	CANOPUS	59°10'		
		04:26:30	ALDEBARAN	41°25'		
3	5-10-43	15:40:00	CAPELLA	56°39'	LAT. 35°09'N LONG. 17°14'E	
		15:50:00	POLARIS	35°15'		
		16:00:00	ALDEBARAN	36°39'		
4	5-15-43	21:50:00	RUCHBAH	61°34'	LAT. 31°17'N LONG. 180°00'	
		21:52:00	MARS	44°47'		
		21:55:00	ENIF	33°28'		
5	5-20-43	09:51:40	ARCTURUS	69°31'	LAT. 39°21'N LONG. 175°43'W	
		09:58:35	DUBHE	54°17'		
		10:07:37	REGULUS	29°02'		
6	5- 5-43	17:01:59	DENEK	50°24'	LAT. 32°19'N LONG. 120°05'W	
		17:03:06	FOMALHAUT	25°58'		
		17:05:02	ALTAIR	27°47'		
7	5-20-43	14:31:48	DUBHE	47°08'	LAT. 35°51'N LONG. 125°11'E	
		14:37:00	SPICA	38°37'		
		14:40:55	DENEKOLA	43°25.5'		
8	5-10-43	04:11:41	CAPELLA	33°30'	LAT. 31°17'N LONG. 140°02'W	
		04:16:50	POLLUX	58°14'		
		04:19:50	PROCYON	45°27'		

PROBLEM WORK NO. 29

TRACK AND GROUND SPEED BETWEEN FIXES

C.A.S. = 157 knots, ALTITUDE = 7000', TEMPERATURE = +10°C, T.A.S. =

(Plot star sights on a small scale chart, I.C. = 0°. Advance or retard lines of position to obtain fix for time of third sight listed in each group.)

FIND TRACK AND GROUND SPEED MADE GOOD FROM PREVIOUS FIX.

CALCULATE DRIFT AND WIND BETWEEN FIXES.

ALTER TRUE COURSE AS INDICATED.

ESTIMATE TRUE HEADING USING PREVIOUS WIND, GIVEN OR CALCULATED.

No.	GCT	TYPE OF FIX	POSITION		TRACK	GROUND SPEED	DRIFT	WIND	TRUE HEADING
			Lat.	Long.					
1	TRUE COURSE 242°								
	07:05	Visual	37°50'N	122°29'W	242°		0°	0°	242°
2	TRUE COURSE 242° Star sights taken:								
	KOCHAB GCT 08:32:00 H _s 50°34'								
ALPHECCA GCT 08:36:00 H _s 76°31'									
VEGA GCT 08:40:00 H _s 44°30'									
08:40	Celestial								
3	TRUE COURSE 247° Star sights taken:								
	ANTARES GCT 09:27:00 H _s 28°15'								
RASALAGUE GCT 09:34:00 H _s 54°37'									
ARCTURUS GCT 09:30:00 H _s 69°00'									
09:30	Celestial								
4	TRUE COURSE 249° Star sights taken:								
	POLARIS GCT 10:13:00 H _s 32°46'								
ALTAIR GCT 10:16:00 H _s 31°24'									
DENEK GCT 10:20:00 H _s 36°14'									
10:20	Celestial								
5	TRUE COURSE 232° Star sights taken:								
	SABIK GCT 11:32:00 H _s 42°44'								
ALIOTH GCT 11:38:00 H _s 40°27'									
SPICA GCT 11:35:00 H _s 22°35'									
11:35	Celestial								

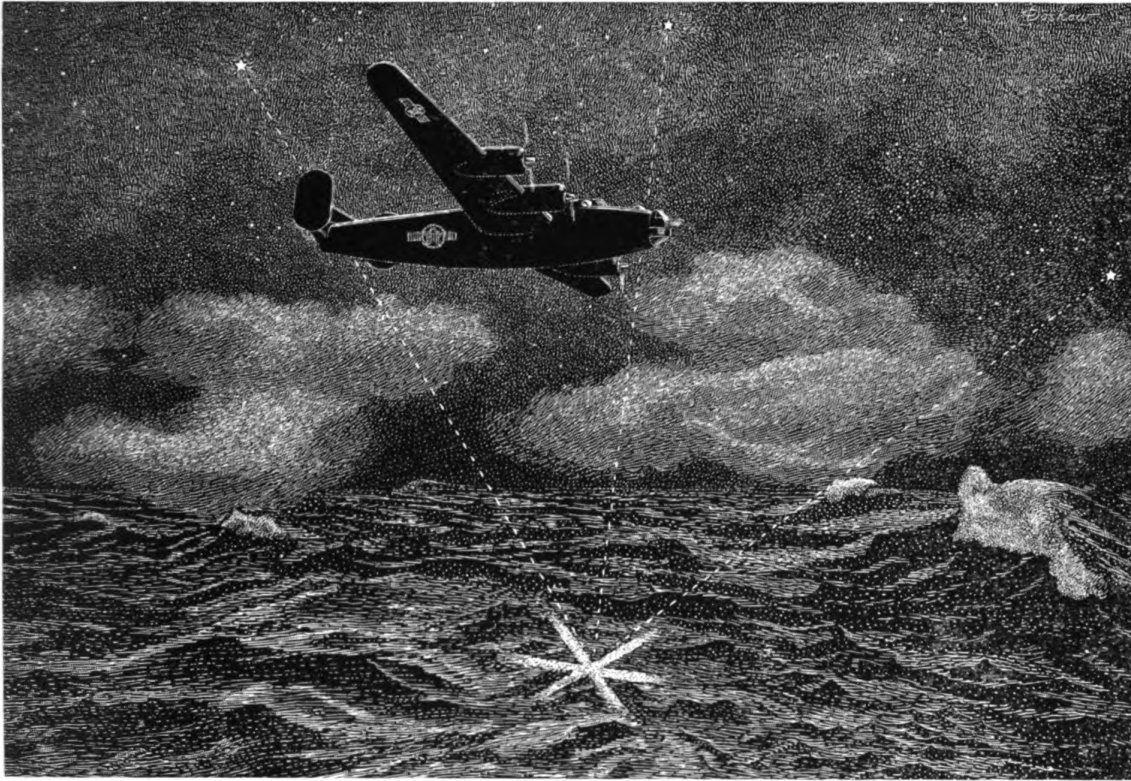
POSITION LINES

CELESTIAL REVIEW EXAMINATION NO. 2

1. Draw Celestial Sphere. Show all parts you know.
2. Define:

(a) Vertical Circle	(f) Altitude
(b) Hour Circle	(g) Local Hour Angle
(c) Azimuth	(h) Ecliptic
(d) Equinoctial	(i) Sidereal Hour Angle
(e) Declination	(j) Geographical Position
3. Show the following by hour angle diagrams:

(a) GHA Sun 120° Long. 40° East LHA Sun	(b) GHA Aries 60° SHA Star 180° Long. 80° East GHA Star LHA Star
---	---
4. Solve:
 - (a) On January 2nd at Long. 120° West, the LCT is 18:31. What is the time and date in Greenwich?
 - (b) On July 15, when GCT is 22:16, what is the Local Civil Time and date at Long. 118° East?
5. Honolulu ZD = $+9\frac{1}{2}$
Canton ZD = $+10\frac{1}{2}$
Fiji ZD = -12
 - (a) An aircraft leaves Honolulu at ZT 10:00 A. M., August 3rd, arriving at Canton 9 hours and 30 minutes later. What is local time and date of arrival?
 - (b) An aircraft leaves Canton at ZT 12:00 noon, January 5th, arriving at Fiji 6 hours later. What is the local zone time and date of arrival?
6. May 1, 1943, GCT $11^h20^m22^s$: Find GHA and declination of Sun, Moon and Sirius.
7. The following altitudes were obtained with a bubble sextant of I.C.— $6'$, at height of 8000 ft., May 10, 1943:
Sun $30^\circ20'$ Venus $60^\circ15'$ Altair $20^\circ12'$ Moon $28^\circ35'$
What are the true altitudes?
8. On May 20, 1943, GCT $11^h20^m20^s$ in longitude $145^\circ18'$ West, the altitude of Polaris (using a bubble sextant) was found to be $32^\circ15'$. If the height of the plane was 5000 ft. and the I.C. 0° , what was the latitude of the observer?
9. At DR latitude $30^\circ15'$ North, longitude $70^\circ25'$ West, on May 30, 1943, at a height of 7000 ft., I.C. 0° , and GCT $03^h16^m07^s$, the observed altitude of Antares was $30^\circ07'$. What are the altitude intercept and azimuth?
10. (a) Describe the process of forming the bubble in a Pioneer octant.
(b) In what relation are the bubble and the body for accurate sights? For inaccurate sights?



☆ 10 ☆

STARS AND THE WEATHER

AWORKING knowledge of the stars is easy to acquire, yet few people make an effort to become familiar with them. The fact that the stars, to the average observer, appear so countless and complex in their arrangement probably accounts for the general ignorance about stars and constellations. As a result, most people view the night sky with the same awe and perplexity with which they might view some complicated maze.

It is true that the thousands of stars which may be seen at a single glance do appear to be a complicated maze. However, on examining the sky more closely, the observer will note that a

few stars—and only a few—are relatively bright, and because of their unusual brilliance they appear to stand out prominently from all the other stars. Only these few bright stars are used in navigation. Therefore, it is very important that the aerial navigator be able to recognize them quickly, because they are literally the tools of his trade.

The astronomer, working in behalf of the navigator, has calculated and tabulated in the Air Almanac the positions in the celestial sphere of the 55 brightest stars and five planets most suitable for observation, so that the navigator may compute from these 60 known celestial positions the location of his aircraft. Of these 60 stars and

planets which are tabulated in the Air Almanac—an infinitely small total compared to the innumerable stars seen at night—the author has used not more than 34 for navigation during his last 500,000 miles of travel.

IDENTIFICATION BY GROUPS

Knowing that only a few of the celestial bodies are needed for navigational purposes, the first step in learning these few is to obtain some good chart of the sky, such as that shown in Figure 136, taken from the Air Almanac. In gaining this knowledge, as in learning the parts of a complicated airplane structure, one may divide the stars into groups, or individual "sub-assemblies of the sky." For a start, Orion's Belt and Sword and the bright stars close by may be considered.

First Group—"Orion's Belt and Sword" (See Group Chart, Figure 137)—The constellation of Orion is easily recognized as a perfectly shaped little dipper. The three brightest stars, which usually are drawn as the belt in Orion, form the bottom of the dipper. This group, definitely recognized in the sky, may be used as a starting point for identifying other nearby stars.

Notice, for instance, that seven bright stars appear to form a circle around Orion; they are Rigel, Sirius, Procyon, Pollux, Castor, Capella, and Aldebaran. Try to learn these stars in this order, and then study each star to learn its individual characteristics.

1. *RIGEL* is just in front of the open end of the dipper.

2. *SIRIUS* is the brightest star in the sky and is located between Rigel and Procyon.

3. *PROCYON* is a very bright star. Nearby is a much dimmer, but easily seen sister star. Also, Procyon is located between Sirius and the sister stars, Pollux and Castor.

4. and 5. *POLLUX* (4) and *CASTOR* (5) are known as sister stars because they are of nearly equal brightness and very close together. Pollux, the brighter, is nearer Procyon.

6. *CAPELLA* is nearly as bright a star as Sirius. It has three little stars nearby in the shape of a triangle. Also, at a little greater distance are four other stars which, together with Capella, form a pentagon.

7. *ALDEBARAN* is a bright, reddish-col-

ored star which forms, with four stars nearby, a perfect letter *A*.

Inside this group of stars, and near the bottom of the dipper, are two other very bright stars, Betelgeux and Bellatrix. Betelgeux is the brighter and has a reddish color.

After these stars have been learned and can be easily located in the sky, the observer will be able to find and name many other stars by their positions relative to this group.

For instance, assume that a very bright star is visible and its name is desired. With relation to known stars, it lies in a line with Procyon and Sirius. It appears to be a little farther away from Sirius than is Procyon, and not quite in a direct line but, rather, slightly offset in a direction away from Orion. Looking at the star chart (Figure 136) the name of this star is determined to be Canopus.

CANOPUS is the second brightest star in the sky.

This same procedure may be used to locate a star in the sky by use of the star chart (Figure 136). Suppose it is desired to find Regulus in the sky. Using the stars already known, the star chart indicates that Regulus is in direct line with Procyon and the three stars forming the belt of Orion.

REGULUS also is distinguished by the fact that it appears as the brightest star in the handle of a small sickle, formed by four dimmer stars nearby.

Note: In describing these stars, the author is endeavoring to record the exact line of reasoning followed when trying to identify the stars in flight. On the charts, and especially the group charts, they appear easy to find, but unless they are definitely located with relation to others known, they may prove difficult to identify.

Second Group—"Big Dipper and Associated Stars" (See group chart, Figure 138)—The Big Dipper is perhaps the easiest constellation to recognize. It is formed by seven stars: Alkaid, Mizar, Alioth, Megrez, Phecda, Merak, and Dubhe. Of these seven, Alkaid, Mizar, and Alioth in the handle, and Dubhe at the end of the cup are the best for navigation. The declination of the Big Dipper is above 50° North, therefore, if the navigator is above 40° North latitude the Big Dipper would never set below the horizon but would appear to revolve en-

EAST

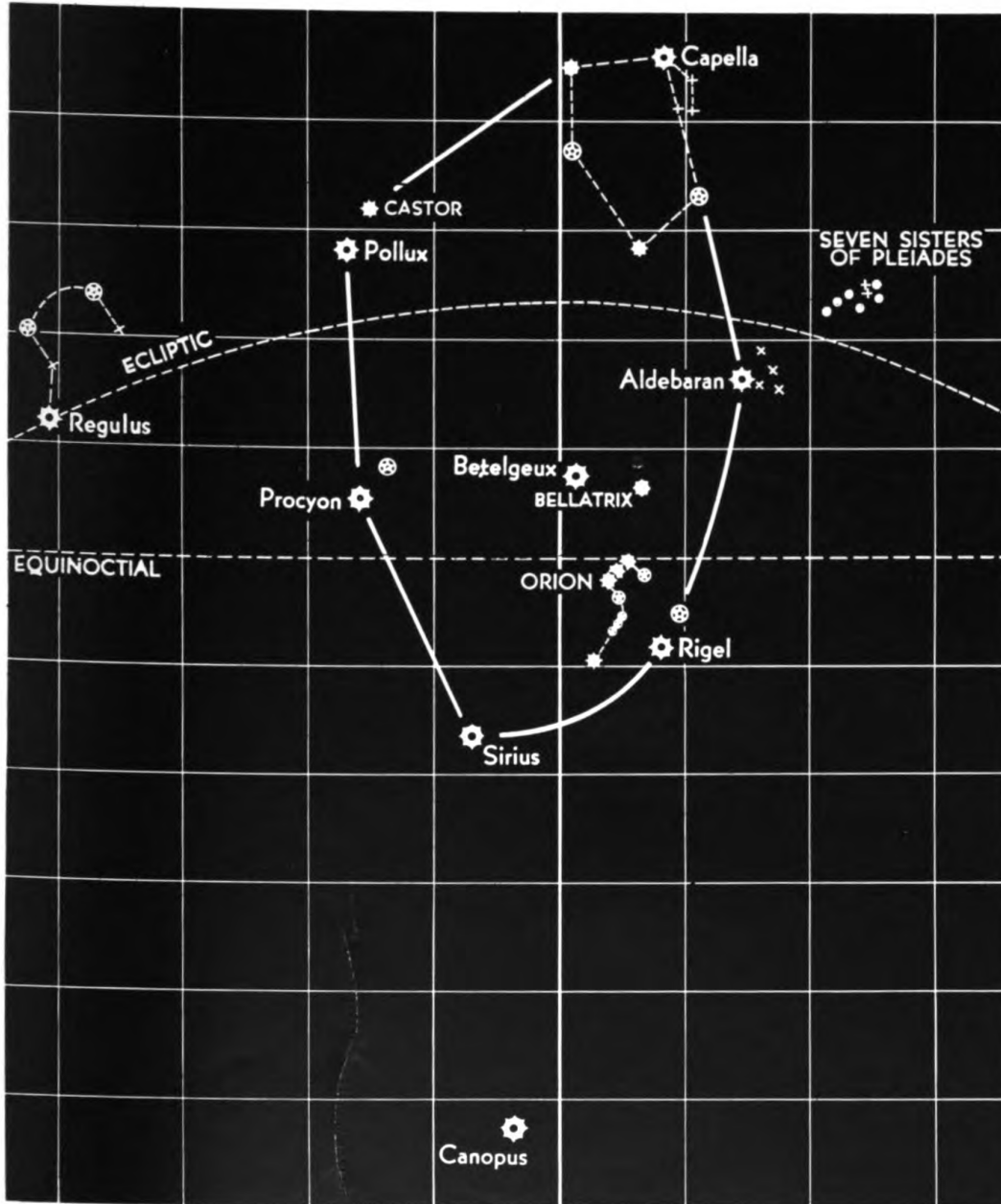


FIG. 137—"ORION'S BELT AND SWORD" GROUP

tirely around the North pole, due to the earth's daily rotation. At one time the Dipper might be seen right side up, as though filling, and then twelve hours later upside down, as though emptying.

To the astronomer, the Big Dipper is known as Ursa Major, but to mariners, for as

long as they have sailed in the Northern hemisphere, the Big Dipper is the North pole indicator. This is because the two stars in the cup, Merak and Dubhe, known as the pointers, point almost directly to Polaris.

POLARIS is the North pole star. Actually it is about 1° from the true North celestial pole.

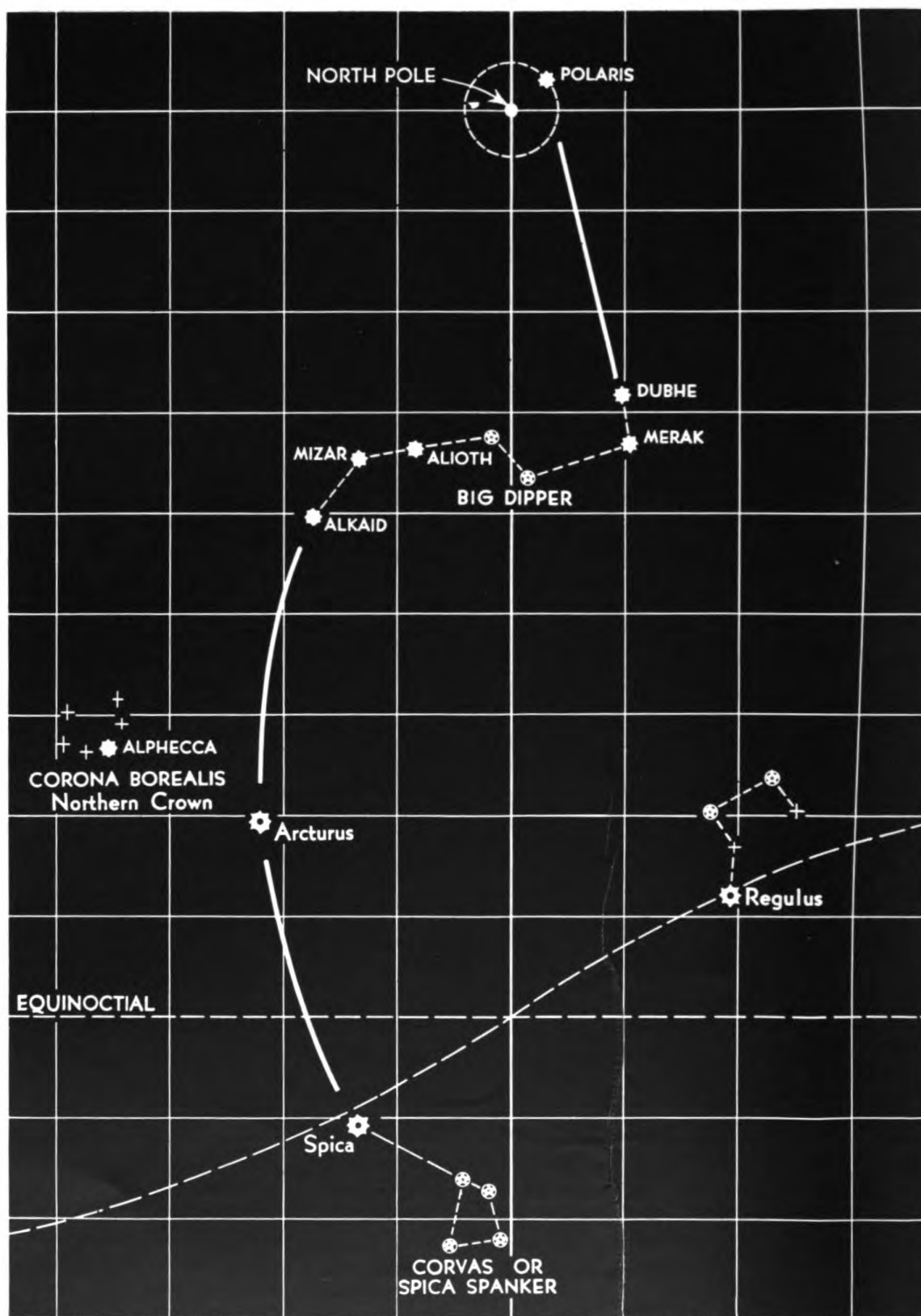


FIG. 138—"BIG DIPPER" GROUP

ARCTURUS and *SPICA* are two very bright stars which may be found by following the curve of the Big Dipper's handle on across the sky. First seen is Arcturus and then Spica. The distance of Spica from Arcturus appears to be very nearly equal to the distance of Arcturus from the end of the handle.

SPICA also may be identified by the constellation Corvus. Corvus is sometimes called Spica Spanker as it resembles a ship's spanker sail. The two stars at the top of the sail point to Spica.

ALPHECCA is the bright star in the constellation Corona Borealis, known as the Northern Crown. Corona Borealis is easily identified by its shape, which resembles a crown, and may be found by searching the skies northwest of Arcturus.

Third Group—"Great Right Triangle" (See group chart, Figure 139)—The Great Right Triangle is formed by the three very bright stars Altair, Vega, and Deneb.

ALTAIR is identified by two dimmer stars, which are located on either side of it in a straight line pointing directly toward Vega.

VEGA is the bright star at the right angle of the great triangle. It may be identified further by two dimmer stars close by, each of which points to an adjacent angle of the great triangle, and which with Vega form a tiny right triangle.

DENEK at the third angle of the great triangle appears slightly dimmer than either Altair or Vega. It also may be identified by two dimmer stars nearby. Each of these stars points to an adjacent angle of the great triangle and form with Deneb a medium-sized equilateral triangle.

Fourth Group—"Square of Pegasus" (See group chart, Figure 139)—All of the stars in this group are of medium brightness, but they are easily recognized in the sky and are useful in placing other stars. The four stars Alpheratz, Scheat, Markab, and Algenib, equally distant from one another, form a great square. Sometimes, however, this constellation is called "The Kite" as there are three other stars lined up with one corner of the square which makes the Square of Pegasus resemble a kite and tail.

ALPHERATZ, the star in the tail corner and *MARKAB*, in the opposite corner are tabulated in the Air Almanac, and are thus useful for navigation.

The Square of Pegasus identifies Fomalhaut and Deneb-Kaitos.

FOMALHAUT is in line with Scheat and Markab.

DENEK-KAITOS is in line with Alpheratz and Algenib.

ENIF affords another example of finding a star by lining it up with stars already known, as it is in line with, and halfway between, Altair and Markab.

These four star groups include most of the useful navigation stars. If the student learns them well, he should have little difficulty in identifying the other bright stars in the sky.

Constellation Groups—There are a few other stars often used in navigation which may be easily identified by their own constellation characteristics, as was Alphecca in Corona Borealis, without resort to grouping. These are, in part:

1. *CAPH*, *SCHEDIR*, and *RUCHBAH* in the constellation Cassiopeia. (See star chart, Figure 136). Cassiopeia is a northern constellation in the shape of a W, M, or a chair and is sometimes called the W, the M, or The Chair depending on its appearance to the observer as it changes position in the sky.

2. *NUNKI* and *KAUS-AUSTRALIS* in the constellation Sagittarius. Sagittarius is a Southern constellation near the ecliptic having the appearance of a little dipper.

3. *RIGIL KENTAURUS* and the *SOUTHERN CROSS*. In the Southern Cross (*CRUX*) the two stars tabulated in the Air Almanac are β and γ Crucis, identified on the chart by the Greek letters β (*BETA*) and γ (*GAMMA*). Rigil Kentaurus and β Kentaurus are called the Southern Cross pointers, as they point directly to the Southern Cross.

IDENTIFICATION BY H. O. 214

Sometimes flight conditions arise where it is necessary for the navigator to take a sight of a celestial body which he is unable to identify by the usual method of its relation to nearby stars. For example, after flying under an overcast sky for several hours the navigator sights a single celestial body through a break in the clouds. In such a case the body may be identified by reference to any altitude-azimuth table or star finder if the altitude and azimuth of the body and the Greenwich civil time are known.

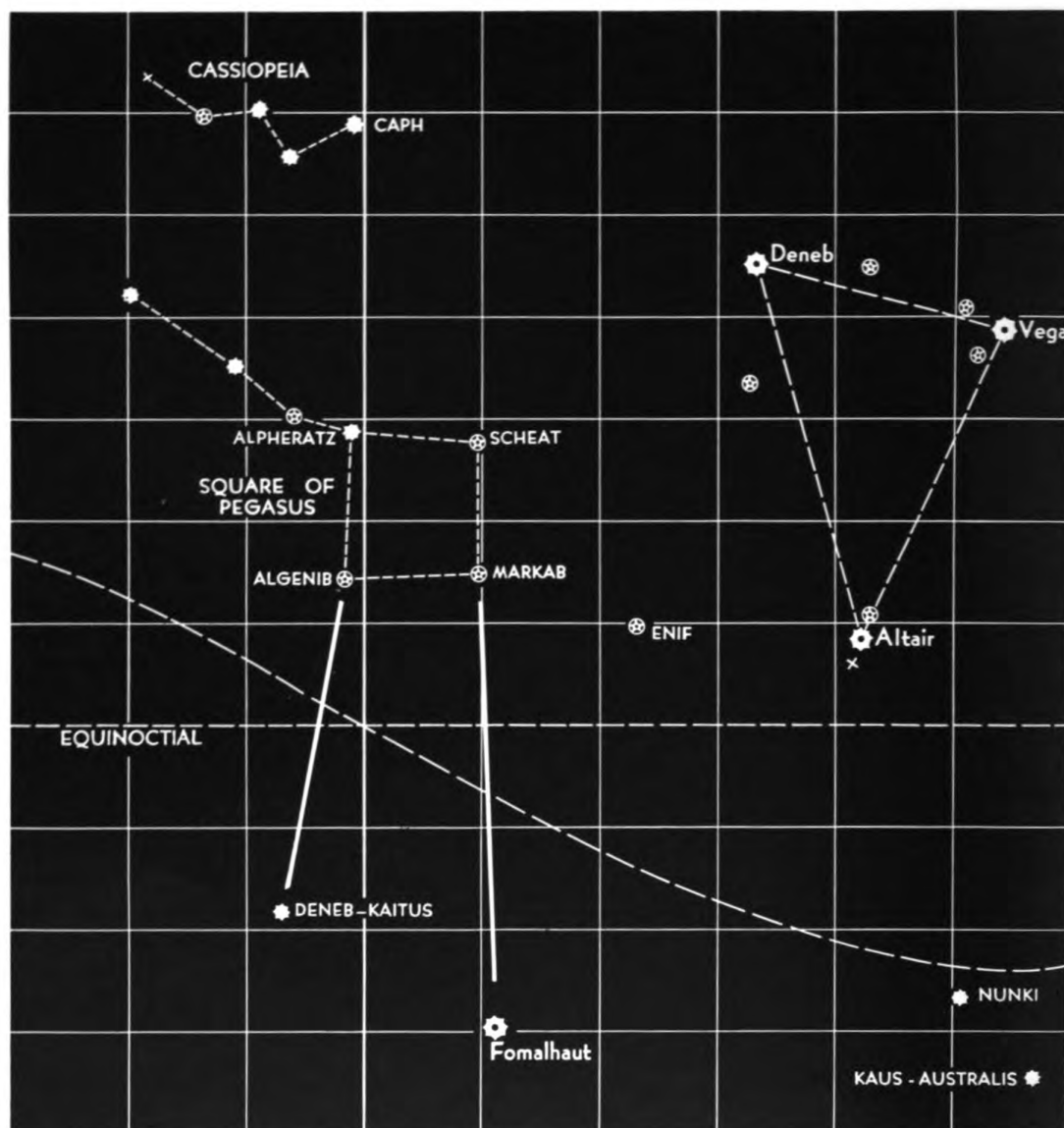


FIG. 139—"GREAT RIGHT TRIANGLE" AND "SQUARE OF PEGASUS" GROUPS

The Procedure using H. O. 214 is as follows:

- Observe altitude of the body.
- Note exact GCT.
- Estimate the body's true bearing and convert true bearing to azimuth (Figure 140).
- Enter Star Identification Table H. O. 214, which immediately follows the proper latitude (latitude of the observer), with the alti-

tude and azimuth of the body, and extract tabulated values of approximate LHA and declination. The LHA is tabulated from 0° to 180° East or West. The azimuth name determines the name of the tabulated LHA. If the azimuth is East, the LHA is East because the body is rising. If the azimuth is West, the LHA is West because the body is setting. Tables show declination North or South.

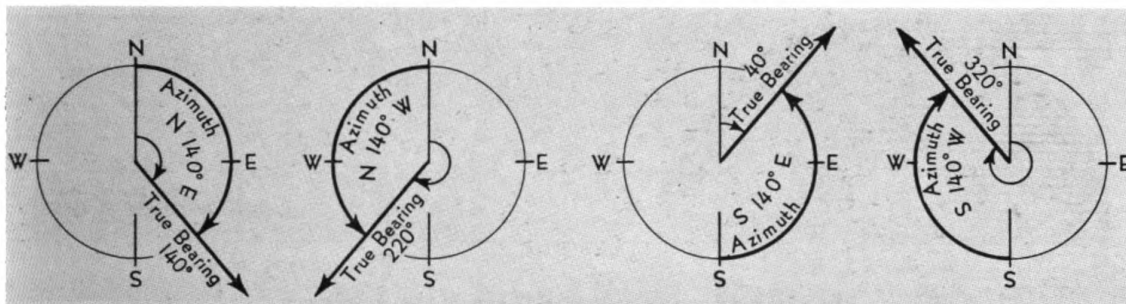


FIG. 140—AZIMUTH MEASURED FROM ELEVATED POLE

e. Convert tabulated LHA if East to LHA West by subtracting LHA East from 360° ($360^\circ - \text{LHA East} = \text{LHA West}$).

f. Apply DR longitude to LHA West to obtain GHA body.

$$\text{GHA} = \text{LHA West} \quad \begin{array}{l} + \text{ West longitude} \\ - \text{ East longitude} \end{array}$$

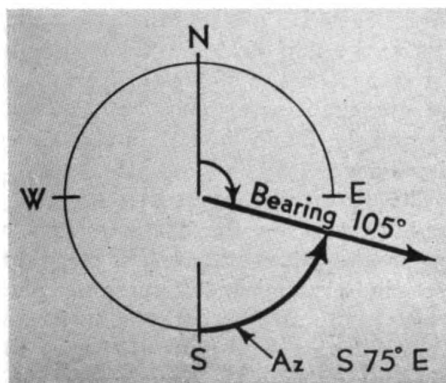


FIG. 141— $Z_n 105^\circ = \text{Az } S 75^\circ E$

g. Extract GHA Υ from Air Almanac for the exact GCT of observation.

h. Find SHA body by subtracting GHA Υ from GHA body (add 360° to GHA body if necessary).

i. Enter Air Almanac, inside back cover, with SHA body and declination. Extract name of star whose coordinates are nearest to calculated SHA and declination.

Example: (Solution of star identification problems is simplified by the use of some such form as that employed below).

May 30: A navigator obtains an altitude of an unknown celestial body of $67^\circ 26'$ at $07^h 52^m 00^s$ GCT. With the aid of the pelorus (instrument for measuring a bearing) he estimates its true bearing to be 105° .

DR position = latitude $32^\circ 16'S$, longitude $117^\circ 28'W$.

$Z_n = 105^\circ$; $\text{Az} = S 75^\circ E$ (Figure 141).

$H_o = 67^\circ 26'$ Enter H. O. 214
 $Z_n = 105^\circ$ (Lat. = $32^\circ S$)
 $\text{Az} = S 75^\circ E$

Declination = $35^\circ S$
 LHA = $26^\circ E$

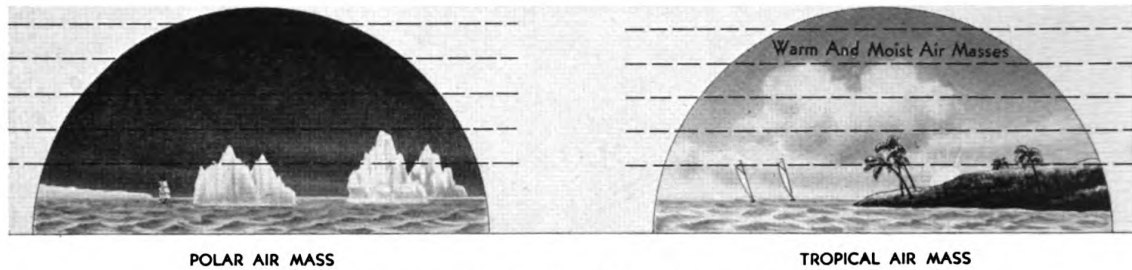
$$(360^\circ - 26^\circ E = \text{LHA } 334^\circ W)$$

GCT = $07^h 52^m 00^s$
 GHA Υ for $07^h 50^m = 4^\circ 27'$
 Corr. for $2^m = 30'$
 GHA $\Upsilon = 4^\circ 57'$

LHA West = 334°
 Long. West = $117^\circ 28'$
 GHA* = $451^\circ 28'$
 $- \text{GHA } \Upsilon = 4^\circ 57'$
 SHA* = $446^\circ 31'$
 $- 360^\circ$
 SHA* = $86^\circ 31'$

Enter Air Almanac
 SHA* $86^\circ 31'$
 Dec. $35^\circ S$

Star is Kaus. Australis



Dotted Lines Illustrate Different Levels Where Temperature, Pressure and Moisture Content Are Relatively the Same Throughout the Air Mass.

FIG. 142—CHARACTERISTICS OF AIR MASSES DERIVED FROM SOURCE REGIONS

PRACTICAL WEATHER MAP FORECASTING

Weather map forecasting is the science of graphically recording possible weather and winds to be expected along a proposed route. This information is predetermined, primarily by reference to known observations of past weather. If the sky were continually clear, weather in itself would be relatively unimportant. But the fact that temperature and pressure changes cause rain, snow, fog, hail, or any number of various other weather phenomena makes a study of meteorology—or at least a general knowledge of its more important aspects—essential to successful flight.

This knowledge is of particular interest to the navigator, who is responsible for seeing that his aircraft reaches its destination safely. It is apparent that an aircraft, having no physical ties to the earth, is subject to every weather condition generated by the atmosphere.

A general knowledge of the fundamentals of meteorology and an understanding of the problems of the meteorologist in forecasting weather, together with the ability to converse with the meteorologist in his own terms, are important qualifications for the air navigator of today. The following is an attempt to familiarize the navigator with the more important aspects of weather analysis in general, and to acquaint him with the causes and effects of weather phenomena as they apply to him in particular.

Air Masses—An air mass is a large body of air approximating horizontal homogeneity. In other words, the physical characteristics are the same over a large area at different levels. These large bodies of air derive their characteristics from, and are named for, their source

regions. That is, air masses originating over polar regions are inherently cold and dry. Those originating in tropical regions tend to be warm and moist (Figure 142). Tropical and polar, therefore, are the names generally applied to warm and cold air masses respectively; however, the source region does not necessarily have to be the tropical or polar region. For instance, large air masses which become stagnant over the Gulf of Mexico take on the warm, moist characteristics of the Gulf water, and in the winter the large air mass between the Cascades and the Rocky Mountain range becomes cold and dry.

The air mass, though it is stagnant long enough to take on the characteristics of its source region, does not remain stagnant, because wind changes and air pressure phenomena cause it to move across the surface of the earth. As the air mass moves it begins to take on the properties of the surface over which it passes. For example, a cold, dry air mass which lingers over a warm body of water takes on moisture and heat. As it warms and expands it is able to hold more moisture. The air mass, however, does not simply mix, but instead the cold air being heavier undercuts the lighter, warmer air causing a discontinuity known as a frontal surface, which chiefly causes storms.

Pressure Systems—A study of the movement of air masses must take into consideration the development of high and low pressure areas. Temperature is the chief cause of changing pressure. The atmosphere contracts at low temperatures and, concentrating over a smaller area, exerts higher pressure. Conversely, high temperatures cause air to expand and the resulting distribution of its weight results in lower pressure (Figure 143).

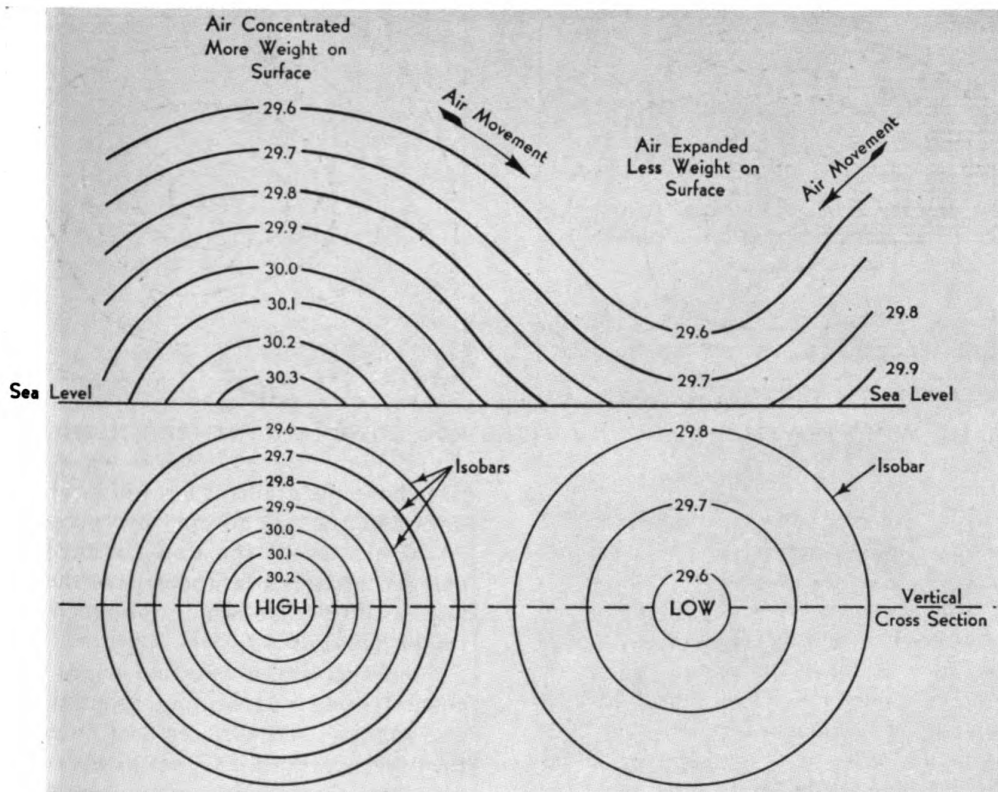


FIG. 143—PRESSURE SYSTEMS AND ISOBARS

Isobars—Isobars are lines connecting points of equal barometric pressure. By reference to these lines on a synoptic chart (a chart which condenses much data into one source), high and low pressure areas may be located in a manner similar to that used to show mountains and valleys on an aeronautical chart by means of contour lines (Figure 143).

Winds—Wind is merely the result of moving air. This movement takes place from an area of high pressure toward an area of low pressure in an attempt to equalize the total atmospheric pressure. Figure 143 shows how air moves in a manner similar to water flowing downhill. The steeper the gradient, the faster the flow. It is obvious, then, that the closer the isobars are together, the faster will be the flow of air indicated, hence the greater the wind velocity. Thus far it would seem, taking into consideration only the gradient force, that the wind would blow in a direction perpendicular to the isobars toward the low pressure area. However, a deflecting element known as

Coriolis force must be considered. This force is a result of the rotation of the earth upon its axis, and turns all winds to the right in the Northern hemisphere and to the left in the Southern hemisphere. It then becomes apparent that, in the Northern hemisphere, winds develop a counterclockwise movement toward the center of a low or cyclonic area, and a clockwise movement outward from a high or anticyclonic area. This is illustrated in Figure 144. The same force causes the reverse situation in the Southern hemisphere. Turning to the left instead of to the right, the movement becomes clockwise in a low and counterclockwise in a high. The direction of the wind is always given in degrees, measured clockwise from true North in the direction from which the wind is blowing.

Surface winds are affected by local high and low pressure areas, while upper air winds more or less conform to the prevailing highs and lows and change with frontal systems. That is, winds in mid-latitudes become prevailing westerlies in both hemispheres where they

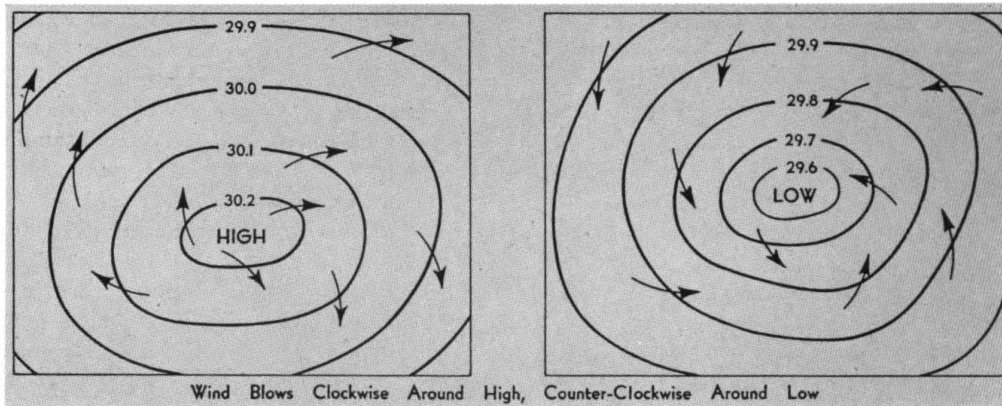


FIG. 144—WIND MOVEMENT AROUND HIGHS AND LOWS IN NORTHERN HEMISPHERE

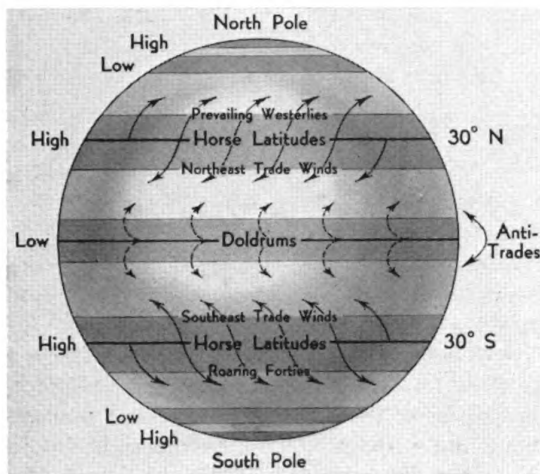


FIG. 145—EARTH'S BASIC WIND CIRCULATION

rise above the gradient force of local highs and lows and are less affected by surface friction. At higher levels, the Coriolis force predominates. Equatorial latitudes have their prevailing easterly trade winds for much the same reason (Figure 145).

Another factor affecting winds is the unequal cooling and heating properties of land and water. Land surfaces heat quickly during the daytime, resulting in sea breezes blowing in over the coast. Conversely, land areas cool faster at night, causing winds to shift and resulting in land breezes out to sea. This phenomenon becomes important when flying close to continental coastlines.

CLOUDS

Cloud Families—Clouds are a visible sign of atmospheric activity, appearing as visible

HIGH CLOUDS (CH) MEAN LOWER LEVEL 20,000 FEET		
1.	Cirrus (Ci)	Thin and Featherlike
2.	Cirro-cumulus (Cc)	Thin—Cotton or Flakelike
3.	Cirro-stratus (Cs)	Very Thin—High Sheet Cloud
MIDDLE CLOUDS (CM) MEAN LEVELS 6500 TO 20,000 FEET		
4.	Alto-cumulus (Ac)	Puffy—Sheep Back
5.	Alto-stratus (As)	Medium High—Uniform Sheet Cloud
LOW CLOUDS (CL) MEAN LEVELS CLOSE TO SURFACE TO 6500 FEET		
6.	Strato-cumulus (Sc)	Globular Masses or Rolls
7.	Stratus (St)	Low Uniform Sheet Cloud
8.	Nimbo-stratus (Ns)	Low Amorphous and Rainy Layer
VERTICAL CLOUDS (CL) MEAN LEVELS 1600 TO 20,000 FEET		
9.	Cumulus (Cu)	Dense—Dome-shaped and Puffy
10.	Cumulo-nimbus (Cb)	Towering Cauliflower—Anvil Top

FIG. 146—CLOUD FAMILIES

moisture condensed from warm, moist air which is cooled by expansion upon rising to higher levels. The two basic cloud types are cumuliform and stratoform, the former being developed vertically in puffy shapes, and the latter appearing as more or less flat sheet clouds. These are further broken down into four family names according to the heights at which they are found. Symbols for these clouds are shown on all weather maps.

Figure 146 shows the cloud families.

Ceiling—The height of the clouds above the surface of the earth is defined as ceiling.

Sky Cover—The amount of clouds covering the dome of the sky is shown by four symbols (Figure 147):

Symbol	Classification	Sky Cover
○	Clear	Total Sky Cover less than one-tenth.
⊖	Scattered	One-tenth to five-tenths covered.
⦶	Broken	More than five-tenths but less than nine-tenths.
⊕	Overcast	Over nine-tenths of sky covered.

FIG. 147—SKY COVER

Fogs—Practically the only difference between a fog and a cloud is the point of view of the observer; fog is nothing more than a stratus cloud on the surface of the earth. There are four principal types of fogs: advection, up-slope, radiation, and frontal.

Advection fog is the result of warm moist air blowing over a cooler surface. Coastal sea fogs are an example of this type.

Up-slope fogs form when parcels of air are mechanically lifted over mountainous terrain and cooled to their dew point. These are obviously restricted to land areas.

Radiation fogs form mainly over land surfaces at night as a result of the earth's cooling. Prerequisites include a temperature inversion and winds of less than eight miles per hour. They are very low-hanging and disappear soon after the sun starts heating the earth once more in the morning.

Frontal fogs occur in frontal zones, principally warm fronts, where warm moist air contacts a body of colder air. They form similarly to the advection type, but are restricted to the smaller area at the front only.

FRONTS

General—A front is simply a line of discontinuity separating two air masses of dissimilar characteristics. Warm moist air pushing up over the surface of colder air is cooled, and precipitation takes place. Heavier cold air squeezing in under warm moist air lifts it to the condensation level and again precipitation occurs. Familiarization with fronts is essential because most of the hazards of flying take place where these air masses of unlike qualities meet. Of particular interest to the pilot is the fact that approximately 90% of all icing takes place in a frontal zone.

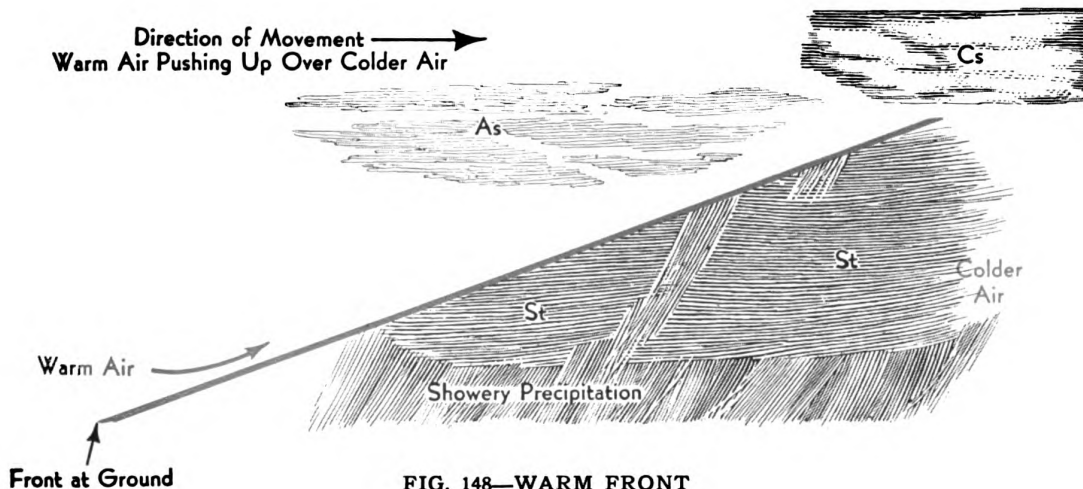


FIG. 148—WARM FRONT

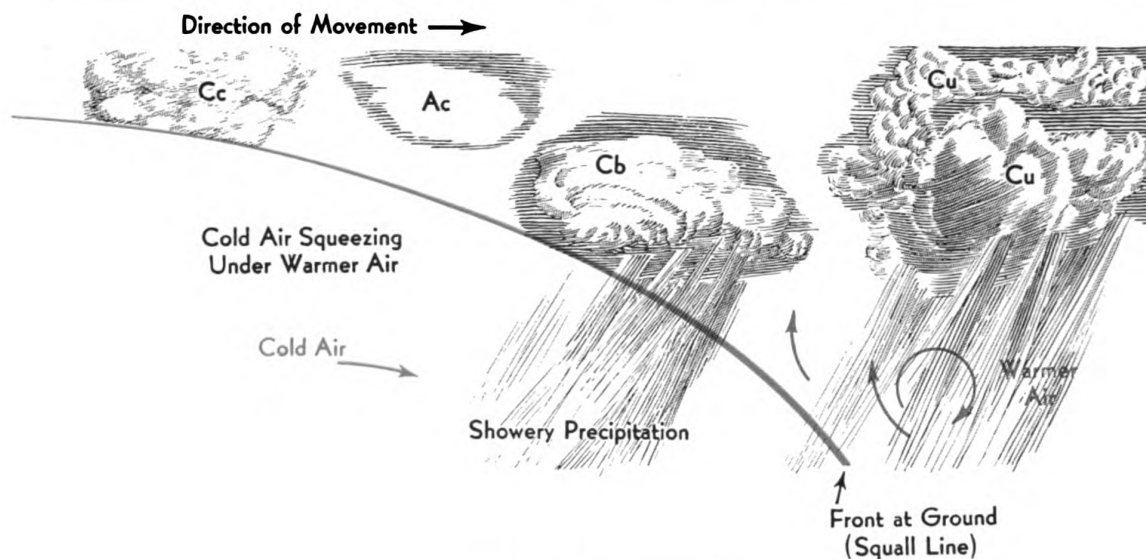


FIG. 149—COLD FRONT

The observer should remember that the fronts are shown on weather maps as a line of discontinuity at the ground. The air navigator must take into consideration the altitude of his aircraft when attempting to determine the point where he will encounter the front, in addition to the movement of the entire frontal system which occurs in an East-South-Easterly direction across the Pacific at a rate of approximately 500 miles per day. Passage of a front is recognized chiefly by a decided temperature change and wind shift; both are items of importance to be considered in air navigation.

Fronts are divided into three main categories: warm, cold, and occluded.

Warm Fronts (Figure 148)—A warm front is indicated where warm air is replacing cold air at the ground. It has a very gentle slope upward over the colder air and is usually accompanied by stratoform clouds, lowering ceilings, and an area of steady precipitation preceding the front. The sequence of clouds observed is usually cirro-stratus, alto-stratus, and stratus. Passage of the front is indicated by an attempted clearing, a decided wind shift, humid air, and mild to moderate turbulence. The principal hazard to aircraft associated with a warm front is fog. Any icing encountered is usually of the rime variety, which is porous and can be taken care of by deicer or anti-icing equipment.

Cold Fronts (Figure 149)—Cold air is replacing warm air at the ground in the case of a cold front. A fast-moving cold front is accompanied by severe turbulence and line squalls which may include frontal thunder storms recognized by the familiar cumulo-nimbus clouds with their anvil tops. Precipitation is more intense but usually confined to the showery type in spotted areas under the cumulus clouds. Hail may be encountered if temperatures are low enough and high swelling cumulo-nimbus clouds are present. In many cases under these conditions, vertical currents beneath these clouds are in excess of 100 miles per hour. It is sufficient to say that they can be dangerous. Passage of the front is indicated by a decided wind shift and drop in temperature followed by clear, cold, good flying weather. Cumuliform clouds follow after the front. The slope is steeper than the warm front.

The principal hazard of the cold front is icing, but also may include severe vertical currents and probable hail showers. Icing is mostly of the clear type which spreads quickly and destroys the airfoil. Vertical currents support larger droplets of water which spread before freezing.

Occluded Fronts—In an occluded front warm air is squeezed off the ground entirely and replaced by colder air. By reference to

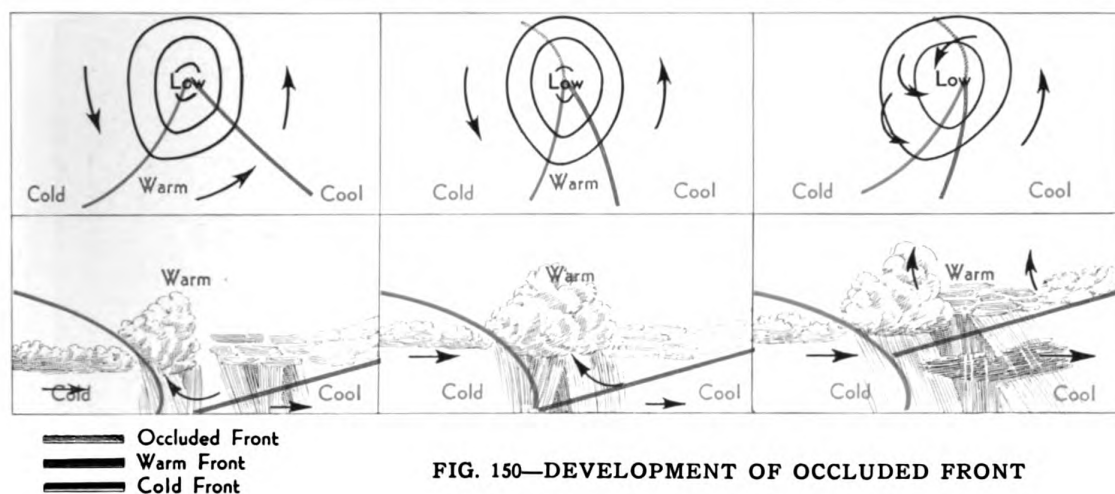


FIG. 150—DEVELOPMENT OF OCCLUDED FRONT

Figure 150 it will be seen that an occluded front is actually the breaking up of a low pressure area, and passage directly through the occluded part will be accompanied by a drop in temperature and very changeable winds. Passage to the south will encounter successive warm and cold fronts according to the direction of flight. Figure 150 shows both top and side view drawings of an occlusion in process.

SUMMARY

Figure 151 represents a typical weather map. The summary below will follow through a flight, and indicate how you, as the navigator, would analyze this information for flight purposes.

First of all, notice the GCT for which the map was drawn, which is important in that the fronts to be crossed on your course are located for 18⁰⁰m GCT only and will move easterly at approximately 21 miles per hour as we progress. It is apparent that temperatures will gradually rise as we leave San Francisco, and that we will be aided by moderately increasing tail winds.

We may expect to encounter the warm front indicated by the heavy red line considerably sooner than indicated, due to the fact that, in addition to the front having moved toward us, its gentle forward slope will cause us to encounter it even sooner at our flight altitude. The definite wind shift from East to Southwest will be accompanied by a slow increase in temperature, weather clearing, and

smooth air after passage through the front. Low stratus clouds will probably be observed preceding the front with an area of continuous precipitation underneath.

A second wind shift may occur with probable head winds as the heavy blue line (cold front) approaches. It will be noted that at 18⁰⁰m GCT, this front is at Honolulu, but will have passed Honolulu before your arrival. Assuming we have a fast plane, we may still pass South of the approaching line squalls which would be visible ahead as a black wall of clouds in the daytime, or evident by the blotting out of low stars at night in the general direction. In any event, we should be prepared to alter course to the left to avoid considerable turbulence and rough weather. This is based on the assumption that the line of discontinuity in question separates air masses of decidedly opposite characteristics. We would be advised of this previous to take-off by the meteorologist in charge.

Assuming that we pass South of the approaching cold front, we may still expect head winds gradually increasing in velocity. This is made evident by the fact that the isobars are closer together, and by the presence of a low pressure area moving in from the Northwest.

The approaching cold front will definitely cross our path and will indicate its passage by a temperature drop, squally weather, and the presence of cumulus type clouds. The intensity of its vertical currents and turbulence may be

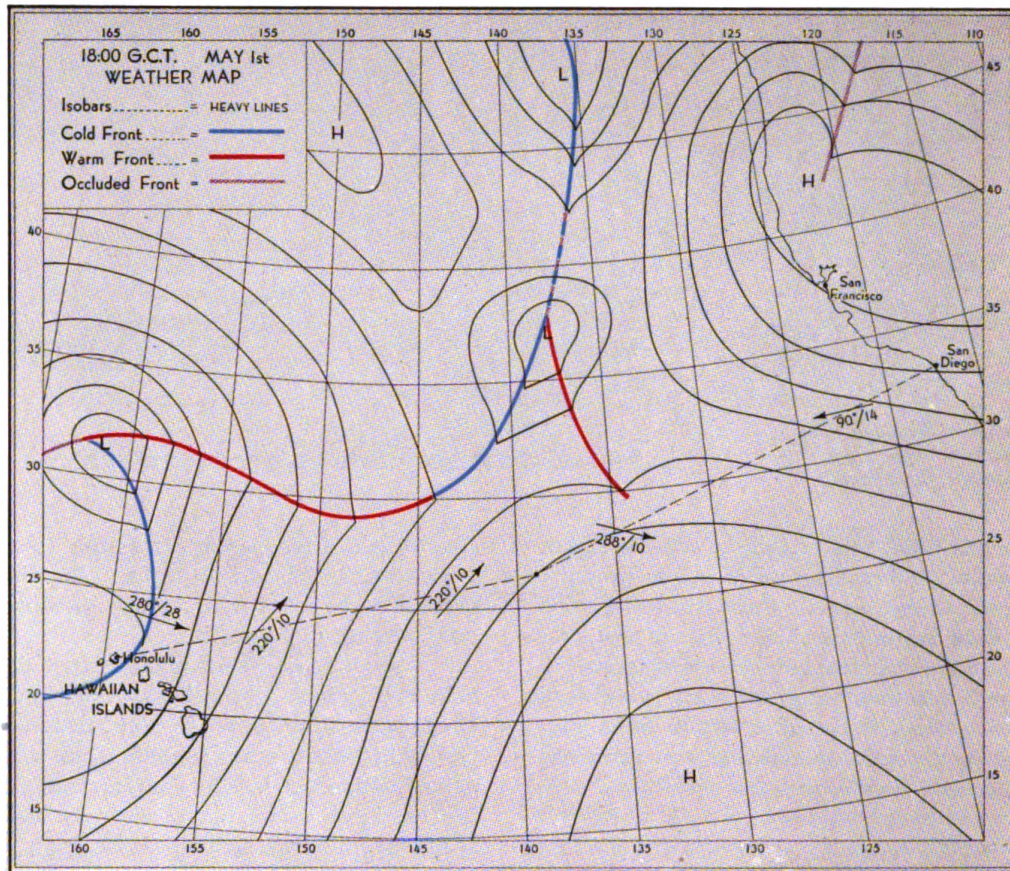


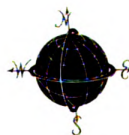
FIG. 151—TYPICAL WEATHER MAP

judged by the vertical swelling of its cloud systems.

The knowledge obtained from the meteorologist before the take-off may be augmented by a radio weather report from Honolulu as to the frontal conditions which we will encounter.

Another wind shift from Southwest to Northwest may be expected after passage of the front. Head winds are still prevailing but the shift may counteract the drift applied on our present heading. A celestial fix at this time should show us any drift tendency, and our destination should be in sight shortly thereafter.

Note: No attempt is made in the foregoing to go deeply into meteorology as a science with its accompanying lapse rates, pressure tendencies, and various weather phenomena. The effort has been directed entirely toward acquainting the air navigator with the basic principles as they apply to the safe flight of his aircraft. Perhaps the air navigator of tomorrow will himself be a competent meteorologist. Until then, he can learn a great deal through intelligent conversation with the weather forecaster if he has been trained to know *what the forecaster is talking about* in addition to knowing *what to look for* and the probable results.



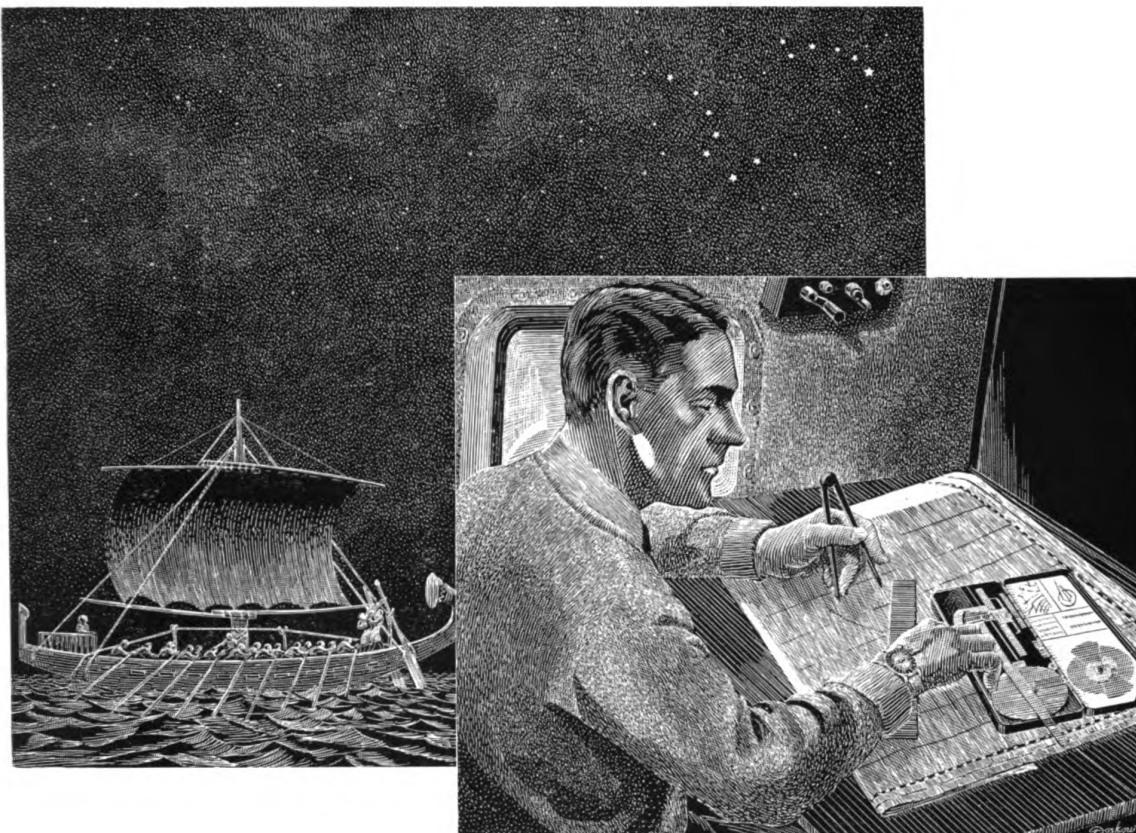
STARS AND THE WEATHER

PROBLEM WORK NO. 30

STAR IDENTIFICATION BY H. O. 214

(Identify stars by H. O. 214 method.)

No.	DATE	GCT	DR POSITION		H ₀	AZIMUTH	STAR
			Lat.	Long.			
1	5- 5-43	13:27:00	37°50'N	121°01'E	19°58'	N 55° E	
2	5- 1-43	13:14:00	39°24'S	110°31'E	37°12'	S 80° W	
3	5-10-43	11:24:00	38°15'N	146°14'E	38°15'	N160° E	
4	5- 5-43	19:43:00	36°14'S	89°45'E	25°00'	S 120° E	
5	5-20-43	13:40:00	32°40'N	123°14'E	38°02'	N100° W	
6	5- 1-43	12:04:00	38°42'S	173°13'W	36°14'	S 25° W	
7	5-25-43	00:49:00	34°17'N	64°50'W	53°08'	N 90° E	
8	5-30-43	02:43:00	35°10'S	84°30'W	51°10'	S 90° E	
9	5-20-43	08:32:00	30°10'S	135°11'W	44°04'	S 80° E	
10	5-15-43	19:24:00	34°48'S	132°12'E	51°34'	S 110° W	
11	5-25-43	20:48:00	33°21'S	56°46'E	44°38'	S 165° E	
12	5-30-43	07:52:00	32°16'S	117°28'W	67°26'	S 75° E	
13	5-25-43	11:36:00	30°03'S	137°17'E	18°08'	S 40° W	
14	5-10-43	10:30:00	33°45'N	179°46'W	59°32'	N120° W	
15	5- 1-43	02:44:00	38°30'N	43°28'W	66°40'	N 35° W	
16	5-15-43	19:33:00	31°38'N	32°32'E	24°26'	N140° E	
17	5-10-43	02:17:00	39°18'S	28°04'W	67°22'	S 85° E	
18	5-20-43	14:32:00	37°10'N	122°44'E	23°46'	N 50° E	
19	5- 5-43	18:42:00	31°26'N	170°23'W	60°04'	N 85° E	
20	5-15-43	18:22:00	36°20'N	146°12'W	34°52'	N170° E	



☆ 11 ☆

SPECIAL PROCEDURES

THE navigator who understands and applies all available navigation procedures, keeps a constant check on his aircraft's track and carefully analyzes all position data for possible errors, will be able to report quickly, and within close limits, the position of his aircraft at any time. Furthermore, he seldom will *feel* in doubt as to his location.

However, certain factors over which he has no control—such as weather, or fatigue—may cause him temporary uncertainty. At such

times, position-finding data appears conflicting and unreliable, and it is then that the navigator must guard against the natural tendency to become excited and confused. Several rules apply under such circumstances:

1. *Remain calm.* A cool mind functions much better in an emergency than one which tries to make decisions under emotional tension.
2. *Act intelligently.* Use common sense.
3. *Continue searching for information.* The best plan, of course, is to try to obtain a celes-

tial fix. If this is not possible, perhaps there are gaps in the overcast which will at least permit securing of a single L.O.P., which can be used for a track or ground speed check. Also, it may be possible to climb above the overcast.

At the same time, the navigator should continue trying for radio bearings which, either alone or combined with celestial lines of position, will enable him to establish his location.

4. *Recheck all work.* Correction of a careless mistake may enable you to establish your position.

5. *Most important of all—do not alter aircraft's heading unless the decision to do so is based upon definite information.*

FIXED SQUARE SEARCH

In the event the E.T.A. has elapsed without the destination having been sighted, the best procedure is to continue on the same heading until certain the objective has been passed. The navigator should keep trying to get celestial lines of position or radio bearings which

will give him definite information as to the aircraft's position. If such information cannot be obtained and the navigator is entirely dependent upon DR, direction of flight should be continued for a period of from 20 to 30 minutes, depending upon the accuracy with which ground speed was known when computing the E.T.A., and also upon the amount of fuel remaining. If the destination has not been sighted by then, nor any new position data obtained, a fixed square search procedure may be instituted. This is a method by which an aircraft can systematically search a fairly large area to locate a small object, such as an island, a disabled ship or a life raft. It is called a "fixed" square search, because it presumes that the object being sought is stationary. Before beginning the search it is necessary to determine the visibility (or the distance, in nautical miles, the observer can see in any direction). This distance is the *visibility factor* (V), and is based on the size of the object being sought, altitude of the aircraft and, principally, the

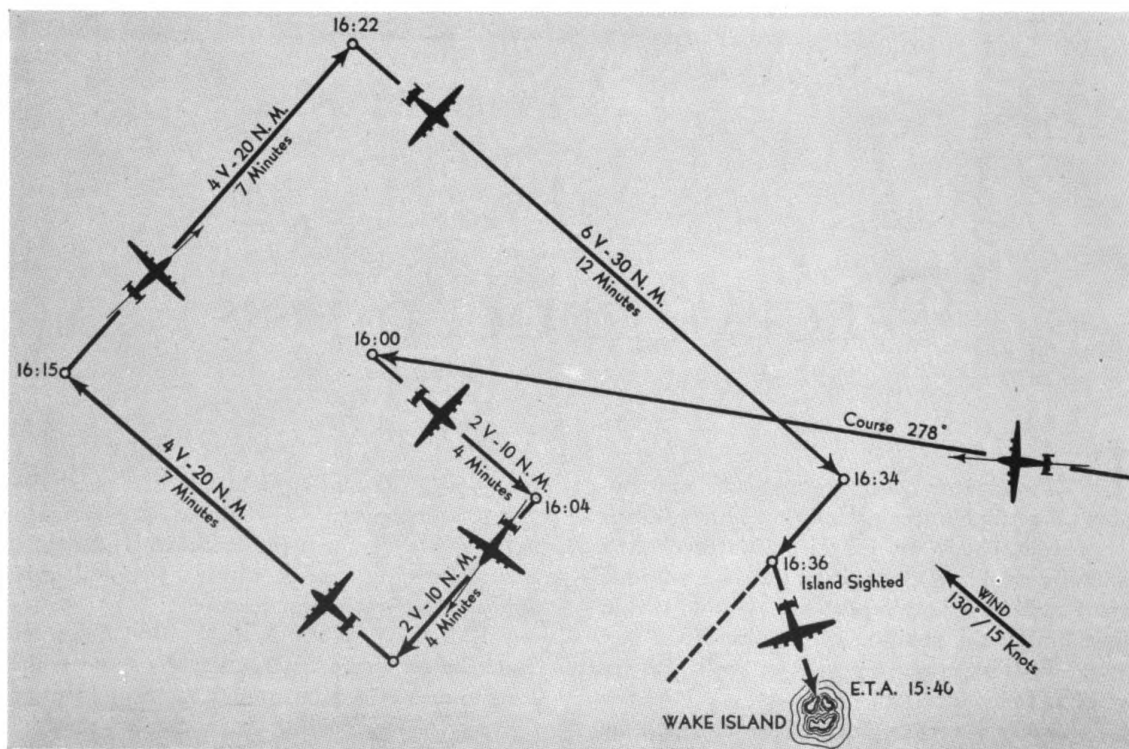


FIG. 152—SQUARE SEARCH PROCEDURE

visibility of the atmosphere. As visibility often is difficult to estimate, it is wise to be conservative in choosing a visibility factor. The determination of this factor, and the application of the square search procedure is illustrated by the following example.

Example (Figure 152) :

The navigator of an aircraft having exceeded his E.T.A. (15:40 GCT) by 20 minutes without sighting his objective (Wake Island), is authorized by the captain to institute a fixed square search procedure. Accordingly, at 16:00 GCT he advises the captain to assume a heading of 130° true in order to fly directly into the wind for the first leg. This upwind heading is flown first, both because it requires no drift calculation and because it reduces the ground speed to a minimum, thus affording more time in which to compute headings for the other legs. The T.A.S. is 165 knots; the wind—determined as carefully as possible by estimate, forecast, or double drift—is $130^\circ/15$ knots. The atmospheric visibility appears to be about eight miles, but to be safe the navigator decides to use a visibility factor of five miles.

The upwind heading is held for a distance equal to 2V (10 miles). And since the aircraft is flying upwind, its ground speed is equal to T.A.S. less the velocity of the wind, or 150 knots. Therefore, it requires 4 minutes to complete the first leg. When the first leg has been completed (16:04), the pilot turns to the right, putting the aircraft on a true heading of 215° . This heading was worked out by the navigator during the first leg, and includes proper drift allowance. This second heading enables the aircraft to make good a track 90° from the first leg.

Note: Instead of turning *right*, as cited in this example, the pilot might have turned *left* to assume a course 90° to the *left* of the original track. Direction of turn is dictated by circumstances under which the search procedure is employed.

This second heading is, in turn, held until a distance equal to 2V (10 miles) has been covered, which requires 4 minutes. When the second leg has been completed (16:08 GCT) the pilot again turns to the right and flies directly downwind (true heading 310°) for the

third leg, a distance equal to 4V (20 miles). At 16:15 GCT, heading is again altered to 045° in order to make good a track 90° from that flown on the third leg, and a distance of 4V (20 miles) is flown on this heading.

Since Wake Island still is not in sight on completion of the fourth leg (16:22 GCT), the navigator prepares to repeat the entire procedure, again increasing the distance to be flown by 2V for each succeeding two legs. Accordingly, at 16:22 GCT a right turn is made to begin the fifth leg, which, like the first leg, is directly upwind. This upwind heading is held for a distance equal to 6V (30 miles). Wake Island is not sighted on the fifth leg, so at 16:34 GCT a right turn is made for the sixth leg. After flying on this leg for two minutes, the island is sighted, bearing 160° true and seven miles distant. On checking his search plan, the navigator finds that Wake Island actually was 26 miles distant, bearing 127° true from the aircraft when the search was started.

RUNNING DOWN A SUN LINE

Running down a sun line is a procedure used by the navigator to reach a difficult objective, such as a small island, when only the sun is available for observation.

For instance, assume that an aircraft is making a daylight flight from Honolulu to Canton Island. During the day, the navigator has been estimating his position solely by sun position lines and drift meter readings. The sun lines have given him a fair estimate of the aircraft's position. However, several hours from Canton Island, he turns on the radio direction finder in order to make sure of his approach, and finds the radio isn't working.

Canton Island is a very small coral atoll, and the navigator knows that if he were only a few miles off course he could easily pass Canton by without seeing it. He therefore recommends to the captain that it would be advisable to run down a sun line.

The procedure is as follows (Figure 153) :

Given: course 209° —G.S. 180 knots.

a. About one hour from Canton Island the navigator observes the sun and obtains the 03:10 position line. From previous position lines and drift meter readings he indicates the

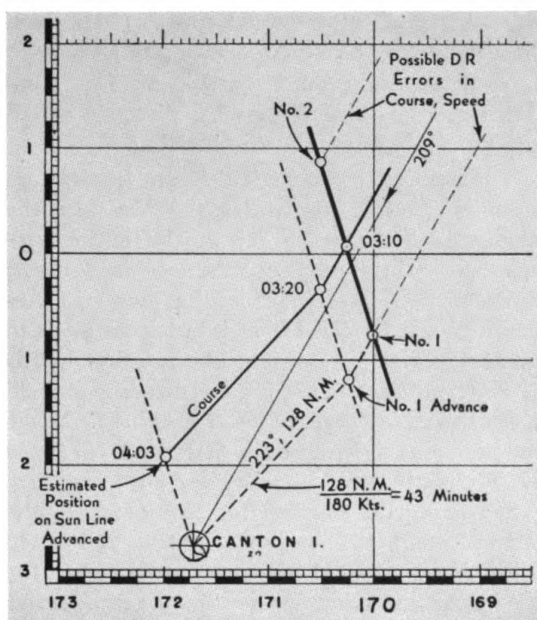


FIG. 153

most probable position on the sun line with a circle.

b. He then estimates possible errors in ground speed and course, and finds positions No. 1 and No. 2, which represent possible position limits on the sun line.

c. 03:20—Since it took ten minutes to calculate and plot the sun line, he advances position No. 1 ten minutes of ground speed and measures the course to Canton as 223° , distance away 128 NM.

d. Altering compass heading to fly 223° true, he then knows the aircraft will be definitely to one side of Canton, and should be somewhere on the position line advanced through Canton Island at 04:03.

e. To check, he again observes the sun and plots another position line at 03:40. Advancing this line through Canton Island, he determines the aircraft will be somewhere on the advanced line at 04:00 (Figure 154).

f. Therefore, at 04:00 he once more alters compass heading to fly 168° true which would run down the sun line, and thus fly over Canton Island.

g. Until Canton Island is sighted, however, he continues to plot sun lines at ten-min-

ute intervals to make certain his course will pass over the island.

LATITUDE BY MERIDIAN ALTITUDE

To take a meridian altitude sight means to measure the altitude of a celestial body when it is on the observer's meridian. This procedure is employed to determine the observer's latitude, and it is one of the simplest problems in celestial navigation. Its solution is derived from the fact that when a body is on the observer's meridian, its local hour angle is zero, hence the astronomical triangle is reduced to a straight line (Figure 155).

Procedure—The procedure necessary to determine latitude by meridian altitude is as follows:

- Observe altitude of celestial body.
- Find zenith distance (Z_d) by subtracting true altitude (H_0) from 90° .

$$90^\circ - H_0 = Z_d$$

- Determine bearing of observer's zenith from the celestial body (North or South) and name zenith distance accordingly. Thus, in example No. 1 (Figure 155), the zenith is North of the celestial body. Zenith distance, therefore, takes the name *North*. In example No. 2 (Figure 155), the zenith is South of the celestial body. Zenith distance, therefore, takes the name *South*.

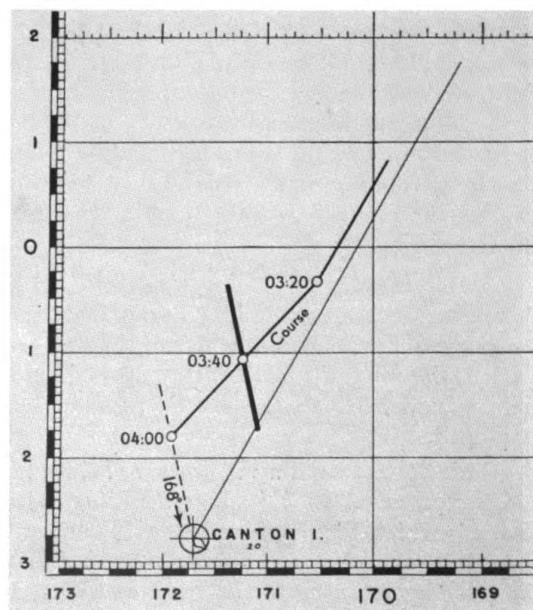
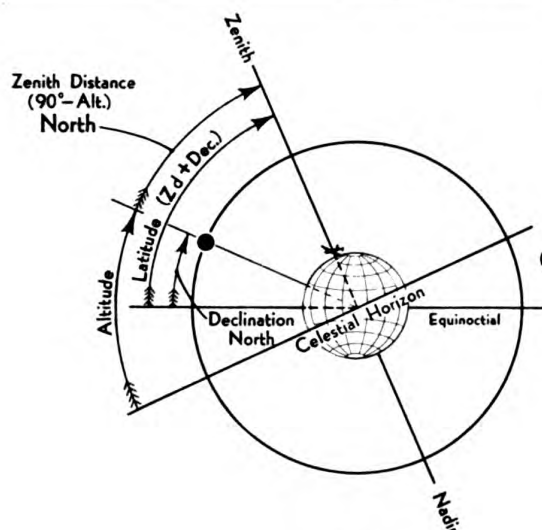
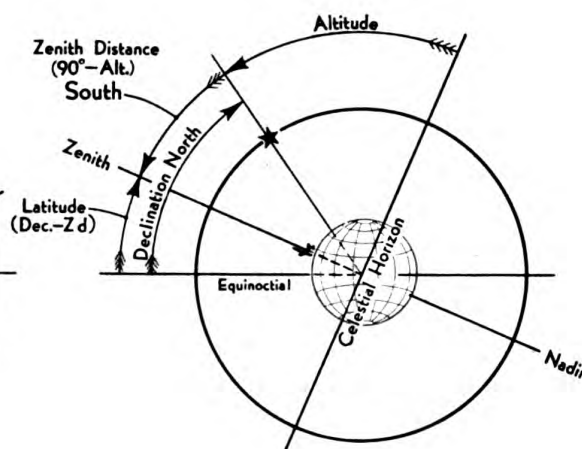


FIG. 154



Example No. 1
Observer's Zenith North of Celestial Body



Example No. 2
Observer's Zenith South of Celestial Body

FIG. 155—LATITUDE BY MERIDIAN ALTITUDE

d. Combine zenith distance and declination to obtain observer's latitude. If values have same name, add them together and give the resulting latitude their common name. If values have contrary names, subtract smaller from larger to obtain their difference which will take the name of the *larger* value and will, again, represent the observer's latitude.

$$Zd \sim dec. = \text{Latitude}$$

Example No. 1

Observer's Zenith North of celestial body

H_s Sun	$= 55^\circ 00'$
Refraction	$= 0'$
H_o	$= 55^\circ 00'$
	from $90^\circ 00'$
ZD	$= 35^\circ 00' \text{ N}$
Declination	$= 12^\circ 00' \text{ N}$
Latitude	$= 47^\circ 00' \text{ N}$

Example No. 2

Observer's Zenith South of celestial body

H_s Star	$= 60^\circ 00'$
Refraction	$= 0'$
H_o	$= 60^\circ 00'$
	from $90^\circ 00'$
ZD	$= 30^\circ 00' \text{ S}$
Declination	$= 43^\circ 00' \text{ N}$
Latitude	$= 13^\circ 00' \text{ N}$

FINDING TIME OF MERIDIAN TRANSIT

When the GHA of a celestial body is the same as the observer's longitude, the LHA is zero. This instant is called the *time of meridian transit*.

As has been previously stated, longitude may be defined as the angle, measured at the *terrestrial* poles, between the Greenwich meridian and the meridian of an observer. Similarly, GHA may be defined as the angle, measured at the *celestial* poles, between the Greenwich (celestial) meridian and the celestial meridian (hour circle) of a celestial body. It is apparent, therefore, that longitude and GHA correspond exactly, the only difference being that GHA is measured to the West through 360° , whereas longitude is measured East or West through 180° .

Graphic Method—The time of meridian transit may be found graphically by advancing the DR position of the observer along the course until his DR longitude is equal to the GHA of the body (found in the Air Almanac) for the same instant of time. This procedure is illustrated in the following example:

Example (Figure 156):

On January 1, at 17^h00^m GCT in longitude $100^\circ 10' \text{ West}$, the navigator desires to precompute the time of meridian transit of the sun.

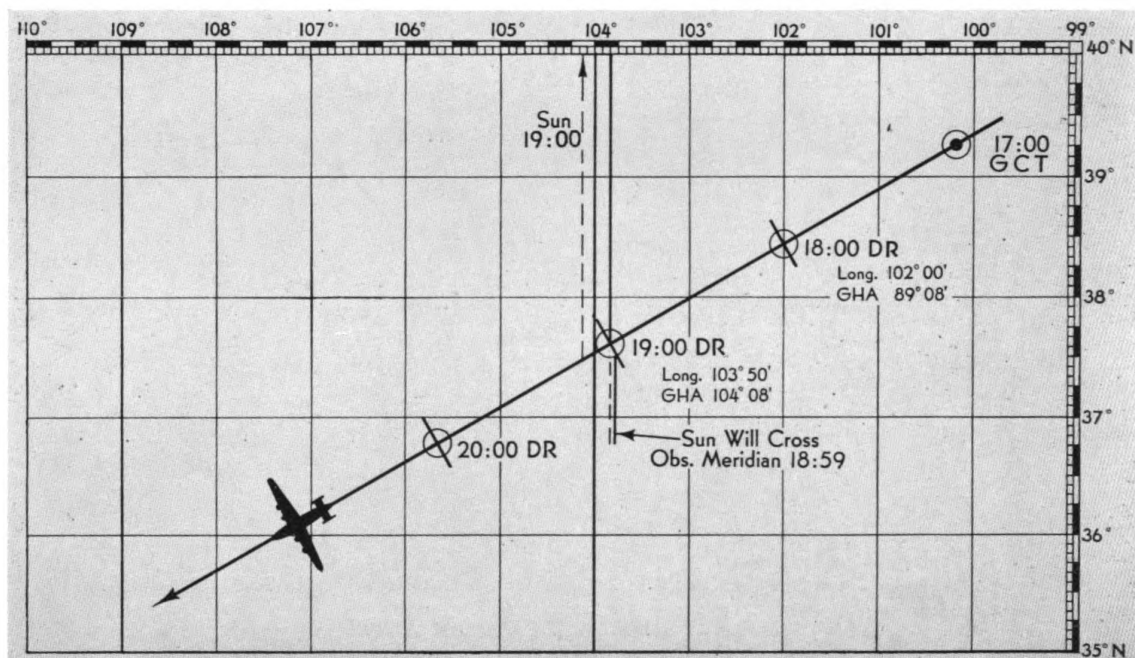


FIG. 156—FINDING TIME OF MERIDIAN TRANSIT

His course is 240° true and ground speed 102 knots.

Referring to the Air Almanac, he finds the GHA to be $89^\circ 08'$ for $18^{\text{h}}00^{\text{m}}$ GCT. His DR longitude, however, he computes $102^\circ 00'$, hence the sun will still not have overtaken him by this time. Accordingly, the navigator again computes his DR longitude, this time for $19^{\text{h}}00^{\text{m}}$ GCT, and finds it to be $103^\circ 50'$. He then notes the tabulated GHA as $104^\circ 08'$, hence the sun will by this time ($19^{\text{h}}00^{\text{m}}$ GCT) be $18'$ past his meridian, or approximately one minute of time.

Note: A celestial body advances in longitude to the West

15° in one hour
1° in four minutes
 $\frac{1}{4}^\circ$, or 15' in one minute

Therefore, he concludes that the time of meridian transit will be very close to $18^{\text{h}}59^{\text{m}}$ GCT.

Rate of Closure Method—The time of meridian transit also may be worked as a *rate of closure* problem (rate at which celestial body is overtaking the aircraft). The procedure is as follows:

a. On an even hour before meridian transit, step off *one hour of ground speed along the course*.

b. Next, measure the aircraft's rate of departure, that is, the number of degrees and minutes of longitude which the aircraft crosses in one hour.

c. Since a celestial body travels westward, traversing longitude at the rate of 15° per hour, find the rate of closure (R of C), or rate at which the sun is overtaking the aircraft, by subtracting the aircraft's rate of departure from 15° .

d. The LHA of the celestial body at the time of departure is the distance in degrees and minutes of arc which the body must close in order to overtake the aircraft. Therefore,

$$\frac{\text{LHA}}{\text{R of C}} + \text{GCT departure} = \text{time of meridian transit.}$$

Example:

Using the same data as in the previous example (Figure 156), the navigator determines the aircraft's rate of departure to be $1^\circ 53'$ of longitude per hour. Therefore, 15° less $1^\circ 53' = 13^\circ 07'$ R of C. (Plane flying away from sun causes sun to close at a rate less than 15° per hour).

SPECIAL PROCEDURES

The navigator then finds the interval of closure (LHA at 17^h00^m GCT) as follows:

$$\begin{array}{rcl} \text{GCT } 17^{\text{h}}00^{\text{m}} & \text{GHA} = & 74^{\circ}09' \text{W} \\ & \text{Longitude} = & 100^{\circ}10' \text{W} \\ & \text{LHA} = & 25^{\circ}01' \end{array}$$

Hence,

$$\begin{array}{l} \frac{25^{\circ}01'}{13^{\circ}07'} = \frac{25.01^{\circ}}{13.11^{\circ}} = 1.91^{\text{h}} = 1^{\text{h}}56^{\text{m}} \\ \text{or, } 17^{\text{h}}00^{\text{m}} + 1^{\text{h}}56^{\text{m}} = 18^{\text{h}}56^{\text{m}} \text{ time of meridian transit.} \end{array}$$

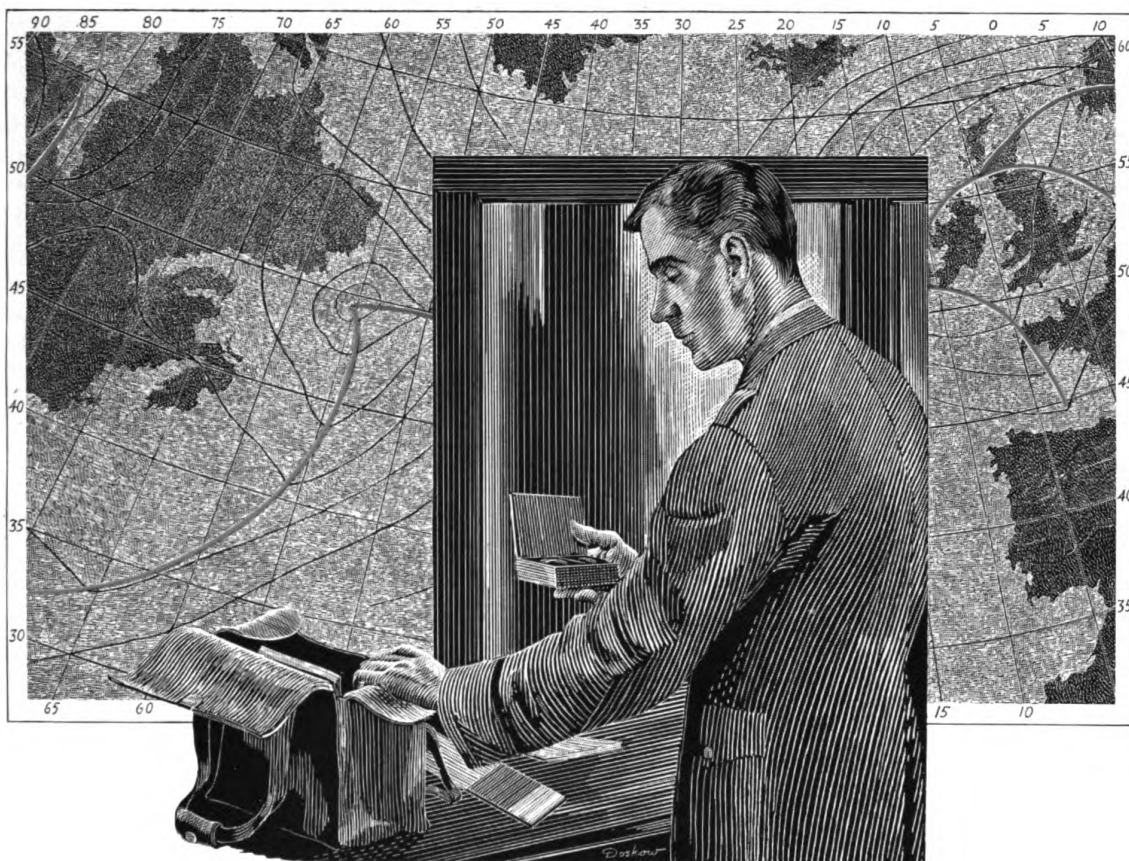


PROBLEM WORK NO. 31

LATITUDE BY MERIDIAN ALTITUDE

(Find latitude by meridian altitude method.)

No.	CELESTIAL BODY	DATE	MTr (GCT)	H ₀	BEARING	LATITUDE
1	SUN	5- 1-43	01:50	78°48'	S	
2	SUN	5-15-43	04:32	74°31'	N	
3	SUN	5-10-43	19:45	61°36'	S	
4	SUN	5-20-43	14:21	56°47'	N	
5	SUN	5- 5-43	22:00	49°48'	N	
6	JUPITER	5- 5-43	02:40	68°31'	S	
7	VEGA	5-10-43	11:00	81°09'	N	
8	MOON	5-20-43	08:14	41°01'	S	
9	ARCTURUS	5- 5-43	07:11	65°29'	N	
10	ANTARES	5-15-43	08:50	28°17'	S	
11	MOON	5-15-43	04:15	62°07'	S	
12	SPICA	5- 5-43	05:45	55°29'	S	
13	REGULUS	5- 1-43	03:45	79°26'	S	
14	ALTAIR	5-20-43	10:31	66°00'	S	
15	ALKAID	5-15-43	07:24	69°43'	N	
16	NUNKI	5- 1-43	15:14	76°13'	S	
17	MIZAR	5-20-43	07:58	76°38'	N	
18	DENEbola	5-10-43	05:00	48°47'	N	
19	RASALAGUE	5-15-43	12:13	71°11'	N	
20	ALPHECCA	5- 5-43	09:30	62°47'	N	



☆ 12 ☆

YOU NAVIGATE TO HONOLULU

THE following notes, experiences and flight data are based upon several actual trips made by the writer. Engine-performance curves, graphs and other factors have been altered, and essential data has been recomputed and brought up to date.

Regardless of these changes, the conditions of flight, information cited and general procedure followed are essentially the same as will apply to any long-range ocean flight and may, therefore, be considered accurate examples.

It is hoped the following description of an actual flight will furnish a more workable con-

cept of the practical application of aerial navigation principles and methods described in earlier chapters.

THE FLIGHT

A phone call from the dispatch office on a May afternoon advises that you will be the navigator on a PBY scheduled for delivery to Honolulu the next day.

Next morning you are up early gathering your gear. By nine o'clock you are busy in the Airway's Navigation Office, checking over all equipment needed for the flight. You have, of course, kept it constantly ready, stowed in a

sturdy briefcase for compact handling, but you know it always pays to make a final check. You remember well the navigator who failed to take along a current Air Almanac. Two hundred miles at sea he discovered, much to his discomfort, that the date of his Almanac had expired. Fortunately, another aircraft was nearby, on the same flight, and so by means of inter-plane radio communication, the navigator of the other aircraft dictated enough data from a current Almanac to enable him to compute celestial fixes and so continue enroute.

Equipment Check—You check your equipment to make certain the following items are included:

- 2 *Air chronometers* (very accurate watches).
- 1 *Aircraft octant* (including spare batteries and bulbs).
- 1 *Air Almanac* (correct date).
- 1 *Set H.O. 214 tables* (one volume for each 10° of latitude to be flown).
- 1 *Weems plotter* (for drawing courses and lines of position).
- 1 *Pair 6" dividers*.
- 6 *Pencils* (sharpened).
- 1 *Scratch pad*.
- 1 *Notebook* (for recording celestial sights and observations).
- 1 *Flashlight*.
- 1 *Roll of masking tape* (to hold charts down).
- 1 *Operations Manual* (containing the graphs and engineering data concerning engine performance and fuel consumption, and navigation aids such as sunrise and sunset tables, courses and distances*, radio aids*, radio station frequencies and identification signals*, schedule of time ticks, radio weather codes* and other essential information).
- 1 *Flight Manual** (containing all detailed information needed regarding check points such as islands and reefs, landing areas, air bases and other facilities along the intended or alternate routes).
- 1 *Set of charts* (including plotting and detail charts of all islands and landfalls along the proposed route, including possible alternates).

*These items are restricted during war, and can be obtained by the captain from Army or Navy officials only, immediately prior to departure.

Note: A plotting chart is most practical when its scale is small ($\frac{3}{4}$ inch to 1 inch equal to 1° of longitude at the equator), because this small scale permits plotting the course for an average point-to-point flight (usually not more than 2000 miles), whereas a larger scale would necessitate changing charts. Furthermore, a small scale chart provides a clearer picture of the aircraft's track, ground speed and drift, and fixes plotted are accurate within a few miles—a negligible error in long-range flying.

Chronometer Check—Before placing the equipment aboard your aircraft, you first wind your chronometers and set them to correct Greenwich civil time (GCT). They are set to coincide with the master watch, kept in the navigation office, and as an additional check, as soon as possible after take-off, the radio operator will pick up a time tick so you may make absolutely certain your watches are set correctly. This is very important, as four seconds of chronometer error will result in an error of one minute of longitude (one mile at the equator).

Later in the morning you board the PBY and arrange your equipment for the flight. With masking tape, you fasten the chart you are going to use to the navigator's table. The rest of the charts you stow in a drawer, and the navigation books are put in handy places for use when needed. You then set the aircraft's watches to GCT and altimeters to zero.

Pre-forecast Check—The aircraft being delivered is a Consolidated Vultee Catalina type flying boat. Take-off has been scheduled for late in the afternoon in order to arrive in Honolulu sometime after sunrise tomorrow. You are not yet certain that the flight will be made today because if the weather forecast is not favorable, the flight will have to be cancelled out. Therefore, you obtain a preliminary forecast which gives you the weather and anticipated flying time, both of which look promising. So you prepare for pre-flight check.

The forecast, to be described in more detail later on, is the weather report obtained from the meteorologist (Weather Bureau in peacetime and the Army or Navy department during war). The forecast includes predicted

weather, winds, clouds and fronts along the proposed route. Applying the force and direction of the wind to the true air speed of the aircraft, an estimate is then made of the average ground speed, and total number of flying hours. If head winds were predicted of such force as to result in the estimated total flying time being very nearly equal to the aircraft's total available fuel hours, the flight would be cancelled out, pending more favorable winds. For a Catalina, 18 to 20 hours is considered a good forecast; over 20 hours, a long forecast.

Pre-flight Check— By 11:00 A. M. the crew is ready to take your Catalina for its final check flight. You still won't leave if it isn't in perfect flying condition. It has had 12 hours of shakedown (flight test) and should be in good shape. However, this is your last opportunity to uncover faults, and if any remain, to have them corrected before departure.

With the crew on board, the captain takes the ship off the water and climbs to 5,000 feet where he levels off. Captain and co-pilot then conduct a series of tests to see if any faults can be shaken out—checking flying performance on first one engine and then the other, with both engines set at various manifold pressures (M.P.) and revolutions per minute (r.p.m.). Magnetos, de-icers, propeller-feathering mechanism and other devices are tested. It looks as though you have a "good airplane," as everything is running smoothly. These final shakedowns are important, as minor flaws revealed in these flights might develop into major failures in long-range flying, resulting in possible disaster for ship and crew.

Radio Compass Check in Flight— When the captain is satisfied that the aircraft is in good operating condition, you then check the radio direction finder loop for calibration error. This is not difficult when relative bearings (R.B.) are computed as outlined in the following procedure.

Fulfilling one of your duties as a navigator you select a small landmark—the South tip of South Coronado Island—and measure the correct magnetic heading between it and radio station KFSD, San Diego. Next, the magnetic headings to fly are computed, as well as the correct relative bearings for each 30° interval

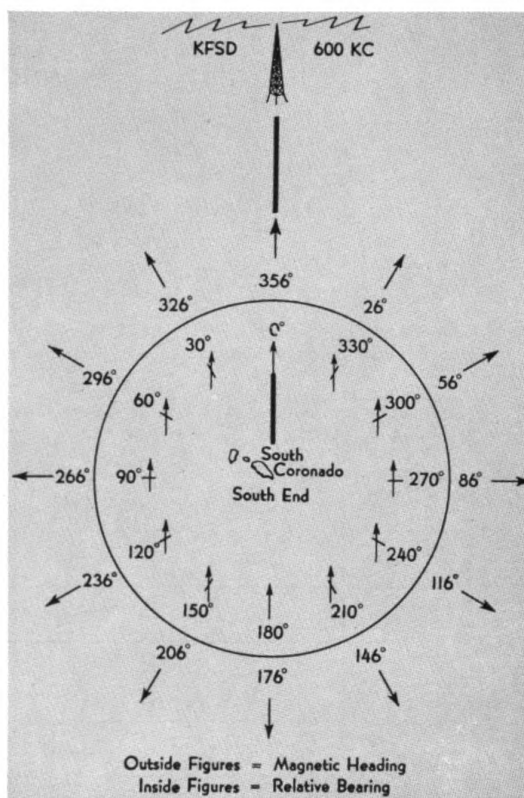


FIG. 157—RADIO COMPASS CHECK

from the first magnetic heading to the radio station, as shown in Figure 157.

To assist you in checking the radio direction finder, your captain steadies the PBY on each of the headings indicated on the diagram. At the instant the ship passes over the selected landmark, on each of the headings, he signals to the radio operator who then takes a relative bearing on KFSD while you double check the compass heading. Knowing the relative bearing, you then quickly determine the amount of calibration error, which equals the difference between the radio bearing taken and the correct relative bearing indicated on the diagram (Figure 157).

With the plane heading 266°, the radio operator took a relative bearing of 96°. By reference to the actual pre-computed bearing, this indicated at once that the direction finder was reading too large a value by 6°. Therefore, the calibration correction on this heading is minus 6°.

AMERICAN AIR NAVIGATOR

WEATHER FORECAST

Date 5-4-43

From San Diego To Honolulu Course 252° Plane Catalina GCT 2250

		ZONE 1	ZONE 2	ZONE 3	ZONE 4	ZONE 5
WEATHER		Clear	Clear	Ptly Cldy	Cldy	Ptly Shrs
CLOUDS	Type	Sctd	Sctd	.4 Sc	.7 Sc Cu	.6 Sc Cu
	Ceil.	1000'	1500'	1500'	2000'	2000'
	Tops	2000'	3500'	6500'	7500'	7000'
	Upper	X	X	Sctd 30,000'	Sctd 14,000'	Sctd 14,000'
WIND	1000'	310/12	320/12	X	X	X
	5000'	330/14	360/14	40/10	170/10	40/12
	7000'	320/12	360/14	120/10	310/14	10/7
	10,000'	X	X	X	X	X

No pronounced Fronts along course.

San Diego—2230 to 0230 GCT, no lower clouds, no upper clouds, surface winds 290°/8 kts., visibility 5/10 smoky, sea smooth.

Terminal: Pearl Harbor—1700 to 1900 GCT May 5th, partly showers, .6 Sc Cu at 2000', tops at 7500', high clouds sctd at 14,000', surface winds 40°/12, visibility unlimited.

Alternate: Hilo—1600 to 1800 GCT May 5th, party showers, 6 Sc Cu at 1500', tops at 8500', high clouds sctd at 14,000', surface winds 40°/14, visibility unlimited.

Remarks—Flight Time Analysis at 7000', 19.8 hours to Pearl Harbor, 18.7 hours to Hilo.

Recommend a drift southward in early portion of flight since winds are more favorable to south of course.

Signed: Weather Officer

FIG. 158

YOU NAVIGATE TO HONOLULU

Calibrating the radio direction finder is an important procedure, for if you can take radio bearings that are reasonably accurate, you will be able to use them to check celestial fixes, or, under proper conditions, to employ such bearings alone to fix the ship's position. You should be able, if necessary, to direct your aircraft to its destination without radio. However, pilots who are accustomed to radio and contact flying over land begin to feel a little worried after flying for ten or fifteen hours without seeing anything but water. At such times, a radio bearing which checks your celestial fix is an added comfort to the crew—and even to yourself. Such bearings are a comfort to the navigator because occasionally a wind will blow your aircraft into an unbelievable position, and though you take celestial sights as often as possible in order to catch these sudden shifts of wind, it is not unusual to be blown 50 miles off course between fixes. You know that your celestial fixes are usually accurate. Nevertheless, when both a celestial fix and a radio bearing check, even if the check establishes your position as off course, everyone on board feels more confidence in your ability.

By 12:00 noon, your check flight is over. Engines, aircraft and radio are in perfect condition; the crew is satisfied they have a "good airplane."

The engineer has elected himself food supply officer and is off to town to obtain a supply of provisions. Since there is nothing

left to check on the aircraft, the crew is temporarily at liberty.

Final Forecast—By 2:00 P. M. you would usually be back at the Airway's office, but due to the war you check in at the Naval Briefing Office for final briefing and weather forecast. Briefing means you'll be given the latest confidential data bearing on the proposed route. This includes instructions for receiving and sending messages and weather reports via radio, information on air base facilities, let-down and approach procedures and other restricted information. You are shown the latest weather map, and given a copy of the forecast (Figure 158) which estimates your flight time as 19.8 hours. However, you will recalculate it yourself and then plot a flight graph of time and fuel consumption. Use of this graph in flight will enable you to determine very quickly if the aircraft is flying according to the estimated forecast. It will indicate whether engines are using too much fuel and, if so, whether you can make your destination, or whether you will have to turn back. The graph also indicates the point of no return (PN), or "splash point," the times of sunrise and sunset, and location of weather fronts. Sometimes, for multi-engine aircraft, when engine data is available, you plot a second curve on the graph to show the PN in the event one of the engines fails.

Note: This weather forecast was based upon the latest weather maps, which, in turn,

ZONE BOUNDARIES—SAN DIEGO TO HONOLULU—COURSE 252°				
ZONE	BOUNDARIES	DISTANCE (Nautical Miles)	COURSES	
			MAGNETIC	TRUE
1	San Diego to 120°W	153	237°	252°
2	120°W to 130°W	540	237°	252°
3	130°W to 140°W	555	237°	252°
4	140°W to 150°W	577	240°	252°
5	150°W to PCY (Pearl Harbor, Honolulu)	494	240°	252°
	Total Distance	2319		

FIG. 159—COURSE ZONES

were based upon all available weather reports received from ships, aircraft and weather stations along the proposed route. Therefore, in order that following flights also may have the benefit of a good forecast, the navigator should carefully and conscientiously observe weather encountered in flight, and report correct weather information by radio.

Course Zones—Looking at the forecast again, note that it is divided into five zones. The report found below each zone provides weather and wind predictions for that particular zone. Weather forecasting, at best, is only an estimate, and since weather reports over vast water areas are few and far between, definite wind predictions cannot be given for every mile of the course. As a result, the weather forecaster has divided the various air routes into several basic segments called zones (see Figure 159). Various zone boundaries are employed. Sometimes the zones are an even 5° of longitude in width; on the other hand they may cover 10° of longitude, depending upon the purposes and procedures of those originating the forecast. However, the general practice is to allow the amount and accuracy of available weather information in each area to determine the width of the zone. Where available weather reports are numerous and

accurate, the area covered by each zone is relatively small. If only a few reports are available, the area included is greater. Usually the weather forecast states the boundaries and areas of each zone. However, zones are practically of standard dimensions, and the Operations Manual which is part of your equipment furnishes this information for your proposed route.

Flight Time Analysis—Your next job as navigator is to analyze the forecast. Analyzing the forecast involves applying the wind predicted for each forecast altitude to pre-determined true air speeds and course, thus gaining an estimate of expected ground speeds and flying time in each zone. When the information is arranged in some such form as illustrated in Figure 161, it is called the *Flight Time Analysis*.

The predetermined true air speeds, which are computed for standard temperature, can be had by inspection of air speed curves, (Figure 160) contained in your Operations Manual.*

*The Operations Manual contains a number of long-range cruising curves which have been computed by the aircraft manufacturer after years of research. These curves indicate engineering and flight data on long-range performance, fuel consumption, engine power settings (M.P. and r.p.m.), etc., for the aircraft under various conditions of gross weight, load and altitude. Of particular interest to the navigator are the air speed curves (Figure 160), and the fuel consumption curves (Figure

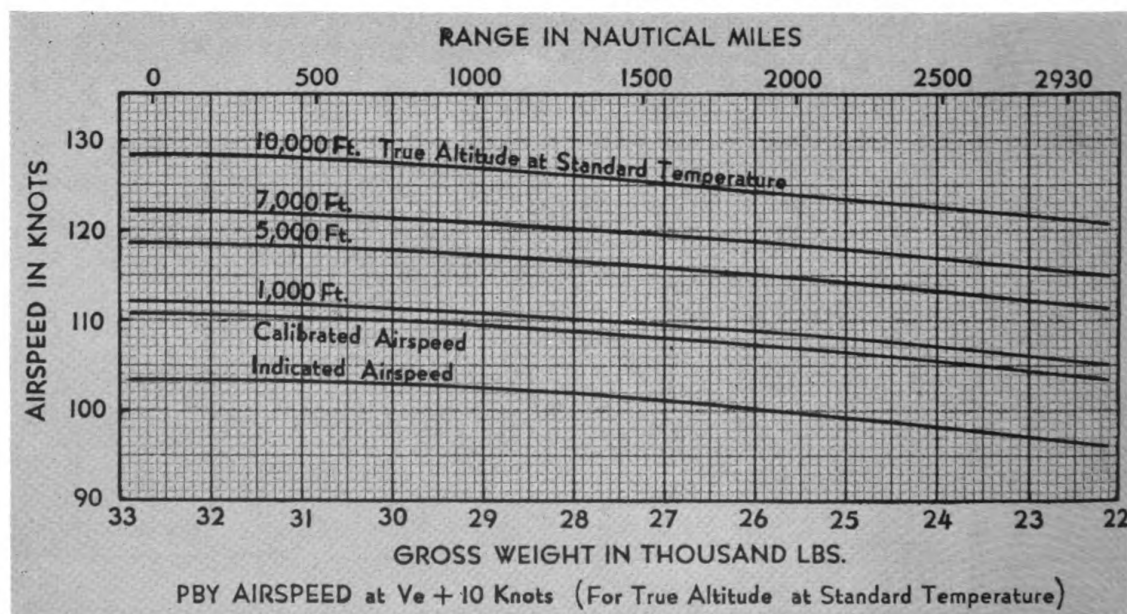
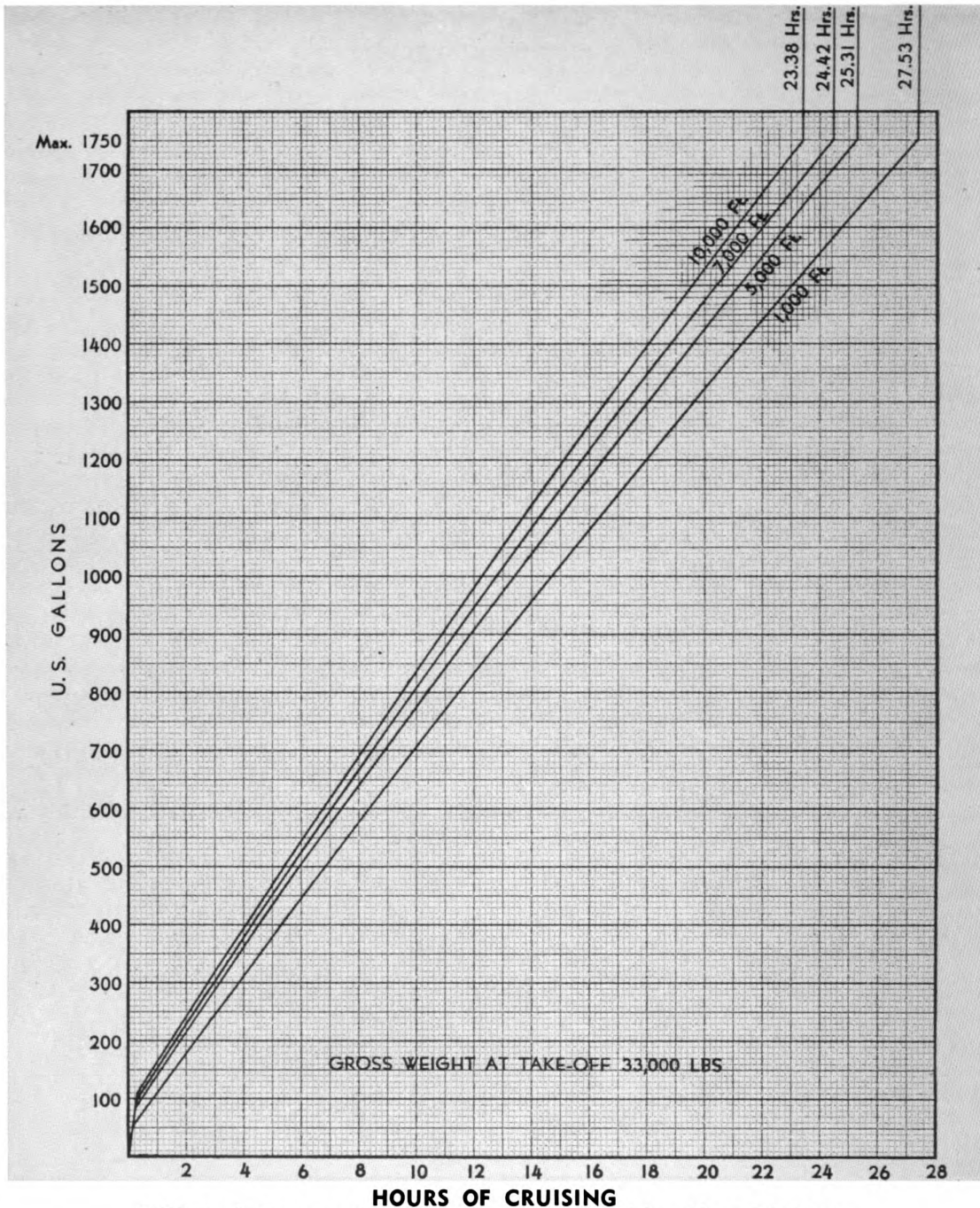


FIG. 160—AIR SPEED CURVES

YOU NAVIGATE TO HONOLULU

FLIGHT TIME ANALYSIS																
		ZONE 1			ZONE 2			ZONE 3			ZONE 4			ZONE 5		
COURSE		252			252			252			252			252		
ZONE DISTANCE		153			540			555			577			494		
TOTAL DISTANCE		153			693			1248			1825			2319		
		TAS	G.S.	TIME	TAS	G.S.	TIME	TAS	G.S.	TIME	TAS	G.S.	TIME	TAS	G.S.	TIME
ALTITUDE	1000'	112	105	1:28	112	107	5:03	111								
	5000'	118	115	1:20	118	122	4:26	117	126	4:25	116	114	5:04	115	125	3:57
	7000'	122	117	1:19	122	126	4:17	121	128	4:20	119	111	5:12	118	121	4:05
	10,000'															
FLIGHT PLAN																
ALTITUDE		7000'			7000'			7000'			7000'			7000'		
ZONE TIME		1:19			4:17			4:20			5:12			4:05		
TOTAL TIME		1:19			5:36			9:56			15:08			19:13		
GAS USED		175			500			805			1160			1425		
GAS REMAIN														325		
<p>RETURN <u>2</u> ENGINES COURSE <u>72°</u></p> <p>ALTITUDE: <u>7000'</u> TOTAL AVAILABLE FUEL HOURS: <u>24.42 hrs. or 24h.25m.</u></p>																
ZONE	T.A.S.	G.S.	ZONE TIME	TOTAL ZONE TIME	TOTAL FUEL HRS. MINUS TOTAL ZONE TIME	FUEL USED										
1	115	119	1:17	1:17	23:08	1670										
2	116	111	4:52	6:09	18:16	1365										
3	117	110	5:03	11:12	13:13	1030										

FIG. 161



PBY Fuel Consumption For Long Range Cruising $V_e + 10$ Knots

FIG. 162—FUEL CONSUMPTION CURVES

162), as these are necessary for long-range forecasting. Notice in Figures 160 and 162 that these curves are for $V_e + 10$ knots performance. V_e indicates the effective velocity (indicated air speed) at which the aircraft would have to be flown in order to make good *maximum flying hours*. V_e curves are used only for *endurance flights* be-

cause it has been proven that when there is a wind blowing, which is usually the case, *greater range* can be attained by increasing the power on the engines slightly in order to indicate an air speed 10 knots faster than V_e , hence $V_e + 10$ knots. For this reason $V_e + 10$ knots curves are generally used for all long-range flights.

Flight Plan—You then carefully analyze the Flight Time Analysis to ascertain the most desirable cruising altitude, and begin to make out the Flight Plan (Figure 161). Normally, it is desirable to select that cruising altitude which will make possible the shortest flying time; however clouds, fronts, temperature and icing level may make it advisable to select a "slower" altitude, where safer, better flying conditions prevail. For today's Flight Plan you use 7000', because at this altitude you should be able to keep above the clouds and stay in clear weather all the way. Totalling the zone times gives you 19^h13^m, indicating that you actually have a 19.2 hour forecast and not 19.8 as stated by the meteorologist.

In order to complete the Flight Plan you refer to the $V_e + 10$ knots fuel consumption curves (Figure 162), also contained in the Operations Manual, and find, using the 7000' curve, the amount of fuel which would be consumed by the end of each zone. The flight plan now indicates that if the flight forecast is accurate, your PBV will use a total of 1425 gallons of fuel and have in reserve 325 gallons on arrival at Honolulu.

Return Flight Plan—A plan for the return flight also is prepared because this information is to be plotted on a flight graph (Figure 163) which provides an ideal method of establishing the amount of reserve fuel needed for safe operation, as well as the radius of action and point of no return.

Several steps are necessary in preparing the Return Flight Plan (Figure 161).

First, you re-compute the Flight Analysis for the first half of the flight, using:

1. Reciprocal course.
2. The forecast winds (Figure 158) at the altitude most favorable for the return course.
3. Pre-determined true air speeds (Figure 160).*

Second, you total the zone times and find the length of time required to return to base

from the end of zone one, zone two, and zone three.

Third, obtain the maximum flying time from the fuel consumption curve (Figure 162). In this case, your aircraft can fly 24.42 hours or 24^h25^m, as this is the time indicated at the end of the 7000' curve on which your forecast is based.

Fourth, subtracting the total zone time (time required to return to base from each zone) from maximum fuel hours, you find the total possible hours you could fly in order to reach the end of each zone, and still be able to reach your base.

Finally, using total flying hours, obtain from the fuel consumption curve (Figure 162) the total fuel that could be used to reach the end of each zone.

Flight Graph—With the flight plans completed, the final step is to plot the computed data in the form of a flight graph (called by many air navigators the "Howgozit"), which will permit a constant, clear check during the flight. The procedure for making the Flight Graph (Figure 163) is as follows:

- a. On a sheet of graph paper of convenient scale, establish a *course distance scale* along the lower edge.
- b. Mark off the *forecast zones* as shown.
- c. Establish a *fuel consumption scale* along the left edge.
- d. On the zone lines, mark the total fuel used at the end of each zone (as computed in the Flight Plan) and draw in the *fuel vs. miles* curve. This curve provides the most important information found on the graph, as it indicates the distance flown by the aircraft with relation to fuel consumed.
- e. Establish a *time scale* along right edge. The scale, on this preliminary layout, should indicate only hourly intervals as the GCT take-off time will have to be known before the actual time scale can be established.

Note that in this instance the time scale is started above the fuel scale, to keep the fuel

*Notice in Figure 160 that the air speed decreases as the range increases. This increase occurs because the aircraft is continuously losing weight as fuel is consumed, and as the weight decreases, flight efficiency increases. That is, the air speed and engine power can be reduced, resulting in less fuel consumption, while the aircraft will

continue to maintain the same cruising performance ($V_e + 10$ knots). Therefore, in order to obtain correct information from the air speed curves (Figure 160), and from the fuel consumption curves (Figure 162), you assume that the aircraft already has flown the first half of its maximum range prior to preparation of the Return Flight Plan.

AMERICAN AIR NAVIGATOR

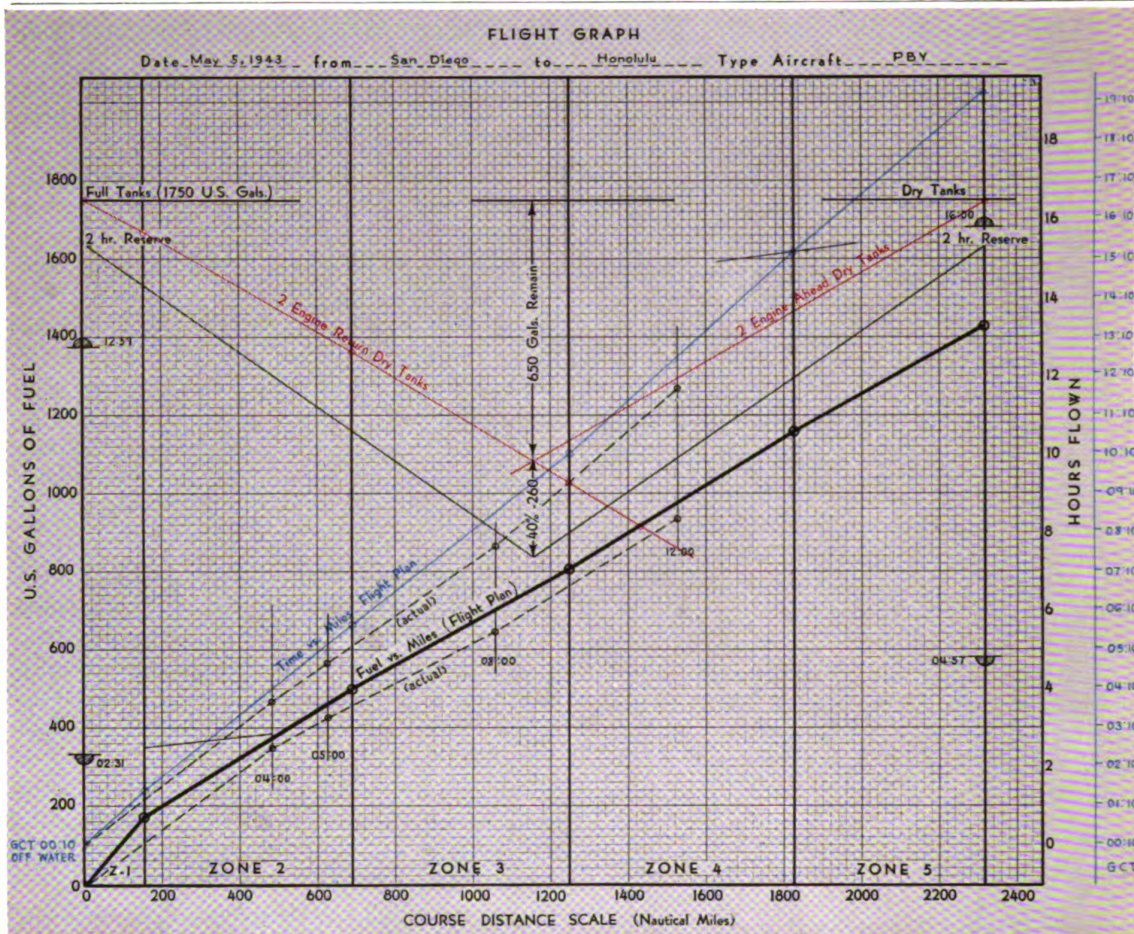


FIG. 163—FLIGHT GRAPH

curve and time curve (to be drawn) separated, thus permitting easier reading.

f. On the zone lines mark the total forecast time to the end of each zone (Flight Plan) and draw in the *time vs. miles* curve.

g. On the zone lines mark the total return fuel used to the end of each return zone (Return Flight Plan), and commencing at point of full tanks, draw in the *two-engine return dry tanks* curve. The intersection of this curve, extended, with the fuel vs. miles curve locates the PN.

h. From the point of dry tanks (at destination), draw a line parallel to the fuel vs. miles curve. This line is the *two engine ahead dry tanks* curve. The point at which the two dry tanks curves intersect represents the absolute maximum amount of fuel the aircraft can consume and still return or go ahead to destination.

i. Determine amount of fuel remaining at point where dry tanks curves intersect, and from this point, at a distance equal to 40% of the remaining fuel, locate a second point vertically beneath the point of intersection.

j. Obtain from fuel consumption curve (Figure 162) amount of fuel required for final two hours cruising. Mark this amount on Flight Graph below *full tanks* and *empty tanks* points and draw in *40% to 2 hours reserve ahead* and *return* curves. These curves represent safe flying curves, and any position below these curves would be considered safe.

k. In airline operations it is especially important also to plot on the Flight Graph the maximum amount of reserve fuel required on arrival at destination. As an example, four hours reserve fuel is ample safe allowance for

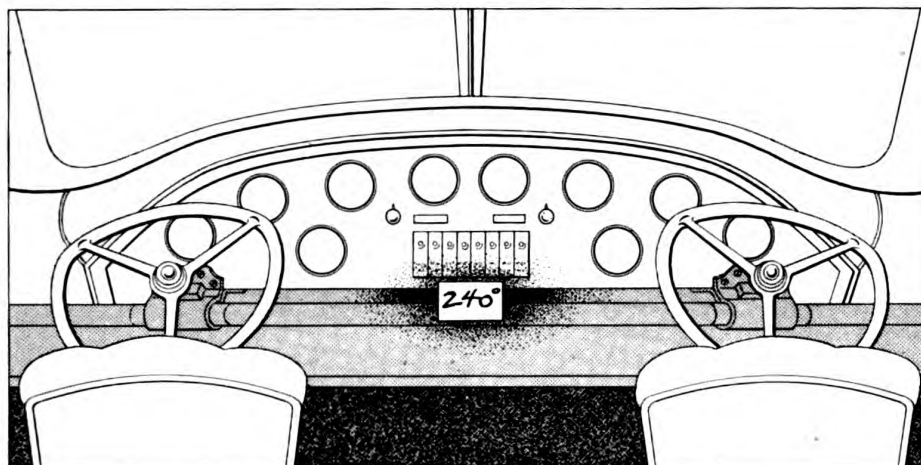


FIG. 164—HEADING MEMORANDUM FOR PILOT

your PBV. Plotting this amount on the Flight Graph you see from the relationship of this amount to the fuel vs. miles curve that you are carrying 75 gallons of fuel over and above the required reserve. Converted into cargo weight, 75 gallons of fuel is equivalent to approximately 450 pounds. In other words, the Flight Graph proves it would be safe to carry 75 gallons less fuel, thus allowing the aircraft to carry two or three extra passengers. It is easy to see that similar additional payloads on all flights would, in the course of a year, amount to many thousands of dollars additional income for the airline.

Note: The average time required to make up one of these flight forecasts and flight graphs is thirty to forty minutes, and its value is inestimable. An airline could not afford to neglect it, and its correct use in flight makes it practically impossible for an aircraft to run out of fuel except through mechanical failure.

Departure Clearance—3:30 P. M.—The co-pilot, radio operator and engineer are at the aircraft waiting for take-off. You and the captain are at the Airway's office receiving last minute instructions, customs papers, bill of health, clearance papers, cargo and mail manifests.

The flight clearance approved, you are on your way. With wide-open throttles your plane roars down the bay, rapidly picking up speed—70 . . . 80 . . . 90 knots. Gradually the spray stops flying past the windshield. The next

moment the ship is clear of the water, and the earth appears to drop away.

Off the Water—May 5, 1943—00:10 GCT. In the excitement of take-off you almost forgot to read the time!

The captain's first question will be, "What is our heading?" You have the answer ready for him, having worked it out in advance. (This and other vector problems will be solved on the computer to save time).

252°	True Course
—15° E	Variation
237°	Magnetic Course
+5° L	Drift (from forecast
	wind—320°/12 knots)
242°	Magnetic Heading
—2° E	Deviation
240°	Compass Heading to fly

You have found from experience that it is a good rule to write the heading on a slip of paper and stick it on the controls where the captain and co-pilot can see it easily (Figure 164).

By now San Diego is far behind. The plane is climbing steadily, higher and higher into the sky. Everyone is in excellent spirits.

00:29 GCT—Nineteen minutes from take-off—your aircraft is now leveling off at 7000' as per forecast. The captain is pulling back on the throttles, adjusting them and the propeller pitch so that the manifold pressure and revolutions per minute are for $V_e + 10$ cruising range, as per engine performance graph.

The reduced power now makes the engines relatively quiet. The engineer has changed the fuel mixture from full rich to automatic lean. The radio operator is sending in the departure time to the shore watch station.

Flight routine has begun.

Navigator's Flight Report—Your first job is to start filling out the Navigator's Flight Report (see example, Figure 166) to which you will refer each hour, filling in the hourly record. From this report you will check courses, positions, and make up the radio reports. After the flight is over it will be given to the Airway's office so they can keep a record of the flight.

In order to save time, the type of position is given a numerical designation as follows:

1. Dead Reckoning Position.
2. Approximate Fix (Single position line and DR, or radio bearings).
3. Fix (Celestial or Visual)

Chart Preparation—Before starting to navigate it is advisable, in order to save time, to draw in the course line marked off in units of 100 nautical miles, and indicate the zone boundaries and forecast winds (Figure 165).

First Position Report—You are now ready to find the 01:00 GCT position. It is a common belief that an aerial navigator has to be extremely fast in his work. While it is true that the ability to calculate rapidly is certainly an asset to the navigator, accuracy and common reasoning are far more important. Navigation is not an exact science, particularly aerial navigation, but it does give an approximate position which is sufficiently accurate with relation to the aircraft's speed for all practical purposes. Bearing in mind the limitations of aerial navigation, it is the navigator's job to know which of the various navigational methods, or combination of methods, will enable him to most accurately determine the aircraft's position with the least effort.

The sun is setting in the West, and if an observation of the sun were taken, it would give you a single line of position which would serve as a fair speed check. However, during the climb you kept track of the indicated air speed, which averaged 98 knots. The indicated air speed now is 103 knots. Since the forecast wind force and direction usually is quite accurate for the first zone, and since you are less

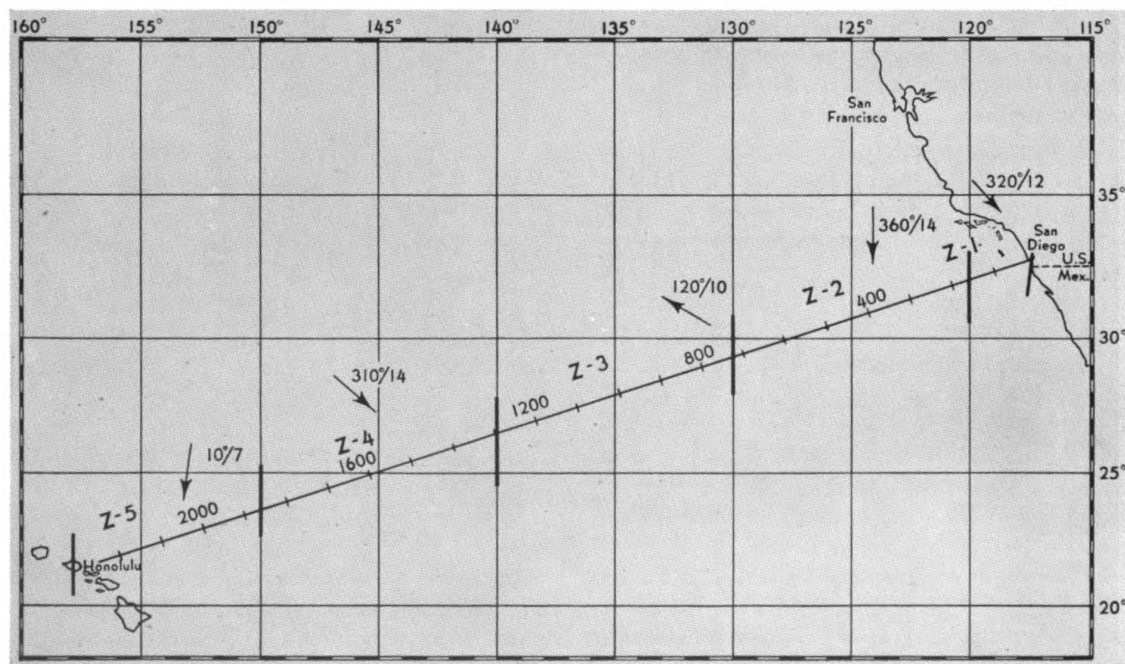


FIG. 165—CHART PREPARATION

YOU NAVIGATE TO HONOLULU

NAVIGATOR'S FLIGHT REPORT																						
TRIP NO.			DATE			CAPTAIN			CO-PILOT													
ORIGIN			DESTINATION			NAVIGATOR			RADIOMAN													
PLANE NO.			TYPE			RADIO CALL			ENGINEER													
TIME G.C.T.	POSITION		AIR SPEED		ALTITUDE °C.	Course or Track	DRIFT	True Head	VARIATION	Comp Head	WIND		GROUND SPEED	TIME ELAPSED	DISTANCE		GASOLINE		WEIGHT	M.P. R.P.M.		
	Lat.	Long.	TYPE	IAS							CAS	TAS			Dir.	Vel.	Flown	To Go			Rate Cons.	Used
00:10	OFF	WATER	34N 150W			252	5L	257	15E	2E	240	310	12	104	0:19	33	2286		1750	33000		
00:29	END	OF CLIMB			7000	252	5L	257	15E	2E	240	320	12	117	0:50	94	2225			280	1980	
01:00	32-15	118-55	1	103	110	252	5L	257	15E	2E	240	320	12	117	0:50	94	2225	120	1630	280	1980	
02:00	31-25	121-10	2	103	110	249	8L	257	15E	2E	240	360	19	127	1:50	220	2099	77	197	1553	280	1980
03:00	30-40	123-30	1	103	110	249	8L	257	15E	2E	240	360	19	127	2:50	350	1969	76	273	1477	280	1980
04:00	29-35	125-55	3	103	110	247	10L	257	15E	2E	240	360	23	128	3:50	485	1834	75	348	1402	309	1270
05:00	28-35	128-40	3	103	110	252	5L	257	15E	2E	240	40	20	139	4:50	630	1689	75	423	1327	270	1950
06:00	28-15	131-10	1	103	110	253	0	253	14E	2E	237	70	16	139	5:50	775	1544	74	497	1253	270	1950
07:00	27-35	133-40	1	102	109	253	0	253	14E	2E	237	70	16	139	6:50	915	1404	74	571	1179	295	1930
08:00	27-15	136-20	3	102	109	252	3R	253	14E	2E	237	100	18	140	7:50	1060	1259	74	645	1105	265	1930
09:00	26-35	138-50	1	102	109	252	0	252	14E	2E	236	70	17	140	8:50	1195	1124	73	718	1032	265	1930
10:00	26-15	140-45	3	102	109	255	3R	252	14E	2E	236	130	07	126	9:50	1300	1019	73	791	959	265	1930
11:00	25-35	142-50	1	102	109	250	1L	251	13E	2E	236	300	05	120	10:50	1420	899	72	863	887	265	1930
12:00	24-45	144-35	3	102	109	248	3L	251	13E	2E	236	280	12	113	11:50	1530	789	71	934	816	265	1930
13:00	24-15	146-40	1	101	108	254	5L	259	12E	2E	245	310	14	116	12:50	1650	669	70	1004	746	265	1930
14:00	23-35	148-45	3	101	108	252	7L	259	12E	2E	245	340	15	122	13:50	1770	549	70	1074	676	265	1880
15:00	22-55	151-00	3	101	108	252	7L	259	12E	2E	245	10	16	128	14:50	1905	414	68	1142	608	260	1880
16:00	22-20	153-15	1	101	108	256	7L	263	12E	2E	249	10	16	128	15:50	2035	284	68	1210	540	260	1880
17:00	21-50	155-25	2	101	108	256	6L	262	11E	2E	249	10	16	128	16:50	2160	159	67	1277	473	260	1880
18:29	ON WATER	HONOLULU												18:09	2319		103	1380	370			

FIG. 166

than an hour out, a DR position will be just as accurate as a celestial line of position, and easier to determine.

Referring to the Performance Manual (Figure 160), the air speed calibration correction is found to be +7 knots. Therefore, the calibrated air speed during the climb was 105 knots, and now is 110 knots. The temperature, read off the instrument panel, is +5°C. Using these calibrated air speeds, and with the computer set at +5°C and at about 3000' for the climb and 7000' for your present altitude, you find the true air speeds flown to be 109 knots on the climb and 123 knots at 7000'.

Applying the forecast wind (Figure 158) to the true air speeds (using the computer) gives the following ground speeds:

- 104 knots—Ground speed during climb
- 118 knots—Ground speed at 7000'

Using this information you compute the distance flown as follows:

- a. 00:10 Off water
- 00:29 End of climb
- 19 minutes elapsed during climb
- b. 19 minutes @ G.S. 104 knots = 33 NM
- c. 00:29 End of climb
- 01:00 Time of position desired
- 31 minutes elapsed since climb
- d. 31 minutes @ G.S. 118 knots = 61 NM
- e. 33 NM
- +61 NM
- 94 NM = Distance flown along course from point of departure to DR position at 01:00 GCT.

Measuring along the course a distance of 94 miles from point of departure, you find that the 01:00 GCT DR position is latitude 32°15' North, longitude 118°55' West (Figure 168).

Fill in the Navigator's Flight Report (Figure 166).

DR Ahead—As it is good practice always to know the airplane's estimated position at least one or two hours in advance, you step off the DR positions for 02:00 and 03:00 o'clock GCT. Both of these positions will be in Zone 2. Until you have a fix to definitely determine the wind, refer back to the forecast wind which is 360°/14 knots. Since the true air speed is



FIG. 167—LEVELED OFF AT 7000'

still 123 knots, and course 252°, you obtain a ground speed of 127 knots and drift of 6° left.

Next, using the mid-latitude scale (Figure 168) opposite the distance to be measured, set the dividers to 127 knots ground speed and step off two points from the 01:00 position. Mark these two points 02:00 DR and 03:00 DR respectively. The actual position of the plane at these two times will probably be south of the course since you are only allowing 5° left drift. However, these are only estimated positions to be revised later when more exact information is found.

01:48 GCT—The radio operator is listening to KFSD, San Diego, so you decide to take a radio bearing which will give you a good check on the drift, as the station lies directly on the course.

A radio bearing taken at the aircraft is the relative bearing of the radio station measured clockwise from the nose of the aircraft. Therefore, at the instant the radio operator takes the bearing of the radio station, he signals and you read the compass heading.

He finds the bearing to be 163°, and the compass heading which you simultaneously noted was 241°. Checking the radio calibration graph you find a correction of +7° for a relative bearing of 163°, making the correct relative bearing 170°.

The next step is to find the true bearing (Figure 169) to which the Mercator correction is then applied in order to obtain the Mercator bearing. This procedure is as follows:

YOU NAVIGATE TO HONOLULU

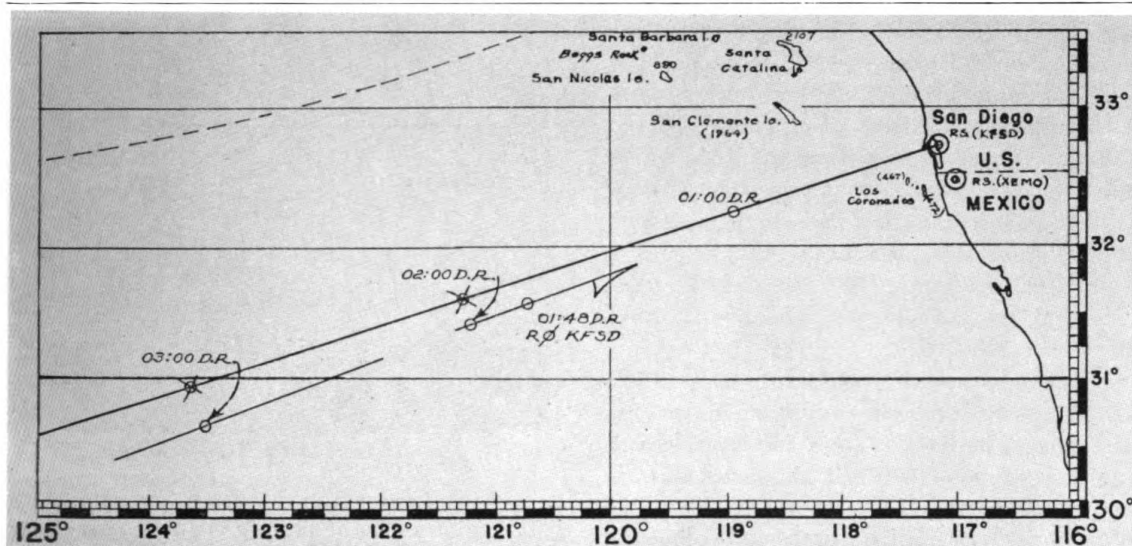


FIG. 168—01:00, 02:00, AND 03:00 GCT POSITIONS

241°	Compass heading
+ 15° E	Variation (found on chart)
256°	Magnetic heading
+ 2° E	Deviation (from deviation card)
258°	True heading
+170°	Correct relative bearing
428°	
-360°	
68°	True (great circle) bearing of KFSD
+ 1°	Mercator correction
69°	Mercator bearing of KFSD from aircraft

The reciprocal of 69° then is plotted on the chart from KFSD, and a DR position established on the bearing line for 01:48, the time the bearing was taken (Figure 168).

You notice that the position thus obtained is about 12 miles south of the course. Assuming that the radio bearing is correct, you conclude that the wind has changed. On measuring the track made good from departure, you find it to be 249°, or 3° less than the track you intended to fly. Therefore, the drift has been 8° left instead of the 5° left you allowed.

To be 12 miles off course is not considered serious in aerial navigation, and since

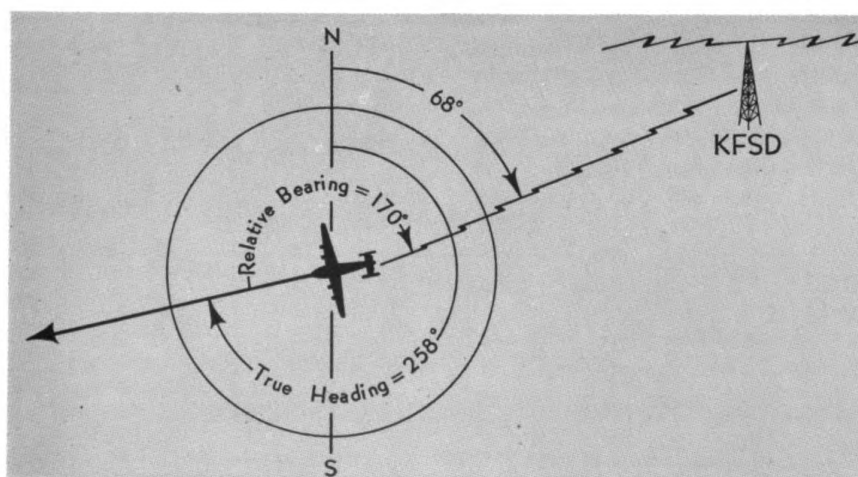


FIG. 169—COMPUTING RADIO BEARING FROM KFSD

the forecast recommended drifting south of the course for more favorable winds, you decide to hold the same heading (240°) until a more definite fix is established. Also to be considered is the forecast wind shift in the next zone which will tend to blow the aircraft back on course.

Advancing the 01:48 position to 02:00 GCT (Figure 168), you fill in the Navigator's Flight Report (Figure 166), entering the forecast wind, track made good and temperature, which now is $+6^\circ\text{C}$.

You now re-establish the 03:00 position and fill in the Navigator's Flight Report for that time. The data (except for latitude and longitude of position) will be identical with that found for the 02:00 report, as there will be no more fixes until the stars are visible.

(Erase the previously estimated 02:00 and 03:00 DR positions).

Using the information on the Navigator's Flight Report sheet, you now prepare the Radio Weather Report using the information obtained at briefing, carefully noting the prevailing weather conditions, especially cloud formations and temperature.

Next, you refer to the Flight Graph (Figure 163) and fill in the GCT scale for the time vs. miles curve, with the GCT time of take-off as starting point. Then you find in the Air Almanac the times of sunrise and sunset for both San Diego and Honolulu, and mark them on the graph as illustrated, using the time and distance scales. Connecting the two sunset and sunrise points with a ruler, you draw short lines where the ruler crosses the time vs. miles curve. These lines indicate the approximate time and position along course at which you may expect sunset and sunrise to occur. They also indicate the duration of night flying. Your curve shows that sunset will occur at about 02:50 GCT.

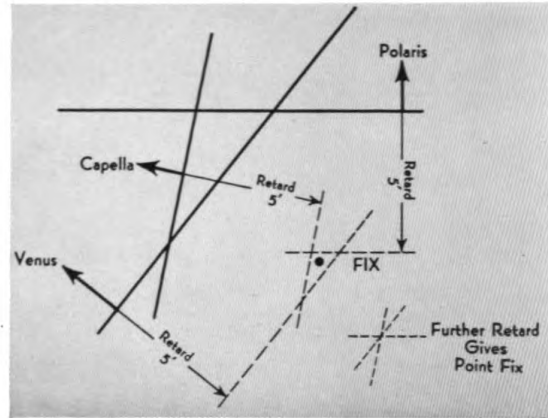


FIG 170—ANALYZING YOUR FIRST FIX

03:00 GCT—The sun has set and the stars are beginning to be plainly visible. You study the sky to determine which of the celestial bodies will afford the best fix. Venus is the first body selected because it is nearly ahead and very bright, thus serving as an excellent speed line sight. (Speed is most important to establish at this time, as it will help check the fuel consumption). Because it is also desirable to know the aircraft's definite position, Polaris, which is easiest to compute, is selected for a latitude and course check. Then a third celestial body, Capella, whose position line will cross the other two at a good angle, is selected to check the other two position lines and make a good three-line fix.

Next, you select a comfortable spot in the aircraft, usually the pilot's or co-pilot's seat, where you can easily observe all three bodies.

Adjusting the octant to the body's approximate altitude, you observe Polaris first, since it is the least brilliant of the three bodies and most difficult to sight in the octant. Then, Capella is shot and, last, the speed sight, Venus.

Their altitudes measured, you proceed to find the resultant lines of position (by H.O. 214 in this case).

POLARIS

GCT 03^h12^m15^s

269°37'

0°35'

GHA Υ = 270°12'W

DR Long. = 124°12'W

LHA Υ = 146° W

H_S = 29°51'

Ref. = -1'

H_O = 29°50'

Corr. LHA Υ = +30'

Latitude = 30°20'N

YOU NAVIGATE TO HONOLULU

CAPELLA

GCT 03^h15^m07^s

	269°37'
	1°17'
	281°53'
GHA	= 552°47'
	—360°
GHA	= 192°47'
Assumed Long.	= 124°47'W
LHA	= 68° W
Assumed Lat.	= 30° N
Dec.	= 45°56'N

Lat. 30°N
Dec. 46°N
Same name

H _s	= 35°39'
Ref.	= —1'
H _O	= 35°38'
H _C	= 35°47.7'
a	= 9.7' away
Δd	= .14' (x 4' = .6')
H	= 35°48.3'
Δd corr.	= —.6'
H _C	= 35°47.7'
Az	= N 52.5°W

VENUS

GCT 03^h18^m07^s

	187°44'
	2°02'
GHA	= 189°46'
Assumed Long.	= 124°46'W
LHA	= 65° W
Assumed Lat.	= 30° N
Dec.	= 25°15'N

Lat. 30°N
Dec. 25°N
Same name

H _s	= 32°47'
Ref.	= —1'
H _O	= 32°46'
H _C	= 32°58.6'
a	= 12.6' away
Δd	= .35 (x 15' = 5.2')
H	= 32°53.4'
Δd corr.	= +5.2'
H _C	= 32°58.6'
Az	= N 78°W

You then plot these three lines of position on the chart (Figure 171), Venus first, then Capella and finally Polaris, advanced to give a fix at 03:18 GCT, the time at which Venus was observed.

Analyzing Fix—You notice that the three position lines form a small triangle, establishing, for all practical purposes, the aircraft's position for the time of the fix as being in the center of the triangle. However, the careful navigator analyzes each of his fixes to determine the reasons, if any, why he did not get a point fix. Rechecking your work carefully you find no mathematical errors. It occurs to you, therefore, that the octant may have some instrument error not accounted for.

Checking the position lines (Figure 170), you quickly see that if each was retarded along its azimuth line, the triangle would be reduced

in size, eventually becoming a point fix. But if these lines were so retarded, you would be presuming that the error was entirely due to unknown instrument error, and that all of your sights were accurate. You feel, however, that, actually, the Polaris sight was somewhat inaccurate since it was dim and consequently difficult to observe. Therefore, you estimate the amount of instrument error as 5' and retard each of the lines by this amount. The resulting triangle is quite small, although lying outside the original triangle, but should much more accurately represent your true position.

This fix shows you to be 30 miles south of the course, making good a track of 247° with a ground speed of 128 knots and drift 10° left. Setting this up on the computer against true air speed you determine the wind to be 360°/23 knots.

Since the speed has increased, you are getting the more favorable wind forecast by keeping south of the course, so you decide to hold the same heading. Advancing this fix to 04:00 GCT on the new course of 247° , at a ground speed of 128 knots, you fill in the Navigator's Flight Report (Figure 166) and make up the Radio Weather Report so it may be sent in on the hour.

Referring to the Flight Graph (Figure 163), you then plot in the actual data in dotted lines alongside the pre-computed curves. This information for 04:00 GCT is as follows:

Time vs. Miles Curve

Time 3^h50^m

Distance 485 NM

Fuel vs. Miles Curve

Fuel 348 gals.

Distance 485 NM

Since both of these actual curves fall below the estimated curves, the ship is making better time and using less fuel than was forecast.

04:30 GCT—Having just eaten a tasty steak sandwich which the engineer cooked up in fine style on the electric grill, you are feeling 100% better. By now members of the crew have settled down to the definite routine of a long-range flight. The motors drone monotonously as you fly on and on, steadily and surely eating up the miles between San Diego and Pearl Harbor. There is little feeling of forward movement when flying, especially at high altitudes or at night. In a train you can note your movement by watching telephone poles whiz by—but there are no phone poles up here!

Anxious now to check up on the 10° left drift you still are allowing for, you scan the sky to see what stars are available for a sight. Though it is beginning to cloud up, there are still plenty of stars out so you decide to shoot Polaris once more for a drift check. Since it is always good policy to select three bodies and plot a definite fix while you are at it, you also shoot Regulus for a speed check and Capella to complete the fix.

POLARIS

GCT $04^h36^m07^s$

$289^\circ41'$

$1^\circ32'$

GHA $\varphi = 291^\circ13'W$

DR Long. $= 127^\circ30'W$

LHA $\varphi = 163^\circ43'W$

$H_S = 28^\circ2'$

I. C. $= -5'$

Ref. $= -1'$

$H_O = 28^\circ23'$

Corr. LHA $\varphi = +45'$

Latitude $= 29^\circ08'N$

CAPELLA

GCT $04^h39^m07^s$

$289^\circ41'$

$2^\circ17'$

$281^\circ53'$

GHA $= 573^\circ51'$

-360°

GHA $= 213^\circ51'$

Assumed Long. $= 127^\circ51'W$

LHA $= 86^\circ W$

Assumed Lat. $= 29^\circ N$

Dec. $= 45^\circ56'N$

Lat. $29^\circ N$

Dec. $46^\circ N$

Same name

$H_S = 23^\circ12'$

I. C. $= -5'$

Ref. $= -2'$

$H_O = 23^\circ05.0'$

$H_C = 23^\circ00.2'$

a $= 4.8'$ toward

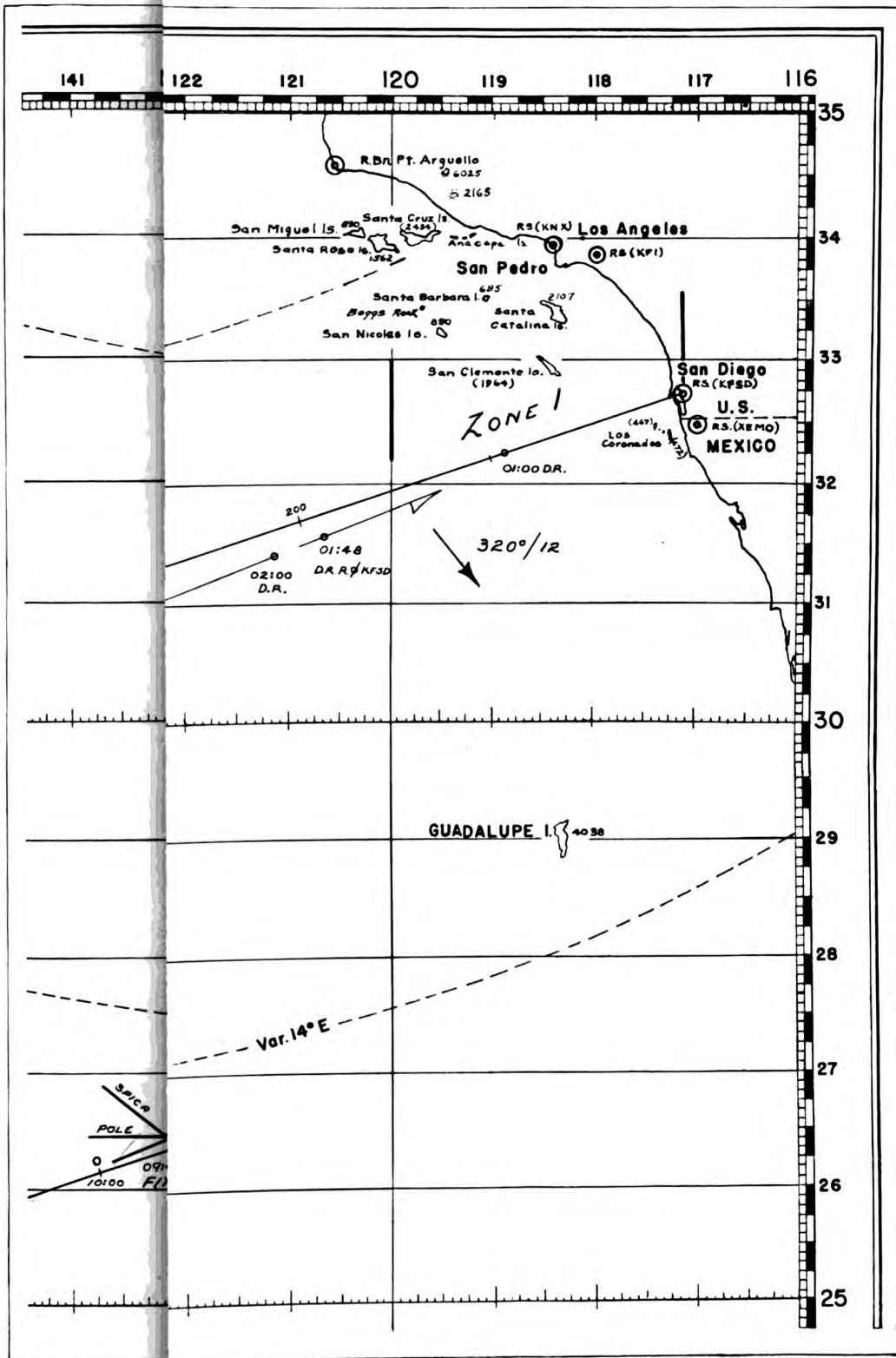
$\Delta d = .32$ ($\times 4' = 1.3'$)

H $= 23^\circ01.5'$

Δd corr. $= -1.3'$

$H_C = 23^\circ00.2'$

Az $= N 48.8^\circ W$



YOU NAVIGATE TO HONOLULU

REGULUS

GCT 04^h42^m07^s

		202°11'
		0°32'
		208°40'
GHA	=	501°23'
		—360°
GHA	=	141°23'
Assumed Long.	=	127°23'W
LHA	=	14° W
Assumed Lat.	=	29° N
Dec.	=	12°15'N

Lat. 29°N
Dec. 12°N
Same name

H _S	=	69°00'
I. C.	=	—5'
Ref.	=	0'
H _O	=	68°55'
H _C	=	68°46.8'
a	=	8.2' toward
Δd	=	.82 (x 15' = 12.3')
H	=	68°34.5'
Δd corr.	=	+12.3'
H _C	=	68°46.8'
Az	=	N 139.6°W

These three lines of position crossing in a point fix prove that you correctly analyzed the last fix, and that your octant actually has an instrument error of about 5'.

Measuring the track and ground speed from the last fix indicates that the wind is shifting around, as forecast.

Advancing the fix to 05:00 GCT you fill in the Navigator's Flight Report, using the track and wind found. Then, after compiling the Radio Weather Report, you plot in the actual data on the Flight Graph.

The temperature has dropped to +4°C. and the sky now is overcast, making it impossible to take any more sights, so you estimate the 06:00 and 07:00 positions (Figure 171).

Since the temperature has lowered, you are probably passing through a mild front which is the reason for the wind shift. Knowing the wind is shifting, you estimate it as 70°/16 knots. Laying out a new course to your destination (253°) you fill in the 06:00 GCT Navigator's Flight Report. Calculating a no-drift heading, you advise the captain to alter compass heading to 237°, which takes into account the change in variation.

06:45 GCT—The temperature has risen to +7°C. The sky is still overcast so you use the 07:00 DR position to fill in the Navigator's Flight Report and Radio Weather Report, using estimated wind.

07:10 GCT — The temperature is now +10°C. and the stars are starting to break

through the clouds. It is possible to take a quick sight between cloud breaks. However, the horizon is rapidly clearing so you decide to wait a few minutes and take a good fix when the sky clears.

07:20 GCT—It is nearly clear, so you start shooting, as you are anxious to obtain a speed line in order to check fuel consumption. You are nearing the PN and a good fuel vs. distance check is necessary in order to determine definitely if you have sufficient fuel to continue to your destination.

This decision as to whether to continue to destination or to turn back is a very important one, and it should be based upon the most reliable information obtainable. Actually, the decision itself will be up to the captain, but he will have to rely largely upon the data with which you supply him. It behooves you, therefore, to keep a very accurate check on ground speed, winds encountered and fuel consumed, so that when it becomes time to decide there will be no cause for hesitation. The more complete and accurate your information, the easier will be the captain's decision, and the faster he can *act*. The more closely you approach the PN, the more critical becomes the time element. A delay of even a few minutes in acting can sometimes mean the difference between a safe arrival and a forced landing at sea.

So far your fuel consumption has not been excessive, but your proximity to the PN makes it important to have a final check. Accordingly, you work out the following sights:

SPICA

GCT 07^h20^m07^s

	332°18'
	02'
	159°27'
GHA	= 491°47'
	—360°
GHA	= 131°47'
Assumed Long.	= 134°47'W
LHA	= 3° E
Assumed Lat.	= 27° N
Dec.	= 10°52'S

Lat. 27°N
Dec. 11°S
Opp. names

H _S	= 51°24'
I. C.	= —5'
Ref.	= 0'
H _O	= 51°19'
H _C	= 52°01.3'
a	= 42.3' away
Δd	= 1.0 (x 8' = 8.0')
H	= 51°53.3'
Δd corr.	= +8.0'
H _C	= 52°01.3'
Az	= N 175.2°E

ARCTURUS

GCT 07^h23^m07^s

	332°18'
	0°47'
	146°44'
GHA	= 479°49'
	—360°
GHA	= 119°49'
Assumed Long.	= 134°49'W
LHA	= 15° E
Assumed Lat.	= 27° N
Dec.	= 19°29'N

Lat. 27°N
Dec. 19°30'N
Same name

H _S	= 74°10'
I. C.	= —5'
Ref.	= 0'
H _O	= 74°05'
H _C	= 74°19.2'
a	= 14.2' away
Δd	= .52 (x 1' = .5')
H	= 74°19.7'
Δd corr.	= —.5'
H _C	= 74°19.2'
Az	= N 115.4°E

REGULUS

GCT 07^h26^m07^s

	332°18'
	1°32'
	208°40'
GHA	= 542°30'
	—360°
GHA	= 182°30'
Assumed Long.	= 134°30'W
LHA	= 48° W
Assumed Lat.	= 27° N
Dec.	= 12°15'N

Lat. 27°N
Dec. 12°N
Same name

H _S	= 43°09'
I. C.	= —5'
Ref.	= —1'
H _O	= 43°03.0'
H _C	= 42°45.8'
a	= 17.2' toward
Δd	= .44 (x 15' = 6.6')
H	= 42°39.2'
Δd corr.	= +6.6'
H _C	= 42°45.8'
Az	= N 98.8°W

YOU NAVIGATE TO HONOLULU

Once again analyzing the fix (Figure 171), it becomes apparent that moving each of the lines slightly away from the body along the azimuth line will result in a point fix at the center of the triangle. The small triangle probably is due to a slight personal error in observation, as most of your sights are fairly close.

07:50 GCT—You find you are making good a track of 256° true from the last fix with wind $100^\circ/18$ knots, resulting in 3° right drift. Since the forecast has proved nearly accurate, you assume the wind will continue shifting around to the north. You decide, therefore, to fly a no-drift heading, parallel to the original course of 252° , estimating the same ground speed. You then advise the captain to alter compass heading to 236° which should keep the aircraft south of the course.

Advancing the fix to 08:00 GCT, you fill in the Navigator's Flight Report, Radio Weather Report and plot the actual position and fuel

consumed on the Flight Graph. The graph indicates that you are using less gasoline than was estimated, and that it will be safe to continue to Honolulu.

09:00 GCT—You feel quite sure of your position, and since you have been busy the last hour catching up on all radio and weather reports, you take time out to cook up a can of soup and some coffee. You then fill in the 09:00 DR position on the Navigator's Flight Report.

09:30 GCT—It is time now to obtain another fix. Your previous fix included a good speed line, which afforded an accurate check on the distance flown—hence average ground speed—and the fuel-distance relationship. For the present, therefore, you are satisfied with the information you have regarding speed and distance. It now seems advisable to check the aircraft's position with reference to the course line, so you take another series of sights as follows:

POLARIS

GCT 09^h34^m07^s

	$4^\circ 53'$
	$1^\circ 02'$
GHA φ	$= 5^\circ 55'$
	$+360^\circ 00'$
GHA φ	$= 365^\circ 55' W$
DR Long.	$= 139^\circ 55' W$
LHA φ	$= 226^\circ W$

H_s	$= 25^\circ 41'$
I. C.	$= -5'$
Ref.	$= -1'$
H_o	$= 25^\circ 35'$
Corr. LHA φ	$= +57'$
Latitude	$= 26^\circ 32' N$

ANTARES

GCT 09^h37^m07^s

	$4^\circ 53'$
	$1^\circ 47'$
	$113^\circ 31'$
GHA	$= 120^\circ 11'$
Assumed Long.	$= 140^\circ 11' W$
LHA	$= 20^\circ E$
Assumed Lat.	$= 26^\circ N$
Dec.	$= 26^\circ 18' S$

Lat. $26^\circ N$
Dec. $26^\circ S$
Opp. names

H_s	$= 34^\circ 02'$
I. C.	$= -5'$
Ref.	$= -1'$
H_o	$= 33^\circ 56'$
H_c	$= 34^\circ 15.5'$
a	$= 19.5' \text{ away}$
Δd	$= .93 (x 18' = 16.7')$
H	$= 34^\circ 32.2'$
$\Delta d \text{ corr.}$	$= -16.7'$
H_c	$= 34^\circ 15.5'$
Az	$= N 158.1^\circ E$

AMERICAN AIR NAVIGATOR

SPICA

GCT 09^h40^m07^s

7°23'

02'

159°27'

GHA = 166°52'

Assumed Long. = 139°52'W

LHA = 27° W

Assumed Lat. = 26° N

Dec. = 10°52'S

Lat. 26°N

Dec. 11°S

Opp. names

H_s = 44°36'

I. C. = -5'

Ref. = -1'

H_o = 44°30.0'

H_c = 44°44.1'

a = 13.9' away

Δd = .82 (x 8' = 6.6')

H = 44°37.5'

Δd corr. = +6.6'

H_c = 44°44.1'

Az = N 141.2°W

Checking the track made good (Figure 171), you find the aircraft is still drifting 3° to the right; however, the ground speed has decreased to 126 knots and the wind has shifted around to 130°/07 knots. You then advance the fix to 10:00 GCT and fill out the Navigator's Flight Report, using wind found and track made good.

Since the aircraft is a little North of the track, you continue to hold the same heading

but estimate the wind will continue shifting around gradually to the Northwest as forecast. Therefore, you establish a DR position for 11:00 GCT (Figure 172) using estimated wind of 300°/05 knots and course of 250°.

11:40 GCT—It is time to compute the ETA (Estimated Time of Arrival) and also to check your course for the estimated wind shift; accordingly, you proceed to take another series of sights.

ANTARES

GCT 11^h44^m07^s

37°28'

1°02'

113°31'

GHA = 152°01'

Assumed Long. = 144°01'W

LHA = 8° W

Assumed Lat. = 25° N

Dec. = 26°18'S

Lat. 25°N

Dec. 26°S

Opp. names

H_s = 38°18'

I. C. = -5'

Ref. = -1'

H_o = 38°12'

H_c = 38°07.3'

a = 4.7' toward

Δd = .99 (x 18' = 17.8')

H = 38°25.1'

Δd corr. = -17.8'

H_c = 38°07.3'

Az = N 170.8°W

VEGA

GCT 11^h47^m07^s

37°28'

1°47'

81°14'

GHA = 120°29'

Assumed Long. = 144°29'W

LHA = 24° E

Assumed Lat. = 25° N

Dec. = 38°44'N

Lat. 25°N

Dec. 38°30'N

Same name

H_s = 65°51'

I. C. = -5'

Ref. = 0'

H_o = 65°46.0'

H_c = 65°32.7'

a = 13.3' toward

Δd = .45 (x 14' = 6.3')

H = 65°39.0'

Δd corr. = -6.3'

H_c = 65°32.7'

Az = N 50°E

YOU NAVIGATE TO HONOLULU

SPICA

GCT 11^h50^m07^s

39°59'

02'

159°27'

GHA = 199°28'

Assumed Long. = 144°28'W

LHA = 55° W

Assumed Lat. = 25° N

Dec. = 10°52'S

Lat. 25°N

Dec. 11°S

Opp. names

H_s = 25°27'

I. C. = -5'

Ref. = -2'

H_O = 25°20.0'

H_C = 25°31.2'

a = 11.2' away

Δd = .57 (x 8' = 4.5')

H = 25°26.7'

Δd corr. = +4.5'

H_C = 25°31.2'

Az = N 117°W

This fix (Figure 172) indicates that you have correctly estimated the wind shift; however, since you now wish to remain on course, you advance the fix to 12:00 GCT and lay out a new course to destination (254°). Estimating 5° left drift for additional wind shift, you then advise the captain to alter compass heading to 245°.

The Navigator's Flight Report and Radio Weather Report then are made out for 12:00 GCT, using wind and track made good.

To determine your ETA, you estimate that

the ground speed probably will average about 125 knots, as the wind is expected to shift to 10°/07 knots. This means you have 6^h18^m more flying time ahead of you. Allowing 10 minutes for let-down procedure, you estimate the ETA to be 18:28 GCT.

Plotting your present actual position on the time vs. miles curve (Flight Graph) provides a good check on the ETA.

The 13:00 GCT report then is made out, using course to destination and estimated wind of 310°/14 knots as forecast.

POLARIS

GCT 13^h24^m07^s

62°32'

1°02'

GHA γ = 63°34'

+360°

GHA γ = 423°34'W

DR Long. = 147°34'W

LHA γ = 276°00'W

H_s = 23°44'

I. C. = -5'

Ref. = -2'

H_O = 23°37'

Corr. LHA γ = +21'

Latitude = 23°58'N

ANTARES

GCT 13^h27^m07^s

62°32'

1°47'

113°31'

GHA = 177°50'

Assumed Long. = 147°50'W

LHA = 30° W

Assumed Lat. = 24° N

Dec. = 26°18'S

Lat. 24°N

Dec. 26°S

Opp. names

H_s = 32°02'

I. C. = -5'

Ref. = -1'

H_O = 31°56'

H_C = 31°56.5'

a = 0.5' away

Δd = .84 (x 18' = 15.1')

H = 32°11.6'

Δd corr. = -15.1'

H_C = 31°56.5'

Az = N 148°W

AMERICAN AIR NAVIGATOR

ARCTURUS

GCT 13^h30^m07^s

65°03'

02'

146°44'

GHA = 211°49'

Assumed Long. = 147°49'W

LHA = 64° W

Assumed Lat. = 24° N

Dec. = 19°29'N

Lat. 24°N
Dec. 19°30'N
Same name

H_s = 30°52'

I. C. = -5'

Ref. = -1'

H_o = 30°46'

H_c = 30°53.2'

a = 7.2' away

Δd = .29 (x 1' = .3')

H = 30°52.9'

Δd corr. = +.3'

H_c = 30°53.2'

Az = N 81.0°W

13:20 GCT—To check the ETA and track, you take another series of sights as follows:

The 13:30 fix (Figure 172) proves that the wind is shifting to the North as forecast. Your ground speed has increased to 122 knots and your ETA is still 18:28 GCT.

Advancing this latest fix to 14:00 GCT, you again fill in the Navigator's Flight Report and Radio Weather Report.

14:50 GCT—Since the sky is starting to get lighter you decide to check the course and speed once more before daybreak, and so obtain the following sights:

VEGA

GCT 14^h52^m07^s

85°06'

0°32'

81°14'

GHA = 166°52'

Assumed Long. = 150°52'W

LHA = 16° W

Assumed Lat. = 23° N

Dec. = 38°44'N

Lat. 23°N
Dec. 38°30'N
Same name

H_s = 69°11'

I. C. = -5'

Ref. = 0'

H_o = 69°06.0'

H_c = 69°11.0'

a = 5.0' away

Δd = .69 (x 14' = 9.7')

H = 69°20.7'

Δd corr. = -9.7'

H_c = 69°11.0'

Az = N 37.7°W

ANTARES

GCT 14^h55^m07^s

85°06'

1°17'

113°31'

GHA = 199°54'

Assumed Long. = 150°54'W

LHA = 49° W

Assumed Lat. = 23° N

Dec. = 26°18'S

Lat. 23°N
Dec. 26°S
Opp. names

H_s = 21°42'

I. C. = -5'

Ref. = -2'

H_o = 21°35'

H_c = 21°36.6'

a = 1.6' away

Δd = .66 (x 18' = 11.9')

H = 21°48.5'

Δd corr. = -11.9'

H_c = 21°36.6'

Az = N 133.1°W

YOU NAVIGATE TO HONOLULU

Checking the track made good, you find the wind has increased so that you are now making a ground speed of 128 knots with 7° left drift.

Advancing the fix to 15:00, and before filling in Navigator's Flight Report, you first fix your present position as of 15:20 GCT and establish a new course to Honolulu. You then advise the captain to alter compass heading to 249° in order to make 256° true.

15:30 GCT—The sun is starting to rise, and although you have felt you could hardly hold your eyes open for the last couple of hours, you now feel exhilarated, your energy renewed. You anxiously anticipate seeing the peaks of the Islands break through the clouds.

Until the sun is higher, it will not be practical to take sights, so you establish a DR position for 16:00 and fill in the Navigator's Flight Report, using present course and last known wind.

16:20 GCT—You turn on the radio compass and listen to the Honolulu radio range. Hearing a steady dash, with an occasional weak dot-dash (A), proves you are right on course. You relax, feeling satisfied that you have done a nice job of navigation.

17:00 GCT—The sun is high enough now for a sight, so you take a shot to check the speed (see below).

The sun line (Figure 172) indicates that you still are making 128 knots.

17:20 GCT—There are clouds obscuring Maui, but you are sure of your position now and the radio range proves you are well clear of the mountain peaks on Maui and Molokai. You tell the captain that if he will start letting down through the clouds, he will see the east end of Molokai abeam at 17:35.

The captain follows your suggestion and sure enough you see the shore line. However, the increased speed of letting down puts Molokai abeam at 17:33, and the crew razz you for being two minutes off. Nonetheless, you know from their smiles that they would trust your navigation anywhere.

For you the flight is as good as over. You start packing your equipment and sit back and relax while the captain brings in the PBY contact and lands on the bay at Honolulu.

The navigator shouldn't always relax, however, upon sighting land. Quite often he can be of immense assistance to the captain by keeping a close check on the landmarks passed and by computing the best headings to the air base. This is especially true in bad weather when the captain must give all of his attention to flying.

18:29 GCT—On water. Finish filling out the Navigator's Flight Report.

Your trip to Honolulu is over.

SUN

GCT 17^h00^m07^s

		75°50'
		02'
GHA	=	75°52'
Assumed Long.	=	154°52'W
LHA	=	79° E
Assumed Lat.	=	22° N
Dec.	=	16°08'N

Lat. 22°N
Dec. 16°N
Same name

H _s	=	15°28'
I. C.	=	—5'
Ref.	=	—3'
H _o	=	15°20'
H _c	=	15°54.3'
a	=	34.3' away
Δd	=	.32 (x 8' = 2.6')
H	=	15°51.7'
Δd corr.	=	+2.6'
H _c	=	15°54.3'
Az	=	N 78.8°E

PROBLEM WORK NO. 32

PBY FLIGHT FROM SAN DIEGO TO HONOLULU

MAY 5, 1943

True Course = 253° Total Distance = 2,290 Nautical Miles

I.A.S. = 92 knots, calibration correction +5 knots.

Fuel aboard at start = 1875 U. S. gallons. Average fuel consumption = 75 g.p.h., during climb = 100 g.p.h.

Flight altitude = 8000'. Temperature +10°C for total trip.

(Follow all procedures in order, thus simulating actual flight. Plot fixes on a small scale chart. Keep running account of flight by filling in flight plan for each hour before proceeding with next fix. Using forecast wind, establish first heading. Do not alter course unless indicated. Deviation is 0°.)

FORECAST WIND FOR 8000'

ZONE 1	SAN DIEGO to 120° W Long.	315° / 20 knots
ZONE 2	120° W to 130° W	315° / 20 knots
ZONE 3	130° W to 140° W	135° / 20 knots
ZONE 4	140° W to 150° W	20° / 20 knots
ZONE 5	150° W to HONOLULU	45° / 15 knots

00:31 GCT off water, compass heading = 246°. (Climb 29 min. at an I.A.S. of 100 knots.)

00:37 Relative radio bearing taken of KFSD, San Diego = 171°. Magnetic heading = 247° (at the instant bearing was taken). Mercator correction = -1°.

No.	STAR SIGHTS TAKEN EN ROUTE—INDEX CORRECTION = 0°					
1	DUBHE	GCT = 03:27:00 H _s = 57° 06'	BETELGEUX	GCT = 03:32:00 H _s = 26° 32'	SIRIUS	GCT = 03:37:00 H _s = 21° 18'
2	PROCYON	GCT = 05:00:00 H _s = 31° 44'	DUBHE	GCT = 04:56:00 H _s = 57° 16'	POLLUX	GCT = 05:05:00 H _s = 41° 15'
3	DUBHE	GCT = 07:14:00 H _s = 49° 12'	POLLUX	GCT = 07:00:00 H _s = 20° 06'	SPICA	GCT = 07:05:00 H _s = 50° 49'
4	REGULUS	GCT = 09:00:00 H _s = 22° 21'	SPICA	GCT = 09:04:00 H _s = 46° 30'	DUBHE	GCT = 09:09:00 H _s = 37° 42'

09:50 GCT Alter course to intercept destination.

5	ANTARES	GCT = 11:55:00 H _s = 37° 35'	ARCTURUS	GCT = 12:00:00 H _s = 46° 42'		
6	ALKAID	GCT = 13:04:00 H _s = 32° 48'	ARCTURUS	GCT = 13:07:30 H _s = 33° 38'	ANTARES	GCT = 13:12:00 H _s = 32° 04'
7	ANTARES	GCT = 14:04:00 H _s = 26° 46'	ARCTURUS	GCT = 14:00:00 H _s = 23° 32'		

16:34 GCT Radio bearing KMZA = 296°, magnetic heading = 263°.

16:36 GCT Radio bearing KNBF = 333°, magnetic heading = 263°.

16:45 GCT ALTER COURSE TO INTERCEPT DESTINATION.

Appreciation is herewith expressed to the U. S. Naval Observatory, and to its superintendent, Capt. J. F. Hellweg, U.S.N., Retired, for permission to reprint the following pages from the Air Almanac.

APPENDIX

APPENDIX

GREENWICH A. M. 1943 JANUARY 1 (FRIDAY)

GCT	SUN	MARS 1.7	JUPITER - 2.3	SATURN 6.0	MOON	☾
	GHA Dec.	GHA Dec.	GHA Dec.	GHA Dec.	GHA Dec.	Par.
0 00	179 14 S23 05	99 46	209 51 S22 16	346 25 N21 59	34 31 N19 36	255 25 S 5 25
10	181 44	102 17	212 21	348 56	37 01	257 50 27
20	184 14	104 47	214 51	351 26	39 31	260 15 29
30	186 43	107 18	217 21	353 57	42 02	262 40 30
40	189 13	109 48	219 51	356 27	44 32	265 05 32
50	191 43	112 18	222 21	358 57	47 03	267 30 34
1 00	194 13 S23 05	114 49	224 51 S22 16	1 28 N21 59	49 33 N19 36	269 55 S 5 36
10	196 43	117 19	227 21	3 58	52 04	272 20 38
20	199 13	119 50	229 51	6 29	54 34	274 45 39
30	201 43	122 20	232 21	8 59	57 05	277 10 41
40	204 13	124 50	234 52	11 30	59 35	279 35 43
50	206 43	127 21	237 22	14 00	62 05	282 01 45
2 00	209 13 S23 05	129 51	239 52 S22 16	16 31 N21 59	64 36 N19 36	284 26 S 5 47
10	211 43	132 22	242 22	19 01	67 06	286 51 48
20	214 13	134 52	244 52	21 32	69 37	289 16 50
30	216 43	137 23	247 22	24 02	72 07	291 41 52
40	219 13	139 53	249 52	26 33	74 38	294 06 54
50	221 43	142 23	252 22	29 03	77 08	296 31 55
3 00	224 13 S23 05	144 54	254 52 S22 17	31 34 N21 59	79 38 N19 36	298 56 S 5 57
10	226 43	147 24	257 22	34 04	82 09	301 21 5 59
20	229 13	149 55	259 53	36 34	84 39	303 46 6 01
30	231 43	152 25	262 23	39 05	87 10	306 11 03
40	234 13	154 55	264 53	41 35	89 40	308 36 04
50	236 42	157 26	267 23	44 06	92 11	311 01 06
4 00	239 12 S23 05	159 56	269 53 S22 17	46 36 N21 59	94 41 N19 36	313 26 S 6 08
10	241 42	162 27	272 23	49 07	97 12	315 51 10
20	244 12	164 57	274 53	51 37	99 42	318 16 11
30	246 42	167 27	277 23	54 08	102 12	320 42 13
40	249 12	169 58	279 53	56 38	104 43	323 07 15
50	251 42	172 28	282 23	59 09	107 13	325 32 17
5 00	254 12 S23 04	174 59	284 53 S22 17	61 39 N21 59	109 44 N19 36	327 57 S 6 19
10	256 42	177 29	287 24	64 10	112 14	330 22 20
20	259 12	180 00	289 54	66 40	114 45	332 47 22
30	261 42	182 30	292 24	69 11	117 15	335 12 24
40	264 12	185 00	294 54	71 41	119 45	337 37 26
50	266 42	187 31	297 24	74 12	122 16	340 02 27
6 00	269 12 S23 04	190 01	299 54 S22 18	76 42 N21 59	124 46 N19 36	342 27 S 6 29
10	271 42	192 32	302 24	79 12	127 17	344 52 31
20	274 12	195 02	304 54	81 43	129 47	347 17 33
30	276 42	197 32	307 24	84 13	132 18	349 42 34
40	279 12	200 03	309 54	86 44	134 48	352 07 36
50	281 42	202 33	312 25	89 14	137 19	354 32 38
7 00	284 12 S23 04	205 04	314 55 S22 18	91 45 N21 59	139 49 N19 36	356 57 S 6 40
10	286 42	207 34	317 25	94 15	142 19	359 22 41
20	289 11	210 04	319 55	96 46	144 50	361 47 43
30	291 41	212 35	322 25	99 16	147 20	364 12 45
40	294 11	215 05	324 55	101 47	149 51	366 37 47
50	296 41	217 36	327 25	104 17	152 21	369 02 49
8 00	299 11 S23 04	220 06	329 55 S22 18	106 48 N21 59	154 52 N19 36	371 27 S 6 50
10	301 41	222 37	332 25	109 18	157 22	373 52 52
20	304 11	225 07	334 55	111 49	159 52	376 17 54
30	306 41	227 37	337 25	114 19	162 23	378 42 56
40	309 11	230 08	339 56	116 49	164 53	381 07 57
50	311 41	232 38	342 26	119 20	167 24	383 32 59
9 00	314 11 S23 04	235 09	344 56 S22 18	121 50 N21 59	169 54 N19 36	385 57 S 7 01
10	316 41	237 39	347 26	124 21	172 25	388 22 03
20	319 11	240 09	349 56	126 51	174 55	390 47 04
30	321 41	242 40	352 26	129 22	177 26	393 12 06
40	324 11	245 10	354 56	131 52	179 56	395 37 08
50	326 41	247 41	357 26	134 23	182 26	398 02 10
10 00	329 11 S23 03	250 11	359 56 S22 19	136 53 N21 59	184 57 N19 36	400 27 S 7 11
10	331 41	252 41	2 26	139 24	187 27	402 52 13
20	334 11	255 12	4 57	141 54	189 58	405 17 15
30	336 41	257 42	7 27	144 25	192 28	407 42 17
40	339 10	260 13	9 57	146 55	194 59	410 07 18
50	341 40	262 43	12 27	149 26	197 29	412 32 20
11 00	344 10 S23 03	265 13	14 57 S22 19	151 56 N21 59	199 59 N19 36	414 57 S 7 22
10	346 40	267 44	17 27	154 26	202 30	417 22 24
20	349 10	270 14	19 57	156 57	205 00	419 47 25
30	351 40	272 45	22 27	159 27	207 31	422 12 27
40	354 10	275 15	24 57	161 58	210 01	424 37 29
50	356 40	277 46	27 27	164 28	212 32	427 02 31
12 00	359 10 S23 03	280 16	29 57 S22 19	166 59 N21 59	215 02 N19 36	429 27 S 7 32

AMERICAN AIR NAVIGATOR

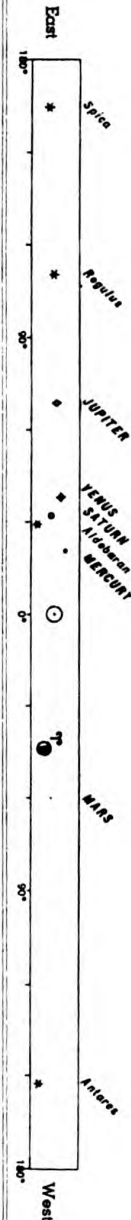
GREENWICH P. M. 1943 JANUARY 1 (FRIDAY)

GCT	☉ SUN		☿ MARS 1.7	♃ JUPITER -2.3	♄ SATURN 0.0	☾ MOON		Lat	Sun- rise	Twil	Moon- rise	Diff.
	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.				
12 00	359 10	S23 03	280 16	29 57	S22 19	166 59	N21 59	215 02	N19 36	69 27	S 7 32	
10	1 40		282 46	32 28		169 29		217 33		71 51	34	
20	4 10		285 17	34 58		172 00		220 03		74 16	36	
30	6 40		287 47	37 28		174 30		222 33		76 41	38	
40	9 10		290 18	39 58		177 01		225 04		79 06	39	
50	11 40		292 48	42 28		179 31		227 34		81 31	41	
13 00	14 10	S23 03	295 18	44 58	S22 19	182 02	N21 59	230 05	N19 36	83 56	S 7 43	60
10	16 40		297 49	47 28		184 32		232 35		86 21	45	58
20	19 10		300 19	49 58		187 03		235 06		88 46	46	56
30	21 40		302 50	52 28		189 33		237 36		91 11	48	54
40	24 10		305 20	54 58		192 04		240 06		93 36	50	52
50	26 40		307 50	57 28		194 34		242 37		96 01	52	50
14 00	29 09	S23 03	310 21	59 59	S22 20	197 04	N21 59	245 07	N19 36	98 26	S 7 53	45
10	31 39		312 51	02 29		199 35		247 38		100 51	55	40
20	34 09		315 22	04 59		202 05		250 08		103 16	57	35
30	36 39		317 52	07 29		204 36		252 39		105 41	59	30
40	39 09		320 23	09 59		207 06		255 09		108 06	00	20
50	41 39		322 53	12 29		209 37		257 39		110 30	02	10
15 00	44 09	S23 02	325 23	74 59	S22 20	212 07	N21 59	260 10	N19 36	112 55	S 8 04	0
10	46 39		327 54	77 29		214 38		262 40		115 20	05	10
20	49 09		330 24	79 59		217 08		265 11		117 45	07	10
30	51 39		332 55	82 29		219 39		267 41		120 10	09	20
40	54 09		335 25	85 00		222 09		270 12		122 35	11	30
50	56 39		337 55	87 30		224 40		272 42		125 00	12	35
16 00	59 09	S23 02	340 26	90 00	S22 20	227 10	N21 59	275 13	N19 36	127 25	S 8 14	40
10	61 39		342 56	92 30		229 41		277 43		129 50	16	45
20	64 09		345 27	95 00		232 11		280 13		132 15	18	50
30	66 39		347 57	97 30		234 41		282 44		134 40	19	52
40	69 09		350 27	100 00		237 12		285 14		137 04	21	54
50	71 39		352 58	102 30		239 42		287 45		139 29	23	56
17 00	74 09	S23 02	355 28	105 00	S22 20	242 13	N21 59	290 15	N19 36	141 54	S 8 24	58
10	76 39		357 59	107 30		244 43		292 46		144 19	26	60
20	79 08		0 29	110 00		247 14		295 16		146 44	28	62
30	81 38		3 00	112 31		249 44		297 46		149 09	30	64
40	84 08		5 30	115 01		252 15		300 17		151 34	31	66
50	86 38		8 00	117 31		254 45		302 47		153 59	33	68
18 00	89 08	S23 02	10 31	120 01	S22 21	257 16	N22 00	305 18	N19 36	156 24	S 8 35	60
10	91 38		13 01	122 31		259 46		307 48		158 48	36	62
20	94 08		15 32	125 01		262 17		310 19		161 13	38	64
30	96 38		18 02	127 31		264 47		312 49		163 38	40	66
40	99 08		20 32	130 01		267 18		315 20		166 03	42	68
50	101 38		23 03	132 31		269 48		317 50		168 28	43	70
19 00	104 08	S23 02	25 33	135 01	S22 21	272 18	N22 00	320 20	N19 36	170 53	S 8 45	60
10	106 38		28 04	137 32		274 49		322 51		173 18	47	58
20	109 08		30 34	140 02		277 19		325 21		175 43	48	56
30	111 38		33 04	142 32		279 50		327 52		178 07	50	54
40	114 08		35 35	145 02		282 20		330 22		180 32	52	52
50	116 38		38 05	147 32		284 51		332 53		182 57	54	50
20 00	119 08	S23 01	40 36	150 02	S22 21	287 21	N22 00	335 23	N19 36	185 22	S 8 55	45
10	121 38		43 06	152 32		289 52		337 53		187 47	57	40
20	124 08		45 36	155 02		292 22		340 24		190 12	59	35
30	126 38		48 07	157 32		294 53		342 54		192 36	00	30
40	129 08		50 37	160 02		297 23		345 25		195 01	02	20
50	131 37		53 08	162 32		299 54		347 55		197 26	04	10
21 00	134 07	S23 01	55 38	165 03	S22 22	302 24	N22 00	350 26	N19 36	199 51	S 9 06	0
10	136 37		58 09	167 33		304 55		352 56		202 16	07	10
20	139 07		60 39	170 03		307 25		355 27		204 41	09	10
30	141 37		63 09	172 33		309 56		357 57		207 06	11	20
40	144 07		65 40	175 03		312 26		0 27		209 30	12	30
50	146 37		68 10	177 33		314 56		2 58		211 55	14	35
22 00	149 07	S23 01	70 41	180 03	S22 22	317 27	N22 00	5 28	N19 36	214 20	S 9 16	40
10	151 37		73 11	182 33		319 57		7 59		216 45	17	45
20	154 07		75 41	185 03		322 28		10 29		219 10	19	50
30	156 37		78 12	187 33		324 58		13 00		221 35	21	52
40	159 07		80 42	190 04		327 29		15 30		223 59	22	54
50	161 37		83 13	192 34		329 59		18 00		226 24	24	56
23 00	164 07	S23 01	85 43	195 04	S22 22	332 30	N22 00	20 31	N19 36	228 49	S 9 26	58
10	166 37		88 13	197 34		335 00		23 01		231 14	27	60
20	169 07		90 44	200 04		337 31		25 32		233 39	29	62
30	171 37		93 14	202 34		340 01		28 02		236 03	31	64
40	174 07		95 45	205 04		342 32		30 33		238 28	33	66
50	176 37		98 15	207 34		345 02		33 03		240 53	34	68
24 00	179 07	S23 01	100 46	210 04	S22 22	347 33	N22 00	35 34	N19 36	243 18	S 9 36	60

APPENDIX

GREENWICH A. M. 1943 MAY 1 (SATURDAY)

GCT	SUN		Dec.	VENUS - 3.5		Dec.	MARS 1.1		Dec.	JUPITER - 1.6		Dec.	MOON		Dec.	Par.	Alt.	Corr.	East	West
	GHA	GHA		GHA	GHA		GHA	GHA		GHA	GHA									
0 00	180 42	114 41	218 03	111 26	N24 45	235 43	S 9 09	107 34	N22 31	219 19	S 3 45									
10	183 12		220 33	143 56		238 13		110 04		221 44		43								
20	185 42		223 04	146 26		240 43		112 35		224 09		41								
30	188 12		225 34	148 56		243 13		115 05		226 34		39								
40	190 42		228 05	151 26		245 43		117 35		228 59		37								
50	193 12		230 35	153 55		248 13		120 06		231 24		35								
1 00	195 42	N14 45	233 05	156 25	N24 45	250 43	S 9 09	122 36	N22 31	233 49	S 3 33									
10	198 12		235 36	158 55		253 13		125 06		236 14		31								
20	200 42		238 06	161 25		255 43		127 37		238 39		30								
30	203 12		240 37	163 55		258 14		130 07		241 04		28								
40	205 42		243 07	166 25		260 44		132 37		243 29		26								
50	208 12		245 38	168 55		263 14		135 08		245 54		24								
2 00	210 42	N14 46	248 08	171 25	N24 46	265 44	S 9 08	137 38	N22 31	248 19	S 3 22									
10	213 12		250 38	173 54		268 14		140 09		250 44		20								
20	215 42		253 09	176 24		270 44		142 39		253 09		18								
30	218 12		255 39	178 54		273 14		145 09		255 34		16								
40	220 42		258 10	181 24		275 44		147 40		257 59		14								
50	223 12		260 40	183 54		278 14		150 10		260 24		12								
3 00	225 42	N14 47	263 10	186 24	N24 46	280 45	S 9 07	152 40	N22 31	262 49	S 3 11									
10	228 12		265 41	188 54		283 15		155 11		265 14		09								
20	230 43		268 11	191 24		285 45		157 41		267 39		07								
30	233 13		270 42	193 54		288 15		160 11		270 04		05								
40	235 43		273 12	196 23		290 45		162 42		272 29		03								
50	238 13		275 42	198 53		293 15		165 12		274 54	3 01	44								
4 00	240 43	N14 47	278 13	201 23	N24 46	295 45	S 9 07	167 42	N22 31	277 20	S 2 59									
10	243 13		280 43	203 53		298 15		170 13		279 45		57								
20	245 43		283 14	206 23		300 46		172 43		282 10		55								
30	248 13		285 44	208 53		303 16		175 13		284 35		53								
40	250 43		288 14	211 23		305 46		177 44		287 00		51								
50	253 13		290 45	213 53		308 16		180 14		289 25		50								
5 00	255 43	N14 48	293 15	216 22	N24 47	310 46	S 9 06	182 44	N22 31	291 50	S 2 48									
10	258 13		295 46	218 52		313 16		185 15		294 15		46								
20	260 43		298 16	221 22		315 46		187 45		296 40		44								
30	263 13		300 47	223 52		318 16		190 16		299 05		42								
40	265 43		303 17	226 22		320 46		192 46		301 30		40								
50	268 13		305 47	228 52		323 17		195 16		303 55		38								
6 00	270 43	N14 49	308 18	231 22	N24 47	325 47	S 9 05	197 47	N22 31	306 20	S 2 36									
10	273 13		310 48	233 52		328 17		200 17		308 45		34								
20	275 43		313 19	236 21		330 47		202 47		311 10		32								
30	278 13		315 49	238 51		333 17		205 18		313 35		30								
40	280 43		318 19	241 21		335 47		207 48		316 00		29								
50	283 13		320 50	243 51		338 17		210 18		318 25		27								
7 00	285 43	N14 50	323 20	246 21	N24 47	340 47	S 9 04	212 49	N22 31	320 50	S 2 25									
10	288 13		325 51	248 51		343 17		215 19		323 15		23								
20	290 43		328 21	251 21		345 48		217 49		325 40		21								
30	293 13		330 51	253 51		348 18		220 20		328 05		19								
40	295 43		333 22	256 21		350 48		222 50		330 30		17								
50	298 13		335 52	258 50		353 18		225 20		332 55		15								
8 00	300 43	N14 50	338 23	261 20	N24 48	355 48	S 9 04	227 51	N22 31	335 20	S 2 13									
10	303 13		340 53	263 50		358 18		230 21		337 45		11								
20	305 43		343 24	266 20		0 48		232 51		340 10		09								
30	308 13		345 54	268 50		3 18		235 22		342 35		07								
40	310 43		348 24	271 20		5 49		237 52		345 00		06								
50	313 13		350 55	273 50		8 19		240 23		347 25		04								
9 00	315 43	N14 51	353 25	276 20	N24 48	10 49	S 9 03	242 53	N22 31	349 50	S 2 02									
10	318 13		355 56	278 49		13 19		245 23		352 15		2 00								
20	320 43		358 26	281 19		15 49		247 54		354 40		1 58								
30	323 13		0 56	283 49		18 19		250 24		357 06		56								
40	325 43		3 27	286 19		20 49		252 54		359 31		54								
50	328 13		5 57	288 49		23 19		255 25		1 56		52								
10 00	330 43	N14 52	8 28	291 19	N24 48	25 49	S 9 02	257 55	N22 31	4 21	S 1 50									
10	333 13		10 58	293 49		28 20		260 25		6 46		48								
20	335 43		13 28	296 19		30 50		262 56		9 11		46								
30	338 13		15 59	298 49		33 20		265 26		11 36		44								
40	340 43		18 29	301 18		35 50		267 56		14 01		43								
50	343 13		21 00	303 48		38 20		270 27		16 26		41								
11 00	345 43	N14 53	23 30	306 18	N24 49	40 50	S 9 02	272 57	N22 31	18 51	S 1 39									
10	348 13		26 01	308 48		43 20		275 27		21 16		37								
20	350 43		28 31	311 18		45 50		277 58		23 41		35								
30	353 13		31 01	313 48		48 20		280 28		26 06		33								
40	355 43		33 32	316 18		50 51		282 58		28 31		31								
50	358 13		36 02	318 48		53 21		285 29		30 56		29								
12 00	0 43	N14 53	38 33	321 17	N24 49	55 51	S 9 01	287 59	N22 31	33 21	S 1 27									



AMERICAN AIR NAVIGATOR

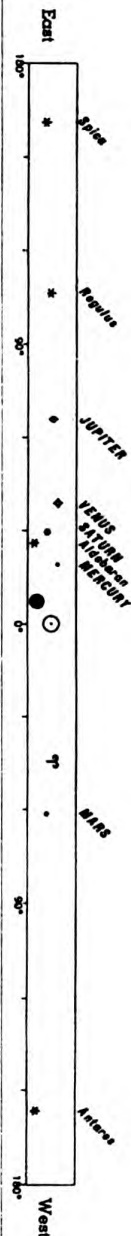
GREENWICH P. M. 1943 MAY 1 (SATURDAY)

GCT	☉ SUN		☿	♀ VENUS +1.5		♂ MARS 1.1		♃ JUPITER -1.6		♄ MOON		Lat.	Sun-rise	Twil.	Moon-rise	Diff.
	GHA	Dec.	GHA	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.					
h m	0 43	N14 53	38 33	321 17	N24 49	55 51	S 9 01	287 59	N22 31	33 21	S 1 27	N				
12 00	3 13		41 03	323 47		58 21		290 30		35 46	25	0	h m	m	h m	m
10	5 43		43 33	326 17		60 51		293 00		38 11	23	70	2 36	NH	4 05	*
20	8 13		46 04	328 47		63 21		295 30		40 36	22	68	3 02	106	4 02	02
30	10 43		48 34	331 17		65 51		298 01		43 01	20	66	21	80	3 59	06
40	13 13		51 05	333 47		68 21		300 31		45 26	18	64	37	66	57	09
50	15 43	N14 54	53 35	336 17	N24 49	70 51	S 9 00	303 01	N22 31	47 52	S 1 16	62	3 50	58	54	13
13 00	18 13		56 05	338 47		73 22		305 32		50 17	14	60	4 01	54	53	15
10	20 43		58 36	341 16		75 52		308 02		52 42	12	58	10	49	51	18
20	23 13		61 06	343 46		78 22		310 32		55 07	10	56	18	45	49	21
30	25 43		63 37	346 16		80 52		313 03		57 32	08	54	26	42	48	22
40	28 13		66 07	348 46		83 22		315 33		59 57	06	52	32	40	47	24
50	30 43	N14 55	68 37	351 16	N24 50	85 52	S 9 00	318 03	N22 31	62 22	S 1 04	50	38	37	46	25
14 00	33 13		71 08	353 46		88 22		320 34		64 47	02	45	4 51	33	43	30
10	35 43		73 38	356 16		90 52		323 04		67 12	1 00	40	5 02	30	41	33
20	38 13		76 09	358 46		93 23		325 34		69 37	0 59	35	10	28	39	36
30	40 43		78 39	1 16		95 53		328 05		72 02	57	30	18	26	38	37
40	43 13		81 10	3 45		98 23		330 35		74 27	55	20	32	24	35	42
50												10	43	22	33	45
15 00	45 43	N14 56	83 40	6 15	N24 50	100 53	S 8 59	333 05	N22 31	76 52	S 0 53	0	5 54	22	30	49
10	48 13		86 10	8 45		103 23		335 36		79 17	51	0				
20	50 43		88 41	11 15		105 53		338 06		81 42	49	10	6 04	22	28	52
30	53 13		91 11	13 45		108 23		340 37		84 07	47	20	15	23	26	56
40	55 43		93 42	16 15		110 53		343 07		86 32	45	30	28	24	23	60
50	58 13		96 12	18 45		113 23		345 37		88 58	43	35	35	26	21	63
16 00	60 43	N14 57	98 42	21 15	N24 51	115 54	S 8 58	348 08	N22 31	91 23	S 0 41	35	35	26	21	63
10	63 14		101 13	23 44		118 24		350 38		93 48	39	40	43	28	20	65
20	65 44		103 43	26 14		120 54		353 08		96 13	37	45	6 53	30	18	68
30	68 14		106 14	28 44		123 24		355 39		98 38	36	50	7 04	33	15	73
40	70 44		108 44	31 14		125 54		358 09		101 03	34	52	09	34	14	74
50	73 14		111 14	33 44		128 24		0 39		103 28	32	54	15	36	13	76
17 00	75 44	N14 57	113 45	36 14	N24 51	130 54	S 8 58	3 10	N22 31	105 53	S 0 30	56	22	38	11	79
10	78 14		116 15	38 44		133 24		5 40		108 18	28	58	29	40	10	81
20	80 44		118 46	41 14		135 55		8 10		110 43	26	60	7 37	43	3 08	84
30	83 14		121 16	43 44		138 25		10 41		113 08	24	S				
40	85 44		123 47	46 13		140 55		13 11		115 33	22					
50	88 14		126 17	48 43		143 25		15 41		117 58	20					
18 00	90 44	N14 58	128 47	51 13	N24 51	145 55	S 8 57	18 12	N22 31	120 23	S 0 18	Lat.	Sun-set	Twil.	Moon-set	Diff.
10	93 14		131 18	53 43		148 25		20 42		122 48	16					
20	95 44		133 48	56 13		150 55		23 12		125 14	14					
30	98 14		136 19	58 43		153 25		25 43		127 39	13					
40	100 44		138 49	61 13		155 55		28 13		130 04	11	N				
50	103 14		141 19	63 43		158 26		30 44		132 29	09	0	h m	m	h m	m
19 00	105 44	N14 59	143 50	66 12	N24 52	160 56	S 8 56	33 14	N22 31	134 54	S 0 07	70	21 22	NH	15 44	105
10	108 14		146 20	68 42		163 26		35 44		137 19	05	68	20 56	109	45	99
20	110 44		148 51	71 12		165 56		38 15		139 44	03	66	36	82	46	94
30	113 14		151 21	73 42		168 26		40 45		142 09	S 0 01	64	20	68	47	89
40	115 44		153 51	76 12		170 56		43 15		144 34	N 0 01	62	20 07	59	47	87
50	118 14		156 22	78 42		173 26		45 46		146 59	03	60	19 55	54	48	83
20 00	120 44	N15 00	158 52	81 12	N24 52	175 56	S 8 55	48 16	N22 30	149 24	N 0 05	58	45	48	48	81
10	123 14		161 23	83 42		178 26		50 46		151 49	07	56	37	44	49	78
20	125 44		163 53	86 11		180 57		53 17		154 14	08	54	30	41	49	76
30	128 14		166 24	88 41		183 27		55 47		156 40	10	52	23	39	50	74
40	130 44		168 54	91 11		185 57		58 17		159 05	12	50	17	36	50	72
50	133 14		171 24	93 41		188 27		60 48		161 30	14	45	19 04	32	50	69
21 00	135 44	N15 01	173 55	96 11	N24 52	190 57	S 8 55	63 18	N22 30	163 55	N 0 16	35	44	27	52	62
10	138 14		176 25	98 41		193 27		65 48		166 20	18	30	37	25	52	60
20	140 44		178 56	101 11		195 57		68 19		168 45	20	20	23	23	53	56
30	143 14		181 26	103 41		198 27		70 49		171 10	22	10	12	22	53	52
40	145 44		183 56	106 11		200 58		73 19		173 35	24					
50	148 14		186 27	108 40		203 28		75 50		176 00	26	0	18 01	22	54	48
22 00	150 44	N15 01	188 57	111 10	N24 53	205 58	S 8 54	78 20	N22 30	178 25	N 0 28	10	17 49	22	54	46
10	153 14		191 28	113 40		208 28		80 51		180 50	30	10	38	23	55	41
20	155 44		193 58	116 10		210 58		83 21		183 15	31	30	25	24	55	38
30	158 14		196 28	118 40		213 28		85 51		185 41	33	30	18	26	56	35
40	160 44		198 59	121 10		215 58		88 22		188 06	35	35	10	28	56	33
50	162 14		201 29	123 40		218 28		90 52		190 31	37	40	10	28	56	33
23 00	165 44	N15 02	204 00	126 10	N24 53	220 58	S 8 53	93 22	N22 30	192 56	N 0 39	45	17 00	31	56	30
10	168 14		206 30	128 39		223 29		95 53		195 21	41	50	16 48	33	57	26
20	170 44		209 00	131 09		225 59		98 23		197 46	43	52	43	35	57	24
30	173 14		211 31	133 39		228 29		100 53		200 11	45	54	37	37	57	23
40	175 44		214 01	136 09		230 59		103 24		202 36	47	56	31	40	58	20
50	178 14		216 32	138 39		233 29		105 54		205 01	49	58	23	42	58	18
24 00	180 44	N15 03	219 02	141 09	N24 53	235 59	S 8 53	108 24	N22 30	207 26	N 0 51	S	16 15	44	15 58	16

APPENDIX

GREENWICH A. M. 1943 MAY 5 (WEDNESDAY)

GCT	☉ SUN		☿	VENUS - 3.5		MARS 1.1	JUPITER - 1.6		● MOON		☾
	GHA	Dec.	GHA	GHA	Dec.	GHA	GHA	Dec.	GHA	Dec.	Par.
h m	° ' "		° ' "	° ' "		° ' "	° ' "		° ' "		
0 00	180 49 N15 56		222 00	140 17 N25 15		236 49 S 8 02	110 54 N22 27		172 17 N13 07		
10	183 19		224 30	142 46		239 19	113 25		174 42		
20	185 49		227 00	145 16		241 50	115 55		177 07		
30	188 19		229 31	147 46		244 20	118 25		179 33		
40	190 49		232 01	150 16		246 50	120 56		181 58		
50	193 19		234 32	152 46		249 20	123 26		184 23		
1 00	195 49 N15 57		237 02	155 16 N25 15		251 50 S 8 02	125 56 N22 27		186 48 N13 15		
10	198 19		239 32	157 46		254 20	128 27		189 13		
20	200 49		242 03	160 16		256 50	130 57		191 38		
30	203 19		244 33	162 45		259 20	133 27		194 03		
40	205 49		247 04	165 15		261 50	135 58		196 28		
50	208 19		249 34	167 45		264 21	138 28		198 53		
2 00	210 49 N15 57		252 04	170 15 N25 15		266 51 S 8 01	140 58 N22 27		201 18 N13 24		
10	213 19		254 35	172 45		269 21	143 29		203 43		
20	215 49		257 05	175 15		271 51	145 59		206 09		
30	218 19		259 36	177 45		274 21	148 29		208 34		
40	220 49		262 06	180 15		276 51	151 00		210 59		
50	223 19		264 37	182 44		279 21	153 30		213 24		
3 00	225 49 N15 58		267 07	185 14 N25 15		281 51 S 8 00	156 01 N22 27		215 49 N13 32		
10	228 19		269 37	187 44		284 22	158 31		218 14		
20	230 49		272 08	190 14		286 52	161 01		220 39		
30	233 19		274 38	192 44		289 22	163 32		223 04		
40	235 49		277 09	195 14		291 52	166 02		225 29		
50	238 19		279 39	197 44		294 22	168 32		227 54		
4 00	240 49 N15 59		282 09	200 14 N25 16		296 52 S 8 00	171 03 N22 27		230 19 N13 40		
10	243 19		284 40	202 43		299 22	173 33		232 45		
20	245 49		287 10	205 13		301 52	176 03		235 10		
30	248 19		289 41	207 43		304 22	178 34		237 35		
40	250 49		292 11	210 13		306 53	181 04		240 00		
50	253 19		294 41	212 43		309 23	183 34		242 25		
5 00	255 49 N16 00		297 12	215 13 N25 16		311 53 S 7 59	186 05 N22 27		244 50 N13 48		
10	258 19		299 42	217 43		314 23	188 35		247 15		
20	260 49		302 13	220 13		316 53	191 05		249 40		
30	263 19		304 43	222 43		319 23	193 36		252 05		
40	265 49		307 14	225 12		321 53	196 06		254 30		
50	268 19		309 44	227 42		324 23	198 36		256 55		
6 00	270 49 N16 00		312 14	230 12 N25 16		326 54 S 7 58	201 07 N22 26		259 21 N13 56		
10	273 19		314 45	232 42		329 24	203 37		261 46		
20	275 49		317 15	235 12		331 54	206 07		264 11		
30	278 19		319 46	237 42		334 24	208 38		266 36		
40	280 49		322 16	240 12		336 54	211 08		269 01		
50	283 19		324 46	242 42		339 24	213 38		271 26		
7 00	285 49 N16 01		327 17	245 11 N25 16		341 54 S 7 57	216 09 N22 26		273 51 N14 03		
10	288 19		329 47	247 41		344 24	218 39		276 16		
20	290 49		332 18	250 11		346 54	221 09		278 41		
30	293 19		334 48	252 41		349 25	223 40		281 06		
40	295 49		337 18	255 11		351 55	226 10		283 31		
50	298 19		339 49	257 41		354 25	228 41		285 56		
8 00	300 50 N16 02		342 19	260 11 N25 17		356 55 S 7 57	231 11 N22 26		288 22 N14 11		
10	303 20		344 50	262 41		359 25	233 41		290 47		
20	305 50		347 20	265 10		1 55	236 12		293 12		
30	308 20		349 50	267 40		4 25	238 42		295 37		
40	310 50		352 21	270 10		6 55	241 12		298 02		
50	313 20		354 51	272 40		9 26	243 43		300 27		
9 00	315 50 N16 02		357 22	275 10 N25 17		11 56 S 7 56	246 13 N22 26		302 52 N14 19		
10	318 20		359 52	277 40		14 26	248 43		305 17		
20	320 50		2 23	280 10		16 56	251 14		307 42		
30	323 20		4 53	282 40		19 26	253 44		310 07		
40	325 50		7 23	285 09		21 56	256 14		312 32		
50	328 20		9 54	287 39		24 26	258 45		314 57		
10 00	330 50 N16 03		12 24	290 09 N25 17		26 56 S 7 55	261 15 N22 26		317 22 N14 26		
10	333 20		14 55	292 39		29 26	263 45		319 47		
20	335 50		17 25	295 09		31 57	266 16		322 13		
30	338 20		19 55	297 39		34 27	268 46		324 38		
40	340 50		22 26	300 09		36 57	271 16		327 03		
50	343 20		24 56	302 39		39 27	273 47		329 28		
11 00	345 50 N16 04		27 27	305 08 N25 17		41 57 S 7 55	276 17 N22 26		331 53 N14 34		
10	348 20		29 57	307 38		44 27	278 47		334 18		
20	350 50		32 27	310 08		46 57	281 18		336 43		
30	353 20		34 58	312 38		49 27	283 48		339 08		
40	355 50		37 28	315 08		51 58	286 18		341 33		
50	358 20		39 59	317 38		54 28	288 49		343 58		
12 00	0 50 N16 05		42 29	320 08 N25 18		56 58 S 7 54	291 19 N22 26		346 23 N14 41		



AMERICAN AIR NAVIGATOR

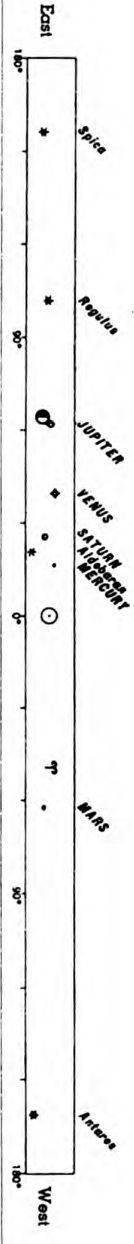
GREENWICH P. M. 1943 MAY 5 (WEDNESDAY)

GCT	SUN GHA Dec.	VENUS - 3.5 GHA Dec.	MARS 1.1 GHA Dec.	JUPITER - 1.6 GHA Dec.	MOON GHA Dec.	Lat	Sun- rise	Lat	Moon- rise	Dist.
h m	° 0' 50" N16 05	42 29 320 08 N25 18	56 58 0 54	291 19 N22 26	346 23 N14 41	N	h m	m	h m	m
12 00	3 20	45 00 322 38	59 28	293 49	348 48 42	70	2 12	NH	3 53	*
10	5 50	47 30 325 08	61 58	296 20	351 14 44	2 43	148	4 12	07	
20	8 20	50 00 327 37	64 28	298 50	353 39 45	66	3 05	89	28	13
30	10 50	52 31 330 07	66 58	301 20	356 04 46	64	23	72	40	19
40	13 20	55 01 332 37	69 28	303 51	358 29 47	62	38	62	4 52	21
50	15 50 N16 05	57 32 335 07 N25 18	71 58 S 7 53	306 21 N22 26	0 54 N14 49	60	3 50	57	5 01	24
13 00	18 20	60 02 337 37	74 29	308 52	3 19 50	58	4 00	51	09	27
10	20 50	62 32 340 07	76 59	311 22	5 44 51	56	10	47	17	28
20	23 20	65 03 342 37	79 29	313 52	8 09 52	54	18	44	23	30
30	25 50	67 33 345 07	81 59	316 23	10 34 53	52	25	41	29	31
40	28 20	70 04 347 36	84 29	318 53	12 59 55	50	31	38	34	33
50	30 50 N16 06	72 34 350 06 N25 18	86 59 S 7 53	321 23 N22 26	15 24 N14 56	48	4 57	31	5 55	38
14 00	33 20	75 04 352 36	89 29	323 54	17 49 57	46	5 06	28	6 03	40
10	35 50	77 35 355 06	91 59	326 24	20 14 58	44	15	26	10	42
20	38 20	80 05 357 36	94 29	328 54	22 40 14	42	20	29	24	44
30	40 50	82 36 0 06	97 00	331 25	25 05 15	40	22	34	27	46
40	43 20	85 06 2 36	99 30	333 55	27 30 02	38	24	39	31	48
50						36	26	44	35	50
15 00	45 50 N16 07	87 37 5 06 N25 18	102 00 S 7 52	336 25 N22 26	29 55 N15 03	34	28	40	37	51
10	48 20	90 07 7 35	104 30	338 56	32 20 04	32	30	42	39	53
20	50 50	92 37 10 05	107 00	341 26	34 45 05	30	32	44	37	55
30	53 20	95 08 12 35	109 30	343 56	37 10 07	28	34	46	35	57
40	55 50	97 38 15 05	112 00	346 27	39 35 08	26	36	48	33	59
50	58 20	100 09 17 35	114 30	348 57	42 00 09	24	38	50	31	61
16 00	60 50 N16 08	102 39 20 05 N25 19	117 01 S 7 51	351 27 N22 26	44 25 N15 10	22	40	52	29	63
10	63 20	105 09 22 35	119 31	353 58	46 50 11	20	42	54	27	65
20	65 50	107 40 25 05	122 01	356 28	49 15 12	18	44	56	25	67
30	68 20	110 10 27 34	124 31	358 58	51 40 14	16	46	58	23	69
40	70 50	112 41 30 04	127 01	1 29	54 05 15	14	48	60	21	71
50	73 20	115 11 32 34	129 31	3 59	56 31 16	12	50	62	19	73
17 00	75 50 N16 08	117 41 35 04 N25 19	132 01 S 7 50	6 29 N22 26	58 56 N15 17	10	52	64	17	75
10	78 20	120 12 37 34	134 31	9 00	61 21 18	8	54	66	15	77
20	80 50	122 42 40 04	137 01	11 30	63 46 19	6	56	68	13	79
30	83 20	125 13 42 34	139 32	14 00	66 11 21	4	58	70	11	81
40	85 50	127 43 45 04	142 02	16 31	68 36 22	2	60	72	9	83
50	88 20	130 13 47 33	144 32	19 01	71 01 23	0	62	74	7	85
18 00	90 50 N16 09	132 44 50 03 N25 19	147 02 S 7 50	21 32 N22 26	73 26 N15 24	0	64	76	5	87
10	93 20	135 14 52 33	149 32	24 02	75 51 25	0	66	78	3	89
20	95 50	137 45 55 03	152 02	26 32	78 16 26	0	68	80	1	91
30	98 20	140 15 57 33	154 32	29 03	80 41 27	0	70	82	0	93
40	100 50	142 46 60 03	157 02	31 33	83 06 29	0	72	84	0	95
50	103 20	145 16 62 33	159 33	34 03	85 31 30	0	74	86	0	97
19 00	105 50 N16 10	147 46 65 03 N25 19	162 03 S 7 49	36 34 N22 26	87 56 N15 31	0	76	88	0	99
10	108 20	150 17 67 32	164 33	39 04	90 21 32	0	78	90	0	101
20	110 50	152 47 70 02	167 03	41 34	92 47 33	0	80	92	0	103
30	113 20	155 18 72 32	169 33	44 05	95 12 34	0	82	94	0	105
40	115 50	157 48 75 02	172 03	46 35	97 37 35	0	84	96	0	107
50	118 20	160 18 77 32	174 33	49 05	100 02 37	0	86	98	0	109
20 00	120 50 N16 10	162 49 80 02 N25 20	177 03 S 7 48	51 36 N22 26	102 27 N15 38	0	88	100	0	111
10	123 20	165 19 82 32	179 33	54 06	104 52 39	0	90	102	0	113
20	125 50	167 50 85 02	182 04	56 36	107 17 40	0	92	104	0	115
30	128 20	170 20 87 32	184 34	59 07	109 42 41	0	94	106	0	117
40	130 50	172 50 90 01	187 04	61 37	112 07 42	0	96	108	0	119
50	133 20	175 21 92 31	189 34	64 07	114 32 43	0	98	110	0	121
21 00	135 50 N16 11	177 51 95 01 N25 20	192 04 S 7 48	66 38 N22 26	116 57 N15 44	0	100	112	0	123
10	138 20	180 22 97 31	194 34	69 08	119 22 45	0	102	114	0	125
20	140 50	182 52 100 01	197 04	71 38	121 47 46	0	104	116	0	127
30	143 20	185 23 102 31	199 34	74 09	124 12 47	0	106	118	0	129
40	145 50	187 53 105 01	202 05	76 39	126 37 48	0	108	120	0	131
50	148 20	190 23 107 31	204 35	79 09	129 02 50	0	110	122	0	133
22 00	150 50 N16 12	192 54 110 00 N25 20	207 05 S 7 47	81 40 N22 26	131 28 N15 51	0	112	124	0	135
10	153 20	195 24 112 30	209 35	84 10	133 53 52	0	114	126	0	137
20	155 50	197 55 115 00	212 05	86 40	136 18 53	0	116	128	0	139
30	158 20	200 25 117 30	214 35	89 11	138 43 54	0	118	130	0	141
40	160 50	202 55 120 00	217 05	91 41	141 08 55	0	120	132	0	143
50	163 20	205 26 122 30	219 35	94 12	143 33 56	0	122	134	0	145
23 00	165 50 N16 13	207 56 125 00 N25 20	222 05 S 7 46	96 42 N22 26	145 58 N15 58	0	124	136	0	147
10	168 20	210 27 127 30	224 36	99 12	148 23 59	0	126	138	0	149
20	170 50	212 57 129 59	227 06	101 43	150 48 60	0	128	140	0	151
30	173 20	215 27 132 29	229 36	104 13	153 13 01	0	130	142	0	153
40	175 50	217 58 134 59	232 06	106 43	155 38 02	0	132	144	0	155
50	178 20	220 28 137 29	234 36	109 14	158 03 03	0	134	146	0	157
24 00	180 50 N16 13	222 59 139 59 N25 21	237 06 S 7 46	111 44 N22 26	160 28 N16 04	0	136	148	0	159

APPENDIX

GREENWICH A. M. 1943 MAY 10 (MONDAY)

GCT	SUN		VENUS - 3.6		MARS 0.9		JUPITER - 1.5		MOON		☾
	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	Par.
0 00	180 55	N17 20	226 55	138 50	N25 37	238 14	S 6 38	115 02	N22 20	113 15	N19 15
10	183 25		229 26	141 19		240 44		117 32		115 40	14
20	185 55		231 56	143 49		243 14		120 02		118 05	14
30	188 25		234 26	146 19		245 45		122 33		120 31	14
40	190 55		236 57	148 49		248 15		125 03		122 56	13
50	193 25		239 27	151 19		250 45		127 33		125 21	13
1 00	195 55	N17 20	241 58	153 49	N25 37	253 15	S 6 37	130 04	N22 20	127 46	N19 12
10	198 25		244 28	156 19		255 45		132 34		130 11	12
20	200 55		246 59	158 49		258 15		135 04		132 36	12
30	203 25		249 29	161 18		260 45		137 35		135 02	11
40	205 55		251 59	163 48		263 15		140 05		137 27	11
50	208 25		254 30	166 18		265 46		142 36		139 52	10
2 00	210 55	N17 21	257 00	168 48	N25 37	268 16	S 6 36	145 06	N22 20	142 17	N19 10
10	213 25		259 31	171 18		270 46		147 36		144 42	09
20	215 55		262 01	173 48		273 16		150 07		147 08	09
30	218 25		264 31	176 18		275 46		152 37		149 33	09
40	220 55		267 02	178 48		278 16		155 07		151 58	08
50	223 25		269 32	181 17		280 46		157 38		154 23	08
3 00	225 55	N17 22	272 03	183 47	N25 37	283 16	S 6 35	160 08	N22 20	156 48	N19 07
10	228 25		274 33	186 17		285 47		162 38		159 14	07
20	230 55		277 03	188 47		288 17		165 09		161 39	06
30	233 25		279 34	191 17		290 47		167 39		164 04	06
40	235 55		282 04	193 47		293 17		170 09		166 29	05
50	238 25		284 35	196 17		295 47		172 40		168 54	05
4 00	240 55	N17 22	287 05	198 47	N25 37	298 17	S 6 35	175 10	N22 20	171 20	N19 04
10	243 25		289 36	201 17		300 47		177 40		173 45	04
20	245 55		292 06	203 46		303 17		180 11		176 10	03
30	248 25		294 36	206 16		305 48		182 41		178 35	03
40	250 55		297 07	208 46		308 18		185 11		181 01	03
50	253 25		299 37	211 16		310 48		187 42		183 26	02
5 00	255 55	N17 23	302 08	213 46	N25 37	313 18	S 6 34	190 12	N22 20	185 51	N19 02
10	258 25		304 38	216 16		315 48		192 42		188 16	01
20	260 55		307 08	218 46		318 18		195 13		190 41	01
30	263 25		309 39	221 16		320 48		197 43		193 07	00
40	265 55		312 09	223 45		323 18		200 13		195 32	19
50	268 25		314 40	226 15		325 48		202 44		197 57	18
6 00	270 55	N17 24	317 10	228 45	N25 37	328 19	S 6 33	205 14	N22 20	200 22	N18 59
10	273 25		319 40	231 15		330 49		207 44		202 47	58
20	275 55		322 11	233 45		333 19		210 15		205 13	58
30	278 25		324 41	236 15		335 49		212 45		207 38	57
40	280 55		327 12	238 45		338 19		215 15		210 03	57
50	283 25		329 42	241 15		340 49		217 46		212 28	56
7 00	285 55	N17 24	332 12	243 45	N25 37	343 19	S 6 33	220 16	N22 20	214 54	N18 56
10	288 25		334 43	246 14		345 49		222 46		217 19	55
20	290 55		337 13	248 44		348 20		225 17		219 44	55
30	293 25		339 44	251 14		350 50		227 47		222 09	54
40	295 55		342 14	253 44		353 20		230 17		224 34	54
50	298 25		344 45	256 14		355 50		232 48		227 00	53
8 00	300 55	N17 25	347 15	258 44	N25 38	358 20	S 6 32	235 18	N22 20	229 25	N18 53
10	303 25		349 45	261 14		0 50		237 48		231 50	52
20	305 55		352 16	263 44		3 20		240 19		234 15	51
30	308 25		354 46	266 13		5 50		242 49		236 41	51
40	310 55		357 17	268 43		8 21		245 19		239 06	50
50	313 25		359 47	271 13		10 51		247 50		241 31	50
9 00	315 55	N17 26	2 17	273 43	N25 38	13 21	S 6 31	250 20	N22 20	243 56	N18 49
10	318 25		4 48	276 13		15 51		252 51		246 21	49
20	320 55		7 18	278 43		18 21		255 21		248 47	48
30	323 25		9 49	281 13		20 51		257 51		251 12	48
40	325 55		12 19	283 43		23 21		260 22		253 37	47
50	328 25		14 49	286 12		25 51		262 52		256 02	47
10 00	330 55	N17 26	17 20	288 42	N25 38	28 21	S 6 30	265 22	N22 20	258 28	N18 46
10	333 25		19 50	291 12		30 52		267 53		260 53	45
20	335 55		22 21	293 42		33 22		270 23		263 18	45
30	338 25		24 51	296 12		35 52		272 53		265 43	44
40	340 55		27 22	298 42		38 22		275 24		268 08	44
50	343 25		29 52	301 12		40 52		277 54		270 34	43
11 00	345 55	N17 27	32 22	303 42	N25 38	43 22	S 6 30	280 24	N22 20	272 59	N18 43
10	348 25		34 53	306 12		45 52		282 55		275 24	42
20	350 55		37 23	308 41		48 22		285 25		277 49	41
30	353 25		39 54	311 11		50 53		287 55		280 15	41
40	355 55		42 24	313 41		53 23		290 26		282 40	40
50	358 25		44 54	316 11		55 53		292 56		285 05	40
12 00	0 55	N17 28	47 25	318 41	N25 38	58 23	S 6 29	295 26	N22 20	287 30	N18 39



AMERICAN AIR NAVIGATOR

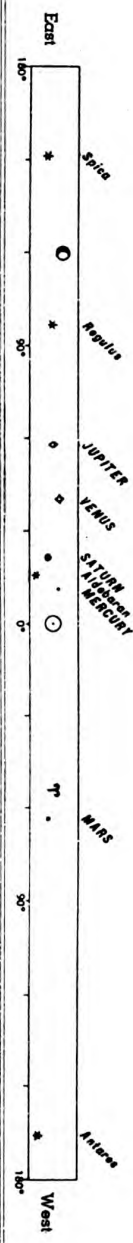
GREENWICH P. M. 1943 MAY 10 (MONDAY)

GCT		O SUN		VENUS - 3.6		MARS 0.9		JUPITER - 1.5		MOON		Lat.	Sun-rise	Twil.	Moon-rise	Diff.	
		GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.						
h	m	0	55	N17	28	47	25	318	41	N25	38						
12	00	3	25			49	55	321	11			N					
10		5	55			52	26	323	41			0	h	m	m	m	
20		8	25			54	56	326	11			1	38	NA	5	57	
30		10	55			57	26	328	40			2	18	NA	6	49	
40		13	25			59	57	331	10			2	46	114	7	21	
50		15	55	N17	28	62	27	333	40	N25	38	3	06	80	7	45	
13	00	18	25			64	58	336	10			3	62	23	67	8	
10		20	55			67	28	338	40			6	23	67	8	03	
20		23	25			69	59	341	10			3	59	49	31	19	
30		25	55			72	29	343	40			4	58	45	32	62	
40		28	25			74	59	346	10			5	49	49	43	60	
50		30	55	N17	29	77	30	348	40	N25	38	16	41	9	02	58	
14	00	33	25			80	00	351	09			2	39	10	10	57	
10		35	55			82	31	353	39			3	38	34	26	55	
20		38	25			85	01	356	09			4	51	31	40	54	
30		40	55			87	31	358	39			5	01	28	9	51	
40		43	25			90	02	1	09			11	26	10	01	52	
50												20	26	24	19	49	
												40	40	22	34	48	
15	00	45	55	N17	30	92	32	3	39	N25	38	0	5	53	22	10	48
10		48	25			95	03	6	09			5	53	22	10	48	47
20		50	55			97	33	8	39								
30		53	25			100	03	11	08			10	6	05	22	11	02
40		55	55			102	34	13	38			20	18	23	17	44	
50		58	25			105	04	16	08			30	34	24	35	41	
16	00	60	55	N17	30	107	35	18	38	N25	38	35	43	26	45	40	
10		63	25			110	05	21	08			4	6	52	28	11	56
20		65	55			112	35	23	38			7	04	31	12	10	
30		68	25			115	06	26	08			18	35	26	36		
40		70	55			117	36	28	38			22	52	25	36	34	
50		73	25			120	07	31	07			34	54	32	38	42	
17	00	75	55	N17	31	122	37	33	37	N25	38	21	56	40	12	52	
10		78	25			125	08	36	07			7	49	43	13	03	
20		80	55			127	38	38	37			8	00	47	13	16	
30		83	25			130	08	41	07			S					
40		85	55			132	39	43	37								
50		88	25			135	09	46	07								
18	00	90	55	N17	32	137	40	48	37	N25	39						
10		93	25			140	10	51	07			Lat.	Sun-set	Twil.	Moon-set	Diff.	
20		95	55			142	40	53	36			N	h	m	m	m	
30		98	25			145	11	56	06			0	h	m	m	m	
40		100	55			147	41	58	36			20	22	21	NA	3	
50		103	25			150	12	61	06			20	49	82	31	19	
19	00	105	55	N17	32	152	42	63	36	N25	39	70	22	21	NA	2	
10		108	25			155	12	66	06			12	68	21	38	NA	2
20		110	55			157	43	68	36			11	66	21	10	121	1
30		113	25			160	13	71	06			11	64	20	49	82	31
40		115	55			162	44	73	35			10	62	32	68	1	12
50		118	25			165	14	76	05			09	60	17	60	0	56
20	00	120	55	N17	33	167	45	78	35	N25	39	09	58	20	05	52	43
10		123	25			170	15	81	05			19	55	47	32	36	
20		125	55			172	45	83	35			19	55	47	32	36	
30		128	25			175	16	86	05			07	52	38	41	13	
40		130	55			177	46	88	35			06	50	30	38	0	
50		133	25			180	17	91	05			05	45	15	34	24	
21	00	135	55	N17	34	182	47	93	35	N25	39	40	19	02	30	15	
10		138	25			185	17	96	04			18	52	27	24	04	
20		140	55			187	48	98	34			30	42	25	23	55	
30		143	25			190	18	101	04			20	26	23	39	42	
40		145	55			192	49	103	34			13	22	24	45		
50		148	25			195	19	106	04			0	18	00	22	23	
22	00	150	55	N17	34	197	49	108	34	N25	39	10	17	47	22	22	
10		153	25			200	20	111	04			33	23	23	43	50	
20		155	55			202	50	113	34			59	20	33	25	52	
30		158	25			205	21	116	03			30	18	25	26	52	
40		160	55			207	51	118	33			35	17	09	27	16	
50		163	25			210	21	121	03			40	16	59	29	22	
23	00	165	55	N17	35	212	52	123	33	N25	39	56	48	32	21	52	
10		168	25			215	22	126	03			55	50	33	35	36	
20		170	55			217	53	128	33			52	27	37	29	60	
30		173	25			220	23	131	03			54	20	39	20	62	
40		175	55			222	54	133	33			56	11	41	11	63	
50		178	25			225	24	136	02			58	16	02	44	21	
24	00	180	55	N17	36	227	54	138	32	N25	39	60	15	52	48	20	

APPENDIX

GREENWICH A. M. 1943 MAY 15 (SATURDAY)

GCT	SUN			VENUS - 3.6			MARS 0.9			JUPITER - 1.5			MOON			Par.
	GHA	Dec.		GHA	Dec.		GHA	Dec.		GHA	Dec.		GHA	Dec.		
h m	° ' "		° ' "	° ' "		° ' "	° ' "		° ' "	° ' "		° ' "	° ' "		° ' "	
0 00	180 56	N18 36	231 51	137 25	N25 42	239 41	S 5 12	119 06	N22 13	57 54	N 5 32					
10	183 26		234 21	139 55		242 11		121 37		60 20						
20	185 56		236 52	142 21		244 41		124 07		62 45						
30	188 26		239 22	144 54		247 11		126 37		65 11						
40	190 56		241 53	147 21		249 41		129 08		67 36						
50	193 26		244 23	149 54		252 12		131 38		70 02						
1 00	195 56	N18 37	246 53	152 24	N25 42	254 42	S 5 11	134 08	N22 13	72 27	N 5 22					
10	198 26		249 24	154 54		257 12		136 39		74 52						
20	200 56		251 54	157 24		259 42		139 09		77 18						
30	203 26		254 25	159 54		262 12		141 39		79 43						
40	205 56		256 55	162 24		264 42		144 10		82 09						
50	208 26		259 25	164 53		267 12		146 40		84 34						
2 00	210 56	N18 37	261 56	167 23	N25 42	269 42	S 5 10	149 10	N22 13	87 00	N 5 11					
10	213 26		264 26	169 53		272 13		151 41		89 25						
20	215 56		266 57	172 23		274 43		154 11		91 51						
30	218 26		269 27	174 53		277 13		156 41		94 16						
40	220 56		271 57	177 23		279 43		159 12		96 41						
50	223 26		274 28	179 53		282 13		161 42		99 07						
3 00	225 56	N18 38	276 58	182 23	N25 41	284 43	S 5 09	164 13	N22 13	101 32	N 5 01					
10	228 26		279 29	184 53		287 13		166 43		103 58						
20	230 56		281 59	187 22		289 43		169 13		106 23						
30	233 26		284 30	189 52		292 14		171 44		108 49						
40	235 56		287 00	192 22		294 44		174 14		111 14						
50	238 26		289 30	194 52		297 14		176 44		113 40						
4 00	240 56	N18 39	292 01	197 22	N25 41	299 44	S 5 09	179 15	N22 13	116 05	N 4 51					
10	243 26		294 31	199 52		302 14		181 45		118 30						
20	245 56		297 02	202 22		304 44		184 15		120 56						
30	248 26		299 32	204 52		307 14		186 46		123 21						
40	250 56		302 02	207 21		309 44		189 16		125 47						
50	253 26		304 33	209 51		312 14		191 46		128 12						
5 00	255 56	N18 39	307 03	212 21	N25 41	314 45	S 5 08	194 17	N22 13	130 38	N 4 40					
10	258 26		309 34	214 51		317 15		196 47		133 03						
20	260 56		312 04	217 21		319 45		199 17		135 29						
30	263 26		314 34	219 51		322 15		201 48		137 54						
40	265 56		317 05	222 21		324 45		204 18		140 19						
50	268 26		319 35	224 51		327 15		206 48		142 45						
6 00	270 56	N18 40	322 06	227 21	N25 41	329 45	S 5 07	209 19	N22 13	145 10	N 4 30					
10	273 26		324 36	229 50		332 15		211 49		147 36						
20	275 56		327 07	232 20		334 46		214 19		150 01						
30	278 26		329 37	234 50		337 16		216 50		152 27						
40	280 56		332 07	237 20		339 46		219 20		154 52						
50	283 26		334 38	239 50		342 16		221 50		157 17						
7 00	285 56	N18 40	337 08	242 20	N25 41	344 46	S 5 07	224 21	N22 13	159 43	N 4 19					
10	288 26		339 39	244 50		347 16		226 51		162 08						
20	290 56		342 09	247 20		349 46		229 21		164 34						
30	293 26		344 39	249 50		352 16		231 52		166 59						
40	295 56		347 10	252 19		354 47		234 22		169 25						
50	298 26		349 40	254 49		357 17		236 52		171 50						
8 00	300 56	N18 41	352 11	257 19	N25 41	359 47	S 5 06	239 23	N22 13	174 16	N 4 09					
10	303 26		354 41	259 49		2 17		241 53		176 41						
20	305 56		357 11	262 19		4 47		244 23		179 06						
30	308 26		359 42	264 49		7 17		246 54		181 32						
40	310 56		2 12	267 19		9 47		249 24		183 57						
50	313 26		4 43	269 49		12 17		251 54		186 23						
9 00	315 56	N18 42	7 13	272 19	N25 41	14 48	S 5 05	254 25	N22 13	188 48	N 3 58					
10	318 26		9 44	274 48		17 18		256 55		191 13						
20	320 56		12 14	277 18		19 48		259 25		193 39						
30	323 26		14 44	279 48		22 18		261 56		196 04						
40	325 56		17 15	282 18		24 48		264 26		198 30						
50	328 26		19 45	284 48		27 18		266 56		200 55						
10 00	330 56	N18 42	22 16	287 18	N25 41	29 48	S 5 04	269 27	N22 13	203 21	N 3 48					
10	333 26		24 46	289 48		32 18		271 57		205 46						
20	335 56		27 16	292 18		34 49		274 27		208 11						
30	338 26		29 47	294 48		37 19		276 58		210 37						
40	340 56		32 17	297 17		39 49		279 28		213 02						
50	343 26		34 48	299 47		42 19		281 58		215 28						
11 00	345 56	N18 43	37 18	302 17	N25 41	44 49	S 5 04	284 29	N22 13	217 53	N 3 37					
10	348 26		39 48	304 47		47 19		286 59		220 19						
20	350 56		42 19	307 17		49 49		289 29		222 44						
30	353 26		44 49	309 47		52 19		292 00		225 09						
40	355 56		47 20	312 17		54 50		294 30		227 35						
50	358 26		49 50	314 47		57 20		297 00		230 00						
12 00	0 57	N18 43	52 21	317 16	N25 41	59 50	S 5 03	299 31	N22 13	232 26	N 3 27					



AMERICAN AIR NAVIGATOR

GREENWICH P. M. 1943 MAY 15 (SATURDAY)

GCT	☉ SUN		☿	♀ VENUS - 3.6		♂ MARS 0.9	♃ JUPITER - 1.5		♄ MOON		Lat.	Sun-rise	Twil.	Moon-rise	Diff.
	GHA	Dec.	GHA	GHA	Dec.	GHA	GHA	Dec.	GHA	Dec.					
h m	° ' "		° ' "	° ' "		° ' "	° ' "		° ' "			h m	m	h m	m
12 00	0 57	N18 43	52 21	317 16	N25 41	59 50	S 5 03	299 31	N22 13	232 26	N 3 27	N			
10	3 27		54 51	319 46		62 20		302 01		234 51		0	5 53	22	36 46
20	5 57		57 21	322 16		64 50		304 31		237 16		70	0 53	14	02 99
30	8 27		59 52	324 46		67 20		307 02		239 42		68	1 52	05	94
40	10 57		62 22	327 16		69 50		309 32		242 07		66	2 26	08	89
50	13 27		64 53	329 46		72 20		312 02		244 33		64	2 50	10	86
13 00	15 57	N18 44	67 23	332 16	N25 41	74 50	S 5 02	314 33	N22 13	246 58	N 3 16	62	3 09	12	82
10	18 27		69 53	334 46		77 21		317 03		249 24		60	25 65	14	79
20	20 57		72 24	337 16		79 51		319 33		251 49		58	38 56	16	76
30	23 27		74 54	339 45		82 21		322 04		254 14		56	49 51	17	75
40	25 57		77 25	342 15		84 51		324 34		256 40		54	3 59	18	73
50	28 27		79 55	344 45		87 21		327 04		259 05		52	4 08	20	70
14 00	30 57	N18 45	82 25	347 15	N25 41	89 51	S 5 02	329 35	N22 13	261 31	N 3 05	50	16 40	21	69
10	33 27		84 56	349 45		92 21		332 05		263 56		48	32 35	23	65
20	35 57		87 26	352 15		94 51		334 35		266 21		46	46 32	25	62
30	38 27		89 57	354 45		97 22		337 06		268 47		44	5 57	27	59
40	40 57		92 27	357 15		99 52		339 36		271 12		42	5 07	28	58
50	43 27		94 57	359 45		102 22		342 06		273 38		40	24 24	31	53
15 00	45 57	N18 45	97 28	2 14	N25 41	104 52	S 5 01	344 37	N22 12	276 03	N 2 55	38	39	23	50
10	48 27		99 58	4 44		107 22		347 07		278 28		36	5 53	22	46
20	50 57		102 29	7 14		109 52		349 38		280 54		34			
30	53 27		104 59	9 44		112 22		352 08		283 19		32	6 06	22	38 43
40	55 57		107 30	12 14		114 52		354 38		285 45		30	21 23	23	40
50	58 27		110 00	14 44		117 23		357 09		288 10		28	37 25	24	36
16 00	60 57	N18 46	112 30	17 14	N25 41	119 53	S 5 00	359 39	N22 12	290 35	N 2 44	26	46	27	34
10	63 27		115 01	19 44		122 23		2 09		293 01		24	6 57	29	31
20	65 57		117 31	22 14		124 53		4 40		295 26		22	7 10	31	28
30	68 27		120 02	24 43		127 23		7 10		297 52		20	25 36	50	25
40	70 57		122 32	27 13		129 53		9 40		300 17		18	33 37	51	23
50	73 27		125 02	29 43		132 23		12 11		302 42		16	40 39	52	22
17 00	75 57	N18 46	127 33	32 13	N25 41	134 53	S 4 59	14 41	N22 12	305 08	N 2 33	14	49 42	54	19
10	78 27		130 03	34 43		137 24		17 11		307 33		12	7 59	44	55
20	80 57		132 34	37 13		139 54		19 42		309 59		10	8 11	48	15
30	83 27		135 04	39 43		142 24		22 12		312 24		8			
40	85 57		137 34	42 13		144 54		24 42		314 49		6			
50	88 27		140 05	44 43		147 24		27 13		317 15		4			
18 00	90 57	N18 47	142 35	47 12	N25 41	149 54	S 4 59	29 43	N22 12	319 40	N 2 23	2			
10	93 27		145 06	49 42		152 24		32 13		322 05		0			
20	95 57		147 36	52 12		154 54		34 44		324 31		0			
30	98 27		150 07	54 42		157 25		37 14		326 56		0			
40	100 57		152 37	57 12		159 55		39 44		329 22		0			
50	103 27		155 07	59 42		162 25		42 15		331 47		0			
19 00	105 57	N18 48	157 38	62 12	N25 41	164 55	S 4 58	44 45	N22 12	334 12	N 2 12	70	23 13	3 09	*
10	108 27		160 08	64 42		167 25		47 15		336 38		68	22 05	3 03	01
20	110 57		162 39	67 12		169 55		49 46		339 03		66	21 30	2 58	06
30	113 27		165 09	69 41		172 25		52 16		341 29		64	21 05	98	54
40	115 57		167 39	72 11		174 55		54 46		343 54		62	20 46	74	51
50	118 27		170 10	74 41		177 26		57 17		346 19		60	29 64	48	14
20 00	120 57	N18 48	172 40	77 11	N25 41	179 56	S 4 57	59 47	N22 12	348 45	N 2 01	58	16 55	45	17
10	123 27		175 11	79 41		182 26		62 17		351 10		56	20 04	50	43
20	125 57		177 41	82 11		184 56		64 48		353 35		54	19 54	46	40
30	128 27		180 11	84 41		187 26		67 18		356 01		52	46 42	39	22
40	130 57		182 42	87 11		189 56		69 48		358 26		50	38 39	37	24
50	133 27		185 12	89 40		192 26		72 19		0 52		48	21 34	33	27
21 00	135 57	N18 49	187 43	92 10	N25 41	194 56	S 4 57	74 49	N22 12	3 17	N 1 50	35	18 56	28	27
10	138 27		190 13	94 40		197 26		77 19		5 42		33	46 26	24	35
20	140 57		192 44	97 10		199 57		79 50		8 08		30	28 24	20	38
30	143 27		195 14	99 40		202 27		82 20		10 33		28	14 22	16	42
40	145 57		197 44	102 10		204 57		84 50		12 58		26			
50	148 27		200 15	104 40		207 27		87 21		15 24		24			
22 00	150 57	N18 49	202 45	107 10	N25 41	209 57	S 4 56	89 51	N22 12	17 49	N 1 39	0	18 00	22	12 45
10	153 27		205 16	109 40		212 27		92 21		20 15		10	17 46	23	08 49
20	155 57		207 46	112 09		214 57		94 52		22 40		20	31 24	24	04 52
30	158 27		210 16	114 39		217 27		97 22		25 05		30	15 25	2	00 56
40	160 57		212 47	117 09		219 58		99 52		27 31		35	17 05	27	57 58
50	163 27		215 17	119 39		222 28		102 23		29 56		40	16 54	29	54 61
23 00	165 57	N18 50	217 48	122 09	N25 41	224 58	S 4 55	104 53	N22 12	32 21	N 1 28	45	41 32	50	64
10	168 27		220 18	124 39		227 28		107 23		34 47		26	50 26	36	46 67
20	170 57		222 48	127 09		229 58		109 54		37 12		25	19 38	44	69
30	173 27		225 19	129 39		232 28		112 24		39 38		23	11 40	42	70
40	175 57		227 49	132 09		234 58		114 54		42 03		21	16 02	43	39 73
50	178 27		230 20	134 38		237 28		117 25		44 28		19	15 51	45	36 76
24 00	180 57	N18 51	232 50	137 08	N25 41	239 59	S 4 54	119 55	N22 12	46 54	N 1 17	8	15 39	49	1 34 77

APPENDIX

GREENWICH A. M. 1943 MAY 20 (THURSDAY)

GCT	SUN				VENUS - 3.6				MARS 0.9				JUPITER - 1.6				MOON				Corr.
	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	Alt.	Corr.	
0 00	180 55	N19 45	236 47	136 04	N25 30	241 09	S 3 45	123 09	N22 06	358 12	S15 15	15	0 37	17	0	17	0	17	0	17	Alt.
10	183 25		239 17	138 34		243 39		125 39		0 37	17		128 10		3 01	18		18		18	Corr.
20	185 55		241 47	141 04		246 09		128 10		3 01	18		130 40		5 26	19		19		19	
30	188 25		244 18	143 34		248 40		133 10		7 50	21		135 41		10 14	22		22		22	
40	190 55		246 48	146 04		251 10		138 11	N22 06	12 39	S15 23	23	140 41		15 03	25		25		25	
50	193 25		249 19	148 34		253 40		143 12		17 27	26		145 42		19 52	27		27		27	
1 00	195 55	N19 45	251 49	151 03	N25 29	256 10	S 3 44	148 12		22 16	29		150 43		24 40	30		30		30	
10	198 25		254 19	153 33		258 40		153 13	N22 06	27 05	S15 31	26	155 43		29 29	33		33		33	
20	200 55		256 50	156 03		261 10		158 14		31 54	34		160 44		34 18	35		35		35	
30	203 25		259 20	158 33		263 40		163 14		36 42	36		165 45		39 07	38		38		38	
40	205 55		261 51	161 03		266 10															
50	208 25		264 21	163 33		268 40															
2 00	210 55	N19 46	266 52	166 03	N25 29	271 11	S 3 44	153 13	N22 06	27 05	S15 31	26	155 43		29 29	33		33		33	
10	213 25		269 22	168 33		273 41		158 14		31 54	34		160 44		34 18	35		35		35	
20	215 55		271 52	171 03		276 11		163 14		36 42	36		165 45		39 07	38		38		38	
30	218 25		274 23	173 33		278 41															
40	220 55		276 53	176 02		281 11															
50	223 25		279 24	178 32		283 41															
3 00	225 55	N19 46	281 54	181 02	N25 29	286 11	S 3 43	168 15	N22 06	41 31	S15 39	38	170 45		43 55	40		40		40	
10	228 25		284 24	183 32		288 41		173 16		46 20	42		175 46		48 44	43		43		43	
20	230 55		286 55	186 02		291 12		178 16		51 08	44		180 47		53 33	45		45		45	
30	233 25		289 25	188 32		293 42		183 17	N22 05	55 57	S15 47	47	185 47		58 21	48		48		48	
40	235 55		291 56	191 02		296 12		188 18		60 46	49		190 48		63 10	50		50		50	
50	238 25		294 26	193 32		298 42		193 18		65 34	52		195 49		67 58	53		53		53	
4 00	240 55	N19 47	296 56	196 02	N25 29	301 12	S 3 42	185 47		70 23	S15 54	54	198 19	N22 05	72 47	55		55		55	
10	243 25		299 27	198 31		303 42		193 18		75 11	57		200 49		77 36	58		58		58	
20	245 55		301 57	201 01		306 12		195 49		80 00	59		203 20		82 24	16		16		16	
30	248 25		304 28	203 31		308 42		198 19	N22 05	84 49	S16 02	62	205 50		87 13	03		03		03	
40	250 55		306 58	206 01		311 13		200 49		89 37	04		208 20		92 02	05		05		05	
50	253 25		309 29	208 31		313 43		213 21	N22 05	94 26	07		215 51		96 50	08		08		08	
5 00	255 55	N19 47	311 59	211 01	N25 29	316 13	S 3 41	218 22		99 15	S16 09	68	223 22		101 39	10		10		10	
10	258 25		314 29	213 31		318 43		223 22		104 03	12		225 53		106 27	13		13		13	
20	260 55		317 00	216 01		321 13		228 23	N22 05	108 52	14		230 53		111 16	15		15		15	
30	263 25		319 30	218 31		323 43		233 24		116 04	18		235 54		118 29	19		19		19	
40	265 55		322 01	221 00		326 13		238 24		120 53	20		240 55		123 17	21		21		21	
50	268 25		324 31	223 30		328 43		243 25	N22 05	125 42	22		248 26		128 47	22		22		22	
6 00	270 55	N19 48	327 01	226 00	N25 28	331 14	S 3 41	245 55		130 30	25		253 26		132 54	26		26		26	
10	273 25		329 32	228 30		333 44		250 56		135 19	27		255 57		137 43	28		28		28	
20	275 55		332 02	231 00		336 14		258 27	N22 05	142 32	S16 31	32	260 57		144 56	32		32		32	
30	278 25		334 33	233 30		338 44		260 57		147 20	33		263 28		149 44	34		34		34	
40	280 55		337 03	236 00		341 14		265 58		152 09	35		268 28		154 33	36		36		36	
50	283 25		339 33	238 30		343 44		268 28		159 21	39		270 59		161 45	40		40		40	
7 00	285 55	N19 48	342 04	241 00	N25 28	346 14	S 3 40	273 29	N22 05	164 10	41		275 59		166 34	42		42		42	
10	288 25		344 34	243 30		348 44		277 30		168 58	43		278 30		168 58	43		43		43	
20	290 55		347 05	245 59		351 15		283 30					281 00								
30	293 25		349 35	248 29		353 45		286 01					283 30								
40	295 55		352 06	250 59		356 15		288 31	N22 05				286 01								
50	298 25		354 36	253 29		358 45		291 01					288 31	N22 05							
8 00	300 55	N19 49	357 06	255 59	N25 28	1 15	S 3 39	293 32					291 01								
10	303 25		359 37	258 29		3 45		296 02					293 32								
20	305 55		2 07	260 59		6 15		298 32					296 02								
30	308 25		4 38	263 29		8 45		301 03					298 32								
40	310 55		7 08	265 59		11 16		303 33	N22 05				301 03								
50	313 25		9 38	268 28		13 46															
9 00	315 55	N19 49	12 09	270 58	N25 28	16 16	S 3 39	258 27	N22 05	128 06	S16 24	24	260 57		130 30	25		25		25	
10	318 25		14 39	273 28		18 46		263 28		132 54	26		263 28		135 19	27		27		27	
20	320 55		17 10	275 58		21 16		268 28		137 43	28		268 28		140 07	30		30		30	
30	323 25		19 40	278 28		23 46		270 59		142 32	S16 31	32	273 29	N22 05	144 56	32		32		32	
40	325 55		22 10	280 58		26 16		275 59		147 20	33		277 30		149 44	34		34		34	
50	328 25		24 41	283 28		28 46		278 30		152 09	35		281 00		154 33	36		36		36	
10 00	330 55	N19 50	27 11	285 58	N25 28	31 17	S 3 38	283 30		159 21	39		283 30		159 21	39		39		39	
10	333 25		29 42	288 28		33 47		286 01		161 45	40		286 01		164 10	41		41		41	
20	335 55		32 12	290 58		36 17		288 31	N22 05	166 34	42		288 31	N22 05	168 58	43		43		43	
30	338 25		34 42	293 27		38 47		291 01					291 01								
40	340 55		37 13	295 57		41 17		293 32					293 32								
50	343 25		39 43	298 27		43 47		296 02					296 02								
11 00	345 55	N19 51	42 14	300 57	N25 27	46 17	S 3 37	298 32					298 32								
10	348 25		44 44	303 27		48 47		301 03					301 03								
20	350 55		47 15	305 57		51 18															
30	353 25		49 45	308 27		53 48															
40	355 55		52 15	310 57		56 18															
50	358 25		54 46	313 27		58 48															
12 00	0 55	N19 51	57 16	315 56	N25 27	61 18	S 3 36	303 33	N22 05	171 22	S16 45										

AMERICAN AIR NAVIGATOR

GREENWICH P. M. 1943 MAY 20 (THURSDAY)

GCT	☉ SUN GHA Dec.	☽ VENUS - 3.6 GHA Dec.	MARS 0.9 GHA Dec.	JUPITER - 1.5 GHA Dec.	☾ MOON GHA Dec.	Lat	Sun- rise	Twil	Moon- rise	Diti.
h m	° ' "	° ' "	° ' "	° ' "	° ' "	N	A m	m	A m	m
12 00	0 55 N19 51	57 16 315 56 N25 27	61 18 S 3 36	303 33 N22 05	171 22 S16 45	N				
10	3 25	59 47 318 26	63 48	306 03	173 47 46					
20	5 55	62 17 320 56	66 18	308 34	176 11 47	70	☐	☐	23 15	133
30	8 25	64 47 323 26	68 48	311 04	178 35 48	68	1 22	NW	22 33	98
40	10 55	67 18 325 56	71 19	313 34	180 59 49	66	2 06	NW	22 05	88
50	13 25	69 48 328 26	73 49	316 05	183 24 50	64	34	125	21 44	83
13 00	15 55 N19 52	72 19 330 56 N25 27	76 19 S 3 36	318 35 N22 05	185 48 S16 51	62	2 56	82	26 80	
10	18 25	74 49 333 26	78 49	321 05	188 12 52	60	3 13	69	12 77	
20	20 55	77 19 335 56	81 19	323 36	190 36 54	58	28	60	21 00	75
30	23 25	79 50 338 25	83 49	326 06	193 00 55	56	40	54	20 49	74
40	25 55	82 20 340 55	86 19	328 36	195 25 56	54	3 51	48	40 73	
50	28 25	84 51 343 25	88 49	331 07	197 49 57	52	4 00	45	32 71	
14 00	30 55 N19 52	87 21 345 55 N25 27	91 20 S 3 35	333 37 N22 05	200 13 S16 58	50	09	42	24 71	
10	33 25	89 52 348 25	93 50	336 07	202 37 16	48	27	36	20 08	69
20	35 55	92 22 350 55	96 20	338 38	205 02 17	46	41	32	19 58	67
30	38 25	94 52 353 25	98 50	341 08	207 26 01	35	4 54	29	45 65	
40	40 55	97 23 355 55	101 20	343 38	209 50 02	30	5 04	27	35 65	
50	43 25	99 53 358 25	103 50	346 09	212 14 03	20	22	24	19 62	
15 00	45 55 N19 53	102 24 0 55 N25 27	106 20 S 3 34	348 39 N22 05	214 38 S17 05	10	38	23	19 04	61
10	48 25	104 54 3 24	108 50	351 09	217 03 06	0	5 53	22	18 51	59
20	50 55	107 24 5 54	111 21	353 40	219 27 07					
30	53 25	109 55 8 24	113 51	356 10	221 51 08	10	6 08	22	38 58	
40	55 55	112 25 10 54	116 21	358 40	224 15 09	20	23	24	24 56	
50	58 25	114 56 13 24	118 51	1 11	226 39 10	30	40	25	18 08	54
16 00	60 55 N19 53	117 26 15 54 N25 27	121 21 S 3 34	1 41 N22 05	229 04 S17 11	35	6 50	27	17 59	52
10	63 25	119 56 18 24	123 51	6 11	231 28 12	40	7 02	29	48 52	
20	65 55	122 27 20 54	126 21	8 42	233 52 13	45	15	32	36 49	
30	68 25	124 57 23 24	128 51	11 12	236 16 14	50	32	36	21 47	
40	70 55	127 28 25 53	131 22	13 42	238 40 15	52	39	38	14 46	
50	73 25	129 58 28 23	133 52	16 13	241 05 16	54	48	40	17 06	45
17 00	75 55 N19 54	132 29 30 53 N25 26	136 22 S 3 33	18 43 N22 05	243 29 S17 17	56	7 57	43	16 58	43
10	78 25	134 59 33 23	138 52	21 13	245 53 18	58	8 08	46	48 42	
20	80 55	137 29 35 53	141 22	23 44	248 17 19	60	8 21	50	16 37	39
30	83 25	140 00 38 23	143 52	26 14	250 41 20	5				
40	85 54	142 30 40 53	146 22	28 44	253 05 21					
50	88 24	145 01 43 23	148 52	31 15	255 30 22					
18 00	90 54 N19 54	147 31 45 53 N25 26	151 23 S 3 32	33 45 N22 05	257 54 S17 24	5	Sun- set	Twil	Moon- set	Diti.
10	93 24	150 01 48 23	153 53	36 15	260 18 25					
20	95 54	152 32 50 52	156 23	38 46	262 42 26					
30	98 24	155 03 53 22	158 53	41 16	265 06 27					
40	100 54	157 33 55 52	161 23	43 46	267 30 28	N				
50	103 24	160 03 58 22	163 53	46 17	269 55 29	0	A m	m	A m	m
19 00	105 54 N19 55	162 33 60 52 N25 26	166 23 S 3 31	48 47 N22 04	272 19 S17 30	70	☐	☐	2 49	*
10	108 24	165 04 63 22	168 53	51 17	274 43 31	68	22 36	NW	3 16	12
20	110 54	167 34 65 52	171 24	53 48	277 07 32	66	21 50	NW	36 21	
30	113 24	170 05 68 22	173 54	56 18	279 31 33	64	21 21	141	3 53	26
40	115 54	172 35 70 52	176 24	58 48	281 55 34	62	20 59	84	4 06	31
50	118 24	175 06 73 21	178 54	61 19	284 20 35	60	40	69	18 34	
20 00	120 54 N19 55	177 36 75 51 N25 26	181 24 S 3 31	63 49 N22 04	286 44 S17 36	58	26	59	28 36	
10	123 24	180 06 78 21	183 54	66 19	289 08 37	56	14	52	37 38	
20	125 54	182 37 80 51	186 24	68 50	291 32 38	54	20 03	47	45 40	
30	128 24	185 07 83 21	188 54	71 20	293 56 39	52	19 53	43	52 41	
40	130 54	187 38 85 51	191 25	73 50	296 20 40	50	44	40	4 58	43
50	133 24	190 08 88 21	193 55	76 21	298 44 41	45	26	35	5 12	45
21 00	135 54 N19 56	192 38 90 51 N25 26	196 25 S 3 30	78 51 N22 04	301 09 S17 42	35	19 00	28	33 49	
10	138 24	195 09 93 21	198 55	81 21	303 33 42	30	18 49	26	41 52	
20	140 54	197 39 95 50	201 25	83 52	305 57 43	20	30	24	5 56	54
30	143 24	200 10 98 20	203 55	86 22	308 21 44	10	15	23	6 09	56
40	145 54	202 40 100 50	206 25	88 52	310 45 45					
50	148 24	205 10 103 20	208 55	91 23	313 09 46	0	18 00	22	21 58	
22 00	150 54 N19 56	207 41 105 50 N25 25	211 26 S 3 29	93 53 N22 04	315 34 S17 47	47	10	17 46	23	33 60
10	153 24	210 11 108 20	213 56	96 23	317 58 48	20	31	24	6 46	62
20	155 54	212 42 110 50	216 26	98 54	320 22 49	20	31	26	7 01	66
30	158 24	215 12 113 20	218 56	101 24	322 46 50	30	13	24	10 66	
40	160 54	217 42 115 50	221 26	103 54	325 10 51	35	17 03	28	10 66	
50	163 24	220 13 118 20	223 56	106 25	327 34 52	40	16 51	30	19 68	
23 00	165 54 N19 57	222 43 120 49 N25 25	226 26 S 3 28	108 55 N22 04	329 58 S17 53	45	38	32	31 70	
10	168 24	225 14 123 19	228 56	111 25	332 22 54	50	21	36	45 73	
20	170 54	227 44 125 49	231 27	113 56	334 46 55	52	13	38	52 74	
30	173 24	230 15 128 19	233 57	116 26	337 11 56	54	16 05	41	7 59	75
40	175 54	232 45 130 49	236 27	118 56	339 35 57	58	15 55	44	8 07	77
50	178 24	235 15 133 19	238 57	121 27	341 59 57	58	44	46	16 79	
24 00	180 54 N19 57	237 46 135 49 N25 25	241 27 S 3 28	123 57 N22 04	344 23 S17 58	5	15 31	51	8 27	81

APPENDIX

GREENWICH A. M. 1943 MAY 25 (TUESDAY)

GCT	SUN		VENUS - 3.6		MARS 0.9		JUPITER - 1.5		MOON		☾	
	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.	GHA	Dec.		
0 00	180 50	N20 45	241 42	134 50	N25 01	242 39	S 2 18	127 09	N21 57	287 04	S16 32	Alt. + Corr.
10	183 20		244 13	137 20		245 09		129 39		289 29	31	
20	185 50		246 43	139 50		247 39		132 09		291 53	30	
30	188 20		249 14	142 20		250 09		134 40		294 17	28	
40	190 50		251 44	144 50		252 39		137 10		296 42	27	
50	193 20		254 14	147 20		255 09		139 40		299 06	26	
1 00	195 50	N20 45	256 45	149 50	N25 01	257 40	S 2 17	142 11	N21 57	301 30	S16 25	0 9 59 58 57 56 55 54 53 52 51 50 49 48 47 46 45 44 43 42 41 40 39 38 37 36 35 34 33 32 31 30 29 28 27 26 25 24 23 22 21 20 19 18 17 16 15 14 13 12 11 10 09 08 07 06 05 04 03 02 01 00
10	198 20		259 15	152 20		260 10		144 41		303 55	24	
20	200 50		261 46	154 50		262 40		147 11		306 19	23	
30	203 20		264 16	157 19		265 10		149 42		308 43	21	
40	205 50		266 46	159 49		267 40		152 12		311 07	20	
50	208 20		269 17	162 19		270 10		154 42		313 32	19	
2 00	210 50	N20 46	271 47	164 49	N25 00	272 40	S 2 17	157 13	N21 57	315 56	S16 18	Spiral
10	213 20		274 18	167 19		275 10		159 43		318 20	17	
20	215 50		276 48	169 49		277 41		162 13		320 45	15	
30	218 20		279 18	172 19		280 11		164 44		323 09	14	
40	220 50		281 49	174 49		282 41		167 14		325 33	13	
50	223 20		284 19	177 19		285 11		169 44		327 58	12	
3 00	225 50	N20 46	286 50	179 49	N25 00	287 41	S 2 16	172 15	N21 57	330 22	S16 11	Regulus
10	228 20		289 20	182 18		290 11		174 45		332 46	09	
20	230 50		291 51	184 48		292 41		177 15		335 11	08	
30	233 20		294 21	187 18		295 11		179 46		337 35	07	
40	235 50		296 51	189 48		297 42		182 16		339 59	06	
50	238 20		299 22	192 18		300 12		184 46		342 24	04	
4 00	240 50	N20 47	301 52	194 48	N25 00	302 42	S 2 15	187 17	N21 57	344 48	S16 03	JUPITER VENUS
10	243 20		304 23	197 18		305 12		189 47		347 12	02	
20	245 50		306 53	199 48		307 42		192 17		349 37	01	
30	248 20		309 23	202 18		310 12		194 48		352 01	16	
40	250 50		311 54	204 48		312 42		197 18		354 25	15	
50	253 20		314 24	207 18		315 12		199 48		356 50	14	
5 00	255 50	N20 47	316 55	209 47	N24 59	317 43	S 2 15	202 19	N21 57	359 14	S15 56	SATURN ALDEBARAN
10	258 20		319 25	212 17		320 13		204 49		1 38	55	
20	260 50		321 55	214 47		322 43		207 19		4 03	53	
30	263 20		324 26	217 17		325 13		209 50		6 27	52	
40	265 50		326 56	219 47		327 43		212 20		8 52	51	
50	268 20		329 27	222 17		330 13		214 50		11 16	50	
6 00	270 50	N20 48	331 57	224 47	N24 59	332 43	S 2 14	217 21	N21 57	13 40	S15 48	MARS
10	273 20		334 28	227 17		335 13		219 51		16 05	47	
20	275 50		336 58	229 47		337 44		222 21		18 29	46	
30	278 20		339 28	232 17		340 14		224 52		20 53	45	
40	280 50		341 59	234 46		342 44		227 22		23 18	43	
50	283 20		344 29	237 16		345 14		229 52		25 42	42	
7 00	285 50	N20 48	347 00	239 46	N24 59	347 44	S 2 13	232 23	N21 57	28 07	S15 41	JUPITER VENUS
10	288 20		349 30	242 16		350 14		234 53		30 31	39	
20	290 50		352 00	244 46		352 44		237 23		32 55	38	
30	293 20		354 31	247 16		355 14		239 54		35 20	37	
40	295 50		357 01	249 46		357 45		242 24		37 44	36	
50	298 20		359 32	252 16		0 15		244 54		40 08	34	
8 00	300 50	N20 48	2 02	254 46	N24 58	2 45	S 2 12	247 25	N21 57	42 33	S15 33	SATURN ALDEBARAN
10	303 20		4 32	257 16		5 15		249 55		44 57	32	
20	305 50		7 03	259 46		7 45		252 25		47 22	30	
30	308 20		9 33	262 15		10 15		254 56		49 46	29	
40	310 50		12 04	264 45		12 45		257 26		52 10	28	
50	313 20		14 34	267 15		15 15		259 56		54 35	27	
9 00	315 50	N20 49	17 05	269 45	N24 58	17 46	S 2 12	262 27	N21 57	56 59	S15 25	SD, ☉
10	318 20		19 35	272 15		20 16		264 57		59 24	24	
20	320 50		22 05	274 45		22 46		267 27		61 48	23	
30	323 20		24 36	277 15		25 16		269 58		64 12	21	
40	325 50		27 06	279 45		27 46		272 28		66 37	20	
50	328 20		29 37	282 15		30 16		274 58		69 01	19	
10 00	330 50	N20 49	32 07	284 45	N24 58	32 46	S 2 11	277 29	N21 57	71 26	S15 17	SD, ☾
10	333 20		34 37	287 14		35 16		279 59		73 50	16	
20	335 49		37 08	289 44		37 47		282 29		76 14	15	
30	338 19		39 38	292 14		40 17		285 00		78 39	13	
40	340 49		42 09	294 44		42 47		287 30		81 03	12	
50	343 19		44 39	297 14		45 17		290 00		83 28	11	
11 00	345 49	N20 50	47 09	299 44	N24 57	47 47	S 2 10	292 31	N21 57	85 52	S15 09	Corr. HA ☾
10	348 19		49 40	302 14		50 17		295 01		88 17	08	
20	350 49		52 10	304 44		52 47		297 31		90 41	07	
30	353 19		54 41	307 14		55 17		300 02		93 05	05	
40	355 49		57 11	309 44		57 48		302 32		95 30	04	
50	358 19		59 41	312 13		60 18		305 02		97 54	03	
12 00	0 49	N20 50	62 12	314 43	N24 57	62 48	S 2 10	307 33	N21 57	100 19	S15 01	☉ Int. ☾

East

100°

Spiral

Regulus

JUPITER VENUS

SATURN ALDEBARAN

MARS

100°

West

Inter.

AMERICAN AIR NAVIGATOR

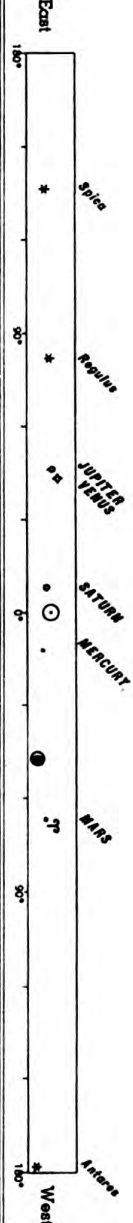
GREENWICH P. M. 1943 MAY 25 (TUESDAY)

GCT	☉ SUN GHA Dec.	☿ VENUS - 3.6 GHA Dec.	♂ MARS 0.9 GHA Dec.	♃ JUPITER - 1.5 GHA Dec.	☾ MOON GHA Dec.	Lat.	Sun- rise	Twil.	Moon- rise	Diff.
h m	° ' N20 50	° ' N24 57	° ' S 2 10	° ' N21 57	° ' S15 01	N	h m	m	h m	m
12 00	0 49	62 12	314 43	62 48	307 33	100 19	7 0	□	2 33	*
10	3 19	64 42	317 13	65 18	310 03	102 43	68	0 44	2 04	07
20	5 49	67 13	319 43	67 48	312 33	105 08	70	1 45	1 42	14
30	8 19	69 43	322 13	70 18	315 04	107 32	68	2 19	25	19
40	10 49	72 14	324 43	72 48	317 34	109 56	66	2 44	1 10	23
50	13 19	74 44	327 13	75 18	320 04	112 21	64	3 03	0 58	27
13 00	15 49	77 14	329 43	77 49	322 35	114 45	62	3 27	48	29
10	18 19	79 45	332 13	80 19	325 05	117 10	60	3 52	38	32
20	20 49	82 15	334 43	82 49	327 35	119 34	58	4 17	30	34
30	23 19	84 46	337 13	85 19	330 06	121 59	56	4 42	23	35
40	25 49	87 16	339 42	87 49	332 36	124 23	54	5 07	16	37
50	28 19	89 46	342 12	90 19	335 06	126 48	52	5 32	0 02	40
14 00	30 49	92 17	344 42	92 49	337 37	129 12	50	5 57	24	42
10	33 19	94 47	347 12	95 19	340 07	131 36	48	6 22	18	44
20	35 49	97 18	349 42	97 50	342 37	134 01	46	6 47	12	46
30	38 19	99 48	352 12	100 20	345 08	136 25	44	7 12	06	48
40	40 49	102 18	354 42	102 50	347 38	138 50	42	7 37	00	50
50	43 19	104 49	357 12	105 20	350 08	141 14	40	8 02	54	52
15 00	45 49	107 19	359 42	107 50	352 39	143 39	38	8 27	48	54
10	48 19	109 50	2 12	110 20	355 09	146 03	36	8 52	42	56
20	50 49	112 20	4 41	112 50	357 39	148 28	34	9 17	36	58
30	53 19	114 51	7 11	115 20	0 10	150 52	32	9 42	30	60
40	55 49	117 21	9 41	117 51	2 40	153 17	30	10 07	24	62
50	58 19	119 51	12 11	120 21	5 10	155 41	28	10 32	18	64
16 00	60 49	122 22	14 41	122 51	7 41	158 06	26	10 57	12	66
10	63 19	124 52	17 11	125 21	10 11	160 30	24	11 22	06	68
20	65 49	127 23	19 41	127 51	12 41	162 55	22	11 47	00	70
30	68 19	129 53	22 11	130 21	15 12	165 19	20	12 12	54	72
40	70 49	132 23	24 41	132 51	17 42	167 44	18	12 37	48	74
50	73 19	134 54	27 11	135 21	20 12	170 08	16	13 02	42	76
17 00	75 49	137 24	29 41	137 52	22 43	172 33	14	13 27	36	78
10	78 19	139 55	32 10	140 22	25 13	174 57	12	13 52	30	80
20	80 49	142 25	34 40	142 52	27 43	177 22	10	14 17	24	82
30	83 19	144 55	37 10	145 22	30 14	179 46	8	14 42	18	84
40	85 49	147 26	39 40	147 52	32 44	182 11	6	15 07	12	86
50	88 19	149 56	42 10	150 22	35 14	184 35	4	15 32	06	88
18 00	90 49	152 27	44 40	152 52	37 45	187 00	2	15 57	00	90
10	93 19	154 57	47 10	155 22	40 15	189 24	0	16 22	54	92
20	95 49	157 28	49 40	157 53	42 45	191 49	0	16 47	48	94
30	98 19	159 58	52 10	160 23	45 16	194 13	0	17 12	42	96
40	100 49	162 28	54 40	162 53	47 46	196 38	0	17 37	36	98
50	103 19	164 59	57 09	165 23	50 16	199 02	0	18 02	30	100
19 00	105 49	167 29	59 39	167 53	52 47	201 27	0	18 27	24	102
10	108 19	170 00	62 09	170 23	55 17	203 51	0	18 52	18	104
20	110 49	172 30	64 39	172 53	57 47	206 16	0	19 17	12	106
30	113 19	175 00	67 09	175 23	60 18	208 40	0	19 42	06	108
40	115 49	177 31	69 39	177 54	62 48	211 05	0	20 07	00	110
50	118 19	180 01	72 09	180 24	65 18	213 29	0	20 32	54	112
20 00	120 49	182 32	74 39	182 54	67 48	215 54	0	20 57	48	114
10	123 19	185 02	77 09	185 24	70 19	218 18	0	21 22	42	116
20	125 49	187 32	79 39	187 54	72 49	220 43	0	21 47	36	118
30	128 19	190 03	82 08	190 24	75 19	223 07	0	22 12	30	120
40	130 49	192 33	84 38	192 54	77 50	225 32	0	22 37	24	122
50	133 19	195 04	87 08	195 24	80 20	227 56	0	23 02	18	124
21 00	135 49	197 34	89 38	197 54	82 50	230 21	0	23 27	12	126
10	138 19	200 04	92 08	200 25	85 21	232 45	0	23 52	06	128
20	140 49	202 35	94 38	202 55	87 51	235 10	0	24 17	00	130
30	143 19	205 05	97 08	205 25	90 21	237 35	0	24 42	54	132
40	145 49	207 36	99 38	207 55	92 52	239 59	0	25 07	48	134
50	148 19	210 06	102 08	210 25	95 22	242 24	0	25 32	42	136
22 00	150 49	212 37	104 38	212 55	97 52	244 48	0	25 57	36	138
10	153 19	215 07	107 08	215 25	100 23	247 13	0	26 22	30	140
20	155 49	217 37	109 37	217 55	102 53	249 37	0	26 47	24	142
30	158 19	220 08	112 07	220 26	105 23	252 02	0	27 12	18	144
40	160 49	222 38	114 37	222 56	107 54	254 26	0	27 37	12	146
50	163 19	225 09	117 07	225 26	110 24	256 51	0	28 02	06	148
23 00	165 49	227 39	119 37	227 56	112 54	259 16	0	28 27	00	150
10	168 19	230 09	122 07	230 26	115 25	261 40	0	28 52	54	152
20	170 49	232 40	124 37	232 56	117 55	264 05	0	29 17	48	154
30	173 19	235 10	127 07	235 26	120 25	266 29	0	29 42	42	156
40	175 49	237 41	129 37	237 56	122 56	268 54	0	30 07	36	158
50	178 19	240 11	132 07	240 27	125 26	271 18	0	30 32	30	160
24 00	180 49	242 41	134 36	242 57	127 56	273 43	0	30 57	24	162

APPENDIX

GREENWICH A. M. 1943 MAY 30 (SUNDAY)

GCT	SUN	VENUS - 2.6	MARS 0.9	JUPITER - 1.5	MOON	☾
	GHA Dec.	GHA Dec.	GHA Dec.	GHA Dec.	GHA Dec.	Par.
0 00	180 42 N21 36	246 38	133 45 N24 17	244 09 S 0 51	131 07 N21 48	225 33 N 4 04
10	183 12	249 08	136 15	246 39	133 37	227 58 06
20	185 42	251 39	138 45	249 10	136 07	230 23 07
30	188 12	254 09	141 15	251 40	138 38	232 48 09
40	190 42	256 40	143 44	254 10	141 08	235 13 11
50	193 12	259 10	146 14	256 40	143 38	237 39 13
1 00	195 42 N21 36	261 40	148 44 N24 16	259 10 S 0 51	146 08 N21 48	240 04 N 4 15
10	198 12	264 11	151 14	261 40	148 39	242 29 17
20	200 42	266 41	153 44	264 10	151 09	244 54 18
30	203 12	269 12	156 14	266 40	153 39	247 20 20
40	205 42	271 42	158 44	269 11	156 10	249 45 22
50	208 12	274 13	161 14	271 41	158 40	252 10 24
2 00	210 42 N21 37	276 43	163 44 N24 16	274 11 S 0 50	161 10 N21 48	254 35 N 4 26
10	213 12	279 13	166 14	276 41	163 41	257 00 28
20	215 42	281 44	168 44	279 11	166 11	259 26 29
30	218 12	284 14	171 14	281 41	168 41	261 51 31
40	220 42	286 45	173 43	284 11	171 12	264 16 33
50	223 12	289 15	176 13	286 41	173 42	266 41 35
3 00	225 42 N21 37	291 45	178 43 N24 15	289 12 S 0 49	176 12 N21 48	269 07 N 4 37
10	228 12	294 16	181 13	291 42	178 43	271 32 39
20	230 42	296 46	183 43	294 12	181 13	273 57 40
30	233 12	299 17	186 13	296 42	183 43	276 22 42
40	235 42	301 47	188 43	299 12	186 14	278 47 44
50	238 12	304 17	191 13	301 42	188 44	281 13 46
4 00	240 42 N21 38	306 48	193 43 N24 15	304 12 S 0 48	191 14 N21 48	283 38 N 4 48
10	243 12	309 18	196 13	306 42	193 45	286 03 49
20	245 42	311 49	198 43	309 13	196 15	288 28 51
30	248 12	314 19	201 12	311 43	198 45	290 54 53
40	250 42	316 50	203 42	314 13	201 16	293 19 55
50	253 12	319 20	206 12	316 43	203 46	295 44 57
5 00	255 42 N21 38	321 50	208 42 N24 14	319 13 S 0 48	206 16 N21 48	298 09 N 4 58
10	258 12	324 21	211 12	321 43	208 47	300 34 5 00
20	260 42	326 51	213 42	324 13	211 17	303 00 02
30	263 12	329 22	216 12	326 43	213 47	305 25 04
40	265 42	331 52	218 42	329 14	216 18	307 50 06
50	268 12	334 22	221 12	331 44	218 48	310 15 07
6 00	270 41 N21 38	336 53	223 42 N24 14	334 14 S 0 47	221 18 N21 48	312 41 N 5 09
10	273 11	339 23	226 12	336 44	223 49	315 06 11
20	275 41	341 54	228 42	339 14	226 19	317 31 13
30	278 11	344 24	231 11	341 44	228 49	319 56 15
40	280 41	346 54	233 41	344 14	231 20	322 22 17
50	283 11	349 25	236 11	346 45	233 50	324 47 18
7 00	285 41 N21 39	351 55	238 41 N24 14	349 15 S 0 46	236 20 N21 48	327 12 N 5 20
10	288 11	354 26	241 11	351 45	238 51	329 37 22
20	290 41	356 56	243 41	354 15	241 21	332 02 24
30	293 11	359 26	246 11	356 45	243 51	334 28 25
40	295 41	1 57	248 41	359 15	246 22	336 53 27
50	298 11	4 27	251 11	1 45	248 52	339 18 29
8 00	300 41 N21 39	6 58	253 41 N24 13	4 15 S 0 46	251 22 N21 48	341 43 N 5 31
10	303 11	9 28	256 11	6 46	253 53	344 09 33
20	305 41	11 59	258 41	9 16	256 23	346 34 34
30	308 11	14 29	261 10	11 46	258 53	348 59 36
40	310 41	16 59	263 40	14 16	261 24	351 24 38
50	313 11	19 30	266 10	16 46	263 54	353 49 40
9 00	315 41 N21 40	22 00	268 40 N24 13	19 16 S 0 45	266 24 N21 48	356 15 N 5 42
10	318 11	24 31	271 10	21 46	268 55	358 40 43
20	320 41	27 01	273 40	24 16	271 25	1 05 45
30	323 11	29 31	276 10	26 47	273 55	3 30 47
40	325 41	32 02	278 40	29 17	276 26	5 56 49
50	328 11	34 32	281 10	31 47	278 56	8 21 51
10 00	330 41 N21 40	37 03	283 40 N24 12	34 17 S 0 44	281 26 N21 48	10 46 N 5 52
10	333 11	39 33	286 10	36 47	283 56	13 11 54
20	335 41	42 03	288 40	39 17	286 27	15 37 56
30	338 11	44 34	291 09	41 47	288 57	18 02 58
40	340 41	47 04	293 39	44 17	291 27	20 27 59
50	343 11	49 35	296 09	46 48	293 58	22 52 6 01
11 00	345 41 N21 40	52 05	298 39 N24 12	49 18 S 0 43	296 28 N21 48	25 17 N 6 03
10	348 11	54 36	301 09	51 48	298 58	27 43 05
20	350 41	57 06	303 39	54 18	301 29	30 08 07
30	353 11	59 36	306 09	56 48	303 59	32 33 08
40	355 41	62 07	308 39	59 18	306 29	34 58 10
50	358 11	64 37	311 09	61 48	309 00	37 24 12
12 00	0 41 N21 41	67 08	313 39 N24 11	64 18 S 0 43	311 30 N21 47	39 49 N 6 14



AMERICAN AIR NAVIGATOR

GREENWICH P. M. 1943 MAY 30 (SUNDAY)

GCT	☉ SUN		☿	♀ VENUS - 3.6		♂ MARS 0.9	♃ JUPITER - 1.5		♄ MOON	☾		Lat	Sun-rise	☾	Moon-rise	Diff.
	GHA	Dec.	GHA	GHA	Dec.	GHA	GHA	Dec.	GHA	Dec.						
12 00	0 41	N21 41	67 08	313 39	N21 11	64 18	311 30	N21 47	39 49	N 6 14	N	h m	h m	h m	h m	*
10	3 11		69 38	316 09		66 49	314 00		42 14	15						
20	5 41		72 08	318 39		69 19	316 31		44 39	17	70	☐	☐	2 14		
30	8 11		74 39	321 09		71 49	319 01		47 05	19	68	☐	☐	19 01		
40	10 41		77 09	323 38		74 19	321 31		49 30	21	66	1 25	☐	23 06		
50	13 11		79 40	326 08		76 49	324 02		51 55	22	64	2 05	☐	26 10		
13 00	15 41	N21 41	82 10	328 38	N21 11	79 19	326 32	N21 47	54 20	N 6 24	62	33	115	30 12		
10	18 11		84 40	331 08		81 49	329 02		56 45	26	60	2 54	80	32 16		
20	20 41		87 11	333 38		84 19	331 33		59 11	28	58	3 11	65	35 18		
30	23 11		89 41	336 08		86 50	334 03		61 36	29	56	25	56	37 20		
40	25 41		92 12	338 38		89 20	336 33		64 01	31	54	38	51	39 22		
50	28 11		94 42	341 08		91 50	339 04		66 26	33	52	49	46	40 25		
14 00	30 41	N21 41	97 13	343 38	N21 10	94 20	341 34	N21 47	68 52	N 6 35	50	3 58	42	42 26		
10	33 11		99 43	346 08		96 50	344 04		71 17	36	45	4 18	37	46 29		
20	35 41		102 13	348 38		99 20	346 35		73 42	38	40	35	32	48 32		
30	38 11		104 44	351 08		101 50	349 05		76 07	40	35	4 48	29	51 35		
40	40 41		107 14	353 37		104 21	351 35		78 32	42	30	5 00	27	53 37		
50	43 11		109 45	356 07		106 51	354 06		80 58	44	20	20	24	2 57 41		
											10	38	23	3 01 44		
15 00	45 41	N21 42	112 15	358 37	N21 10	109 21	356 36	N21 47	83 23	N 6 45		5 54	22	04 48		
10	48 11		114 45	1 07		111 51	359 06		85 48	47						
20	50 41		117 16	3 37		114 21	1 37		88 13	49						
30	53 11		119 46	6 07		116 51	4 07		90 39	51	10	6 09	23	08 50		
40	55 41		122 17	8 37		119 21	6 37		93 04	52	20	26	24	11 54		
50	58 11		124 47	11 07		121 51	9 08		95 29	54	30	46	26	15 58		
16 00	60 41	N21 42	127 17	13 37	N21 10	124 22	11 38	N21 47	97 54	N 6 56	35	6 57	28	18 60		
10	63 11		129 48	16 07		126 52	14 08		100 20	57	40	7 10	30	21 62		
20	65 41		132 18	18 37		129 22	16 39		102 45	6 59	45	25	33	24 66		
30	68 11		134 49	21 07		131 52	19 09		105 10	7 01	50	43	37	28 69		
40	70 41		137 19	23 36		134 22	21 39		107 35	03	52	7 52	38	29 72		
50	73 11		139 49	26 06		136 52	24 10		110 00	04	54	8 02	41	31 73		
17 00	75 41	N21 43	142 20	28 36	N21 09	139 22	26 40	N21 47	112 26	N 7 06	56	13	44	34 75		
10	78 11		144 50	31 06		141 52	29 10		114 51	08	58	26	48	36 77		
20	80 40		147 21	33 36		144 23	31 41		117 16	10	60	8 41	53	3 38 81		
30	83 10		149 51	36 06		146 53	34 11		119 41	11	S					
40	85 40		152 22	38 36		149 23	36 41		122 07	13						
50	88 10		154 52	41 06		151 53	39 12		124 32	15						
18 00	90 40	N21 43	157 22	43 36	N21 09	154 23	41 42	N21 47	126 57	N 7 17						
10	93 10		159 53	46 06		156 53	44 12		129 22	18						
20	95 40		162 23	48 36		159 23	46 43		131 47	20						
30	98 10		164 54	51 06		161 53	49 13		134 13	22						
40	100 40		167 24	53 35		164 24	51 43		136 38	23						
50	103 10		169 54	56 05		166 54	54 14		139 03	25						
19 00	105 40	N21 43	172 25	58 35	N21 08	169 24	56 44	N21 47	141 28	N 7 27	70	☐	☐	16 48	102	
10	108 10		174 55	61 05		171 54	59 14		143 54	29	68	☐	☐	40 95		
20	110 40		177 26	63 35		174 24	61 44		146 19	30	66	22 34	☐	33 90		
30	113 10		179 56	66 05		176 54	64 15		148 44	32	64	21 52	☐	26 86		
40	115 40		182 26	68 35		179 24	66 45		151 09	34	62	24 120		22 82		
50	118 10		184 57	71 05		181 54	69 15		153 34	35	60	21 02	80	17 79		
20 00	120 40	N21 44	187 27	73 35	N21 08	184 25	71 46	N21 47	156 00	N 7 37	58	20 44	65	13 77		
10	123 10		189 58	76 05		186 55	74 16		158 25	39	56	30	57	10 74		
20	125 40		192 28	78 35		189 25	76 46		160 50	41	54	17	51	06 73		
30	128 10		194 59	81 05		191 55	79 17		163 15	42	52	20 06	45	04 70		
40	130 40		197 29	83 35		194 25	81 47		165 41	44	50	19 57	42	16 01 69		
50	133 10		199 59	86 04		196 55	84 17		168 06	46	45	36	36	15 55 66		
											40	20	32	50 63		
21 00	135 40	N21 44	202 30	88 34	N21 07	199 25	86 48	N21 47	170 31	N 7 48	35	19 07	29	47 59		
10	138 10		205 00	91 04		201 55	89 18		172 56	49	30	18 55	27	43 58		
20	140 40		207 31	93 34		204 26	91 48		175 22	51	20	34	24	37 54		
30	143 10		210 01	96 04		206 56	94 19		177 47	53	10	17	23	32 50		
40	145 40		212 31	98 34		209 26	96 49		180 12	54						
50	148 10		215 02	101 04		211 56	99 19		182 37	56						
22 00	150 40	N21 45	217 32	103 34	N21 07	214 26	101 50	N21 47	185 02	N 7 58						
10	153 10		220 03	106 04		216 56	104 20		187 28	7 59	10	17 44	23	22 44		
20	155 40		222 33	108 34		219 26	106 50		189 53	8 01	20	27	24	16 42		
30	158 10		225 03	111 04		221 57	109 21		192 18	03	30	17 08	26	10 38		
40	160 40		227 34	113 34		224 27	111 51		194 43	05	35	16 57	28	07 35		
50	163 10		230 04	116 03		226 57	114 21		197 09	06	40	44	30	15 03 33		
23 00	165 40	N21 45	232 35	118 33	N21 06	229 27	116 52	N21 47	199 34	N 8 08	45	28	33	14 59 30		
10	168 10		235 05	121 03		231 57	119 22		201 59	10	50	10	37	53 27		
20	170 40		237 36	123 33		234 27	121 52		204 24	11	52	16 01	39	51 25		
30	173 10		240 06	126 03		236 57	124 23		206 49	13	54	15 51	42	48 23		
40	175 40		242 36	128 33		239 27	126 53		209 15	15	56	40	45	45 21		
50	178 10		245 07	131 03		241 58	129 23		211 40	16	58	27	49	42 19		
											60	15 12	54	14 38 17		
24 00	180 40	N21 45	247 37	133 33	N21 06	244 28	131 54	N21 47	214 05	N 8 18	S					

APPENDIX

CONVERSION OF ARC TO TIME

°	h	m	°	h	m	°	h	m	°	h	m	°	h	m	°	h	m	'	m	s	"	s
0	0	0	60	4	0	120	8	0	180	12	0	240	16	0	300	20	0	0	0	0	0	0.00
1	0	4	61	4	4	121	8	4	181	12	4	241	16	4	301	20	4	1	0	4	1	0.07
2	0	8	62	4	8	122	8	8	182	12	8	242	16	8	302	20	8	2	0	8	2	0.13
3	0	12	63	4	12	123	8	12	183	12	12	243	16	12	303	20	12	3	0	12	3	0.20
4	0	16	64	4	16	124	8	16	184	12	16	244	16	16	304	20	16	4	0	16	4	0.27
5	0	20	65	4	20	125	8	20	185	12	20	245	16	20	305	20	20	5	0	20	5	0.33
6	0	24	66	4	24	126	8	24	186	12	24	246	16	24	306	20	24	6	0	24	6	0.40
7	0	28	67	4	28	127	8	28	187	12	28	247	16	28	307	20	28	7	0	28	7	0.47
8	0	32	68	4	32	128	8	32	188	12	32	248	16	32	308	20	32	8	0	32	8	0.53
9	0	36	69	4	36	129	8	36	189	12	36	249	16	36	309	20	36	9	0	36	9	0.60
10	0	40	70	4	40	130	8	40	190	12	40	250	16	40	310	20	40	10	0	40	10	0.67
11	0	44	71	4	44	131	8	44	191	12	44	251	16	44	311	20	44	11	0	44	11	0.73
12	0	48	72	4	48	132	8	48	192	12	48	252	16	48	312	20	48	12	0	48	12	0.80
13	0	52	73	4	52	133	8	52	193	12	52	253	16	52	313	20	52	13	0	52	13	0.87
14	0	56	74	4	56	134	8	56	194	12	56	254	16	56	314	20	56	14	0	56	14	0.93
15	1	0	75	5	0	135	9	0	195	13	0	255	17	0	315	21	0	15	1	0	15	1.00
16	1	4	76	5	4	136	9	4	196	13	4	256	17	4	316	21	4	16	1	4	16	1.07
17	1	8	77	5	8	137	9	8	197	13	8	257	17	8	317	21	8	17	1	8	17	1.13
18	1	12	78	5	12	138	9	12	198	13	12	258	17	12	318	21	12	18	1	12	18	1.20
19	1	16	79	5	16	139	9	16	199	13	16	259	17	16	319	21	16	19	1	16	19	1.27
20	1	20	80	5	20	140	9	20	200	13	20	260	17	20	320	21	20	20	1	20	20	1.33
21	1	24	81	5	24	141	9	24	201	13	24	261	17	24	321	21	24	21	1	24	21	1.40
22	1	28	82	5	28	142	9	28	202	13	28	262	17	28	322	21	28	22	1	28	22	1.47
23	1	32	83	5	32	143	9	32	203	13	32	263	17	32	323	21	32	23	1	32	23	1.53
24	1	36	84	5	36	144	9	36	204	13	36	264	17	36	324	21	36	24	1	36	24	1.60
25	1	40	85	5	40	145	9	40	205	13	40	265	17	40	325	21	40	25	1	40	25	1.67
26	1	44	86	5	44	146	9	44	206	13	44	266	17	44	326	21	44	26	1	44	26	1.73
27	1	48	87	5	48	147	9	48	207	13	48	267	17	48	327	21	48	27	1	48	27	1.80
28	1	52	88	5	52	148	9	52	208	13	52	268	17	52	328	21	52	28	1	52	28	1.87
29	1	56	89	5	56	149	9	56	209	13	56	269	17	56	329	21	56	29	1	56	29	1.93
30	2	0	90	6	0	150	10	0	210	14	0	270	18	0	330	22	0	30	2	0	30	2.00
31	2	4	91	6	4	151	10	4	211	14	4	271	18	4	331	22	4	31	2	4	31	2.07
32	2	8	92	6	8	152	10	8	212	14	8	272	18	8	332	22	8	32	2	8	32	2.13
33	2	12	93	6	12	153	10	12	213	14	12	273	18	12	333	22	12	33	2	12	33	2.20
34	2	16	94	6	16	154	10	16	214	14	16	274	18	16	334	22	16	34	2	16	34	2.27
35	2	20	95	6	20	155	10	20	215	14	20	275	18	20	335	22	20	35	2	20	35	2.33
36	2	24	96	6	24	156	10	24	216	14	24	276	18	24	336	22	24	36	2	24	36	2.40
37	2	28	97	6	28	157	10	28	217	14	28	277	18	28	337	22	28	37	2	28	37	2.47
38	2	32	98	6	32	158	10	32	218	14	32	278	18	32	338	22	32	38	2	32	38	2.53
39	2	36	99	6	36	159	10	36	219	14	36	279	18	36	339	22	36	39	2	36	39	2.60
40	2	40	100	6	40	160	10	40	220	14	40	280	18	40	340	22	40	40	2	40	40	2.67
41	2	44	101	6	44	161	10	44	221	14	44	281	18	44	341	22	44	41	2	44	41	2.73
42	2	48	102	6	48	162	10	48	222	14	48	282	18	48	342	22	48	42	2	48	42	2.80
43	2	52	103	6	52	163	10	52	223	14	52	283	18	52	343	22	52	43	2	52	43	2.87
44	2	56	104	6	56	164	10	56	224	14	56	284	18	56	344	22	56	44	2	56	44	2.93
45	3	0	105	7	0	165	11	0	225	15	0	285	19	0	345	23	0	45	3	0	45	3.00
46	3	4	106	7	4	166	11	4	226	15	4	286	19	4	346	23	4	46	3	4	46	3.07
47	3	8	107	7	8	167	11	8	227	15	8	287	19	8	347	23	8	47	3	8	47	3.13
48	3	12	108	7	12	168	11	12	228	15	12	288	19	12	348	23	12	48	3	12	48	3.20
49	3	16	109	7	16	169	11	16	229	15	16	289	19	16	349	23	16	49	3	16	49	3.27
50	3	20	110	7	20	170	11	20	230	15	20	290	19	20	350	23	20	50	3	20	50	3.33
51	3	24	111	7	24	171	11	24	231	15	24	291	19	24	351	23	24	51	3	24	51	3.40
52	3	28	112	7	28	172	11	28	232	15	28	292	19	28	352	23	28	52	3	28	52	3.47
53	3	32	113	7	32	173	11	32	233	15	32	293	19	32	353	23	32	53	3	32	53	3.53
54	3	36	114	7	36	174	11	36	234	15	36	294	19	36	354	23	36	54	3	36	54	3.60
55	3	40	115	7	40	175	11	40	235	15	40	295	19	40	355	23	40	55	3	40	55	3.67
56	3	44	116	7	44	176	11	44	236	15	44	296	19	44	356	23	44	56	3	44	56	3.73
57	3	48	117	7	48	177	11	48	237	15	48	297	19	48	357	23	48	57	3	48	57	3.80
58	3	52	118	7	52	178	11	52	238	15	52	298	19	52	358	23	52	58	3	52	58	3.87
59	3	56	119	7	56	179	11	56	239	15	56	299	19	56	359	23	56	59	3	56	59	3.93
60	4	0	120	8	0	180	12	0	240	16	0	300	20	0	360	24	0	60	4	0	60	4.00

AMERICAN AIR NAVIGATOR

POLARIS

LHA T Corr.	LHA T Corr.	LHA T Corr.	LHA T Corr.	LHA T Corr.	LHA T Corr.
° / '	° / '	° / '	° / '	° / '	° / '
358 56 -54	89 43 -26	128 40 +14	178 28 +54	270 37 +26	309 39 -14
1 02 -55	90 46 -25	129 39 +15	180 37 +55	271 41 +25	310 37 -15
3 20 -56	91 48 -24	130 38 +16	182 56 +56	272 43 +24	311 36 -16
5 52 -57	92 50 -23	131 37 +17	185 31 +57	273 46 +23	312 35 -17
8 45 -58	93 52 -22	132 36 +18	188 27 +58	274 48 +22	313 34 -18
12 11 -59	94 53 -21	133 36 +19	191 57 +59	275 49 +21	314 34 -19
16 47 -60	95 53 -20	134 37 +20	196 37 +60	276 50 +20	315 34 -20
35 41 -59	96 53 -19	135 37 +21	215 51 +59	277 51 +19	316 34 -21
40 16 -58	97 53 -18	136 38 +22	220 31 +58	278 51 +18	317 34 -22
43 42 -57	98 53 -17	137 39 +23	224 00 +57	279 51 +17	318 35 -23
46 35 -56	99 52 -16	138 41 +24	226 56 +56	280 50 +16	319 37 -24
49 07 -55	100 51 -15	139 44 +25	229 31 +55	281 49 +15	320 39 -25
51 25 -54	101 50 -14	140 46 +26	231 51 +54	282 48 +14	321 41 -26
53 31 -53	102 48 -13	141 50 +27	233 59 +53	283 47 +13	322 44 -27
55 30 -52	103 47 -12	142 54 +28	236 00 +52	284 46 +12	323 48 -28
57 21 -51	104 45 -11	143 58 +29	237 53 +51	285 44 +11	324 52 -29
59 07 -50	105 42 -10	145 04 +30	239 40 +50	286 42 +10	325 56 -30
60 48 -49	106 40 -9	146 10 +31	241 23 +49	287 40 +9	327 02 -31
62 25 -48	107 38 -8	147 16 +32	243 01 +48	288 37 +8	328 08 -32
63 59 -47	108 35 -7	148 24 +33	244 36 +47	289 35 +7	329 15 -33
65 29 -46	109 32 -6	149 32 +34	246 08 +46	290 32 +6	330 22 -34
66 57 -45	110 30 -5	150 42 +35	247 36 +45	291 30 +5	331 31 -35
68 22 -44	111 27 -4	151 52 +36	249 02 +44	292 27 +4	332 41 -36
69 44 -43	112 24 -3	153 03 +37	250 26 +43	293 24 +3	333 51 -37
71 05 -42	113 21 -2	154 16 +38	251 48 +42	294 21 +2	335 03 -38
72 24 -41	114 18 -1	155 30 +39	253 08 +41	295 18 +1	336 16 -39
73 41 -40	115 15 -0	156 45 +40	254 26 +40	296 15 +0	337 30 -40
74 57 -39	116 12 +1	158 01 +41	255 42 +39	297 12 -1	338 46 -41
76 11 -38	117 09 +2	159 19 +42	256 58 +38	298 09 -2	340 03 -42
77 24 -37	118 06 +3	160 39 +43	258 11 +37	299 06 -3	341 22 -43
78 36 -36	119 03 +4	162 01 +44	259 24 +36	300 03 -4	342 43 -44
79 46 -35	120 00 +5	163 25 +45	260 35 +35	301 00 -5	344 06 -45
80 56 -34	120 57 +6	164 51 +46	261 46 +34	301 57 -6	345 31 -46
82 05 -33	121 55 +7	166 20 +47	262 55 +33	302 55 -7	346 58 -47
83 12 -32	122 52 +8	167 51 +48	264 03 +32	303 52 -8	348 28 -48
84 19 -31	123 50 +9	169 26 +49	265 11 +31	304 49 -9	350 02 -49
85 25 -30	124 47 +10	171 04 +50	266 17 +30	305 47 -10	351 39 -50
86 31 -29	125 45 +11	172 47 +51	267 23 +29	306 45 -11	353 20 -51
87 36 -28	126 43 +12	174 34 +52	268 29 +28	307 43 -12	355 06 -52
88 40 -27	127 42 +13	176 27 +53	269 33 +27	308 41 -13	356 57 -53
89 43 -26	128 40 +14	178 28 +54	270 37 +26	309 39 -14	358 56 -54

LHA T measured westward. Refraction not included: Use Table A.

ADDITIONAL STARS

Name	Meg.	SHA	Dec.
		° / '	° / '
Alkaid	1.9	153 40	N49 36
γ Argus	1.9	238 04	S47 10
δ Can. Maj. †	2.0	253 29	S26 18
Castor	1.6	247 16	N32 01
β Centauri . .	0.9	150 03	S60 06
El Nath . . . †	1.8	279 20	N28 34
γ Gem . . . †	1.9	261 24	N16 27
ζ Orionis . . . †	1.9	275 32	S 1 58
Schedir	2.1-2.6	350 41	N56 13

† The planetary section of the *Astronomical Navigation Tables* may be used for these stars.

May-Aug., 1943

APPENDIX

STARS

Alphabetical order				Order of SHA			
Name	Mag.	SHA	Dec.	SHA	Dec.	RA	Name
Acamar	3.4	315 59	S40 32	14 31	N14 54	23 02	Markab
Achernar . . . 1	0.6	336 06	S57 31	16 22	S29 55	22 55	Fomalhaut
Acrux 2	1.1	174 08	S62 47	28 50	S47 14	22 05	Al Na'ir
Adhara †	1.6	255 54	S28 54	34 39	N 9 37	21 41	Enif
Aldebaran . . 3	1.1	291 50	N16 24	50 07	N45 05	20 40	Deneb
Alioth	1.7	167 07	N56 16	54 42	S56 55	20 21	Peacock
Al Na'ir	2.2	28 50	S47 14	63 00	N 8 43	19 48	Altair
Alnilam †	1.8	276 40	S 1 14	77 04	S26 22	18 52	Nunki
Alphard †	2.2	218 48	S 8 25	81 14	N38 44	18 35	Vega
Alphecca . . . †	2.3	126 56	N26 54	84 54	S34 25	18 20	Kaus Aust.
Alpheratz . . . 4	2.2	358 38	N28 47	91 10	N51 30	17 55	Etamin
Al Suhail	2.2	223 32	S43 12	96 55	N12 36	17 32	Rasalague
Altair 5	0.9	63 00	N 8 43	97 33	S37 04	17 30	Shaula
Antares 6	1.2	113 31	S26 13	103 13	S15 39	17 07	Sabik
Arcturus . . . 7	0.2	146 44	N19 29	(109 20)	S68 56	16 43	α Tri. Aust.
ϵ Argus	1.7	234 40	S59 20	113 31	S26 18	16 26	Antares
Bellatrix †	1.7	279 29	N 6 18	120 45	S22 28	15 57	Dschubba
Betelgeux . . . 8	0.1-1.2	271 59	N 7 24	126 56	N26 54	15 32	Alphecca
Canopus 9	-0.9	264 20	S52 40	(137 17)	N74 24	14 51	Kochab
Capella 10	0.2	281 53	N45 56	141 04	S60 36	14 36	Rigel Kent.
Caph	2.4	358 28	N58 50	146 44	N19 29	14 13	Arcturus
θ Centauri	2.3	149 10	S36 06	149 10	S36 06	14 03	θ Centauri
β Crucis	1.5	168 54	S59 23	159 27	S10 52	13 22	Spica
γ Crucis	1.6	173 00	S56 48	159 35	N55 14	13 22	Mizar
Deneb 11	1.3	50 07	N45 05	167 07	N56 16	12 52	Alioth
Deneb Kait. . . †	2.2	349 49	S18 18	168 54	S59 23	12 44	β Crucis
Denebola . . . †	2.2	183 28	N14 53	173 00	S56 48	12 28	γ Crucis
Dschubba	2.5	120 45	S22 28	174 08	S62 47	12 23	Acrux
Dubhe 12	2.0	194 57	N62 04	183 28	N14 53	11 46	Denebola
Enif	2.5	34 39	N 9 37	194 57	N62 04	11 00	Dubhe
Etamin	2.4	91 10	N51 30	208 40	N12 15	10 05	Regulus
Fomalhaut . . 13	1.3	16 22	S29 55	218 48	S 8 25	9 25	Alphard
Hamal †	2.2	329 01	N23 12	(221 52)	S69 29	9 13	Miaplacidus
Kaus Aust. . . .	2.0	84 54	S34 25	223 32	S43 12	9 06	Al Suhail
Kochab	2.2	(137 17)	N74 24	234 40	S59 20	8 21	ϵ Argus
Marfak	1.9	309 56	N49 39	244 33	N28 10	7 42	Pollux
Markab	2.6	14 31	N14 54	245 55	N 5 22	7 36	Procyon
Miaplacidus . . .	1.8	(221 52)	S69 29	255 54	S28 54	6 56	Adhara
Mizar	2.4	159 35	N55 14	259 21	S16 38	6 43	Sirius
Nunki	2.1	77 04	S26 22	264 20	S52 40	6 23	Canopus
Peacock 14	2.1	54 42	S56 55	271 59	N 7 24	5 52	Betelgeux
Polaris	2.1	(333 51)	N88 59	276 40	S 1 14	5 33	Alnilam
Pollux 15	1.2	244 33	N28 10	279 29	N 6 18	5 22	Bellatrix
Procyon 16	0.5	245 55	N 5 22	281 53	N45 56	5 12	Capella
Rasalague . . . †	2.1	96 55	N12 36	282 03	S 8 16	5 12	Rigel
Regulus 17	1.3	208 40	N12 15	291 50	N16 24	4 33	Aldebaran
Rigel 18	0.3	282 03	S 8 16	309 56	N49 39	3 20	Marfak
Rigel Kent. . . 19	0.3	141 04	S60 36	315 59	S40 32	2 56	Acamar
Ruchbah	2.8	339 29	N59 56	329 01	N23 12	2 04	Hamal
Sabik	2.6	103 13	S15 39	(333 51)	N88 59	1 45	Polaris
Shaula	1.7	97 33	S37 04	336 06	S57 31	1 36	Achernar
Sirius 20	-1.6	259 21	S16 38	339 29	N59 56	1 22	Ruchbah
Spica 21	1.2	159 27	S10 52	349 49	S18 18	0 41	Deneb Kait.
α Tri. Aust. . . .	1.9	(109 20)	S68 56	358 28	N58 50	0 06	Caph
Vega 22	0.1	81 14	N38 44	358 38	N28 47	0 05	Alpheratz

SHA = $360^\circ - \text{RA}$ GHA* = GHA \mp SHA*

May-Aug., 1943

AMERICAN AIR NAVIGATOR

INTERPOLATION OF GHA

SUN, PLANETS, ♄						MOON					
Int.	Corr.	Int.	Corr.	Int.	Corr.	Int.	Corr.	Int.	Corr.	Int.	Corr.
00 00	0 00	03 17	0 50	06 37	1 40	00 00	0 00	03 20	0 49	06 39	1 37
01 00	0 01	21 0 51		41 1 41		02 00	0 01	24 0 50		43 1 38	
05 00	0 02	25 0 52		45 1 42		06 00	0 02	29 0 51		47 1 39	
09 00	0 03	29 0 53		49 1 43		10 00	0 03	33 0 52		52 1 40	
13 00	0 04	33 0 54		53 1 44		14 00	0 04	37 0 53		56 1 41	
17 00	0 05	37 0 55		57 1 45		18 00	0 05	41 0 54		07 00	1 42
21 00	0 06	41 0 56		07 01 46		22 00	0 06	45 0 55		04 1 43	
25 00	0 07	45 0 57		05 1 47		26 00	0 07	49 0 56		08 1 44	
29 00	0 08	49 0 58		13 1 48		31 00	0 08	53 0 57		12 1 45	
33 00	0 09	53 0 59		17 1 49		35 00	0 09	58 0 58		16 1 46	
37 00	0 10	01 1 00		21 1 50		39 00	0 10	02 0 59		20 1 47	
41 00	0 11	05 1 01		25 1 51		43 00	0 11	06 0 00		25 1 48	
45 00	0 12	09 1 02		29 1 52		47 00	0 12	10 0 01		30 1 49	
49 00	0 13	13 1 03		33 1 53		51 00	0 13	14 0 02		33 1 50	
53 00	0 14	17 1 04		37 1 54		55 00	0 14	18 0 03		37 1 51	
57 00	0 15	21 1 05		41 1 55		01 00	0 15	22 0 04		41 1 52	
01 01	0 16	25 1 06		45 1 56		04 00	0 16	27 0 05		45 1 53	
05 01	0 17	29 1 07		49 1 57		08 00	0 17	31 0 06		49 1 54	
09 01	0 18	33 1 08		53 1 58		12 00	0 18	35 0 07		54 1 55	
13 01	0 19	37 1 09		57 1 59		16 00	0 19	39 0 08		58 1 56	
17 01	0 20	41 1 10		08 01 2 00		20 00	0 20	43 0 09		08 02 1 57	
21 01	0 21	45 1 11		05 2 01		24 00	0 21	47 0 10		10 1 58	
25 01	0 22	49 1 12		09 2 02		29 00	0 22	51 0 11		14 1 59	
29 01	0 23	53 1 13		13 2 03		33 00	0 23	56 0 12		18 2 00	
33 01	0 24	57 1 14		17 2 04		37 00	0 24	01 0 13		23 2 01	
37 01	0 25	05 1 15		21 2 05		41 00	0 25	04 0 14		27 2 02	
41 01	0 26	09 1 16		25 2 06		45 00	0 26	08 0 15		31 2 03	
45 01	0 27	13 1 17		29 2 07		49 00	0 27	12 0 16		35 2 04	
49 01	0 28	17 1 18		33 2 08		53 00	0 28	16 0 17		39 2 05	
53 01	0 29	21 1 19		37 2 09		58 00	0 29	20 0 18		43 2 06	
57 01	0 30	25 1 20		41 2 10		02 00	0 30	25 0 19		47 2 07	
02 01	0 31	29 1 21		45 2 11		06 00	0 31	29 0 20		52 2 08	
05 01	0 32	33 1 22		49 2 12		10 00	0 32	33 0 21		56 2 09	
09 01	0 33	37 1 23		53 2 13		14 00	0 33	37 0 22		09 00 2 10	
13 01	0 34	41 1 24		57 2 14		18 00	0 34	41 0 23		04 2 11	
17 01	0 35	45 1 25		01 2 15		22 00	0 35	45 0 24		08 2 12	
21 01	0 36	49 1 26		05 2 16		26 00	0 36	49 0 25		12 2 13	
25 01	0 37	53 1 27		09 2 17		31 00	0 37	54 0 26		16 2 14	
29 01	0 38	57 1 28		13 2 18		35 00	0 38	58 0 27		21 2 15	
33 01	0 39	01 1 29		17 2 19		39 00	0 39	02 0 28		25 2 16	
37 01	0 40	05 1 30		21 2 20		43 00	0 40	06 0 29		29 2 17	
41 01	0 41	09 1 31		25 2 21		47 00	0 41	10 0 30		33 2 18	
45 01	0 42	13 1 32		29 2 22		51 00	0 42	14 0 31		37 2 19	
49 01	0 43	17 1 33		33 2 23		55 00	0 43	18 0 32		41 2 20	
53 01	0 44	21 1 34		37 2 24		03 00	0 44	23 0 33		45 2 21	
57 01	0 45	25 1 35		41 2 25		04 00	0 45	27 0 34		50 2 22	
03 01	0 46	29 1 36		45 2 26		08 00	0 46	31 0 35		54 2 23	
05 01	0 47	33 1 37		49 2 27		12 00	0 47	35 0 36		58 2 24	
09 01	0 48	37 1 38		53 2 28		16 00	0 48	39 0 37		10 00 2 25	
13 01	0 49	41 1 39		57 2 29		20 00	0 49				
17 01	0 50			10 00 2 30		24 00	0 50				

Correction to be added to GHA for interval of GCT

DIP

Subtract from altitude observed with sea horizon.

Height	Corr.	Height	Corr.	Height	Corr.	Height	Corr.
<i>Fl.</i>	<i>'</i>	<i>Fl.</i>	<i>'</i>	<i>Fl.</i>	<i>'</i>	<i>Fl.</i>	<i>'</i>
0	1	160	13	620	25	1380	37
2	2	180	14	670	26	1460	38
6	3	210	15	730	27	1540	39
12	3	250	16	780	28	1620	40
21	4	280	17	840	29	1700	41
31	6	310	18	900	30	1790	42
43	7	350	19	960	31	1870	43
58	8	390	20	1030	32	1960	44
75	9	430	21	1090	33	2060	45
93	10	480	22	1160	34	2150	46
114	11	520	23	1230	35	2250	47
137	12	570	24	1310	36	2340	48
162		620		1380		2440	

REFRACTION

A. Total correction.—For use with H. O. 208, H. O. 209, H. O. 211, H. O. 214, and the Polaris Table. Subtract from observed altitude.

Height in feet	Observed altitude						
	5°	10°	15°	20°	30°	45°	60°
0	10	5	4	3	2	1	1
5,000	8	5	3	2	1	1	0
10,000	7	4	3	2	1	1	0
15,000	6	3	2	2	1	1	0
20,000	5	3	2	1	1	1	0
25,000	4	2	2	1	1	0	0
30,000	3	2	1	1	1	0	0
35,000	3	2	1	1	1	0	0
40,000	2	1	1	1	0	0	0

GLOSSARY OF NAVIGATION TERMS

AGONIC LINE—Line connecting points of zero magnetic variation.

AIR NAVIGATION—Determining, and keeping an account of, an aircraft's position while flying from point to point.

Pilotage—Navigation with reference to visible ground objects.

Dead Reckoning — Navigation based on known values of time, speed, distance and direction.

Radio—Navigation by use of radio aids such as radio ranges, marker beacons and direction finding stations.

Celestial—Navigation with reference to the positions of the celestial bodies.

AIR SPEED—Speed of an aircraft with reference to the surrounding air.

Indicated—Air speed shown by the air speed indicator.

Calibrated—Indicated air speed corrected for instrument and installation errors.

True—Calibrated air speed corrected for temperature and pressure (altitude).

ALTITUDE (celestial) — Angular distance from the celestial horizon to the center of a celestial body measured upon the vertical circle passing through the body. It is also the angle at the center of the earth subtended by this arc.

ALTITUDE INTERCEPT—Angular difference between computed altitude of a body and its observed (true) altitude. Distance (in nautical miles) of observer's line of position from his assumed position.

ASTRONOMICAL TRIANGLE—Triangle on the celestial sphere bounded by the hour and vertical circles through a body, and the observer's (celestial) meridian.

AZIMUTH—The angle at the zenith (or arc of the horizon) measured from the observer's meridian to the vertical circle

passing through a body. It is measured from the North branch of the meridian if observer is in North latitude and from the South branch if he is in South latitude, and toward the East if body is rising or West if body is setting, through 180°.

AXIS

Earth—Centerline about which the earth turns.

Celestial Sphere—Earth's axis extended to the celestial sphere.

BEARING—Direction of one object to another measured clockwise through 360°.

Relative—Direction of an object with reference to the centerline of the aircraft.

True—Direction of an object with reference to true North.

Magnetic—Direction of an object with reference to magnetic North.

CHART—Representation of the earth's surface, or a portion thereof, on a flat surface. Maps are primarily concerned with the land, while charts usually supply information about water and land bounded by water.

CIRCLE OF EQUAL ALTITUDE—Circle on the earth's surface from which any observer will observe a given celestial body at the same altitude for a given instant of time.

COLLIMATION—Act of bringing a celestial body and bubble into coincidence in the field of an aircraft octant.

COURSE—Intended track.

True—Course measured with reference to true North.

Magnetic—Course measured with reference to magnetic North.

Compass—Course measured with reference to compass North (direction in which compass needle points).

DECLINATION—Arc of the hour circle measured from the equinoctial to the celestial body, North or South through 90°.

DEVIATION—Influence of local magnetic disturbances within the aircraft on the magnetic compass needle, causing the needle to point to other than magnetic North. The angular difference between magnetic North and compass North, which changes with heading of aircraft.

DIP

Compass—Deflection of compass needle from the horizontal due to the influence of the vertical component of the earth's magnetic field.

Celestial—A sextant altitude correction caused by depression of the horizon in proportion to aircraft's height. Must be considered when using sea horizon as horizontal reference plane.

DRIFT—Effect of crosswind. (Also see DRIFT ANGLE.)

DRIFT ANGLE—Angular difference between heading and track.

DRIFT CORRECTION ANGLE—Angular difference between heading and course.

ECLIPTIC—Path of earth in its revolution around the sun, or path of sun in its *apparent* revolution around the earth.

EQUATOR—Great circle on the earth's surface lying midway between the poles. Used as reference line for the measurement of latitude.

EQUINOCTIAL—Projection of the earth's equator onto the celestial sphere. Reference line for the measurement of declination.

EQUINOX—Intersection of the equinoctial and the ecliptic.

Vernal—Point where the sun changes declination from South to North (First Point of Aries).

Autumnal—Point where the sun changes declination from North to South (First Point of Libra).

FIRST POINT OF ARIES—(See Equinox, Vernal).

FIX—Geographical location of aircraft as determined by two or more lines of position.

GEOGRAPHICAL POSITION—Point where a line connecting the center of the earth with a body would intersect the surface of the earth.

GREAT CIRCLE—Circle on the surface of a sphere, the plane of which passes through the center of the sphere.

GROUND SPEED—Speed of the aircraft with reference to the ground.

HEADING—Direction in which the aircraft is pointed.

True—Heading with reference to true North.

Magnetic—Heading with reference to magnetic North.

Compass—Heading with reference to compass North.

HORIZON

Celestial (true) — Great circle on the celestial sphere, the plane of which is perpendicular to the zenith-nadir line.

Observer's—Plane passing through the eye of the observer, parallel to the celestial horizon.

Note: Celestial horizon and observer's horizon are so close, relatively, that for observations of all celestial bodies other than the moon they are considered to coincide.

HOOR ANGLE

Local—An arc of the equinoctial measured from the observer's meridian (upper branch) to the hour circle passing through a celestial body, over West through 360°.

Note: When used as an argument to enter H. O. 214 (or any other table of precomputed solutions of the astronomical triangle), local hour angle is measured only through 180° and is then named *East* if the body is *East* of the observer, and *West* if it is *West* of him.

Greenwich—Arc of the equinoctial measured from the Greenwich meridian to the hour circle passing through the celestial body, over West through 360°.

Sidereal—Arc of the equinoctial measured from the First Point of Aries to the hour

circle passing through the body, over West through 360° .

HOURLY CIRCLE—Great circle on the celestial sphere passing through both poles and a celestial body.

INDEX CORRECTION—Angular amount by which the counter on a sextant or octant is in error in recording the altitude of a body due to misalignment of index glass and arm, slippage or other mechanical faults within the instrument.

INSTRUMENT CORRECTION—Total angular correction which must be applied to each altitude reading because of index and bubble error in a sextant or octant.

ISOGONIC LINE—Line connecting points of equal magnetic variation.

KNOT—Unit of speed equal to one nautical mile per hour.

LATITUDE—Angular distance North and South of the equator measured in degrees of arc from 0° to 90° .

LINE OF POSITION

General—Straight line on the earth's surface representing all possible positions of an aircraft at a given instant of time.

Celestial—Small arc of a circle of equal altitude on which the observer is located. For short distances this arc is considered to be a straight line.

LONGITUDE—Angular distance East and West of the prime meridian, measured in degrees of arc from 0° to 180° .

MAP—(see CHART).

MERIDIAN—Great circle on the earth's surface passing through both poles. Projected onto the celestial sphere it becomes a celestial meridian.

Prime (Greenwich)—Meridian used as reference line for the measurement of longitude. Passes through Greenwich, England.

Observer's—Meridian passing through the observer's position.

NADIR—Point on the celestial sphere directly beneath the observer.

NAUTICAL MILE—Unit of distance equal to approximately 6080 feet, or one minute

of arc of a great circle on the earth's surface, hence one minute of latitude, or one minute of longitude at the equator.

OCTANT—Instrument used to measure the altitudes of celestial bodies. Specifically, one capable of measuring angles up to and including 90° . However, as commonly used, the term is synonymous with the term "sextant."

PARALLAX—Angle subtended at the center of the moon by the radius of the earth.

PARALLEL (of latitude)—Division of latitude parallel to the equator. Parallels are small circles.

PELORUS—A dummy compass card equipped with sighting vanes for taking bearings on terrestrial or celestial objects.

POLES

Earth—Ends of the earth's axis.

Celestial—Earth's poles projected onto the celestial sphere.

POLAR DISTANCE—Arc of the hour circle intercepted between a celestial body and the elevated pole. One side of the astronomical triangle. Equal to 90° minus declination of the body.

RADIUS OF ACTION—Distance an aircraft can fly with a given amount of fuel and under given wind conditions and still return to the same or an alternate base.

RHUMB LINE—Line which intersects all meridians at the same angle.

REFRACTION

General—Bending of light as it passes from one medium to another of different density.

Celestial—Bending of light rays from a celestial body on entering the earth's atmosphere, causing an error in measuring the altitude of the body.

RIGHT ASCENSION—An arc of the equinoctial lying between the First Point of Aries and the hour circle passing through a body, measured East through 24 hours.

SOLSTICE—Point of maximum declination of the sun. If maximum northerly declination it is known as summer solstice; if maximum southerly declination, it is winter solstice.

SPHERE

General—A body bounded by a surface, all points of which are equidistant from a point within called the center.

Celestial—A theoretical globe of infinite radius whose point of origin is considered to be the center of the earth.

SEMIDIAMETER—A sextant altitude correction which must be taken into account when using either the upper or lower limb (edge) of the sun or moon to achieve collimation. Ordinarily, therefore, this correction applies only to observations made with a marine sextant.

SEXTANT—Same as Octant except that it is capable of measuring angles up to 120°. Term commonly used to signify any type of altitude-measuring instrument.

SMALL CIRCLE—Circle on the earth's surface, the plane of which does not pass through the center of the earth.

STATUTE MILE—An arbitrary unit of distance equal to 5280 feet.

TRACK—Actual path made good by an aircraft over the ground.

True—Track measured with reference to true North.

Magnetic—Track measured with reference to magnetic North.

Compass—Track measured with reference to compass North.

Great Circle—Shortest distance between two points on the earth's surface.

TIME—Lapse between two events.

Solar (Apparent)—Time derived from passage of the true sun.

Civil (Mean)—Time derived from passage of the mean sun (an imaginary sun which is considered to move along the equinoctial at a constant rate throughout the year).

Sidereal (Star)—Time derived from passage of the First Point of Aries (or a star).

VARIATION—Angular difference between true North and magnetic North. Caused by the fact that true (geographic) North and magnetic North are not located at the same position on the earth's surface. Changes with aircraft's geographical position.

VECTOR—Any quantity having direction and magnitude which are represented graphically by a directed straight line segment. In navigation the principal vector quantities considered are heading and air speed of the aircraft, track (intended or actual) and ground speed, and the direction and velocity of the wind.

VERTICAL CIRCLE—Great circle on the celestial sphere passing through zenith, nadir and a celestial body.

ZENITH—Point on the celestial sphere directly above the observer.

Distance—Arc of the vertical circle lying between the celestial body and zenith. Equal to 90° minus altitude of the body. One side of the astronomical triangle.



ABBREVIATIONS

a	Altitude intercept
alt. (or h)	Altitude
A.M.	Antemeridian or forenoon
A.S.	Air speed
Az	Azimuth
C	Course
Cel.	Celestial
C.A.S.	Calibrated air speed
CC	Compass course
CE	Compass error
CH	Compass heading

Corr.	Correction
D.A.	Drift angle
dec.	Declination
dist.	Distance
DLo	Difference of longitude
Dep.	Departure
Dev.	Deviation
DL	Difference of latitude
DR	Dead Reckoning
E	East
ETA	Estimated time of arrival

GLOSSARY OF NAVIGATION TERMS

G.P.	Geographical position
GAT	Greenwich apparent time
GCT	Greenwich civil time
GHA	Greenwich hour angle
G.S.	Ground speed
GST	Greenwich sidereal time
HA	Hour angle
h(H)	Altitude
h _c	Computed altitude
h _o	True (observed) altitude
h _s	Sextant or octant altitude
I.A.S.	Indicated air speed
I.C.	Instrument correction
Kts	Knots
Lat.	Latitude
Long.	Longitude
m	Difference of meridional parts
M	Meridional parts
min.	minutes (of time)
MH	Magnetic heading
Mag.	Magnetic
MC	Magnetic course
m.p.h.	Miles per hour (statute miles)
MTr	Meridian transit
N	North
Na.	Nadir
L.O.P.	Line of position
NM	Nautical miles
P	Pole
Pd	Polar distance
par.	Parallax
P.M.	Postmeridian or afternoon
P _n	North celestial pole
PN	Point of no return
P _s	South celestial pole
R.A.	Radius of action
RA	Right ascension
R.B.	Relative bearing
Ref.	Refraction
S	South
SD	Semidiameter

SHA	Sidereal hour angle
TH	True heading
T.A.S.	True air speed
Temp.	Temperature
TC	True course
Tk	Track
T/t	Time to turn
Var.	Variation
W	West
Z	Zenith
Zd	Zenith distance
ZD	Zone description
Z _n	True bearing of a celestial body (measured from true north through 360°)
ZT	Zone time



SYMBOLS

☉	The sun
☾	The moon
*	A star or planet
☉	Altitude lower limb
☿	Altitude upper limb
♈	Vernal Equinox or First Point of Aries
♊	Bearing (Phi)
♋	Longitude (Lambda)
♌	Beta
♍	Delta
♎	Gamma



POSITION LEGEND

- 1—DR position
- 2—Approximate fix
- 3—Fix



ANSWERS TO PROBLEM WORK

Exact agreement cannot be expected with problems involving vector diagrams and chart work since the plotted angles or information obtained on a computer may vary slightly. Therefore, altitudes should be within ± 50 feet; air speed ± 2 knots or 2 m.p.h.; latitude and longitude ± 5 minutes, and courses and bearings within one degree of the results tabulated herewith.

PROBLEM WORK NO. 1 CHART DRAWINGS		PROBLEM WORK NO. 2 MERCATOR CHART		PROBLEM WORK NO. 3 AIRCRAFT COMPASS DRAWING	
PROBLEM WORK NO. 4 ALTIMETER CORRECTION		PROBLEM WORK NO. 5 AIR SPEED CORRECTION			
No.	TRUE ALTITUDE	No.	I.A.S. (m.p.h.)	C.A.S. (knots)	T.A.S. (knots)
1	8300'	1		129	154
2	3080'	2		146	164
3	23,300'	3	188	169	
4	11,200'	4	187	168	
5	5180'	5		149	164
6	19,100'	6	179	160	
7	1990'	7		132	144
8	9000'	8	185	165	
9	17,750'	9		144	146
10	13,850'	10		126	131
11	6170'	11		137	163
12	1410'	12	200	177	
13	2830'	13		150	162
14	10,200'	14		161	202
15	8500'	15		167	205
16	6270'	16		143	148
17	7500'	17	185	166	
18	9950'	18	187	168	
19	8700'	19		161	178
20	12,200'	20		109	111

PROBLEM ANSWERS

**PROBLEM WORK NO. 6
APPLYING COMPASS ERRORS**

No.	TRUE COURSE	VARIATION	MAGNETIC COURSE	DEVIATION	COMPASS COURSE
1		10° E			80°
2	270°			3° W	
3			24°		23°
4	2°				346°
5		5° W		5° W	
6			343°	4° E	
7		0°	122°		
8	359°		346°		
9		10° W		1° E	
10		15° E	100°		
11	223°		237°		
12		9° E			219°
13	147°				154°
14		11° W		9° E	
15			315°	5° W	
16			201°		200°
17	312°			6° W	
18		8° E	326°		
19	3°				343°
20		5° W			100°

**PROBLEM WORK NO. 7
TRACK AND HEADING**

No.	TRACK TRUE	VAR.	TRACK MAGNETIC	DRIFT	HEADING MAGNETIC	DEV.	HEADING	
							COMPASS	TRUE
1			103°		101°		104°	115°
2	231°			4° R		2° E		227°
3		14° W		4° L	65°		68°	
4		10° E		6° R		0°		336°
5			81°	3° R		7° E		93°
6	276°			2° R		9° W		274°
7			121°	1° R	120°		124°	
8	242°			2° R			223°	240°
9		16° E			155°	6° W		171°
10	209°		221°			4° E		209°
11	173°		183°				185°	175°
12	198°	16° E		4° R		5° W		
13	114°		131°		124°	4° E		
14		11° E		1° L	346°			357°
15		14° E		1° L			3°	7°
16	300°				281°	6° E		297°
17		12° W			262°		276°	250°
18	222°		236°		232°		237°	
19	200°	24° E	176°				190°	
20	118°			4° R		6° E		114°

AMERICAN AIR NAVIGATOR

PROBLEM WORK NO. 8 TIME—SPEED—DISTANCE

No.	DISTANCE (Nautical Miles)	No.	SPEED (Knots)	No.	TIME
1	181.25	9	118.05	16	1h26m
2	213.11	10	157.85	17	1h56m
3	347.64	11	109.12	18	0h35m
4	111.46	12	142.09	19	1h20m
5	468.05	13	141.96	20	1h42m
6	240.30	14	185.36		
7	284.40	15	148.15		
8	152.54				

PROBLEM WORK NO. 9-A VECTOR DIAGRAMS

PROBLEM WORK NO. 9-B VECTOR DIAGRAMS

PROBLEM WORK NO. 10 DOUBLE DRIFT

No.	HEADING		GROUND SPEED	DRIFT	No.	WIND	DRIFT	No.	WIND
	TRUE	COMPASS							
1	69°		104	16°R	1	22°/28	10°R	1	168°/11
2	152°		102	21°R	2	64°/47	25°R	2	316°/23
3	155°		92	15°L	3	236°/27	10°R	3	163°/17
4	315°		130	15°L	4	296°/25	12°L	4	102°/40
5	216°		141	8°L	5	116°/26	14°L	5	122°/23
6	162°		176	14°L	6	119°/19	7°L	6	43°/25
7	162°		176	8°R	7	344°/12	6°R	7	180°/19
8	248°		125	8°L	8	2°/43	9°L	8	192°/26
9	45°		165	8°L	9	59°/41	13°R	9	24°/33
10	348°		178	12°R	10	197°/41	16°R	10	339°/25
11	359°		158	9°L	11	27°/16	8°L	11	305°/19
12	24°		196	8°L	12	161°/30	11°L	12	355°/14
13	92°		139	8°R	13	336°/38	12°R	13	193°/41
14	71°		132	11°L	14	310°/17	7°R	14	114°/23
15	335°		163	5°L	15	142°/36	18°L	15	233°/21
16	195°	192°	142	9°R	16	23°/33	12°R	16	41°/25
17	24°	26°	151	7°R	17	98°/52	22°L	17	125°/17
18	82°	87°	168	6°L	18	264°/22	11°L	18	251°/36
19	199°	185°	127	9°L	19	213°/10	4°L	19	136°/18
20	280°	286°	140	10°L	20	338°/29	12°R	20	156°/22

DEAD RECKONING REVIEW TEST NO. 1

- 4 (a) Magnetic course = 306°. True course = 296°.
 (b) True track = 206°. Compass heading = 197°. Magnetic course = 190°.
 (c) True heading = 29°. Compass course = 15°. Magnetic course = 21°.
- 5 True altitude = 9300 feet.
- 6 (a) Calibrated air speed = 158 knots. True air speed = 191 knots.
 (b) True air speed = 193 knots.
- 7 (a) 05:44 GCT. (b) 118 knots. (c) 690 Nautical miles.
- 8 (a) Ground speed = 163 knots. True heading = 282°. Drift = 10°L.
 (b) Ground speed = 94 knots. Track = 74°. Drift = 14°R.
 (c) Wind = 313°/56 knots. Drift = 15°L.
- 9 Wind = 238°/22 knots.
- 10 Distance out = 529 Nautical miles. Time to turn = 12:21 GCT.

PROBLEM ANSWERS

PROBLEM WORK NO. 11-A RADIUS OF ACTION TO A FIXED BASE					PROBLEM WORK NO. 11-B RADIUS OF ACTION TO ALTERNATE BASE				
No.	TRUE HEADING		RADIUS OF ACTION	TIME TO TURN	No.	True Heading to		DISTANCE OUT	TIME TO TURN
	OUT	IN				Destination	Alternate		
1	78°	266°	524	12:33	1	38°	177°	293	2:39
2	301°	139°	301	12:21	2	222°	163°	57	:49
3	206°	10°	262	15:37	3	39°	249°	219	2:19
4	256°	65°	751	22:02	4	280°	137°	280	2:02
5	280°	82°	231	12:08	5	138°	281°	534	2:47
6	185°	37°	230	8:57	6	334°	227°	159	1:35
7	48°	234°	514	10:25	7	51°	158°	95	1:42
8	22°	238°	222	17:37	8	123°	332°	194	1:09
9	354°	227°	384	3:51	9	186°	284°	143	:51
10	123°	328°	516	4:02	10	296°	194°	63	:38
PROBLEM WORK NO. 12 RADIO BEARING DRAWING					PROBLEM WORK NO. 15 CELESTIAL SPHERE				
PROBLEM WORK NO. 13 MERCATOR SIGN CORRECTION					PROBLEM WORK NO. 16 ASTRONOMICAL TRIANGLE				
PROBLEM WORK NO. 14 RADIO BEARINGS		PROBLEM WORK NO. 17 GHA AND DECLINATION			PROBLEM WORK NO. 18 HOUR ANGLE DIAGRAMS				
No.	Mercator Bearing	No.	GHA	DEC.	No.	GHA ARIES	SHA	GHA BODY	LHA
1	54°	1	185°00'	N 14°44'	1				105°02'E
2	348°	2	152°52'	N 16°12'	2			330°41'	133°34'E
3	240°	3	195°57'	N 17°20'	3			66°42'	106°42'W
4	232°	4	193°46'	N 18°37'	4			122°43'	67°17'E
5	69°	5	76°12'	N 19°54'	5			234°15'	117°04'W
6	247°	6	41°59'	N 21°47'	6			93°45'	160°45'E
7	305°	7	334°38'	N 22°13'	7			28°51'	34°21'W
8	155°	8	353°10'	N 22°31'	8			114°34'	24°34'W
9	343°	9	316°32'	N 22°20'	9			347°15'	150°12'E
10	87°	10	35°48'	N 22°05'	10		50°07'	310°07'	64°53'E
11	290°	11	196°46'	N 8°06'	11	283°14'	259°21'	182°35'	1°25'W
12	165°	12	192°58'	N 39°	12	27°16'	63°00'	90°16'	19°46'W
13	311°	13	58°25'	N 5°32'	13	207°56'	291°50'	139°46'	130°09'E
14	78°	14	28°28'	S 7°55'	14	104°44'	113°31'	218°15'	97°52'W
15	126°	15	61°05'	S 5°03'	15	210°06'	16°22'	226°28'	113°17'E
16	50°	16	11°49'	S 29°55'	16	113°45'	245°55'	359°40'	2°35'E
17	359°	17	344°15'	N 7°24'	17	250°08'	77°04'	327°12'	3°09'E
18	30°	18	238°24'	N 19°29'	18	32°17'	159°27'	191°44'	96°29'W
19	63°	19	47°51'	N 45°05'					
20	31°	20	86°17'	N 8°43'					

AMERICAN AIR NAVIGATOR

CELESTIAL REVIEW TEST NO. 1

3 (a) LHA = 21° West.

(b) LHA = 70° East.

(c) LHA = 13° West.

(d) LHA = 17° East.

(e) LHA = 42° West.

PROBLEM WORK NO. 19 ALTITUDE CORRECTIONS		PROBLEM WORK NO. 20 LATITUDE BY POLARIS		PROBLEM WORK NO. 21 LOCAL ZONE TIME		
No.	H ₀	No.	LATITUDE	No.	ARRIVAL TIME AND DATE	
					GCT	ZONE TIME
1	49°11'	1	32°32'N	1	22:00 May 1	12:30 May 1
2	57°49'	2	38°51'N	2	06:00 May 6	20:30 May 5
3	36°17'	3	37°05'N	3	21:00 May 10	09:00 May 11
4	52°41'	4	39°45'N	4	17:30 May 15	03:30 May 16
5	39°01'	5	30°44'N	5	05:00 May 21	16:00 May 21
6	34°22'	6	37°19'N	6	18:45 May 25	08:15 May 25
7	30°48'	7	43°06'N	7	12:00 May 31	05:00 May 31
8	38°06'	8	23°56'N	8	00:15 May 16	13:45 May 15
9	23°17'	9	31°48'N	9	22:00 May 1	10:00 May 2
10	39°41'	10	25°01'N	10	22:30 May 10	09:30 May 11
11	41°14'	11	34°10'N	11	15:30 May 5	01:30 May 6
12	27°27'	12	26°45'N	12	12:00 May 20	23:00 May 20
13	42°38'	13	43°15'N	13	00:15 May 26	12:15 May 26
14	35°22'	14	29°52'N	14	23:00 May 15	12:30 May 15
15	28°51'	15	44°13'N	15	18:00 May 30	08:30 May 30
16	39°16'	16	37°16'N	16	03:30 May 2	20:30 May 1
17	30°32'	17	28°41'N	17	18:54 May 5	09:24 May 5
18	45°07'	18	44°12'N	18	01:00 May 11	18:00 May 10
19	41°58'	19	33°32'N			
20	41°27'	20	38°59'N			

PROBLEM WORK NO. 22

LCT — GCT

No.	LCT		No.	GCT	
	TIME	DATE		TIME	DATE
1	04:30:20 PM	May 1	11	06:14:40	May 6
2	05:16:00 AM	May 5	12	06:21:20	May 1
3	10:34:00 PM	May 9	13	06:28:40	May 10
4	11:39:30 AM	May 16	14	06:54:00	May 15
5	08:54:20 AM	May 20	15	00:50:40	May 21
6	09:33:20 AM	May 25	16	00:41:44	May 26
7	10:43:30 PM	May 29	17	00:26:40	May 30
8	00:30:00 AM	May 11	18	17:16:00	May 15
9	08:26:10 PM	May 1	19	10:53:40	May 6
10	02:26:56 AM	May 16	20	18:47:16	May 19

PROBLEM ANSWERS

PROBLEM WORK NO. 23 ASSUMED POSITION—LHA—DECLINATION					PROBLEM WORK NO. 24 ALTITUDE AND AZIMUTH(H.O. 214)		
No.	ASSUMED POSITION		LHA	Dec.	No.	H _c	AZIMUTH
	Lat.	Long.					
1	34°N	110°48'E	92°W	N 19°29'	1	28°18.2'	N 91.3°W
2	23°S	172°20'W	18°E	N 16°12'	2	46°45.9'	N 93.8°E
3	14°N	117°28'W	57°E	N 22°19'	3	14°13.5'	N 117.9°E
4	22°N	178°11'E	34°W	N 18°37'	4	56°23.9'	N 107.2°W
5	11°S	124°04'E	77°W	S 60°36'	5	49°17.9'	N 105.1°W
6	15°N	117°26'E	80°W	N 5°22'	6	62°47.9'	S 34.8°W
7	33°N	168°31'W	20°W	N 14°45'	7	37°40.8'	N 104.2°E
8	18°S	104°51'E	48°W	N 25°17'	8	17°57.8'	N 142.8°W
9	15°S	167°46'E	64°E	S 26°22'	9	60°38.7'	N 105.9°E
10	32°S	114°57'W	78°E	N 1°25'	10	44°02.0'	N 121.8°W
11	19°N	127°21'W	42°E	S 10°52'	11	64°56.2'	N 45.8°E
12	7°S	91°58'W	89°W	N 16°24'	12	55°06.4'	N 55.8°W
13	20°N	116°16'W	98°W	N 7°24'	13	73°36.6'	S 88.7°E
14	4°N	152°26'W	40°E	S 8°16'	14	20°15.8'	S 152.8°W
15	17°S	86°32'E	63°E	S 52°40'	15	40°33.0'	N 116.5°E
16	31°S	178°01'W	37°W	N 18°37'	16	32°06.8'	S 133.3°E
17	20°N	116°53'W	82°W	N 6°18'	17	45°34.0'	S 114.5°W
18	18°S	172°39'E	17°E	N 21°45'	18	35°45.5'	N 147.5°E
19	14°S	178°46'W	22°E	S 60°36'	19	24°18.1'	N 129.3°W
20	24°N	162°27'W	54°W	N 5°22'	20	66°18.1'	N 66.3°E

PROBLEM WORK NO. 25 AND NO. 26 ASSUMED POSITION—INTERCEPT—AZIMUTH (PLOT ON CHART)					PROBLEM WORK NO. 27 ADVANCING SINGLE LINES OF POSITION			
No.	ASSUMED POSITION		INTERCEPT	AZIMUTH	No.	Az	Intercept	N. M. Advance
	Lat.	Long.						
1	35°S	155°45'E	18.6' toward	S 70.7°E	1	N 104°E	28' away	15
2	35°S	153°23'E	0.6' away	S 11.9°W	2	S 146°W	46' toward	33
3	35°S	153°49'E	8.6' away	S 131.1°E	3	S 138°W	9' toward	133
4	30°N	160°26'W	28.1' away	N 85.1°W	4	N 123°W	20' toward	82
5	30°N	158°36'W	19.9' toward	N 96.0°E	5	S 172°E	7' toward	53
6	30°N	161°31'W	39.1' away	N 163.0°E	6	S 156°E	28' toward	25
7	39°N	152°25'E	7.2' away	N 116.7°W	7	N 109°E	28' toward	6
8	39°N	151°14'E	0.0	0.0°	8	S 105°E	34' away	18
9	39°N	151°04'E	14.2' toward	N 164.1°W	9	N 125°W	40' away	40
10	33°N	135°30'W	21.0' toward	N 81.6°E	10	S 140°W	37' toward	6
11	33°N	136°01'W	3.8' toward	N 140.4°W	11	N 147°E	44' away	51
12	33°N	136°03'W	22.0' away	N 117.4°W	12	S 116°W	47' away	7
13	36°S	145°00'W	7.2' away	S 141.5°W	13	N 87°W	23' away	14
14	36°S	146°06'W	9.0' toward	S 177.7°W	14	N 88°W	31' away	60
15	36°S	146°45'W	18.7' toward	S 144.0°E	15	N 123°E	7' toward	24
16	32°S	155°36'E	14.6' away	S 99.2°E	16	Latitude 35°42'N		6
17	32°S	156°59'E	2.2' away	S 127.2°W	17	N 108°E	23' away	17
18	32°S	156°18'E	14.8' away	S 172.1°E	18	N 76°E	20' toward	11
19	38°S	133°47'W	4.1' away	S 147.2°W	19	S 121°W	27' toward	9
20	38°S	135°20'W	7.8' toward	S 173.6°W	20	N 33°W	10' toward	29

CELESTIAL REVIEW TEST NO. 2

- 3 (a) LHA Sun = 160° West. (b) LHA Star = 40° East. GHA Star = 240°.
 4 (a) GCT = 02:31, January 3rd. (b) LCT = 6:08 AM July 16th.
 5 (a) LZT = 6:30 PM August 3rd. (b) LZT = 4:30 PM January 6th.
 6 GHA Sun = 350°49', Declination = 14°53'N.
 GHA Moon = 23°46', Declination = 1°35'S.
 GHA_{*} Sirius = 287°58', Declination = 16°38'S.
 7 H₀ Sun = 30°13'. H₀ Venus = 60°09'. H₀ Altair = 20°04'. H₀ Moon = 29°15'.
 8 Latitude = 32°48' North.
 9 Altitude Intercept = 7.1 away, Azimuth = N 158° E.

**PROBLEM WORK
NO. 28
THREE STAR FIXES**

PROBLEM WORK NO. 29

TRACK AND GROUND SPEED BETWEEN FIXES

No.	FIX		No.	POSITION		Track	Ground Speed	Drift	Wind	True Heading
	Lat.	Long.		Lat.	Long.					
1	31°57'N	9°19'W	1				177			
2	32°08'S	136°33'E	2	34°48'N	127°15'W	231°	185	11°L	340°/34	242°
3	35°08'N	16°59'E	3	33°25'N	130°07'W	240°	200	18°L	358°/63	258°
4	31°31'N	179°40'W	4	33°27'N	133°41'W	271°	213	4°R	108°/38	267°
5	39°36'N	175°27'W	5	31°31'N	136°54'W	235°	160	7°R	181°/26	228°
6	32°38'N	119°25'W								
7	36°08'N	125°33'E								
8	31°33'N	139°25'W								

**PROBLEM WORK NO. 30
STAR IDENTIFICATION BY H. O. 214**

**PROBLEM WORK NO. 31
LATITUDE BY MERIDIAN ALTITUDE**

No.	STAR	No.	STAR	No.	LATITUDE	No.	LATITUDE
1	Vega	11	Rasalague	1	25°58'N	11	32°41'N
2	Adhara	12	Kaus-Australis	2	3°10'N	12	23°39'N
3	Spica	13	Canopus	3	45°57'N	13	22°49'N
4	Altair	14	Denebola	4	13°21'S	14	32°43'N
5	Regulus	15	Alioth	5	24°00'S	15	29°19'N
6	Miaplacidus	16	Dschubba	6	43°56'N	16	12°35'S
7	Alphecca	17	Shaula	7	29°53'N	17	41°52'N
8	Antares	18	Deneb	8	32°41'N	18	26°20'S
9	Nunki	19	Alpheratz	9	5°02'S	19	6°13'S
10	Sabik	20	Deneb-Kaitos	10	35°25'N	20	0°19'S

INDEX

- Abbreviations, List of, 216
- Agonic line, 19
- Aircraft octant, 101
 - accuracy of, 102
 - altitude corrections, 80
 - errors, 106
 - handling, 108
 - optical principle of, 104
 - types of, 103
 - use, 102
- Air speed, 34
 - calibrated, 21
 - indicated, 21
 - indicator, 20
 - true, 22
- Almanac, Air, 77
 - dip, 82
 - extracts from, 193
 - finding GHA and declination, 78
 - latitude by Polaris, 84
 - parallax, 81
 - planets and stars along Ecliptic, 83
 - refraction, 80
 - semidiameter, 82
 - sunrise, sunset, moonrise and moonset, 83
- Altimeter, 20
 - errors, 20
 - uses, 20
- Altitude, 67, 80
 - circle of equal, 112
 - computed, 80
 - corrections, 80
 - sextant, 80
 - true, 80
- Analyzing fix, 179
- Answers to problem work, 218
- Aperiodic compass, 15
- Apparent solar time, 93
- Astronomical triangle, 72
- Autumnal equinox, 71
- Azimuth, 70
 - plotting, 121
- Bendix radio compass, 55
- Bubble acceleration error, 108
- Bubble error, 107
 - determining, 107
- Bubble octant, 103 (also see octant)
 - collimation, 106
 - forming bubble, 104
 - types of bubble, 105
- Calibration, compass, 17
- Calibration error, radio, 57
- Celestial coordinates, 68
 - equinoctial, 69
 - horizon, 69
- Celestial horizon, 67
- Celestial equator, 68
- Celestial sphere, 65
 - terms, 66
- Charts, 3
 - definition of, 3
 - development of, 5
 - gnomonic, 10
 - Lambert, conformal conic, 9
 - Mercator, 5
 - polyconic, 11
 - preparation for flight, 174
- Checking procedures, 164
 - equipment, 164
 - chronometer, 164
 - pre-flight, 165
 - pre-forecast, 164
 - radio compass, 165
- Chronometer, aircraft, 22
- Circles, 2
 - great, 2
 - hour, 68
 - of equal altitude, 112
 - small, 2
- Civil time, 94
 - description of, 94
 - Greenwich, 23
 - local, 98
- Clouds, 148
 - sky cover, 149
 - types of, 148

- Collimation, 106
- Compass, 14
 - aperiodic, 15
 - calibration, 17
 - compensation of, 17
 - deviation, 19
 - gyro, 14
 - magnetic, 14
 - pilots', 15
 - radio, 54
 - sun, 14
 - variation, 17
- Compensation, 17
 - by compass rose, 17
 - by Pelorus, 17
- Coriolis error, 108
- Coriolis force, 147
- Course, 31
 - compass, 31
 - drift-heading relationship, 33
 - magnetic, 30
 - true, 31
- Course conversion, 32
- Dead reckoning, 29
 - air speed, 34
 - compass errors, 31
 - course conversion, 32
 - course-drift-heading relationship, 33
 - ground speed, 34
 - radius of action, 39
 - time-speed-distance relationship, 34
 - vector diagrams, 35
- Declination, 68, 78
- Deviation, 19
- Dip, 14, 82
 - celestial, 82
 - magnetic, 15
- Distances, measuring, 6
- Double drift, 38
 - procedure, 38
- Drift angle, 33
- Drift meter, 23
 - alignment of, 24
 - simple type, 23
 - type B-3, 23
- Earth, 1
 - orbit of, 94
 - shape of, 2
 - terms, 2
- Ecliptic, 70
 - obliquity of, 94
 - positions of stars and planets along, 83
- Emergency procedure, 155
- Equator, 2, 68
 - celestial, 68
 - earth, 2
- Equinoctial, 68
- Equinoxes, 71
 - autumnal, 71
 - vernal, 71
- Equipment check list, 164
- Errors, octant, 106
 - bubble, 107
 - bubble acceleration, 108
 - coriolis, 108
 - index, 107
 - instrument, 21, 107
 - transient, 108
- Errors, compass, 31
 - applying, 31
- Fixes, 126
 - analyzing, 179
 - by any two position lines, 112, 128
 - by radio bearings, 62
 - running, 127
 - three star, 129
 - track and ground speed between, 129
- Flight graph, 171
 - preparation of, 171
- Flight time analysis, 168
- Flight plan, 171
 - return, preparation of, 171
- Fog, types of, 149
- Forecast, flight, 166
- Front, 149
 - cold, 150
 - occluded, 150
 - warm, 150
- Geographical position, 113
 - relation of observer to, 113
- Glossary of navigation terms, 213
- Gnomonic projection, 10
- Great circle, 2
 - radio bearing, 53
 - track, 3
- Greenwich civil time, 23
- Greenwich hour angle, 70, 78

INDEX

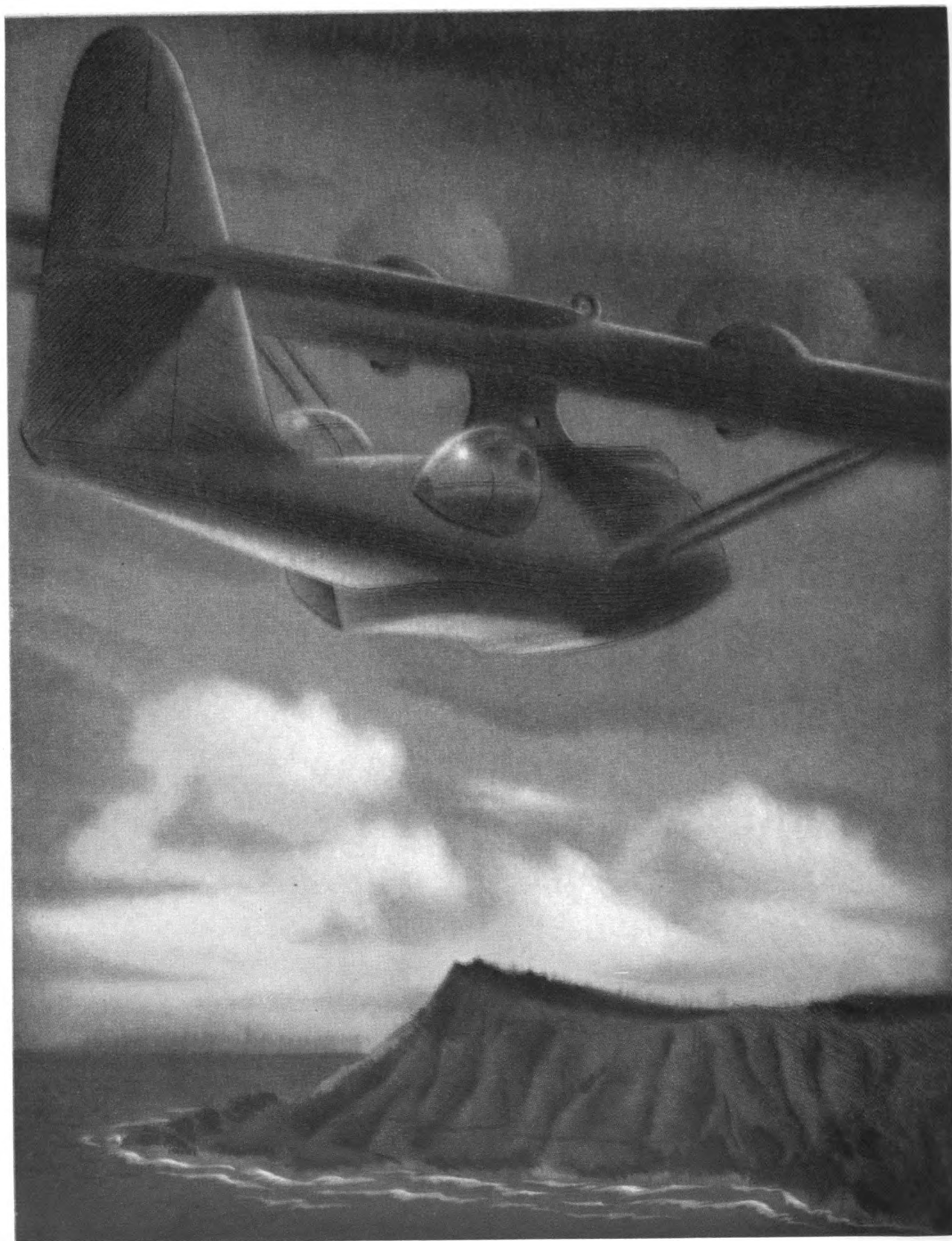
- Ground speed, 34
 - determining, 24, 126, 129
- Gyro compass, 14
- Heading, 33
 - compass, 33
 - course-drift relationship, 33
 - magnetic, 33
 - true, 33
- H.O. 214, 114
 - computing position lines by, 116
 - obtaining local hour angle, 116
- Homing, 56
 - aural-null, 56
 - procedure, 56
 - visual, 56
- Horizon, celestial, 67
- Hour angle, 70
 - diagrams, 79
 - Greenwich, 70, 78
 - local, 70
 - sidereal, 72, 78
- Hour circles, 68
- Index error, 107
- Indicator, airspeed, 20
 - left-right, 56
- Instruments, aircraft navigation, 13
- Instrument correction, 80
- Instrument error, 21, 107
- International date line, 97
- Isogonic lines, 19
- Knot, 3
- Lambert projection, 9
- Latitude, 2, 84, 158
 - by Meridian altitude, 158
 - by Polaris, 84
 - definition of, 2
 - parallels of, 2
- Left-right indicator, 56
- Lines of position, 111
 - advancing or retarding, 126
 - assumed position, 121
 - computing, 114
 - plotting, 121
 - single, 125
 - use to determine ground speed, 126
 - use to determine track, 126
- Local hour angle, 70
 - finding, 116
- Longitude, 2
- Loop radio, 55
 - automatically operated, 55
 - manually operated, 55
- Magnetic compass, 14
 - compensation, 16
 - types of, 15
- Magnetic pole, 17, 19
- Map (see Chart)
- Mercator chart, 5
 - construction of, 6
 - plotting radio bearing on, 58
- Mercator correction, 58
 - determining sign of, 58
- Meridian altitude, latitude by, 158
- Meridian transit, finding time of, 159
 - graphic method, 159
 - procedure, 158
 - rate of closure, 160
- Meridians, 2
 - prime, 2
- Meteorology, 146
- Mile, 3
 - nautical, 3
 - statute, 3
- Moonrise and moonset, determining time of, 83
- Nadir, 67
- Nautical mile, 3
- Navigator's flight report, 174
- Night effect, 53
- Octant, (also see bubble octant)
 - accuracy of, 102
 - aircraft, 101
 - altitude corrections, 80
 - errors, 106
 - evolution of, 101
 - handling, 108
 - optical principle of, 104
 - types of, 103
 - use, 102
- Parallax, 81
- Parallels of latitude, 2
- Pelorus, 17
- Pilots' compass, 15
 - principal parts, 15

- Pitot tube, 21
- Plotting, 121
 assumed position, 121
 azimuth, 121
 line of position, 123
 with Weems plotter, 122
- Point of no return, 39
- Polaris, latitude by, 84
- Polyconic projection, 11
- Position legend, 217
- Pre-flight check, 165
- Prime meridian, 2
- Prime vertical, 67
- Projections, 5
 gnomonic, 10
 Lambert conformal conic, 9
 Mercator, 5
 polyconic, 11
- Radio, 53
 bearings, 57, 60
 fix, 62
 homing with, 56
 loop, 55
 navigation, 53
 night effect, 53
 position finding by, 60
 ranges, 59
 wave characteristics, 53
- Radio compass, 54
 check, 165
 description, 54
- Radio compass, Bendix, 55
 for bearings, 56
 for homing, 56
- Radio range, 59
- Radius of action, 39
 to an alternate base, 41
 to same base, 40
- Refraction, 80
- Relative radio bearings, 57
 calibration error, 57
 Mercator correction, 58
 obtaining fix, 62
 procedure, 57, 60
 properties and limitations, 60
 true bearings, 57
- Rhumb line, 3
- Right ascension, 72
- Running down a sun line, 157
- Semi-diameter, 82
 correction for, 79
- Sextant (see octant)
- Sidereal hour angle, 72, 78
- Sidereal time, 95
- Sights, procedure for taking, 108
- Small circle, 2
- Solstices, 71
- Special procedures, 155
 emergency, 155
 latitude by meridian altitude, 158
 meridian transit, 159
 running down a sun line, 157
 square search, 156
- Speed, 34
 air, 34
 ground, 34
 -time-distance relationship, 34
- Sphere, celestial, 65
- Square search procedure, 156
- Star groups, 140
- Star identification, 140
 by H.O. 214, 143
- Statute mile, 3
- Sun compass, 14
- Sunrise and sunset, determining time of, 83
- Symbols, 217
- Thermometer, 22
- Time, 91
 apparent solar, 93
 civil, 94
 conversion to arc, 92
 equation of, 95
 importance of, 91
 kinds of, 93
 measurement of, 91
 relationship between local and Greenwich, 98
 hour angle, longitude relationship to, 92
 sidereal, 95
 zone, 96
- Time-speed-distance relationship, 34
- Track, 3, 31
 between fixes, 129
 determined by single position line, 126
 great circle, 3
 rhumb line, 3

INDEX

- Transient error, 108
- Transit instrument, 92
- Triangle of velocities (see vector diagrams)
- Variation, compass, 17
- Vector diagrams, 35
 - basic types, 35
- Vernal equinox, 71
- Vertical circle, 67
 - prime vertical, 67
- Weather, forecasting, 146
 - air masses, 146
 - isobars, 147
 - pressure systems, 146
 - winds, 147
- Weather map analysis, 151
- Zenith, 67
- Zone description, 97
- Zones, forecast, 167
- Zone time, 96
 - conversion of, 97





89090518663



B89090518663A



Coronado



Catalina Amphibian



Vengeance



Liberator Express



✓

rator

Relian

Sentinel "Flying Bo

Valiant

89090518663



b89090518663a